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(NASA-CR-141800) RESULTS OF A 0.03- SCALE N75-33162
AERODYNAMIC CHARACTERISTICS INVESTIGATION OF
BOEING 747 CARRIER (MODEL NO. AX 1319 I-1).
MATED WITH A SPACE SHUTTLE ORBITER (MODEL Unclas
45-0) CONDUCTED IN THE BOEING TRANSONIC WIND G3/18 42994

SPACE SHUTTLE

AEROTHERMODYNAMIC DATA REPORT

JOHNSON SPACE CENTER
HOUSTON, TEXAS



DMS-DR-2211 NASA CR-141,800

VOLUME 1 of 3

RESULTS OF A 0.03-SCALE AERODYNAMIC CHARACTERISTICS

INVESTIGATION OF A BOEING 747 CARRIER

(MODEL NO. AX 1319 I-1) MATED WITH A SPACE

SHUTTLE ORBITER (MODEL 45-0) CONDUCTED IN

THE BOEING TRANSONIĆ WIND TUNNEL (CA5)

by

747 Aerodynamics, 747 Flight Controls and Wind Tunnel Test Group, Boeing Aerospace Company

Prepared under NASA Contract Number NAS9-13247

bу

Data Management Services Chrysler Corporation Space Division New Orleans, La. 70189

for

Engineering Analysis Division

Johnson Space Center National Aeronautics and Space Administration Houston, Texas

WIND TUNNEL TEST SPECIFICS:

Test Number: BTWT 1431

NASA Series Number: CA5

Model Number: AX 1319 I-1 (Carrier), 45-0 (Orbiter) Test Dates: September 20 through September 30, 1974

Occupancy Hours: 65.3

FACILITY COORDINATOR:

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ABSTRACT

This report presents results of an experimental aerodynamic investigation of the Boeing 747/Space Shuttle Carrier aircraft conducted in the Boeing Transonic Wind Tunnel during September 1974. The objective of these tests was to determine performance, stability, and control characteristics of various carrier aircraft configurations. Aerodynamic characteristics of the carrier mated with the Orbiter, carrier alone, and Orbiter alone were investigated. Carrier support system tare and interference effects were determined.

Six-component force and moment data were recorded for the carrier and Orbiter. Buffet onset characteristics of the carrier vertical tail and horizontal tail were recorded. Angles of attack from -3° through 26° and angles of sideslip between +12° and -12° were investigated at Mach numbers from 0.15 through 0.70.

This report for CA5 consists of three volumes: Volume 1 - data figures 1 to 200; Volume 2 - data figures 201 to 260; Volume 3 - tabulated source data.

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	FIG. 136	747 + ORB (8 DEG) + TIP FINS + STD SPD BRK, FLAPS UP LAUNCH, ALPI = 6, RUD = 10/10	747	. D	509-511
	FIG. 137	747 + ORB (8 DEG) + TIP FINS + STD SPD BRK, FLAPS UP LAUNCH, ALPI = 6, RUD = 10/10	ORB	C	512-514
	FIG. 138	747 + ORB (8 DEG) + TIP FINS, FLAPS UP LAUNCH, ALPHAI = 6 DEG	TOTAL	С	515-517

FIGURE NUMBER	TITLE		COEFFICIENT SCHEDULE	PAGES
FIG. 139	747 + ORB (8 DEG) + TIP FINS, FLAPS UP LAUNCH, ALPHAI = 6 DEG	747	D	518-520
FIG. 140	747 + ORB (8 DEG) + TIP FINS, FLAPS UP LAUNCH, ALPHAI = 6 DEG	ORB	С	521-523
FIG. 141	747 + ORB (8 DEG) + TIP FINS + STD SPD BRK, FLAPS UP LAUNCH, ALPHAI = 6, TC OFF	TOTAL	С	524-526
FIG. 142	747 + ORB (8 DEG) + TIP FINS + STD SPD BRK, FLAPS UP LAUNCH, ALPHAI = 6, TC OFF	747	D	527-529
FIG. 143	747 + ORB (8 DEG) + TIP FINS + STD SPD BRK, FLAPS UP LAUNCH, ALPHAI = 6, TC OFF	ORB	С	530-532
FIG. 144	747 + ORB (8 DEG) + TIP FINS + STD SPD BRK, FLAPUP LNCH, ALP = 6, TC OFF, R = 10/10	TOTAL	С	533-535
FIG. 145	747 + ORB (8 DEG) + TIP FINS + STD SPD BRK, FLAPUP LNCH, ALP = 6, TC OFF, R = 10/10	747	· D	536-53 <u>8</u>
FIG. 146	747 + ORB (8 DEG) + TIP FINS + STD SPD BRK, FLAPUP LNCH, ALP = 6, TC OFF, R = 10/10	ORB	С	539-541
FIG. 147	747 + ORB (10 DEG) + TIP FINS + STD SPD BRK, FLAPS UP LAUNCH, ALPHAI = 6 DEG	TOTAL	ċ	542-544
FIG. 148	747 + ORB (10 DEG) + TIP FINS + STD SPD BRK, FLAPS UP LAUNCH, ALPHAI = 6 DEG	747	D	545.–547
FIG. 149	747 + ORB (10 DEG) + TIP FINS + STD SPD BRK, FLAPS UP LAUNCH, ALPHAI = 6 DEG	ORB	C	548-550

FIGURE NUMBER	TITLE		COEFFICIENT SCHEDULE	PAGES
FIG. 150	747 + ORB (8 DEG) + STD SP, VERT TAIL OFF, FLAP UP LNCH, ALPHAI = 6, TC OFF	TOTAL	С	551-553
FIG. 151	747 + ORB (8 DEG) + STD SP, VERT TAIL OFF, FLAP UP LNCH, ALPHAI = 6, TC OFF	747	D	554-556
FIG. 152	747 + ORB (8 DEG) + STD SP, VERT TAIL OFF, FLAP UP LNCH, ALPHAI = 6, TC OFF	ORB	С	557-559
FIG. 153	747 + ORB (8 DEG) + TIP FINS + STD SP, FLAPS UP LNCH, ALPI = 6, ORB ELEV = 5	TOTAL	С	560-562
FIG. 154	747 + ORB (8 DEG) + TIP FINS + STD SP, FLAPS UP LNCH, ALPI = 6, ORB ELEV = 5	747	D	563-565
FIG. 155	747 + ORB (8 DEG) + TIP FINS + STD SP, FLAPS UP LNCH, ALPI = 6, ORB ELEV = 5	ORB '	С	566-568
FIG. 156	747 + ORB (8 DEG) + STD SP, VERT TAIL OFF, FLAPS UP, ALPI = 2, DE = 0/0 (S = -1)	TOTAL	С	569-571
FIG. 157	747 + ORB (8 DEG) + STD SP, VERT TAIL OFF, FLAPS UP, ALPI = 2, DE = $0/0 (S = -1)$	747	D	572-574
FIG. 158	747 + ORB (8 DEG) + STD SP, VERT TAIL OFF, FLAPS UP, ALPI = 2, DE = $0/0$ (S = -1)	ORB	С	575-577
FIG. 159	747 + ORB (8 DEG) + STD SP, VERT TAIL OFF, FLAPS UP, ALPI = 6, DE = 0/0 (S = -1)	TOTAL	С	578-580
FIG. 160	747 + ORB (8 DEG) + STD SP, VERT TAIL OFF, FLAPS UP, ALPI = 6, DE = $0/0$ (S = -1)	747	D	581-583

FIGURE NUMBER	TITLE		COEFFICIENT SCHEDULE	PAGES
FIG. 161	747 + ORB (8 DEG) + STD SP, VERT TAIL OFF, FLAPS UP, ALPI = 6, DE = 0/0 (S = -1)	ORB	Ċ	584-586
FIG. 162	747 + ORB (8 DEG) + STD SP, VERT TAIL OFF, FLAPS UP, ALPI = 6, DE = 0/0 (S = +1)	TOTAL	. c .	587-589
FIG. 163	747 + ORB (8 DEG) + STD SP, VERT TAIL OFF, FLAPS UP, ALPI = 6, DE = 0/0 (S = +1)	747	D	590-592
FIG. 164	747 + ORB (8 DEG) + STD SP, VERT TAIL OFF, FLAPS UP, ALPI = 6, DE = 0/0 (S = +1)	ORB	С	593-595
FIG. 165	747 + ORB (4 DEG), VERT TAIL OFF, FLAPS UP FERRY, ALPHAI = 2 (S = ~2)	TOTAL	С	596-598
FIG. 166	747 + ORB (4 DEG), VERT TAIL OFF, FLAPS UP FERRY, ALPHAI = $2 (S = -2)$	747	D	599-601
FIG. 167	747 + ORB (4 DEG), VERT TAIL OFF, FLAPS UP FERRY, ALPHAI = 2 (S = -2)	ORB	. c .	602-604
FIG. 168	747 + ORB (4 DEG), VERT TAIL OFF, FLAPS UP FERRY, ALPHAI = 6 (S = -2)	TOTAL	С	605-607
FIG. 169	747 + ORB (4 DEG), VERT TAIL OFF, FLAPS UP FERRY, ALPHAI = 6 (S = -2)	747	D	608-610
FIG. 170	747 + ORB (4 DEG), VERT TAIL OFF, FLAPS UP FERRY, ALPHAI = 6 (S = -2)	ORB	С	611-613
FIG. 171	747 + ORB (4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI = 8 (S = -2)	TOTAL	C	614-616

FIGURE NUMBER	TITLE	COEFFICIENT SCHEDULE	PAGES
FIG. 172	747 + ORB (4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI = 8 (S = -2)	.7 D	617-619
FIG. 173	747 + ORB (4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI = 8 (S = -2)	RB C	620-622
FIG. 174	747 + ORB (4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI = 2 (S = ~3.2)	L C	623-625
FIG. 175	747 + ORB (4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI = 2 (S =-3.2)	.7 D	626-628
FIG. 176	747 + ORB (4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI = 2 (S =-3.2)	B C	629-631
FIG. 177	747 + ORB (4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI = 6 (S =-3.2)	ıL C	632-634
FIG. 178	747 + ORB (4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI = 6 (S = -3.2)	.7 D	635-637
FIG. 179	747 + ORB (4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI = 6 (S = -3.2)	B C·	638-640
FIG. 180	747 + ORB (4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI = 8 (S = -3.2)	L C	641-643
FIG. 181	747 + ORB (4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI = 8 (S = ~3.2)	.7` D	644-646
FļG. 182	747 + ORB (4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI = 8 (S = -3.2)	RB C	647-649

FIGURE NUMBER	TITLE	COEFFICIENT SCHEDULE	PAGES
FIG. 183	747 + ORB (4 DEG) + TIP FINS, FLAPS UP FERRY, ALPHAI = 2 (S = -3.2) TOTAL	, c	650-652
FIG. 184	747 + ORB (4 DEG) + TIP FINS, FLAPS UP FERRY, ALPHAI = 2 (S = -3.2) 747	D	653-655
FIG. 185	747 + ORB (4 DEG) + TIP FINS, FLAPS UP FERRY, ALPHAI = 2 (S = -3.2) ORB	C	656-658
FIG. 186	747 + ORB (4 DEG) + TIP FINS, FLAPS UP FERRY, ALPHAI = 6 (S = ~3.2) TOTAL	C	659`-661
FIG. 187	747 + ORB (4 DEG) + TIP FINS, FLAPS UP FERRY, ALPHAI = 6 (S = -3.2) 747	D	662-664
FIG. 188	747 + ORB (4 DEG) + TIP FINS, FLAPS UP FERRY, ALPHAI = 6 (S = -3.2) ORB	С	665-667
FIG. 189	747 + ORB (4 DEG) + TIP FINS, FLAPS UP FERRY, ALPHAI = 8 (S = ~3.2) TOTAL	. c	668-670
FIG. 190	747 + ORB (4 DEG) + TIP FINS, FLAPS UP FERRY, ALPHAI = 8 (S = -3.2) 747	D	671-673
FIG. 191	747 + ORB (4 DEG) + TIP FINS, FLAPS UP FERRY, ALPHAI = 8 (S = -3.2) ORB	С	674-676
FIG. 192	747 + ORB (4 DEG) + TIP FINS, FLAPS UP FERRY, ALPHAI = 6 (S = -2) TOTAL	c	677-679
FļG. 193	747 + ORB (4 DEG) + TIP FINS, FLAPS UP FERRY, ALPHAI = 6 (S = -2) 747	D	680-682

FIGURE NUMBER	TITLE		COEFFICIENT SCHEDULE.	PAGES
FIG. 194	747 + ORB (4 DEG) + TIP FINS, FLAPS UP FERRY, ALPHAI = 6 (S = -2).	ORB	С	683-685
FIG. 195	747 + ORB (4 DEG) + TIP FINS, FLAPS UP, ALPHAI = 6, RUDDER = 10/10	 TOTAL	С	686-688
FIG. 196	747 + ORB (4 DEG) + TIP FINS, FLAPS UP, ALPHAI = 6, RUDDER = 10/10	747	D	689-691
FIG. 197	747 + ORB (4 DEG) + TIP FINS, FLAPS UP, ALPHAI = 6, RUDDER = 10/10	ORB	C	692-694
FIG. 198	747 + ORB (4 DEG) + TIP FINS, FLAPS UP, ALPHAI = 2, RUDDER = 10/10	TOTAL	С	695-697
FIG. 199	747 + ORB (4 DEG) + TIP FINS, FLAPS UP, ALPHAI = 2, RUDDER = 10/10.	747	D.	698-700
	747 + ORB (4 DEG) + TIP FINS, FLAPS UP, ALPHAI = 2, RUDDER = 10/10	ORB	C.	701-703
<u>VOLUME 2</u> FIG. 201	_747 + ORB (4 DEG) + TIP FINS, FLAPS UP, ALPHAI = 8, RUDDER = 10/10	TOTAL	С	704-706
FIG. 202	747 + ORB (4 DEG) + TIP FINS, FLAPS UP, ALPHAI = 8, RUDDER = 10/10	747	D	707-709
FÍG. 203	747 + ORB (4 DEG) + TIP FINS, FLAPS UP, ALPHAI = 8, RUDDER = 10/10	ORB	С	710-712
FIG. 204	747 + ORB (4 DEG) + TIP FINS, FLAPS UP, ALPHAI = 6, RUDDER = 10/10	TOTAL	c,	713-715

FIGURE NUMBER	TITLE		COEFFICIENT SCHEDULE	PAGES
FIG. 205	747 + ORB (4 DEG) + TIP FINS, FLAPS UP, ALPHAI = 6, RUDDER = 10/10	747	D	716-718
FIG. 206	747 + ORB (4 DEG) + TIP FINS, FLAPS UP, ALPHAI = 6, RUDDER = 10/10	ORB	С	719-721
FIG. 207	747 + ORB (4 DEG) + TIP FINS, FŁAPS UP, ALPHAI = 6, TAIL CONE OFF	TOTAL	C	722-724
FIG. 208	747 + ORB (4 DEG) + TIP FINS, FLAPS UP, ALPHAI = 6, TAIL CONE OFF	747	D	725-727
FIG. 209	747 + ORB (4 DEG) + TIP FINS, FLAPS UP, ALPHAI = 6, TAIL CONE OFF	ORB	· c	728-730
FIG. 210	ORBITER ALONE, TAIL CONE OFF, ALPHAI = 8 DEG		С	731 -733
FIG. 211	ORBITER ALONE, TAIL CONE OFF, ALPHAI = 15 DEG		С	734-736
FIG. 212	ORBITER ALONE, TAIL CONE ON, ALPHAI = 8 DEG		С	737-739
FIG. 213	ORBITER ALONE, TAIL CONE ON, ALPHAI = 15 DEG		С	740-742
FIG. 214	747 ALONE, EFFECT OF STD IN-FLIGHT SPEED BRAKES, FLAPS UP		Α	743-750
FIG. 215	747 + ORB (8 DEG), EFFECT OF STD IN-FLIGHT SPEED BRAKES, FLAPS UP		А	751-766
FIG. 216	BASIC 747 + ORB (10 DEG), EFFECT OF FULL 45 DEG SPOILERS, FLAPS 20		А	767-778
FIG. 217	747 ALONE, EFFECT OF FULL 45 DEG SPOILERS, FLAPS 20		Α	779~782

FIGURE NUMBER	TITLE	COEFFICIENT SCHEDULE	, PAGES
FIG. 218	747 + ORB (10 DEG), EFFECT OF STD BRAKES OR FULL 45 SPOILERS, FLAPS 20	А	783-794
FIG. 219	747 + ORB, EFFECT OF ORB INCIDENCE, FLAPS UP FERRY CONFIGURATION	A	795–810
⊸FIG. 220	747 + ORB, EFFECT OF ORB INCIDENCE, FLAPS 20 LAUNCH CONFIGURATION	A	811-822
FIG. 221	747 + ORB, EFFECT OF ORB INCIDENCE, FLAPS UP LAUNCH CONFIGURATION	А	823-838
	747 + ORB (4 DEG), EFFECT OF TAIL CONE, FLAPS UP FERRY CONFIG.	Α	839-854
FIG. 223	747 + ORB (8 DEG), EFFECT OF TAIL CONE, FLAPS UP LAUNCH CONFIG.	Α	855-870
FIG. 224	747 ALONE, EFFECT OF HORIZ TIP FINS, FLAPS UP	Α	871-890
FIG. 225	747 + ORB (4 DEG), EFFECT OF HORIZ TIP FINS, FLAPS UP FERRY	A	891-906
FIG. 226	747 ALONE, EFFECT OF HORIZ TIP FINS, FLAPS 20	Α	907-922
FIG. 227	747 + ORB (10 DEG), EFFECT OF HORIZ TIP FINS, FLAPS 20 LAUNCH	. A	923-934
FIG. 228	747 ALONE, EFFECT OF ORB SUPPORTS, FLAPS UP	A	935-950
FÍG. 229	747 ALONE, STABILIZER EFFECT, FLAPS UP	Α	951-970
FIG. 230	747 + ORB (8 DEG), STABILIZER EFFECT, FLAPS UP LAUNCH	Ų	971-990
FIG. 231	747 + ORB (4 DEG), STABILIZER EFFECT, FLAPS UP FERRY	А	991-1010

•	FIGURE NUMBER	TITLE	COEFFICIENTSCHEDULE	PAGES
•	FIG. 232	747 + ORB (10 DEG), STABILIZER EFFECT, FLAPS 20 LAUNCH	Α	1011-1022
	FIG. 233	747 ALONE, ELEVATOR EFFECT FLAPS UP	Α	1023-1026
	FIG. 234	747 ALONE, EFFECT OF 5 DEG SIDESLIP, FLAPS UP	Е	1027-1040
	FIG. 235	747 ALONE, RUDDER EFFECT (BETA = 0), FLAPS UP	E	1041-1047
	FIG. 236	747 + ORB (8 DEG), ELEVATOR EFFECT, FLAPS UP LAUNCH	А	1048-1063
	FIG. 237	747 + ORB (8 DEG), ELEVATOR EFFECT, FLAPS UP LAUNCH, TAIL CONE OFF	Α	1064-1079
	FIG. 238	747 + ORB (4 DEG), ELEVATOR EFFECT, FLAPS UP FERRY	Α	1080-1095
သ ိ	FIG. 239	747 ALONE, ELEVATOR EFFECT, FLAPS 20	Α	1096-1099
	FIG. 240	747 + ORB (8 DEG), EFFECT OF ORBITER ELEVON, FLAPS UP	Α	1100-1107
	FIG. 241	747 + ORB (8 DEG), STABILIZER EFFECT, FLAPS UP LAUNCH, TAIL CONE OFF	А	1108-1123
	FIG. 242	BASIC 747 + ORB (4 DEG), STABILIZER EFFECT, FLAPS UP FERRY	А	1124-1139
	FIG. 243	747 + ORB, EFFECT OF FLAPS ON LAUNCH CONFIG.	А	1140-1151
	FIG. 244	747 ALONE, EFFECT OF HORIZ TIP FINS, FLAPS UP	С	1152-1166
	FIG. 245	747 + ORB (4 DEG), EFFECT OF HORIZ TIP FINS, FLAPS UP FERRY	.C	1167-1181
	FIG. 246	747 ALONE, EFFECT OF HORIZ TIP FINS, FLAPS 20	С	1182-1190
	FIG. 247	747 + ORB (10 DEG), EFFECT OF HORIZ TIP FINS, FLAPS 20 LAUNC	Н С	1191-1199

	FIGURE NUMBER	ŢĨŢĹĘ	COEFFICIENȚ SCHEDULE	PAGES
	FIG. 248	747 ALONE, RUDDER EFFECT WITH ORB SUPPORTS, FLAPS UP, ALPHAI = 6 DEG	C	1200-1205
	FIG. 249	747 + ORB (4 DEG), RUDDER EFFECT, FLAPS UP FERRY, ALPHAI = 6 DEG	С	1206-1217
	FIG. 250	747 + ORB (4 DEG), EFFECT OF ALPHA, FLAPS UP FERRY.	С	1218-1232
	FIG. 251	747 + ORB (8 DEG), RUDDER EFFECT, FLAPS UP LÄUNCH, ALPHAI = 6 DEG	С	· ¦ 1233-1244
	FIG. 252	747 + ORB (8 DEG), RUDDER EFFECT, TAIL CONE OFF, FLAPS UP LAUNCH, ALPHAI = 6	С	1245-1256
24	FIG. 253	747 + ORB (8 DEG), EFFECT OF ORB ELEVONS, FLAPS UP LAUNCH	Ċ	1257-1262
	FIG. 254	747 + ORB (4 DEG), EFFECT OF TAIL CONE, FLAPS UP FERRY	С	1263-1274
	FIG. 255	747 + ORB (8 DEG), EFFECT OF TAIL CONE, FLAPS UP LAUNCH	С	1275-1286
	FIG. 256	747 + ORB, EFFECT OF ORB INCIDENCE, FLAPS UP LAUNCH	C	1287-1298
	FIG. 257	747 + ORB, EFFECT OF ORB INCIDENCE, FLAPS 20 L'AUNCH	C	1299-1307
	FIG. 258	747 + ORB, EFFECT OF ORB INCIDENCE, FLAPS UP FERRY	С	1308-1319
	FIG. 259	747 + ORB (10 DEG), RUDDER EFFECT, FLAPS 20 LAUNCH, ALPHAI = 6 DEG	C	1320-1328
	FIG. 260	747 + ORB (10 DEG), EFFECT OF FULL 45 DEG SPOILERS, FLAPS 20	. c	1329-1337

COEFFICIENT SCHEDULE:

- A: CL vs. ALPHAW; CL vs. CD; CL, ALPHAW vs. CLM
- B: DCL vs. ALPHAW; DCL vs. DCD; DCL, ALPHAW vs. DCLM
- C: CY, CLN, CSL vs. BETA
- D: DCY, DCLN, DCSL vs. BETA
- E: CL vs. ALPHAW; CL vs. CD; CL, ALPHAW vs. CLM; CY, CLN, CSL vs. ALPHAW
 - F: CL vs. ALPHA; CL vs. CD; .CL, ALPHA vs. CLM

NOMENCIATURE General

SYMBOL	PLOT SYMBOL	DEFINITION
a		speed of sound; m/sec, ft/sec
c_p	CP .	pressure coefficient; $(p_1 - p_{\omega})/q$
М	MACH	Mach number; V/a
р		pressure; N/m ² , psf
q	Q(NSM) Q(PSF)	dynamic pressure; $1/2\rho V^2$, N/m^2 , psf
rn/l	\mathtt{RN}/\mathtt{L}	unit Reynolds number; per m, per ft
V	•	velocity; m/sec, ft/sec
α	ALPHA	angle of attack, degrees
β	BETA	angle of sideslip, degrees
ψ	PSI	angle of yaw, degrees
$oldsymbol{\phi}$	PHI	angle of roll, degrees
· p		mass density; kg/m3, slugs/ft3
	<u>R</u>	eference & C.G. Definitions
Ab 、		base area; m ² , ft ²
þ	BREF	wing span or reference span; m, ft
c.g.		center of gravity
$oldsymbol{\ell_{ ext{REF}}}{ar{ ext{c}}}$	LREF	reference length or wing mean aerodynamic chord; m, ft
S	SREF	wing area or reference area; m ² , ft ²
	MRP	moment reference point
	XMRP	moment reference point on X axis
	YMRP	moment reference point on Y axis
	ZMRP	moment reference point on Z axis
SUBSCRIP b 1 s t	<u>rs</u>	base local static conditions total conditions free stream

NOMENCLATURE (Continued)

Body-Axis System

SYMBOL	SYMBOL SYMBOL	DEFINITION
. C ^M	CN	normal-force coefficient; $\frac{\text{normal force}}{\text{qS}}$
$\mathbf{c}_\mathtt{A}$	CA	axial-force coefficient: axial force qS
$c_{\underline{\Upsilon}}$	CY	side-force coefficient; side force
c _{Ab}	CAB	base-force coefficient; base force qS -Ab(p_b - p_{∞})/ qS
$\mathtt{C}_{\mathbb{A}_{\widehat{\mathbf{f}}}}$	CAF	forebody axial force coefficient, \mathtt{C}_{A} - $\mathtt{C}_{A_{b}}$
C _m	CIM'	pitching-moment coefficient; pitching moment $qS \ell_{REF}$
c_n	CYN	yaving-moment coefficient; yaving moment qSb
c _l	CBL	rolling-moment coefficient; rolling moment qSb
		Stability-Axis System
C _I ,	\mathbf{CL}	lift coefficient; $\frac{\text{lift}}{\text{qS}}$
$\mathbf{c}^{\mathtt{D}}$	CD	drag coefficient; drag
c_{D_b}	CDB	base-drag coefficient; base drag
$\mathbf{c}_{\mathbf{D_f}}$	CDF	forebody drag coefficient; C_{D} - $C_{D_{b}}$
$C_{\underline{Y}}$	CY .	side-force coefficient; side force qS
C _m	CIM	pitching-moment coefficient; $\frac{\text{pitching moment}}{\text{qS}}_{REF}$
c _n	CLN	yaving-moment coefficient; yaving moment qSb
$^{\mathrm{c}}\!\ell$	CŠT	rolling-moment coefficient; rolling moment
r/d	r/d	lift-to-drag ratio; $c_{ m L}/c_{ m D}$
$\mathrm{L/D}_{\mathfrak{T}}$	L/DF	lift to forebody drag ratio; $c_{ m I}/c_{ m Df}$

NOMENCLATURE (Continued) Additions to Standard List

Symbo 1	Plot Symbol	Definition
BSTA		carrier fuselage station, in.
BWL .		carrier water line, in.
FS		fuselage station, in.
ⁱ ORB	IORB	Orbiter incidence angle, deg.
MS		model station, in.
S _{i-j}	Si-j	spoiler nos. i through j deflection angle, deg.
WL		waterline, in.
WP		water plane, in.
Хс	хс	carrier longitudinal station, in.
Хо	х0	Orbiter longitudinal station, in.
Yc	YC	carrier lateral station, in.
Yo	Y0	Orbiter lateral station, in.
Zc	ZC	carrier vertical station, in.
Zo	ZO .	Orbiter vertical station, in.
α _C	ALPHAC	carrier fuselage angle of attack, deg.
^α 0	ALPHA	Orbiter angle of attack, $\alpha_0 = i_{ORB} - 2^\circ + \alpha_W + \Delta i_{ORB}$, deg.
$^{\alpha}$ W	ALPHAW	carrier wing angle of attack, $\alpha_W = \alpha_C + 2^\circ$, deg.
βc	BETA	carrier sideslip angle, deg.
·β0	BETA	Orbiter sideslip angle, deg.

NOMENCLATURE (Continued) Additions to Standard List

Symbol	Plot Symbol	<u>Definition</u>
٠ وژ٦	DCL	incremental lift force coefficient (stability axis), DCL
$\delta C_{\overline{D}}$.	DCD	incremental drag force coefficient (stability axis), DCD
δC _m	DCLM	incremental pitching moment coefficient (stability axis), DCLM
δCγ	DCY	<pre>incremental side force coefficient (stability axis), DCY</pre>
δC_{η}	DCLN	incremental yawing moment coefficient (stability axis), DCLN
δC	DCSL	incremental rolling moment coefficient (stability axis), DCSL
δe	ORBELV	Orbiter elevator deflection angle, ORBELV, deg.
α_{I}	ALPHAI	nominal value of ALPHAW, ALPHAI ALPHAI = ALPHAW - sting balance correction
CL _T	CL-T	main balance strut tare lift force coefficient, CL-T
\mathtt{CD}_T	CD-T	main balance strut tare drag force coefficient, CD-T
c _{mT}	CLM-T	main balance strut tare pitching moment coefficient, CLM-T
c_{YT}	CA-L	main balance strut tare side force coefficient, CY-T
c_{nT}	CLN-T	yawing moment coefficient, CLN-T
$C_{L\overline{I}}$	CSL-T	main balance strut tare rolling moment coefficient, CSL-T

NOMENCLATURE (Concluded) Additions.to Standard List

<u>Symbol</u>	Plot Symbol	Definition
Δ ⁱ ORB	DIORB	change in Orbiter incidence due to support strut and balance deflections, deg.
^{δa} IL	AIL-IL	carrier left inboard aileron deflection angle, deg.
^{δa} IR	AIL-IR	carrier right inboard aileron deflection angle, deg.
^{δa} 0L	AIL-OL	carrier left outboard aileron deflection angle, deg.
^{δa} OR	AIL-OR	carrier right outboard aileron deflection angle, deg.
δ გ	AILRON	Orbiter aileron deflection angle, deg.
δe _{IB}	ELV-IB	carrier inboard elevator deflection, deg.
δe OB	ELV-OB	carrier outboard elevator deflection, deg.
δe	ELEVON -	Orbiter elevon deflection angle, deg.
⁸ BF	BDFLAP	Orbiter body flap deflection angle, deg.
$^{\delta}$ RL	RUD-L	carrier lower rudder panel deflection angle, deg.
^δ RU	RUD-U	carrier upper rudder panel deflection angle, deg.
δR	RUDDER	Orbiter rudder deflection angle, deg.
δSB	SPDBRK	Orbiter speedbrake deflection angle, deg.
δ_{WPP}	STAB	carrier horizontal -stabilizer deflection angle, deg.

CONFIGURATIONS INVESTIGATED

The test articles were 0.030-scale representations of the Boeing 747 carrier and Space Shuttle Orbiter vehicles. The carrier and Orbiter were tested both alone and mated. Carrier alone testing included a strut tare and interference investigation which utilized a dummy support strut. The BTWT external balance measured total force and moment data for the mated vehicle and carrier forces and moments for carrier alone testing. Orbiter forces and moments were measured by a six-component internal balance (660) mounted in the Orbiter.

The carrier model (no. AX 1319 I-1) represented the NASA 905 carrier aircraft which is a Boeing 747-100 aircraft with 200 ft. 2 vertical tip fins on the horizontal tail. Configuration designations used were:

$$K_1 = B_{27.8} M_{26.8}^{25} N_{57}^{58} T_{19} W_{44.1}$$

$$F_{20} = F_{8.1}^{20} F_{8.2}^{20} J_{11.10.2}^{60} J_{13.8.1}^{50} L_{9.8.3}^{70}$$

	1 27.0 20.0 07 13 11.1
	$F_{20} = F_{8.1}^{20} F_{8.2}^{20} J_{11.10.2}^{60} J_{13.8.1}^{50} L_{9.8.3}^{70}$
^B 27.8	fuselage
Fo	all flaps retracted at 0° deflection
F ²⁰ 8.1	inboard flaps deflected 20°
F _{8.2}	outboard flaps deflected 20°
H _{15.1A}	basic horizontal tail
H _{15.6}	horizontal tail with 200 ft. ² vertical tip fins
IS	image strut for support strut tare and interference studies
^J 11.10.2 ⁵⁰ J3.8.1	mid-span leading edge flaps deflected 60°
. ^{J50} 13.8.1	outboard leading edge flaps deflected 50°

CONFIGURATIONS INVESTIGATED (Continued)

 $L_{9.8.3}^{70}$ inboard leading edge Kruegers deflected 70°

M₂₅ inboard nacelle struts

M_{26.8} outboard nacelle struts

N₅₇ inboard nacelle

N₅₈ outboard nacelle

S₁₋₁₂ spoilers

T₁₉ flap track fairing

V_{9.1} vertical tail

W_{44.1} wing

The Orbiter model (no. 45-0) represented the space shuttle Orbiter configuration 140A/B lines. The following configuration designations were used:

ORB = $AT_{39} B_{26} M_{16} R_5 V_8 W_{116} E_{44}$

AT₃₈ Orbiter/carrier forward support strut

AT₃₉ Orbiter/carrier aft support strut

B₂₆ fuselage

M₁₆ OMS pods

N₂₄ MPS nozzles

N₂₈ OMS nozzles

R₅ rudder

CONFIGURATIONS INVESTIGATED (Concluded)

 ${\it TC}_5$ tail cone fairing

V₈ vertical tail

W₁₁₆ wing

E₄₄ slotted elevon

C₉ canopy

F₈ body flap

TEST FACILITY DESCRIPTION

The Boeing Transonic Wind Tunnel (BTWT) is a continuous flow, closed circuit, single return, atmospheric facility with the following characteristics:

Test Section Flow P	arameters	Test Section Dimens	ions
Freestream Condition	Range	Description	Value
Mach number	0 thru 1.15	Cross-section (minus	
Dynamic pressure, psia	0 thru 6.3	corner fillets), ft.	8 x 12
Static pressure, psia	15 to 5.4	Length, ft.	14.5
Stagnation pressure	atmospheric	Area, ft ²	88
Maximum unit Reynolds number, per foot	4 x 10 ⁶		
Maximum total temperature, °F	160		

The test section can be operated with either solid or slotted walls. The slotted wall configuration consists of 16 slots which can vary wall porosity from 3.5% to 11%.

Test data acquisition, recording, computations, and display are done by an XDS-9300 computer and Astro data sub-system.

DATA REDUCTION

Aerodynamic forces and moments measured by the external balance were reduced to coefficient form in body and stability axis systems, utilizing carrier reference dimensions:

Symbol	Description	Model <u>Scale</u>	Full Scale
\$.	747 wing area, ft. ²	4.95	5500
b	747 wing span, in.	70.44	2348
č	747 wing mean aerodynamic chord, in.	9.834	327.8
X _{MRP}	747 longitudinal moment reference point, in. X_c	40.197	1339.91
Y _{MRP}	747 lateral moment reference point, in. $Y_{\rm C}$	0.0	0.0
Z _{MRP}	747 vertical moment reference point, in. Z_c	5.7225	190.75

Aerodynamic forces and moments measured by the internal balance were reduced to coefficient form in body and stability axis systems utilizing both the carrier reference dimensions (described above) to create one set of data, and Orbiter reference dimensions to create a second set of data. Orbiter reference dimensions are:

Symbol	Description	Model <u>Scale</u>	Full Scale
S	Orbiter wing area, ft. ²	2.42	2690
Ь	Orbiter wing span, in.	28.10	936.68
Ĉ	Orbiter wing mean aerodynamic chord, in.	14.244	474. 81
X _{MRP}	Orbiter longitudinal moment reference point, in. X_0	33.270	1109.0

DATA REDUCTION (Concluded).

		_Model Scale	Eull - Scale
YMRP	Orbiter lateral moment reference point, in. Y_0	0.0	0.0
Z _{MRP}	Orbiter vertical moment reference point, in. Z _o	11.25	375.0

Support strut tare and interference corrections were applied to carrier data. These corrections (datasets RGMCO1-RGMC18), were obtained by subtracting image strut in minus image strut out datasets (RGMOO1-RGMO38).

Table IV summarizes the data set notation used for different types of data.

TABLE I .

TEST : BTWT 1431	(CA5)		DATE: 4/25/75
	TEST CO	INDITIONS	
MACH NUMBER	REYNOLDS NUMBER (per foot)	DYNAMIC PRESSURE (pounds/sq.ft)	STAGNATION TEMPERATURE (degrees Fahrenheit)
0.15	.98 x 10 ⁶	32.8	106
0.30	1.75·x 10 ⁶	126.0	130
0.50	2.75 x 10 ⁶	312.1	145
0.60	3.125 x 10 ⁶	418.2	147.5
0.70	3.575 x 10 ⁶	523.3	148.5
	,		,
		<u> </u>	
		1	
-			
BALANCE UTILIZED:	Main Bal - BTWT Ex Internal Bal - Boe		-
	CAPACI"	ТҮ	COEFFICIENT
NF	MAIN 10,000 1b	· IŃTERNAL 1780 lb	TOLERANCE:
SF	5,000 lb	1335 1b	
AF	1,000 lb	140 1b	
PM	100,000 in-lb	4266 in-1b	
RM	25,000 in-1b	2014 in-lb	
YM	25,000 in-1b	2014 in-1b	
COMMENTS:			
	37		

TABLE II.

		•						ABL	- Ť	1.										
TEST: B	TWT 1431 CA-5		[DATA	SEŤ	· /RU	N NL	JMBE	R	COLI	_ATIO	n sumi	MARY		DATE	; <u>7-</u>	ノーウ、	5		_
	1	T 66	HD.	<u> </u>				DAR	ME	TERS	/VAĽU	ES .			1	M	ACH NL	MBERS	4,0	\neg
DATA SET	I CONFIGURATION			STAB	Ř99-	ELV-	\$17.				IORB				.15	.30	.50	. 60	.70	
was a superior of the superior		14>	0	-2	0	9/3	0	120	7	0					31	17	16.	15	13	
	K1+F0 H15.1A V9.1 + IS	-				1	 	┪	+	Ť				+	1	/8	19	20	21	
02		2	<u> </u>	-2	0		╁╌┾	╅	-				+	 	30	25	24	23	22	
03		6	Ê	-2	0		╂╼┼╸	+	_	_		 :	 	 	100	29	28	27	26	
04	Y	8	Â	-ż	0	<u> </u>	┼┼-		+	<u> </u>		-	 	 	_		 · · · · · · · · · · · · · · · · · · ·	36	35	ŀ
. 05	K1+F0 V9.1 + IS	顶	0		0		<u> </u>		_		<u> </u>	ļ	<u> </u>	-	.34	38	37			!
. 06	K1+F0 H15,1A + IS	2	◬	-2		0/3						ļ	<u> </u>		ļ	42.	41	40	39	TES
07	,	6	A	-2									<u> </u>		47	4-6	45	44	43	77
08	V .	8	A	-2									_			51	50	49	48	ž
1 1	K1+H15.1A F20 + IS	4	Â	+1												57	56	.55	54	Z
W 10		4	A	+1												61	60	59	58	BEF
 		6		+1_		H^-	+	++	\dashv	+		 	 	1		65	64	63	62	55
<u>} </u>		8		+1			++	++	+	_	\vdash	1	†	- 	1	69	68	67	66	
12	<u> </u>	+	1	7.		1	+	+	+	+	╁┼┼	 	-	 	-	73	72	7/	70	
-	K1 + F20 Vq.1 + IS	1.4>	0		0	-	H	+++	_			<u> </u>		 		77/18	1	75/90	74	
	K1+F20 H15.1A V9.1 +IS	ΙÀ	0	+1	٥	ं/3	+-		-		 - 	 	-		91	1/8	, , ,	170	79	
<u>; 15</u>		-3,4		+1	0	-	 		_	<u> </u>			4	 		-	27	81/85	80	ł
. 1,6		4.25		+1	0	<u> </u>	4	. .	_ _		1	ļ	↓	<u> </u>	<u> </u>	83	 	 		
17		6	2	+1	Ò								4		_	89	88	87	86	ľ
Y 18		8		÷Ť	Ó	Ψ.		1.1		<u> </u>	<u> </u>					95	94	93	92	
,	7 13 19			25	•	31		37			, 43 ``	4	19 .	55		61	•	(BE 67(AL)	TH) -	5 76
C, L, IC, D,	ICILM CYLICIEN, CIST	-, D	,I,O	$R_{i}B_{i}$	DCL	-1 1 ^D	C.D.	ع الا	LN			CLN	ی,۵٫۵	4.19	55	MAC	H	ALP	HAW	
CDT, CLMT,	CYT, CLNT CSLT = W = A FOR A DATASETS	=1/3	-2-2	712,	7=10		OEF	FICE	JTS			L	14/4.4	B a A.	25 N 58	IDV	AR (1)	16VA	R (2)	ИDV
•	$Q = \Delta = 12$	ر مر=	<u> </u>	$\frac{3}{4} - \frac{3}{4}$	-2 -	1.0	1,2,	3, 4.	6.	8.10.1	2	F20 =	F20	F 20	J11.10.2	J 50	1 49.5	7.3		_
, SCHED	DULES . 8 - 2 - 12,-10	ره - د	٠,	, ,	<u> </u>	., 5,	., -,	_ , , ,	٠,٠	.,,,,,		20.		<u> </u>	V (1.10.6	- 10.0.	0087			

FO = FLAPS RETRACTED 0; HIS.1A = BASIC HORIE. TAIL; HIS.6 = HORIE. TAIL WEOD FC VERT. TIP FINS; V9.1 = VERT. TAIL; AT38 = ORB/CARR. SUPPORT STRUTS

TC5 = TAIL CONE FAIRING; Ne8 = OMS NORTLES; Ne4 = MRS NORTLES; IS = IMAGE STRUTS; S = Spoiler DEFLECTIONS; IORB = ORB. INCIDENCE;

STAB = STAB, DEFLECTION; RUD-U/L = RUDDER DEFLECTION; ELV-IB/OB = ELEV. DEFLECTION

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TABLE II. - Continued. DATE: 7-1-75 TEST: BTWT 143/ CA-5 DATA SET/RUN NUMBER COLLATION SUMMARY MACH NUMBERS PARAMETERS/VALUES SÇHD. DATA SET CONFIGURATION α β STAB RUD- ELV- S172 S3.4 S-8 IORB DENTIFIER .1.5 .3 . 5 .61 .7 99 100 98 0/3 101 130 0 102 RGM019 K1 F20 V9.1 H15.1A 0 0 0 0 +1 106 105 A 104 103 +1 ٥ 20 109 107 2 110 108 +1 0 21 8 114 /// +1 //3 112 0 22 115/116 14> 118 117 0 0 23 KI F20 V9.1 4 0/3 122 121 120 24 KI F20 H15.1A Æ /23 126 124 125 6 A 127 25 +1 128 +1 2 131 130 /29 26 134 2 2 736 /35 27 KI FO H15,1A +1 /37 142 138 6 141 140 139 (à +1 28 145 144 143 8 A +1 146 29 148 147 1A> 0 150 149 15/ O 30 K1 FO V9.1 155 154 152 153 0 +1 14> 9/3 156 31 K1 FO V9-1 H15.1A 0 159 158 157 14> 160 0 -2 ٥ 161 32 165 164 162 2 2 -2 163 0 33 166 6 E 169 168 167 -2 Ø 170 34 174 173 172 171 8 -2 0 35 177 176 175 178 2 2 -2 36 KI FO H15.1 A 75 76 37 € 1 13 19 31 43 49 55 25 IE JAR 12 YOV COEFFICENTS IC CAR (1) a OF B SCHEDULES

	DATA SET	<u> </u>	431 CA-5	ا	нD.	DAT.	A SE	T/RU	N NL		-	LATION ZVALUES		RY	10010	; 7		UMBERS	
	ENTIFIER	CO	NFIGURATION .	ď		STAB	200-	製品	S1,32	S3,4	55-8	IORB		-	.75	المراجع والمساولين	.5	.6	1.7
{ RG	-M037	K1 F	6 H15.1A	6	A	-2		9/3	0	0	O				/83	782	181	180	179
	38	١	,	8	æ	-2	_	9/3			:					187	186	185	184
	39	K1 F	~ Y9.1	A	0	_	0								194	193	192	191	189
	40	K1 Fo	V9.1 H15.1 A	A	0	-2		9/3							202	20/	200	199	198
L	41	١	,	6	Æ.	-2									. 207	206	205	204	203
	42	K1 Fo	V9.1 H15.6	A	0	-2									212	2//	210	209	208
	43	·		6	A	-2									217	216	215	214	213
	44			A	0	+ L				Ý	Ý.				222	155	220	219	218
<u> </u>	45		· · · · · · · · · · · · · · · · · · ·	A	٥	+ 1				45	20				, in the second	<u> </u>	224	223	
	46			◬	+5	+1		1		;						<u> </u>	226	225	
	47			A	0	+1		19/13				1				<u> </u>		227	
	48			A	0	+ 1	Y	-19/-7									·	228	<u> </u>
	49	\		Æ.	0	+ 1	15/15		<u> </u>	!						<u> </u>		229.	<u> </u>
	50	K1 F0 }	9.1 HIS. 6AT38 AT39	6	A	+1	15/15	ļ								<u> </u>	233	232	<u> </u>
	51			6	ඣ	+ 1	0	⁶ /3								ļ		235	
<u> </u>	52	<u> </u>	•	A	0	+ T	0	0/3								240		238	<u> </u>
	53	KIF019.	1 H15.6 AT38,1 AT39	Â	0	+ 1	0	9/3	Y.	<u> </u>	<u> </u>					244	243	242	
	V 54	K1 Fo	19.1 H15.6	A	0	-1.9	0	0/3	0	45	20	l y l				<u> </u>	246	245	name (
<u> </u>		•	13 19			25	···	31		37		43	49		55	61		٤٦	
-	11111	<u></u>	1111111	لسلسة	لسيا	ــــــــــــــــــــــــــــــــــــــ		- ا	OE F F	ICEN*	<u></u>	ــــــــــــــــــــــــــــــــــــــ				1111	AR (1)	1c.A	P 12
-	α OR	β	· · · · · · · · · · · · · · · · · · ·																

TABLE II. - Continued.

E31:01	WT 1431 CA-5			DAT	A SE	T/RU	N NU	МВЕ	R CC)LL/	ATION	i SUMI	MARY		DATE	· 7-	-/-/	<u> </u>	
DATA SET	CONFIGURATION		НĎ.		-dus 1	I ELV-					/ALUE	S	7	www.com	<u> </u>	THE RESERVE AND ADDRESS OF THE PERSON NAMED IN COLUMN	CHARLES NAMED IN	UMBERS	; M emorous cas
			T-^~		UZL		r — —	T	-		IORB	**********	<u> </u>	-	.15	.3	.5	, 6	247
	K1 F0 H15.1A	6	 			 	0		0	1	<u> </u>		ļ	 	251	250	249	 	
	K1 F20 H15,1 A	6	 			ļ		-			_			-		ļ <u>.</u>	255	254	253
	KI F20 H15,1A V9.1	6	 			 		-	- -	_			ļ	<u> </u>		ļ	258	257	256
58	<u> </u>	A	0	+1	0		-	 	4-4	_					264	263	262/265	261	260
	K1 Fe0 H15.6 V9,1	A	٥	 ♣ +1 0 9/3							269		268	267	266				
60		6	<u>A</u>		1	1			· · · · · · · · · · · · · · · · · · ·			2		272	271	270			
61		A	0	+1	0	9/3	45	45	45	5							273	<u></u>	<u> </u>
62	Y	Â	0	+1	0	19/13					¥				County of		274		
63	K1 F20 H15.6 Y9.1 + ORB	Â	0	+1	0	19/13				5	3.7	•					277	278	279
64	+ TC5 + AT 38./	A	A 0 +1 0 9/3 V V A 0 +1 0 9/3 O V A 0 +1 0 9/3 O O B A +1 0 9/3 O O B A +1 0 9/3 O O			T						282	281	280					
65		A.	0	+1.	0	9/3			11	_				 	1			284	283
66		<u> </u>	0	4 <u>1</u>	0	9/3	-		†	_	1 1			 			285		
67		6		—	0	0/3	_		1	\dashv	1 1						288	287	286
68		A				0/3			┿	十							291	290	289
69		6	<u> </u>			0/3			++		++			<u> </u>			296	295	294
70		A	\vdash				<u>L</u>	W	120					 			299	298	297
		<u> </u>	H								1 1			 					300
7/	1		 							\dashv	! !	<u> </u>	,	 	306	 	302	36/	303
72	Y	8	A 0 +1 0 9/3 V V V A 0 +1 0 9/3 V V V A 0 +1 0 9/3 V V V A 0 +1 0 9/3 A 5 A 5 A 5 A 0 +1 0 9/3 A 6 A 7 A 0 +1 0 9/3 A 9 A 0 9/3 A 9 A 0 9/3 A 9 9		1	**************************************		-	306		305	304	305						
	13 19			43	Э	49	·	55		61		٤ 7	<u></u>						
ابتبلت			لـــــ	سنا			1-1-			11		<u>l</u>		<u>. L.</u>		سبا		ببيل	لحب
α OR	β			O +1 0 0/3				ORE	3= W/	16 B26	V8 R5	M16	4739 L	44	+ TS0				

TABLE II. - Continued.

TES	T: <i>BT</i>	WT /	431 CA-5	1		DAT.	V (E.	_			COLL	LATION	ı cılıaı	LADV		DATE	; ₇ -	・ 「+フ	ى	·	•
		7		<u>.</u> .		DAJI	4 JE	17KU		·				IAKI		T	3/4 / / Marie	AACI NI	JMBERS		
	TA SET ITIFIER	(2)	NFIGURATION .	o SC	нD. В	STAB	RUP-	ELV-				/VALUE	228 228			1.15	1.3	1.5	.6	·7	
Rei	7073	K1 F20 F	115.6 Vq.1 + ORB + AT38.1	6	A	+1	0	9/3	C)	ا م	0	9.7				310		309	308	307	
	74		HI5.1A Vq.1 + ORB + AT38.1		0	+1.	0	0/3	0	0	0	9.7						3/3	312	311	
	75	1		A	0	+1	0	9/3	45	45	45	9.7		·		Ì		316	.315	314	
	.76		,	6	2	+1	.0.	0/3	45	45	45	9.7					<u> </u>	319	318	3.17	
	フフ	K1 F20 V9	.,+ORB+TC5+AT38.1	A	0		0	_	45	45	45	9.7					323	32,2	321	320	
		}	.IA+ORB+TC5+AT38, 1	-3	A	+1		ं/3	45	45	45	9.7				327		326	325	324	<u> </u>
	79	KIF20 HIS	6 V9, 1 +088 +TC5+AT	, 6	2	+1	0	0/3	4-5	45	45	7.85						330	329	328	15
	20	,	,	Æ.	0	+1	0	0/3	45	4-5	45	7.85						333 335	332 334	331	RUN
	18	K1 F0 H15.4	.V4.1+088+TC5+AT39.2	A	0	+1	0	9/3	0	45	20	7.85				340	339	338	337	336	Z
	82			2	2	+ T	0	0/3	0	45	20	7.85				345	344	343	34-2	34/	1967
	83			6	È	+1	0	0/3	0	45	20	7,85	,			350	349	348	347	346) [©]
	84	·		A	0	-2	0	9/3	0	45	20	7.85					354	3 <i>5</i> ,3	352	351	
	85	,		Æ.	0	+1	0	10/13	0	45	20	7.85					358	357	356	355	<u> </u>
	86	. \	/	6	Æ	+1	10/10	%3	0	45	20	7.85	Ý				362	36/	360	359	
	87	K1F0H15.6	V9.1+088+TC5+AT38 2	A	0	+1	%	93	0	45	20	7.85	5					364	363		
	88		·	6	ඬ	1.+	%	9/3	0	45	20	7.85	5	., . ,				366	365		
	89			6	â	+1	0	0/3	.0	0	0	7.85	٥			•	370	369	368	367	
Y	90	\		A	0	+ <u>1</u>	0	0/3	0	0	٥	7.85	Ø				37 4	373	372	37/	
	•	7	13 19			25	···	31		37		43	49		55		61		t. ⁷		5 76
	4444	4, 1, 1, 1,	<u> </u>		ليب			ببا	- . 	بب	- 	1111	سلس		ll		حيبا	<u> </u>	10 . At		1EV
	α OR SCHEDU	•						C ·	DEFFI	CENT.							101	AR (IÌ	IL , A	114	

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TABLE II. - Continued.

TE	ST : 87	WT 14.	3/ CA-5			DAT.	A SE	r/RU	N NU	мвег	R COLI	LATION	SUMMARY	·	DATE	; <u>7</u> -	1-7	5		
D.	ATA SET	COV	FIGURATION		нD,					PARAN	/ETERS	/VALUE	3		T	۸	IACH N	UMBERS	*************************************	-
IDE	NTIFIER			α¥	β	STAB	RUP- U/L	ELV- IB/OB	51,78	9716	S-8	IORB			.15	-3	, 5	.6	1.7	T
RG.	MO91	KIFOHI +Neg+N	5.6 Vq. 1 + ORB 24 + AT38,2	Â	0	-2.	0	9/3	0	45	20	7.85				378	377	376	375	
	92			â	0	+1	0	9/3	0	45	20	7,85				385	381	380	379	decoposado
	93			6	4	+1	0	9/3	0	45	20	7.85				386	385	384	383	1
	94			6	à	+1	10/10	9/3	0	45	20	7.85				390	 	388	387	1
	95	,	ļ	A	٥	+1.	%	19/13	0	45	20	7.85				394	393	392	391	1
	96	KIFO HIE.6	Vq.1+ ORB+TCS+ AT 39.2	Â	0	+1	%	9/0	0	45	20	9,7				398	397	396	395	4
	97			6	A	+1	%	90	0	45	20	9,7				402	401	400	399	LST
	98	K,F0 V9.1+C	X8+TC5+AT58.2A	Æ.	0		0/0	A-14-0	0.	45	20	7.85			407	406	405	404	403	1 2
	77 1	+N20+N2	ORB + AT38,2 A	幽	0		9/0		0	45	20	7.85				411	410	409	408	Ž
	100	KIROHISIA + Neg +	+ ORB + FT38.2 A	6	<u>a</u>	-1		G/8	0	45	20	7.85				415	414	4/3	412	X D
1	101		+ORE-TC5+ AT 38.2 A	٤	Ê.	-1		%	0	45	20	7.85	~~~	····		419	4-18	417	416	12.
1	102			6	2	-1		%	0	45	20	7.85		_	424	423	422	421	420	
	/03	v	^	6	iż.	+1		%	0	45	20	7.85			429	428	427	426	425	
	104	KIFOHIS.IA+	-0RB+ TC5+ AT38.3	2	ريج	-2		9/3	0	0	0	4.0		1		<i>4</i> 35	431/ ₄₃₄	433	4-32	
	105			Ġ	نغ	-2		9/3	0	0	0	4.0		1	440	4-39	438	437	436	
;	106	Y		8	Ź	-2		0/3	0	0	0	4.0				444	443	442	441	differ market
	107	(2/oVq,,+0R	8+7C5+AT39.3	A	٥		0/0		0	0	0	4,0		 	449	448	447	446	445	
Y	108	CLFOHIS,IAK	7.1+0RB+TC5+AT 78.3	A	0	-3,2		%	0	0	٥	4.0				4 53	452	451	450	
	7		13 19		2	5		31		37		43	49	55		6 J	MARKET MENTAL STATE	Ł. ⁷	-	5 7 (
	4.1.1.4	<u> </u>	<u> </u>					<u> </u>	L. .	LL		LLL				لسا		Li	ıl	- <u>i</u> -
	a or Schepu	•	***************************************	·*********		· · · · · · · · · · · · · · · · · · ·		co	E F F I	CENT					,	FDVA	(R·I)	IÜ , AF	₹ 12 .	, C \

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TABLE II. - Continued.

TE	S ⊺: <i>B⊤</i>	WT 143	1 · CA-5	Ì	,	DAT.	A SET	Γ/RU	N NL	JMB	ER	COLL	.ATI	ON SUMMARY	DA	TE	フ -	/- フ	5		
D	ATA SET	CONTI	GURATION	sc	HD.							ETERS,				70 year-	Ą	IACH N	JMBERS		
	NTIFIER			α	β	STAB	RUP- U/L	TE/08	11) 15 21 15	§	10	z=-8	IOR	В		15	.3	.s	1.6	1.7	
RG	M/09	K1FoH15,1AVq	1+0RB+TC5+AT 38.3	2	a	-3.2	0	%3	٥		2	0	4.0)			457	456	455	454	1
	110			6	æ.	-3.2	0	0/3									461	460	459	458	
	[[[/	8	A	-3.2	Ø	9/3									465	464	463	462	
	112,	k1F0H15.6V4.1+	ORB+TC5+AT38.3	A	0	-3.2	Ö	9/3							4	70	469	468	467	466	
	113			2	▲	-3.2	0	%3							4	75	474	473	472	471	
	114			6	A	-3.2	0	9/3							4	84	4-79	478	477	476	<u>ן</u>
	115			8	A	-3.2	0	9/3				,					4 83	482	481	480	EST
	116			A	0	+1	0	9/3				-;				•	488	487	486	485	R V V
	117			A	0	- 2	0	0/3		\Box		!			1 4	793	492	491	490	489	Z
	118			6	<u> </u>	-2	0	%			1				4	-98	497	496	495	494	N.B.E
	119		· · · · · · · · · · · · · · · · · · ·	A	0	-2	0	-19/7			1						502	50/	500	499	ਨ
	120			A	0	-2	0	10/13				1			202		506	505	504	503	١.
	121			6		-2	19/10	%				!					509	510	508	507	
	122			2	A	-2	10/10	<i>0</i> /3									514	513	512	51/	
	123		,	8	A	-2	10/10	%		1				•			518	517	516	515	
	124	Ý		6	Δ	-2,	10/10	°/3									585	524	523	255	
	125	KEFOHIS.IAY9.	1 + OR8+7C5+ AT39, 3	Ø.	0	-2	%	%		:		1					529	528	527	526	
7	126	KIFOHI5.6V AT38.3 + NE	9.1 +ORB + 28 + N24	A	O	-5	0/0	0/3	Ϋ́	Τ̈́	1	Ÿ	V			,	533	532.	53/	530	
)	7		3 19		2	25		31		37			43	49 _	55		61		£ 7	-	5 76
	1111	لببيا	<u>,1</u>		<u>ــــــــــــــــــــــــــــــــــــ</u>			بالم	<u> </u>	<u></u>	<u> </u>	ــــــــــــــــــــــــــــــــــــــ			1111	1.1	حينا	<u> </u>	ببيا		
	a OR SCHEDU	-						C)E F F	ICEN	175						IDV	AR (1)	1A, 31	? (2, '	1 L V

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TEST: BT	NT 1431 CA-5			DAT	4 SE	T/RUI	וטא א	MBER	R COLI	_ _AOITA_	SUMMAR	₹Y	DATE	; 7	7- /-	75		_
DATA SET			нD.							/VALUE	S			٨	ACH N	MBERS	· · · · · · · · · · · · · · · · · · ·	
DENTIFIER	CONFIGURATION	ď	β	STAB	RUP U/L	INOB	S1,2 11,12	33,4 95/0	ჰ₅-გ	TORB			.15	.3	. 5	,6	.7	
RGM127	KI FO HIS, 6 V9.1 + ORB+ AT38, 3 + NaR + Na4	6		-2.	%	9/3	ò	0	0	4,0				538	537	}	535	
/ż8	ORB+ N28+ NE4	A	0										<u> </u>	543	542	ļ	540	
129		A	0				· 		1						545			
/30		8	ئے						;				ļ	549	548	547	546	4
/3/	<u> </u>	15	<u> </u>											553	552	551	550/55A	ă.
/32	ORB + TC5	◬	0											558	557	556	555	4
183		.8	12,			.			!					562	56/	560		
Y 134	<u> </u>	15	Œ.	¥	Y	V	<u> </u>	Ϋ́	Ý	Ÿ				566	565	564	563	1
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	13 19 25 21 37 43 49 55 (1 € 75.76) COEFFICENTS COEF																	
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	13 19			25		31	·····	37	<u>, , , , , , , , , , , , , , , , , , , </u>	43	49	55		£I		د٦		5
	 		لــــــــــــــــــــــــــــــــــــــ							1111				سبا	17.1	سبا		
a OR	B d(B)= +11.	حزسہ	+2	6		Ç ()E	CENT	; 					16\ 	AR (1)	1A 7:	7 (2 '	10
SCHEDU	\int_{ES} $\alpha(C) = -3$	2+2	0 B	Y /	, 20	->2	6 B)	, 2	0							, ,		_

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TABLE III. - MODEL DIMENSIONAL DATA

a. Carrier Model

MODEL COMPONENT: B27.8	- Sesseries is their at Sone & A more	
GENERAL DESCRIPTION: BODY, 747 Proj	ect with A.P.V. e	xit.
		•
MODEL SCALE: 0.03		
DRAWING NUMBER: 65-69716	•	
DIMENSIONS:	FULL SCALE	MODEL SCALE
Length, in.	2702	81.06
Max. Width, in.	255	7.65
Max. Depth		
Fineness Ratio		
Area		
Max. Cross-Sectional		
Planform		<u> </u>
Wetted, ft ²	424	.3816
Base		

MODEL COMPONENT : Fô		
GENERAL DESCRIPTIONClean Wing	,	
Flaps up	and the second s	
		\
DRAWING NUMBER		
DIMENSIONS	FULL SCALE	MODEL SCALE
·Area		
Span (equivalent)		
inb'd equivalent chord		
Outb'd equivalent chord	Market Market Control of the Control	
Ratio movable surface chord/ total surface chord	***	Name of the last o
At Inb'd equiv. chord	**************************************	
At Outb'd equiv. chord		
Sweep Back Angles, degrees		· · · · · · · · · · · · · · · · · · ·
Leading Edge		
Trailing Edge		
Hingeline		##
Area Moment (Normal to hinge line)		4,000

MODEL COMPONENT F20 8.1		
GENERAL DESCRIPTION _ Flap inboard		emporate 5 II - specificane in in
deflected 20°	•	
,		
•	•	
DRAWING NUMBER . 5.0. 1319-4 thru	- 10 & -38	,
DIMENSIONS	FULL SCALE	MODEL SCALE
Area		
Span (equivalent)		WBL 3.854 to WBL 13.351
Inb'd equivalent chord		,
Outb'd equivalent chord	-	• •
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord		
At Outb'd equiv. chord		•
Sweep Back Angles, degrees	·	
Leading Edge		
Trailing Edgé	The state of the s	***************************************
Hingeline	***	44.
Area Moment (Normal to hinge line)		

MODEL COMPONENT : F8.2		
GENERAL DESCRIPTIONFlap, Outb	d deflected 20°	
DRAWING NUMBER 5.0 1319-20 thru	u -25 & -42	
DIMENSIONS	FULL SCALE	MODEL SCALE
Area .		·
Span (equivalent)		WBL 15.450 to WBL 24.660
Inb'd equivalent chord		
Outb'd equivalent chord		
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord		
At Outh'd equiv. chord		
Sweep Back Angles, degrees		
Leading Edge		-
Trailing Edge		
Hingeline		
Area Moment (Normal to hinge line)		

MODEL COMPONENT: Horizontal Tail - H15.1A		
GENERAL DESCRIPTION: Swept Horizontal Tail	With	
Planform Radius Fillet at L.E Body Inte	rsection.	
MODEL SCALE= 0.03		
DRAWING NUMBER: 65-66630, 69-49180, 1007-4	77	
<u>DIMENSIONS</u> :	Full Scale	Model Scale
EXPOSED DATA (one side)		
Area- ft ²	<u>736</u>	0.6625
Span- in.	436,5	13.095
Aspect Ratio		
Taper Ratio	·	
Dihedral Angle-deg.	7	7
Incidence Angle-deg.		
Sweep Back Angle-deg. L. E.	43.08	43.08
Chords-in.		
Root	388	11.64
Tip	<u> </u>	2.91
MAC	•	
Apex Location-in.		
X _o ·		
Yo		
Z _G		
.25 MAC Location-in.		
X _C Station	2564	76.920
Y.o		
Z _o		····

MODEL COMPONENT: Horizontal Tail	H15.6	
GENERAL DESCRIPTION: Horizontal	tail with vertical fins	
on each tip at body B.L. 12.82	······	•
MODEL SCALE: 0.03		
DRAWING NUMBER:		
DIMENSIONS:	Full Scale	Model Scale
Exposed Data (one side)		
Area - ft ²	200	0.18
Span - in.		•
Aspect Ratio		-
· Taper Ratio		
Dihedral Angle-deg.		
Incidence Angle-deg.		
Sweep Back Angle-deg.		·
Chords' - in.		

MODEL COMPONENT :		
GENERAL DESCRIPTION Middle L.E. STat	deflected 60°	
Model Scale: 0.03		•
DRAWING NUMBER 5.0.1319-26, & -44		
		, *
DIMENSIONS:	FULL SCALE	MODEL SCALE
Area		
From WBL 15.389 to MAC Strut Span (equivalent), in.	434	13.022
Inb'd equivalent chord		
Chord	31.6	948 -
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	,	
At Outb'd equiv. chord		•
Sweep Back Angles, degrees		
Leading Edge		
Trailing Edge		
Hingeline	· ·	
Area Moment (Normal to hinge line)		

MODEL COMPONENT		
GENERAL DESCRIPTIONOutboard L.E.	Slat deflected	50°
	•	
Model Scale: 0.03		
DRAWING NUMBER		
DIMENSIONS	FULL SCALE	MODEL SCALE
` Area		
(From WBL 26.374) Span (equivalent),in	360,5	10.814
Inb'd equivalent chord	•	
Outh'd equivalent chord, in	24.1	.722
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord		·
At Outb'd equiv. chord		·
Sweep Back Angles, degrees		
Leading Edge		
Trailing Edge		
Hingeline	·	
Area Moment (Normal to hinge line)	-	

MODEL COMPONENT: L9.8.3		
GENERAL DESCRIPTION: Inboard Leading.	edge_Krueger-f-lap;-def-le	cted-70°
Model Scale: 0.03	•	
Drawing NUMBER: 5.0. 1319-26 -44		
		- · · · · · · · · · · · · · · · · · · ·
DIMENSIONS:	Full Scale	Model Scale
Span from WBL 7.322 to Nacelle Strut, in	304.8	9.144
Inb'd equivalent chord, in	315.3	9.46.

MODEL COMPONENT: M25			
GENERAL DESCRIPTION: Inboard 747,	J _{T9D} Nacelle	•	
strut			_
Model Scale: 0.03			
DRAWING NUMBER			
DIMENSIONS:	FULL SCALE	MODEL SCALE	
Wing B.L. of nacelle β, in.	470.0	14.100	
Cont angle deg inboard	_ 2	2	

MODEL COMPONENT: M26.8	<u>.</u>	<u>. </u>
GENERAL DESCRIPTION: Outboard 747, & JT9D Nacel'le	·	<i>,</i>
Strut	,	
Model Scale: 0.03		
DRAWING NUMBER 937-590	·	,

TABLE IIIa. - Continued. MODEL DIMENSIONAL DATA

MODEL COMPONENT: N57	·
GENERAL DESCRIPTION Inboard Fan Cowl and Primary	
747 Nacelle, flow thru type	
Model Scale: 0.03	
DRAWING NUMBER 5.0 1007-96, 97	

TABLE IIIa. - Continued. MODEL DIMENSIONAL DATA

MODEL COMPONENT: N ₅₈	
GENERAL DESCRIPTION: Outboard fan cowl and primary	
747 Nacelle, flow thru type	
Model Scale: 0.03	
DRAWING NUMBER 5.0. 1.007-96, -97	

TABLE IIIa. - Continued.

MODEL COMPONENT: SPOILERS - S1-12			
GENERAL DESCRIPTION: Multi-panel flight spoilers. Four outboard and			
two inboard spoilers per side. Subscrip	t denotes spoiler	panel ^S l is the	
most outboard L.H. panel and \$12 is most			
747 MODEL SCALE: 0.030	MODEL: 1065	-	
DRAWING NO.: 65-71450, S. O. 1065-51,	-59, -81, -173		
DIMENSIONS: (one panel)	FULL SCALE	MODEL SCALE	
. Outboard S_{1-4} and S_{9-12} (Ft ²)	21.48	0.0193	
Span (equivalent), in.	6.25	0.1875	
Chord, in.	3.44	0.1032	
Inboard S_{5-6} and S_{7-8} (Ft ²)	35.31	0.0318	
Span (equivalent), in.	7.50	0.225	
Chord, in.	4.71	0.1413	

MODEL COMPONENT: T19		
GENERAL DESCRIPTION: Flap track fai	rings,	
4 on each side.	-	
Model Scale: 0.03		
DRAWING NUMBER <u>5.0. 1007-403</u>		
•	• •	
DIMENSIONS &	Full Scale	Model Scale
	235.3	7.06
WBL of Track no. 1, in.		•
WBL of Track no. 2, in.	353.	<u>·10.59</u>
WBL of Track no. 3, in.	652.	19.56
WBL of Track no. 4, in.	743.6	22.31
Distance from wing trailing edge, in.	50.	1.5

MODEL COMPONENT: V9.1	*	**
GENERAL DESCRIPTION: Swept Vertical Ta	il	
Model Scale: 0.03		
DRAWING NUMBER: 65-6.9716; 1007-26, -61	0; 937-319	
DIMENSIONS:	FULL SCALE	MODEL SCALE
TOTÁL DATA		
Area (Theo)- Ft ²	830	.7470
Planform Span (Theo) - In. Aspect Ratio Rate of Taper	386.5	11.595
Taper Ratio Sweep-Back Angles, Degrees Leading Edge Trailing Edge 0.25 Element Line Chords: Root (Theo) WP-in. Tip (Theo) WP-in. MAC Fus. Sta. of .25 MAC	50.12 461.67 157 2529.6	50.12 13.85 4.71 75.888
W. P. of .25 MAC B.L. of .25 MAC	· · · · · · · · · · · · · · · · · · ·	•
Airfoil Section Leading Wedge Angle-Deg. Trailing Wedge Angle-Deg. Leading Edge Radius Void Area		
Blanketed Area		

TABLË IIIa. - Concluded.

GENERAL DESCRIPTION: Swept 747 Wing		
MODEL SCALE: 0.03		
	DWG. NO.:	65-89585
DIMENSIONS:	FULL SCALE	MODEL SCALE
ÒTAL		
Area (Theo.) Ft ²	· 5500	. 4.95
Planform	2348	70.44
Span (Theo. In.)	6.96	6.96
Aspect Ratio	<u> </u>	
Rate of Taper		
Taper Ratio		.7
Dihedral Angle, degrees		
Incidence Angle, degrees		
Aerodynamic Twist, degrees		
Swept Back Angles, degrees		<u>.</u>
Leading Edge .	· .	
Trailing Edge		• ,
0.25 Element Line		
Chords:		
Root (Theo) B.P.O.O.		
Tip, (Theo) B.P.		-
MAC	327.8	9.834
Fus. Sta. of .25 MAC	1339.91	40.197
W. P. of .25 MAC	190.75	5.7225
B. L. of +25 MAC		
		

TABLE III. MODEL DIMENSIONAL DATA

. b. Orbiter Model

MODEL COMPONENT: ATTACH STRUCTURE - AT38

GENERAL DESCRIPTION: Orbiter forward attachment struts.

MODEL SCALE: 0.030

DRAWING NO.: Boeing 1319-43

DIMENSIONS:

HEIGHT OF ACTUATOR RED TIP ABOVE STRUT UPPER JUNCTION, IN.

SYMBOL.	FULL SCALE	MODEL SCALE
AT38	15.5	0.465
AT38.1.	91.67	2.75
^{AT} 38.2	75.00	2.25
^{AT} 38.2A	75.0	2.25
AT38.3	ROD REMOVED	ROD REMOVED

MODEL COMPONENT: ATTACH STRUCTURE - AT39

GENERAL DESCRIPTION: Orbiter aft attachment, pitch adjustable from

0 to 10 deg.

MODEL SCALE: 0.030

DRAWING NO.: Boeing 50 1319-37.

DIMENSIONS:	FULL SCALE	MODEL SCALE
Pivot location:		
In., X _C	400.0	12.0
In., Z _C	160.7	4₹82 1
Equivalent Span: (At O deg iorb)		
Centerline orbiter, in.	521.0	15.63

MODEL COMPONENT : BODY - B26			
GENERAL DESCRIPTION:Configuration 140A/B orbiter fuselage.			
NOTE: B26 is identical to B24 except underside of fuselage has been			
refaired to accept W116.			
MODEL SCALE: 0.030			
DRAWING NUMBER: <u>VL70-000143B</u> -000200 -000140A000140	. - 000205, -006 B	089000145.	
DIMENSIONS: Length (OML: Fund Sta. Xo=235) In	FULL SCALE . 1293.3	MODEL SCALE 38.799	
Length (IML: Fwd Sta. X _o =238) In	1290.3	38.709	
Max Width (@ $X_0 = 1528.3$) In.	264.00	7.920	
Max Depth (@ $X_0 = 1464$) In.	250.00	7.500	
Fineness Ratio	0.264	0.264	
Area - Ft ²		***	
Max. Cross-Sectional	340.88	0.3068	
Planform	· · · · · · · · · · · · · · · · · · ·	•	
Wetted			
Base		`	

MODEL COMPONENT : OMS POD - M16	· · · · · · · · · · · · · · · · · · ·	
GENERAL DESCRIPTION :Configurati	on 1400 orbiter	OMS pod - short
pod.		
	Þ	
MODEL SCALE: 0.030	•	
DRAWING NUMBER :	0	·
	•	
,		
DIMENSION'S:	FULL SCALE	MODEL SCALE
Length (OMS Fwd Sta. X _o =1310.5)	In. 258.50	7.755
Max Width (@ .X _o = 1511), In.	136.8	4.104
Max Depth (@ X = 1511). In.	74.70	2.241
Fineness Ratio	2.484	2.484
Area - Ft ²	· ·	
Max. Cross—Sectional	58.864	0.053
·Planform .		
Wetted		
Base		

MODEL COMPONENT: MP3 NOZZLES - N24			
GENERAL DESCRIPTION: . Configuration 140A/B Orbiter MPS Nozzles			
-			
	•		
MODEL SCALE: 0.030	MODEL DWG: SS-A00147.	Release 12 .	
DRAWING NUMBER: <u>VL70-005030A</u> , -00014	OA		
DIMENSIONS:	FULL SCALE	MODEL SCALE	
Length - In.			
Gimbal Point to Exit Plane Throat to Exit Plane	<u>157.0</u>	4.71 2.976	
Diameter - In. Exit Throat	91.000	2.73	
Inlet	***		
Area - ft ² Exit Throat	45.166	0.0407	
Gimbal Point (Station) In. Upper Nozzle			
Y Z	0.0 0.0 443.00	43.35 0.0 13.29	
Lower Nozzles X	1468.170	44.045	
Y Z	$\frac{\pm}{342.640}$ \pm	1.59 10.279	
Null Position - Deg. Upper Nozzle	-,		
Pitch Yaw	<u> 16</u> 0	16 0	
Lower Nozzle Pitch	10	10	
Yaw	3.5	3.5	

MODEL COMPONENT: OMS NOZZLES - N28	·	
GENERAL DESCRIPTION: Configuration 140A/B Orbiter OMS Nozzles :		
•		
MODEL SCALE: 0.030	•	<i>,</i> ·
DRAWING NUMBER: VL70-000140A (Location), SS-A00	106. Release	5 · (Contour)
DIMENSIONS:	FULL SCALE	MODEL SCALE
•		
Length - In. Gimbal Point to Exit Plane		
Throat to Exit Plane		
Diameter - In.		
· Exit Throat.		
Inlet		,—————————————————————————————————————
Área - ft ²		
EXXX Max. Cross-Sectional (1 nozzle) Throat		
Gimbal Point (Station) - In.		
Left Manar Nozzle		
Yo.	1518.0 -88.0	<u>45.54</u> - 2.64
\overline{z}_{o}^{-0}	492.0	14.76
Right kaner Nozzles	•	
Xo	1518.0	45.54
Υ _o Z _o	<u>88.0</u> 492.0	<u>2.64</u> 14.76
Null Position - Deg.		
Leftzeer Nozzle		
Pitch	15°491	150491
Yaw	12°17'	<u> 12°17'</u>
Right ZXXXX Nozzle	n #0	O
Pitch Yaw	15°49' 12°17'	15°49! 12°17!

MODEL COMPONENT: RUDDER - R5		
GENERAL DESCRIPTION: Configuration 1400 orbit	er rudder (id	entical to
MODEL SCALE: 0.030		
DRAWING NUMBER: VL70-000146B00	0095	
DIMENSIONS:	FULL SCALE	MODEL SCALE
Area - Ft ²	100.15	0.090
Span (equivalent). In.	201.00	6.030
Inb'd equivalent chord. In.	91.585	2.748
Outb'd equivalent chord . In.	50.833	1.525
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	0.400	0.400
At Outb'd equiv. chord	0.400	0.400
Sweep Back Angles, degrees		
Leading Edge	34.83	34.83
Tailing Edge	26.25	26.25
Hingeline (Product of larea & c)	34.83	34.83
Area Moment (Normaly toxinger line). Ft.	610.92	0.165
Mean Aerodynamic Chord, In.	73.2	2.196

	SCRIPTION: Fairing mount		
ferry missio	ns.		
MODEL SCALE:	0.030	······································	;- -
DRAWING NUM	BER: Boeing Drawing 1319	- 71.	
DIMENSIONS :		FULL SCALE	MODEL 'SCALI
		•	•
Leng	th , in.	445.83	13.375
Max I	Width, in.	303.33	9.10
Max	Height, in.	265.00	7.95
Finer	iess Ratio		
4rea	- Ft ²		
1	Projected frontal area Max. Cross-Sectional	324.105	0.2917
	Planform	-	,
•	´ Wetted	· ,	
	Base		

GENERAL DESCRIPTION: Configuration 140C o	rbiter vertical ta	<u>il (identic</u> a
to configuration 140A/B vertical tail).		
	·	
MODEL SCALE: 0.030		
DRAWING NUMBER: VI70-000140C -000144B		
dimensions:	FULL SCALE	MODEL SCAI
TOTAL DATA .		
Area (Theo) - Ft ² Planform Span (Theo) - In. Aspect Ratio Rate of Taper Taper Ratio Sweep-Back Angles, Degrees. Leading Edge Trailing Edge O.25 Element Line Chords: Root (Theo) WP Tip (Theo) WP MAC Fus. Sta. of .25 MAC W.P. of .25 MAC B.L. of .25 MAC	413.243 315.72 1.675 0.507 0.404 45.00 26.25 41.13 268.50 108.47 199.81 1463.35 635.52 0.00	0.372 9.472 1.675 0.507 0.404 45.00 26.25 41.13 8.055 3.254 5.994 43.901 19.066 0.00
Airfoil Section Leading Wedge Angle - Deg. Trailing Wedge Angle - Deg. Leading Edge Radius	10.00 14.92 2.00	10.00 14.92 0.060
Void Area - Ft ² Blanketed Area - Ft ²	<u>13.17</u>	0.0

MODEL COMPONENT: WING-W 116		
GENERAL DESCRIPTION: Configuration 4		
NOTE: Identical to W11/ except airfoil thickness.	Dihedral angle	is along
trailing edge of wing.		
MODEL SCALE: 0.030		· • · · · · · · · · · · · · · · · · · ·
TEST NO.	DWG. NO. VL7	0-000140A, -000200
DIMENSIONS:	FULL-SCALE	MODEL SCALE
Area (Theo.) Ft2 Planform Span (Theo In. Aspect Ratio Rate of Taper Taper Ratio Dihedral Angle, degrees Incidence Angle, degrees Aerodynamic Twist, degrees Sweep Back Angles, degrees Leading Edge Trailing Edge O.25 Element Line Chords: Root (Theo) B.P.O.O. Tip, (Theo) B.P. MAC Fus. Sta. of .25 MAC B.L. of .25 MAC B.L. of .25 MAC EXPOSED DATA Area (Theo) Ft2 Span, (Theo) In. BP108 Aspect Ratio Chords Root BP108 Tip 1.00 b MAC Fus. Sta. of .25 MAC Area of .25 MAC Area (Theo) Ft2 Span (Theo) In. BP108 Aspect Ratio Chords Root BP108 Tip 1.00 b MAC Airfoil Section (Rockwell Mod NASA) XXXX-64 Root b Tip b Tip b Table Courter Theory Ft2 Data for (1) of (2) Sides Leading Edge Cuff 2	2690.00 936.68 2.265 1.177 0.200 3.500 0.500 3.000 45.000 - 10.056 35.209 - 689.24 137.85 474.81 1136.83 290.58 182.13 1751.50 720.68 2.059 0.245 562.09 137.85 392.83 1185.98 294.30 251.77	2.421 28.10 2.265 1.177 0.200 3.500 0.500 3.000 45.000 -10.056 35.209 20.677 4.136 14.244 34.105 8.717 5.464 1.576 21.620 2.059 0.245 16.863 4.136 11.785 35.579 8.829 7.553
Planform Area Ft ⁴ Leading Edge Intersects Fus M. L. 0 Sta Leading Edge Intersects Wing 0 Sta 72	113.18 500.00 1024.00	0.102 15.00 30.72

MODEL COMPONENT: SLOTTED ELEVON (6 -INCH GAP) - Ehh		
GENERAL DESCRIPTION: Configuration 140A/1	B Orbiter elevon.	
MOTE: Ehlis a slotted version of Egg. 1	Data are for one s	ide.
MODEL SCALE: 0.030		
DRAWING NUMBER: VL70-000200 -006	089006092	
DIMENSIONS:	FULL SCALE	MODEL SCALE
Area - Ft ²	210.0	0.189
Span (equivalent) In.	349.2	10.476
Inb'd equivalent chord . In.	118.004	3.540
Outb'd equivalent chord . In.	55.192	1.656
<pre>Ratio movable surface chord/ total surface chord</pre>	,	
At Inb'd equiv. chord	0.2096	0.2096
At Outb'd equiv. chord	0.4004	0.4004
Sweep Back Angles, degrees		,
Leading Edge	0.00	0,00
Tailing Edge	- 10.056	10.056
Hingeline (Product of area and \overline{c})	0.00	0,00
Area, Moment (Normalkxtoxtkingaxkine) Ft3	1587.25	0,043
Mean Aerodynamic Chord, In.	90.7	2.721

MODEL COMPONENT : CANOPY - C9				
GENERAL DESCRIPTION: Configuration 3A. Canopy used with fuselage				
. B ₂₆ .	• ,			•
MODEL SCALE: 0.030	MODEL	DWG:	SS-A001	47, Release 12
DRAWING NUMBER: VL70-000143A				
DIMENSIONS:		FULL	SCALE	MODEL SCALE
Length ($X_0 = 434.643$ to 578),In.	14	3.357	4.301
Max Width (@ $X_0 = 513.127$),	In.	15:	2.412	4.572
Max Depth (@ $X_0 = 485.0$).	In.	2	5.00	0.750
Fineness Ratio		·		
Area		,	<u> </u>	
Max. Cross—Sectional		···		•
Planform		4		
Wetted			***	
Base				

TABLE IIIb. (Concluded)

MODEL COMPONENT : BODY FLAP - FR			
GENERAL DESCRIPTION: Configuration 140A/B orbiter body flap			
Hingeline located at $X_0 = 1528.3$, $Z_0 = 2$	284.3		
•		· · · · · · · · · · · · · · · · · · ·	
MODEL SCALE: 0.030 MODEL	L DWG: SS-A00147	. Release 12	
DRAWING NUMBER: VL70-000140A, -0001	145		
DIMENSIONS:	FULL SCALE	MODEL SCALE	
Length ($X_0=1520 \text{ to } X_0=1613$) In.	93.00	2.79	
Max Width (In.)	262.0	7.86	
Max Depth (@ $X_0 = 1520$), In.	23.00	0.69	
Fineness Ratio			
Area - Ft ²			
Max. Cross—Sectional	-		
Planform	150.525	0.135	
Wetted	**************************************		
Base .	41.847	0.038	

TABLE IV.. - CA5 DATASET DESCRIPTION $\,\cdot\,$

-DATASET- TYPE	DESCRIPTION
RGMXXX :	Main balance data which have been corrected for main balance mounting system tares and contain wall corrections.
AGMXXX	Orbiter (internal) balance "basic" data which contains "standard" wind tunnel corrections including a wall correction.
BGMXXX	Main balance data (corrected for main balance mounting system tares) minus orbiter (internal) balance "basic" data (RGMXXX-AGMXXX).
RGMCXX	Main balance mounting system tare data.

Notes:

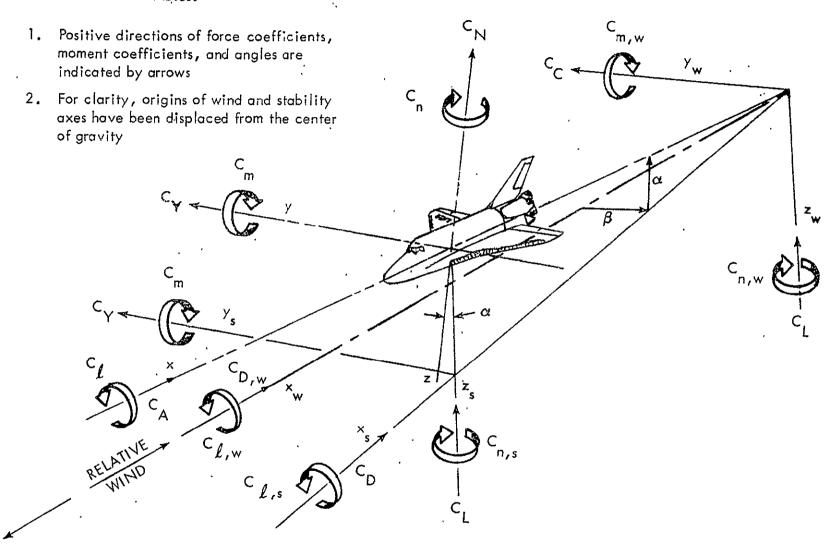
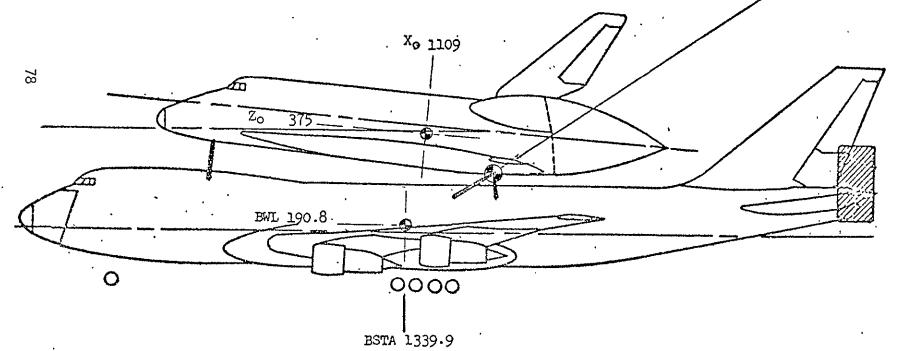


Figure A. - Axis systems.

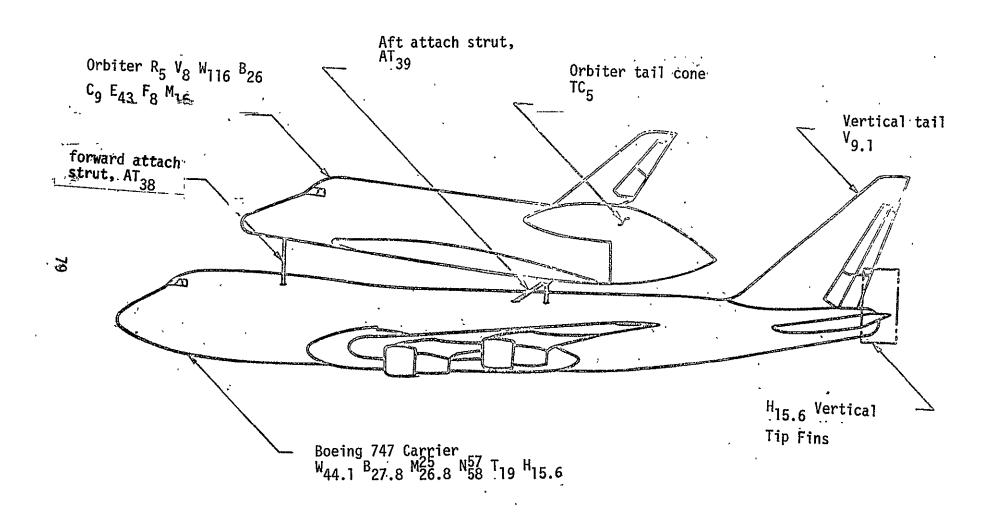
REFERENCE DIMENSIONS (FS)

,	ORBITER	747 CARRIER
WING AREA ∼Ft ²	2690	5500
MAC $(\bar{c}) \sim \text{INCHES}$	474.81	327.78
SPAN (b) ~ INCHES	936.68	2348.04
MOMENT REFERENCE CENTER	67.5% LB	25.0 % C
F.S. ~ INCHES W.P. ~ INCHES	1109.0 375.0	1339.9 190.8

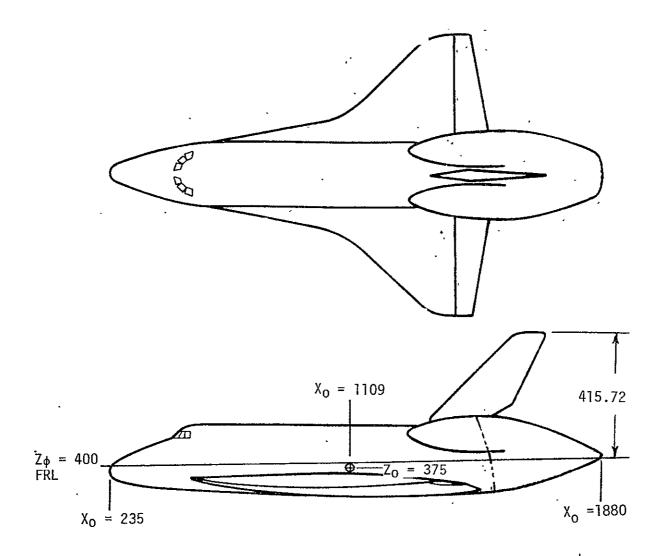
BWL 400 (20 267.5 BSTA 1607 (X0 1317)

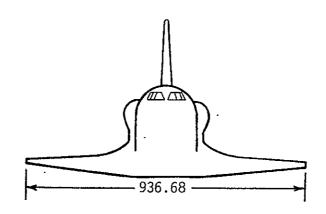


a. Orbiter/747 Flight Test Configuration Reference Dimensions Figure B. - Model sketches.



b. Orbiter/Carrier Model NomenclatureFigure B. - Continued.



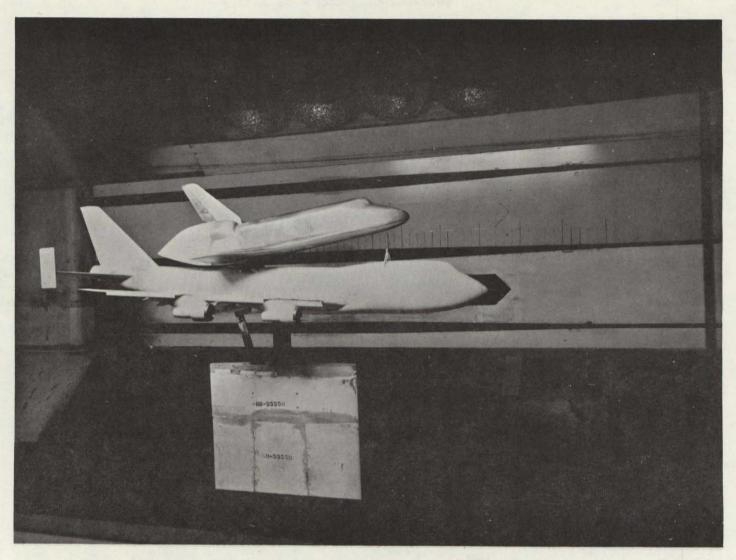


c. Orbiter ConfigurationFigure B. - Continued.



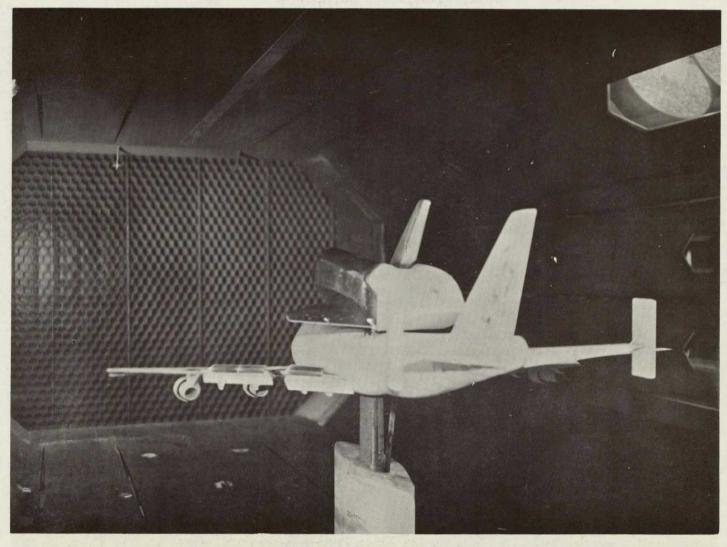
a. Orbiter/Carrier Mated Configuration Side View $i_{ORB} = 8^{\circ}$, Orbiter without Tail Cone Fairing

Figure C. - Model photographs.

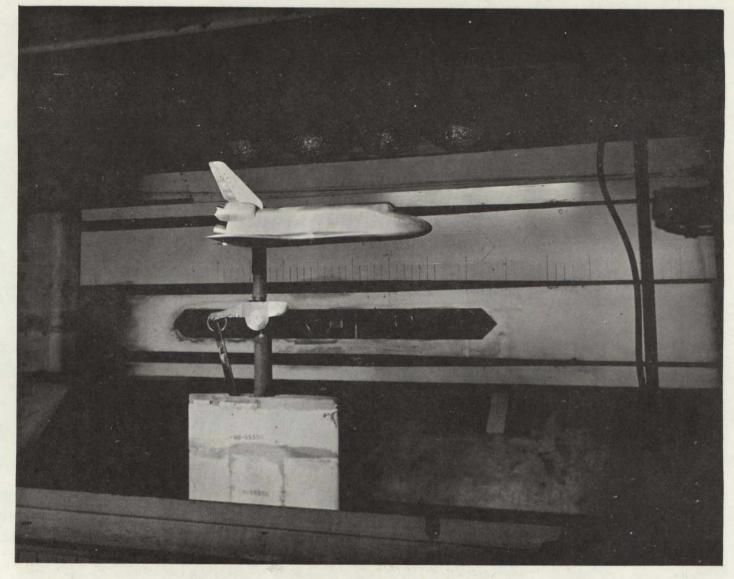


b. Orbiter/Carrier Mated Configuration Side View $i_{ORB} = 10^{\circ}$, Orbiter with Tail Cone Fairing

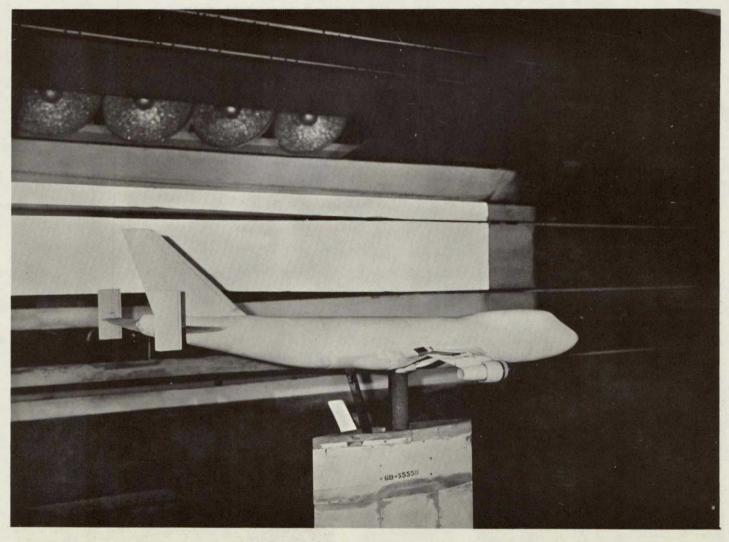
Figure C. - Continued.



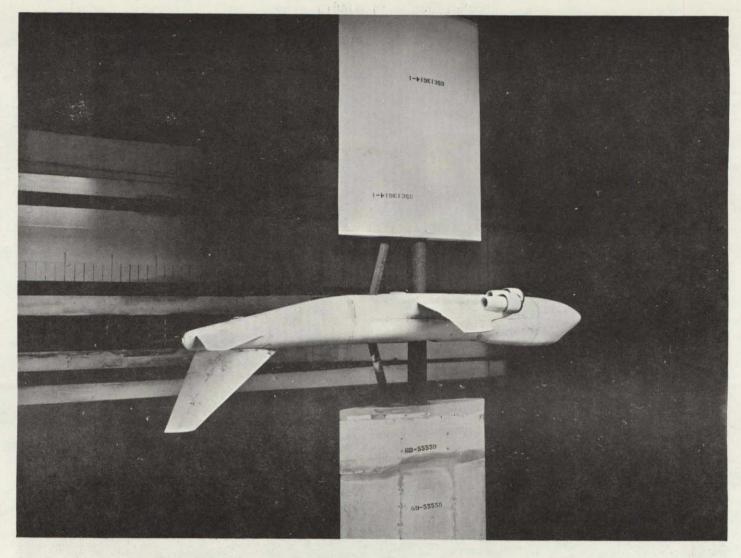
c. Orbiter/Carrier Rear View
Figure C. - Continued.



d. Orbiter Alone InstallationFigure C. - Continued.



e. Carrier Alone Installation Figure C. - Continued.



f. Carrier Inverted with Image Strut for Determination of Wind Tunnel Strut Tare and Interference Effects

Figure C. - Concluded.

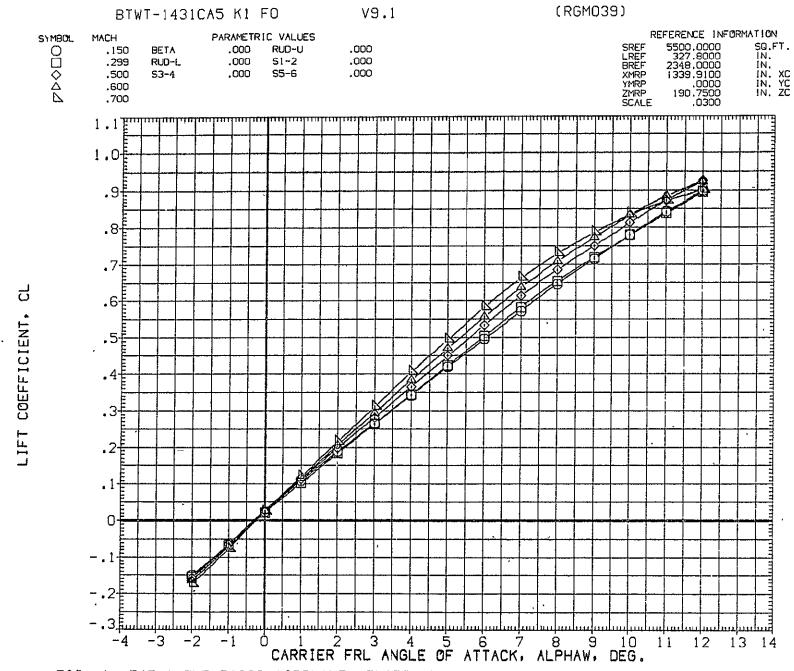


FIG. 1 747 ALONE, BASIC AIRPLANE FLAPS UP, HORIZ TAIL OFF

PAGE

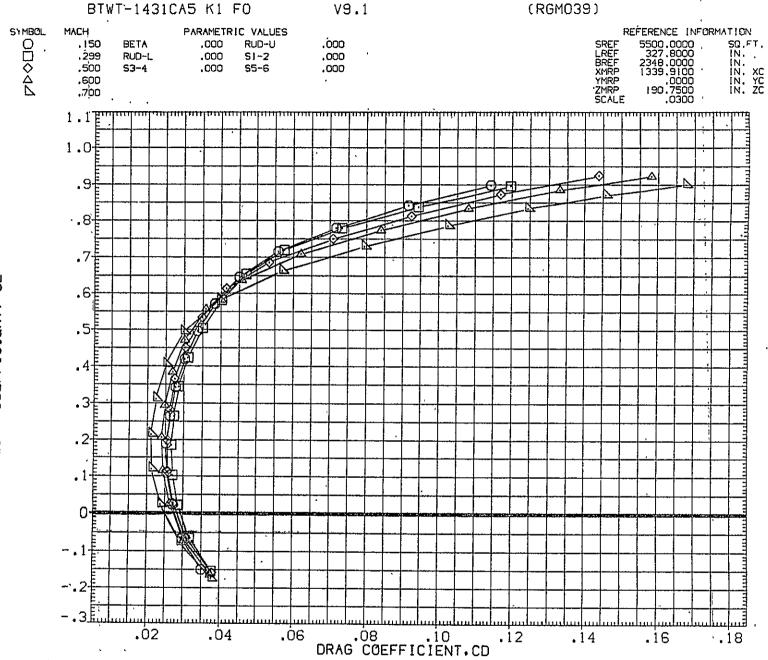


FIG. 1 747 ALONE, BASIC AIRPLANE FLAPS UP, HORIZ TAIL OFF

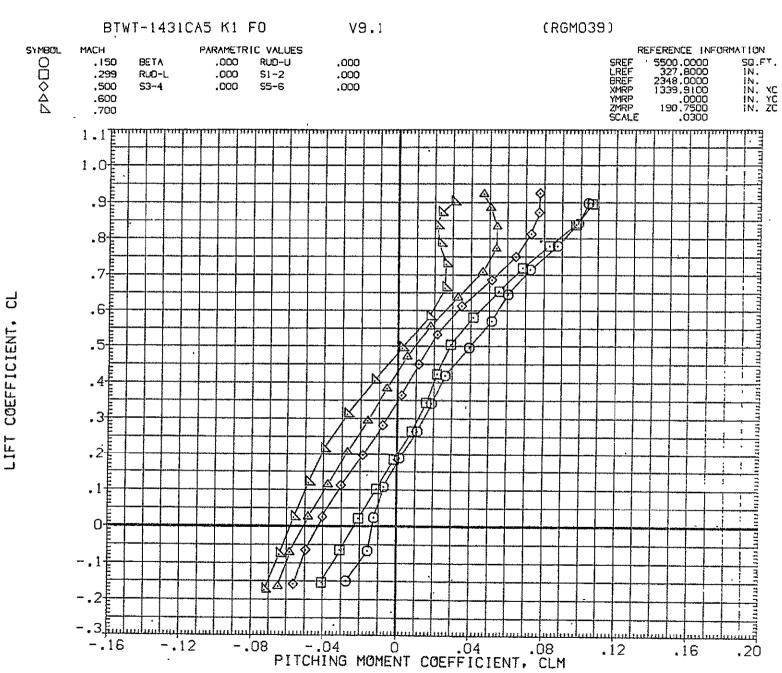


FIG. 1 747 ALONE, BASIC AIRPLANE FLAPS UP, HORIZ TAIL OFF

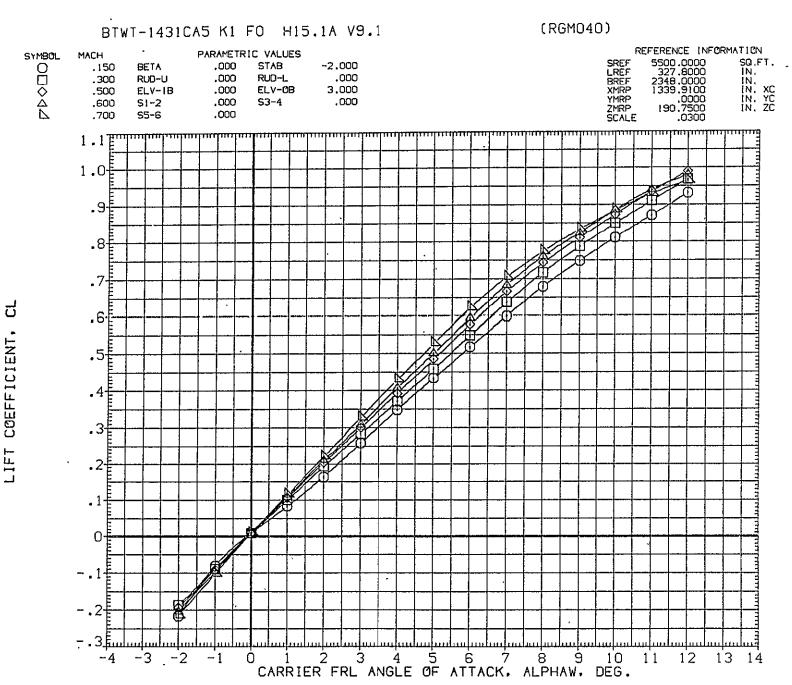
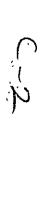


FIG. 2 747 ALONE, BASIC AIRPLANE FLAPS UP, TAIL ON(S=-2)



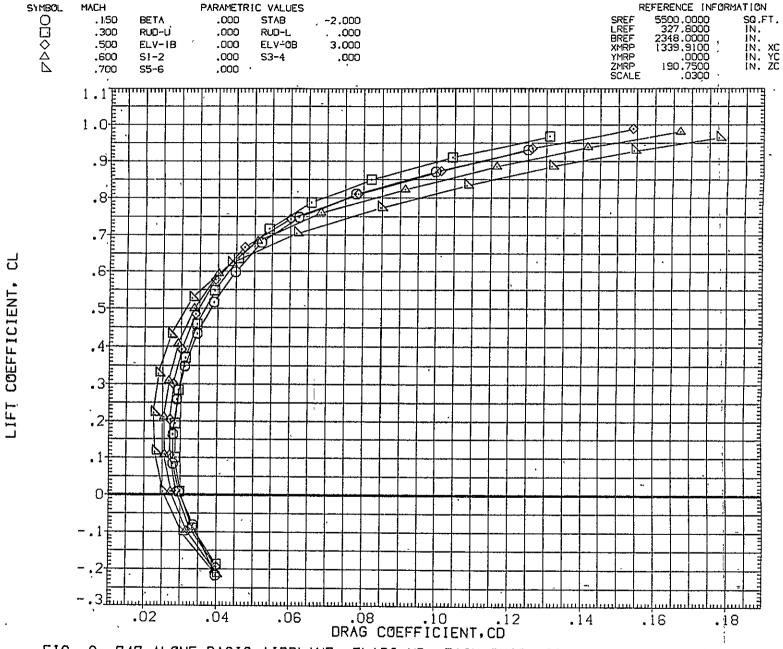


FIG. 2 747 ALONE, BASIC AIRPLANE FLAPS UP, TAIL ON(S=-2)

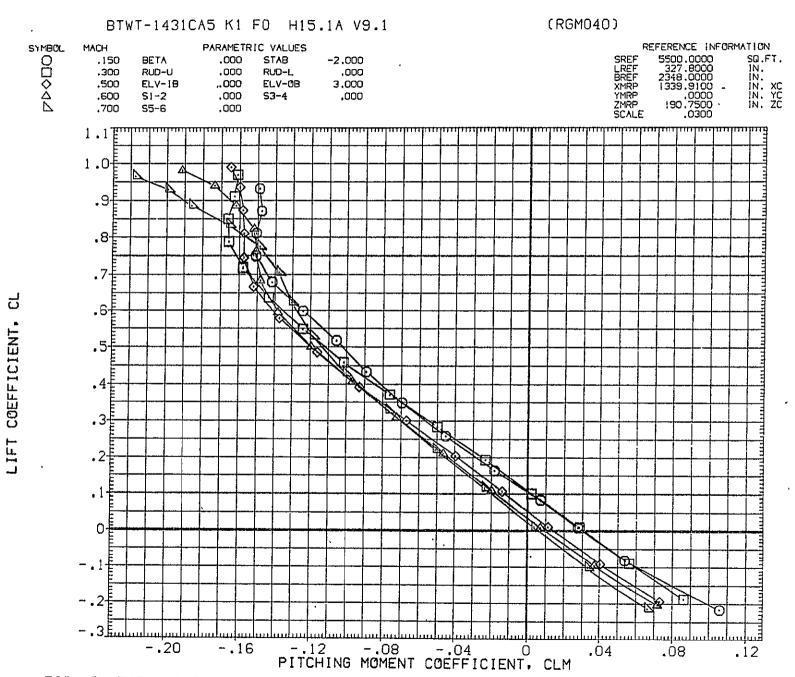


FIG. 2 747 ALONE, BASIC AIRPLANE FLAPS UP, TAIL ON(S=-2)

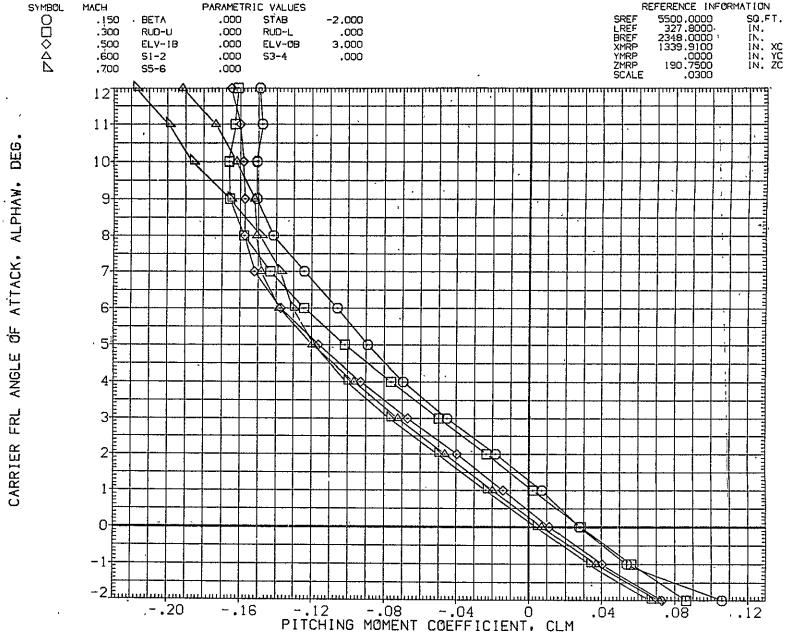


FIG. 2 747 ALONE, BASIC AIRPLANE FLAPS UP, TAIL ON(S=-2)

CARRIER FRL ANGLE OF ATTACK, ALPHAW, DEG.

FIG. 3 747 ALONE+TIP FINS FLAPS UP, TAIL ON(S=-2)

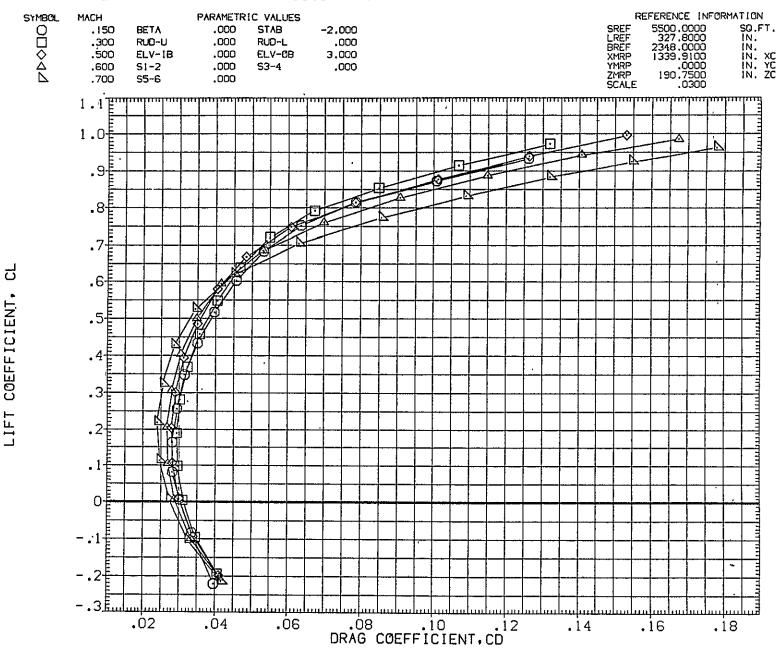


FIG. 3 747 ALONE+TIP FINS FLAPS UP, TAIL ON(S=-2)

10

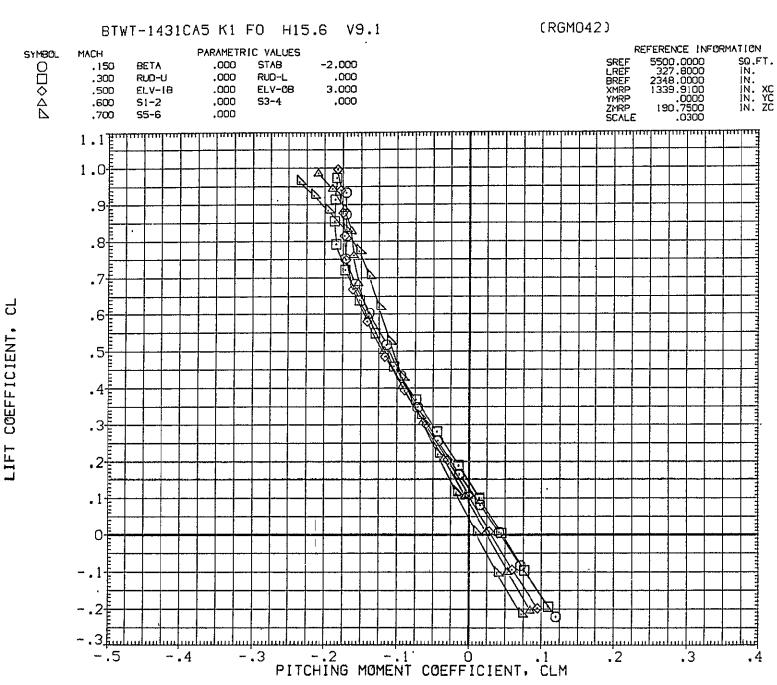


FIG. 3 747 ALONE+TIP FINS FLAPS UP, TAIL ON(S=-2)

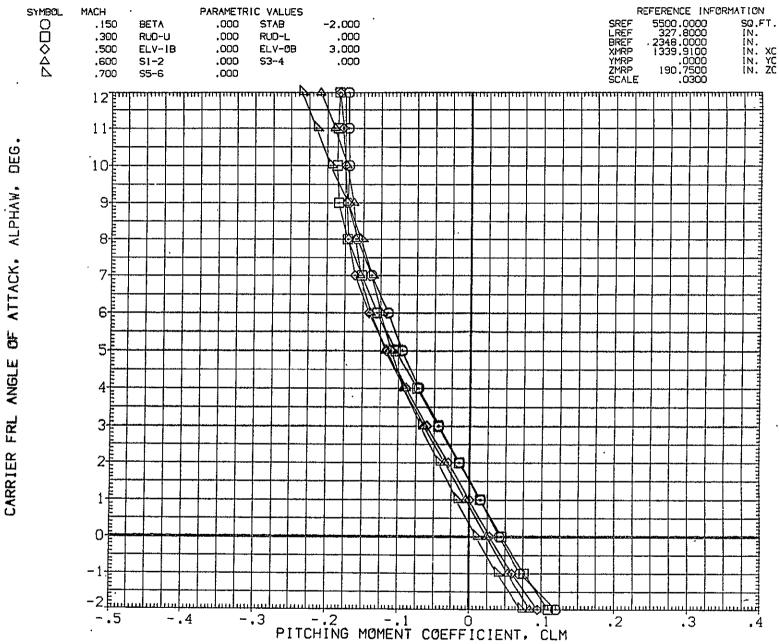


FIG. 3 747 ALONE+TIP FINS FLAPS UP, TAIL ON(S=-2)

12

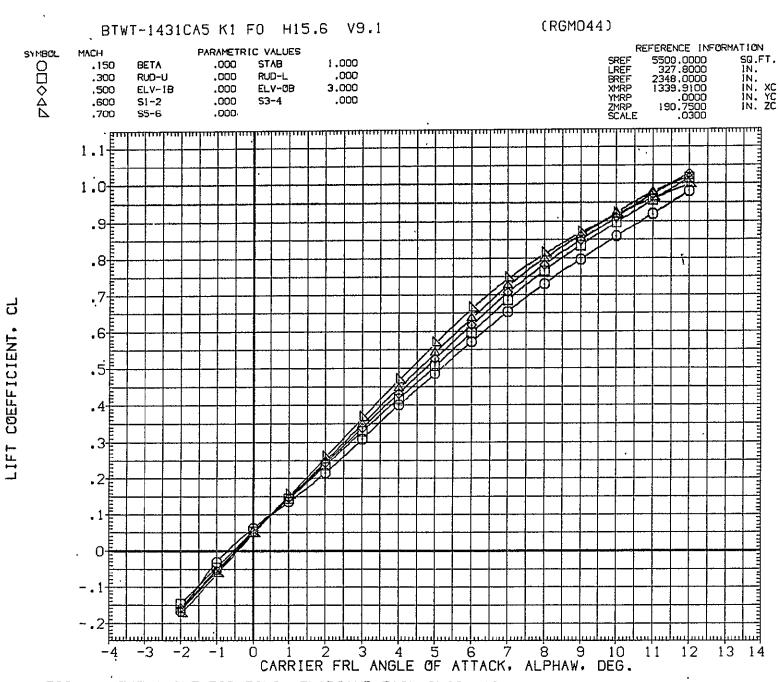


FIG. 4 747 ALONE+TIP FINS FLAPS UP, TAIL ON(S=+1)

FIG. 4 747 ALONE+TIP FINS FLAPS UP, TAIL ON(S=+1)

FIG. 4 747 ALONE+TIP FINS FLAPS UP, TAIL ON(S=+1)

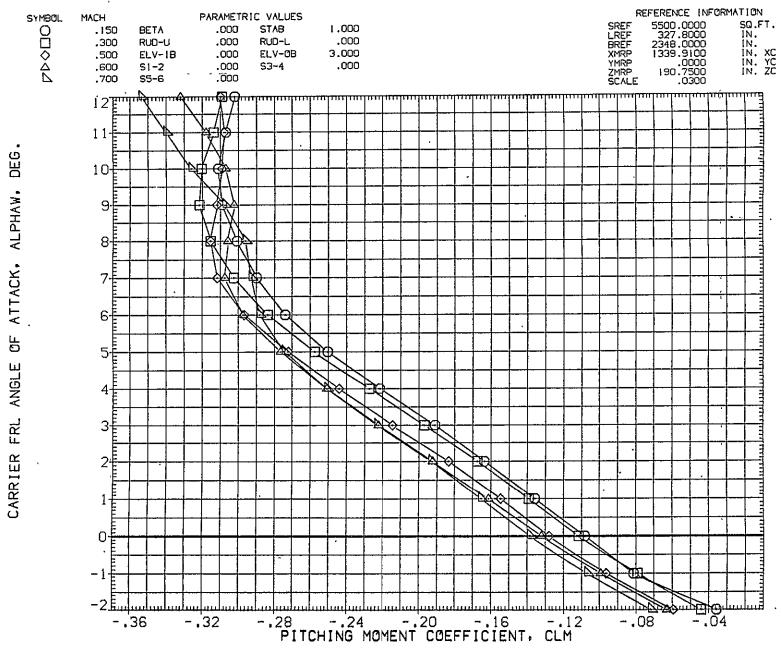


FIG. 4 747 ALONE+TIP FINS FLAPS UP. TAIL ON(S=+1)

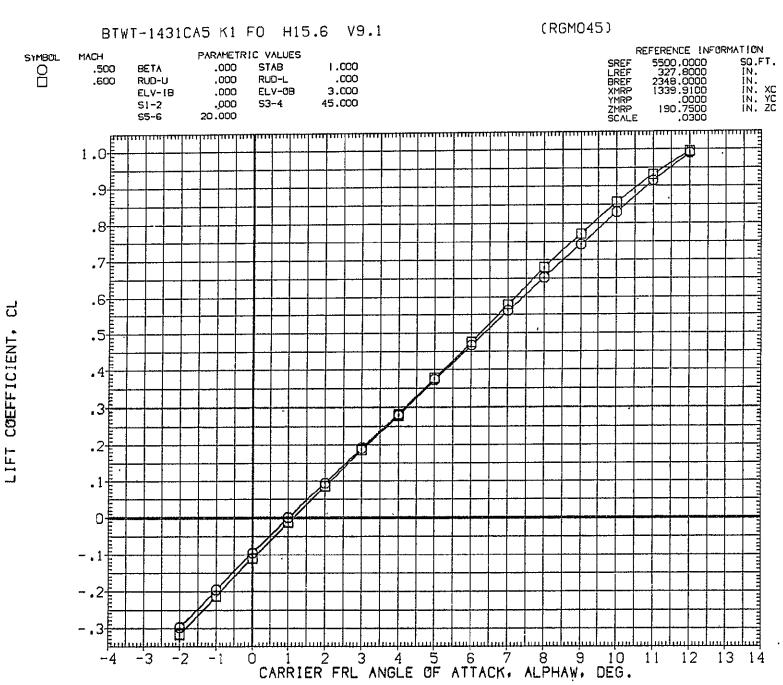


FIG. 5 747 ALONE+TIP FINS+STD SPD BRAKES FLAPS UP, TAIL ON(S=+1)

17

FIG. 5 747 ALONE+TIP FINS+STD SPD BRAKES FLAPS UP, TAIL ON(S=+1)

.10 .12

DRAG COEFFICIENT, CD

.08

.04

.06

.20

.18

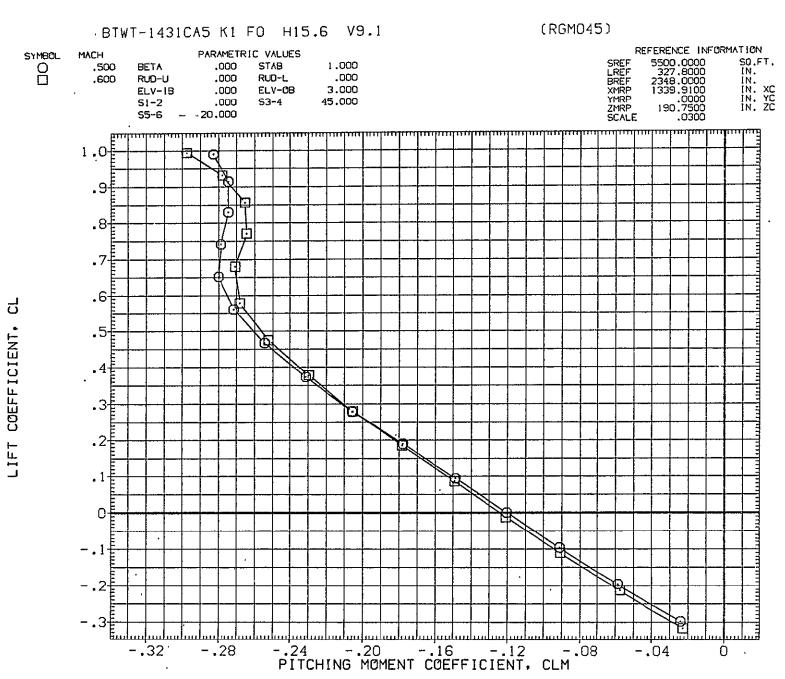


FIG. 5 747 ALONE+TIP FINS+STD SPD BRAKES FLAPS UP, TAIL ON(S=+1)

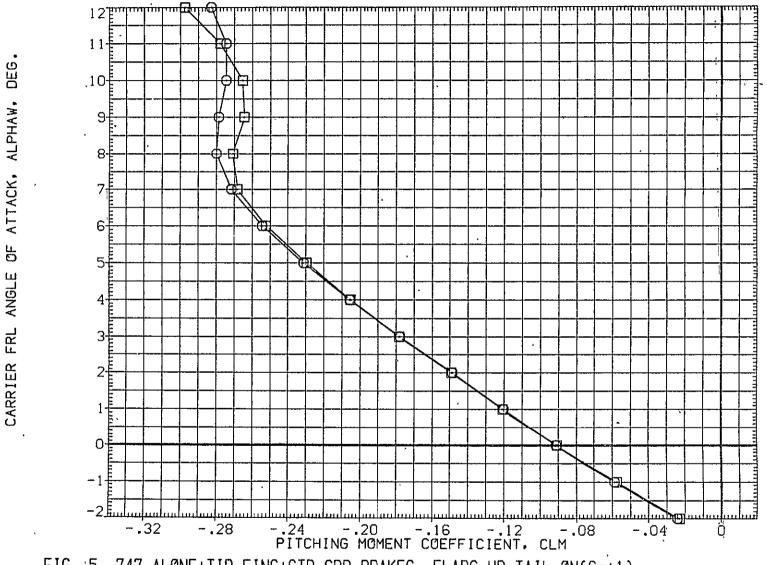


FIG. 5 747 ALONE+TIP FINS+STD SPD BRAKES FLAPS UP, TAIL ON(S=+1)

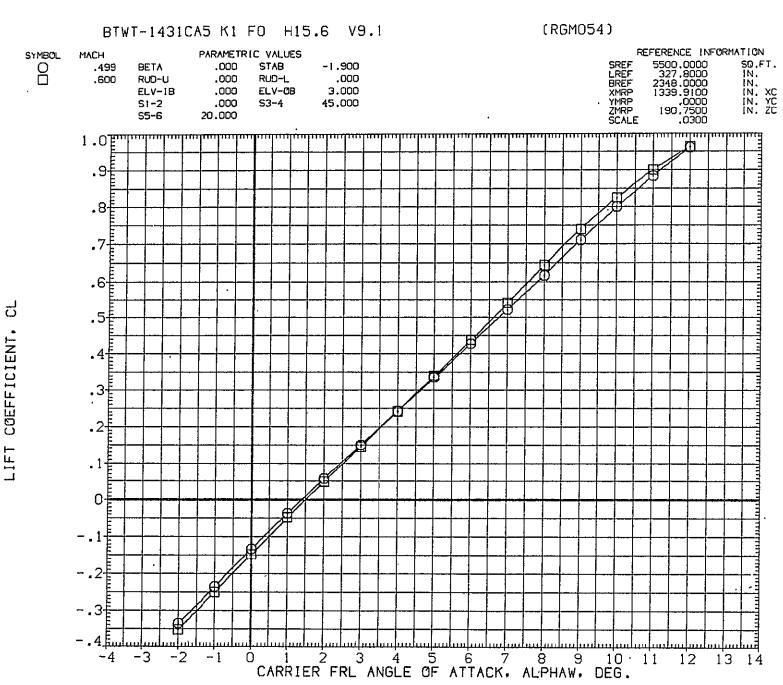


FIG. 6 747 ALONE+TIP FINS+STD SPD BRAKES FLAPS UP. TAIL ON(S=-2)

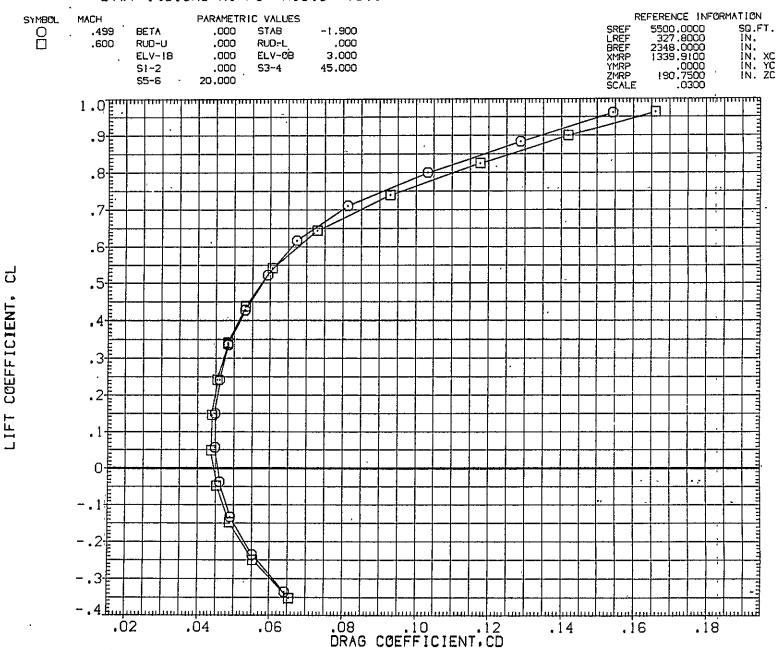


FIG. 6 747 ALONE+TIP FINS+STD SPD BRAKES FLAPS UP, TAIL ON(S=-2)

22

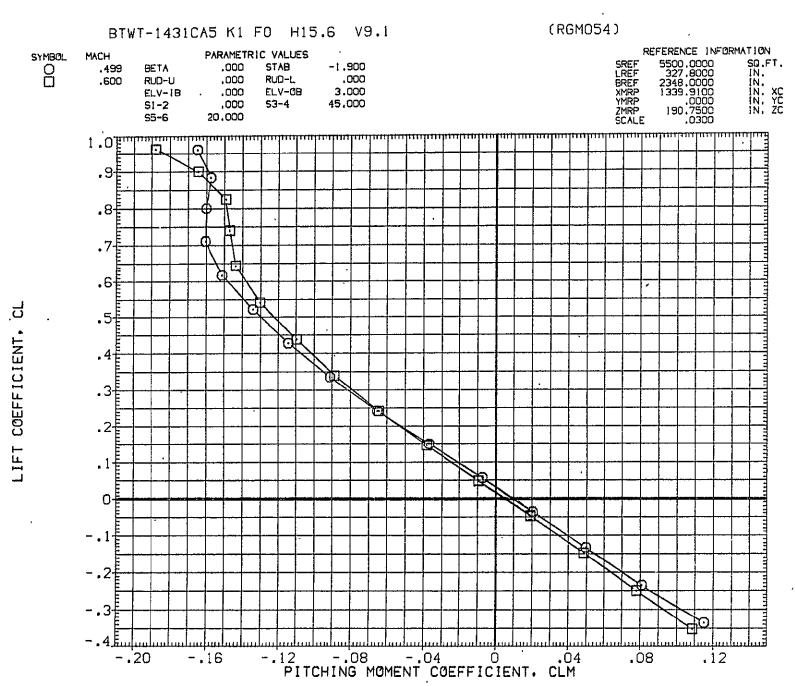


FIG. 6 747 ALONE+TIP FINS+STD SPD BRAKES FLAPS UP, TAIL ON(S=-2)

FIG. 6 747 ALONE+TIP FINS+STD_SPD BRAKES FLAPS UP, TAIL ON(S=-2)

-.08

- .04

PITCHING MOMENT COEFFICIENT, CLM

-.16

-.12

,i2

.Ò8

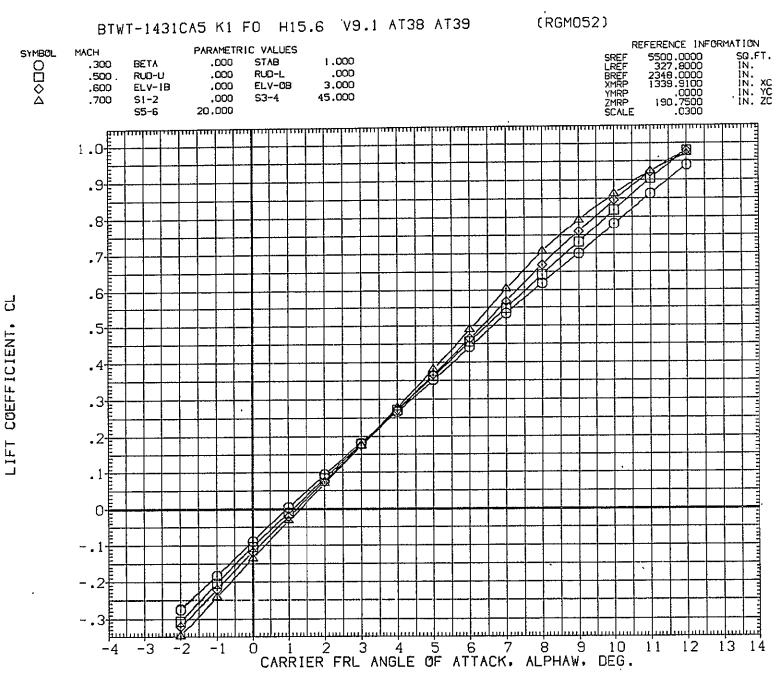


FIG. 7 747 ALONE+TIP FINS+STD SPD BRAKES+ORB SUPPORTS STRUTS(LO)

FIG. 7 747 ALONE+TIP FINS+STD SPD BRAKES+ORB SUPPORTS STRUTS(LO)

.08

.04

DRAG COEFFICIENT.CD

.20

.18

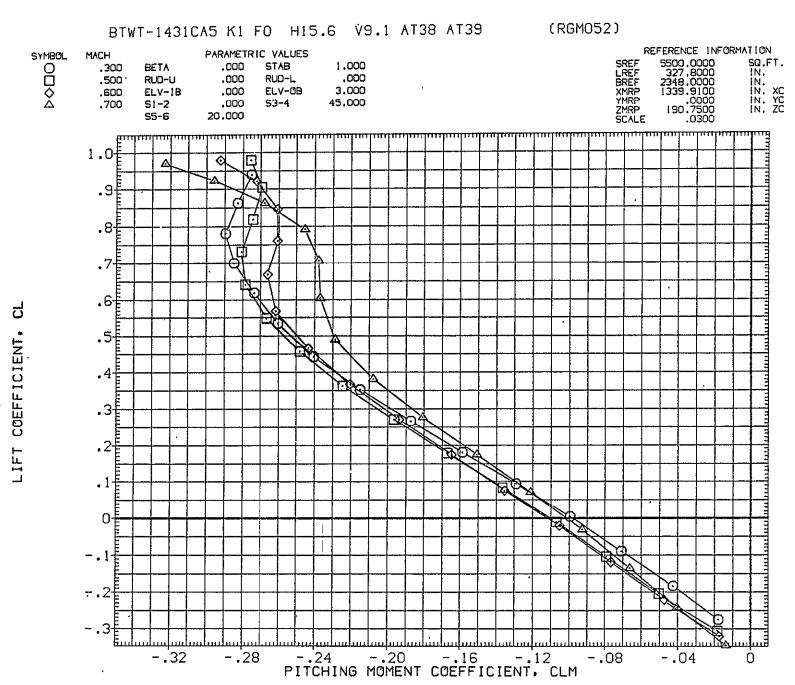


FIG. 7 747 ALONE+TIP FINS+STD SPD BRAKES+ORB SUPPORTS STRUTS(LO)

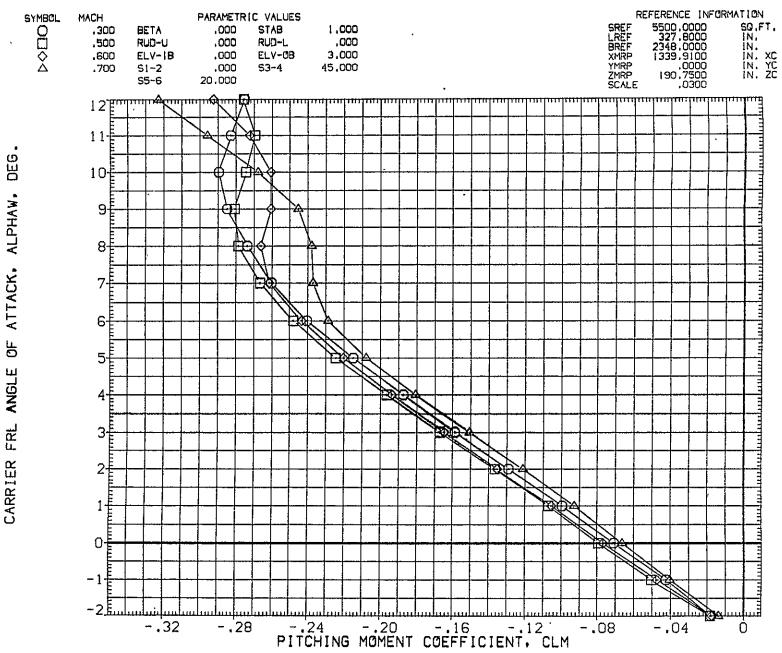


FIG. 7 747 ALONE+TIP FINS+STD SPD BRAKES+ORB SUPPORTS STRUTS(LO)

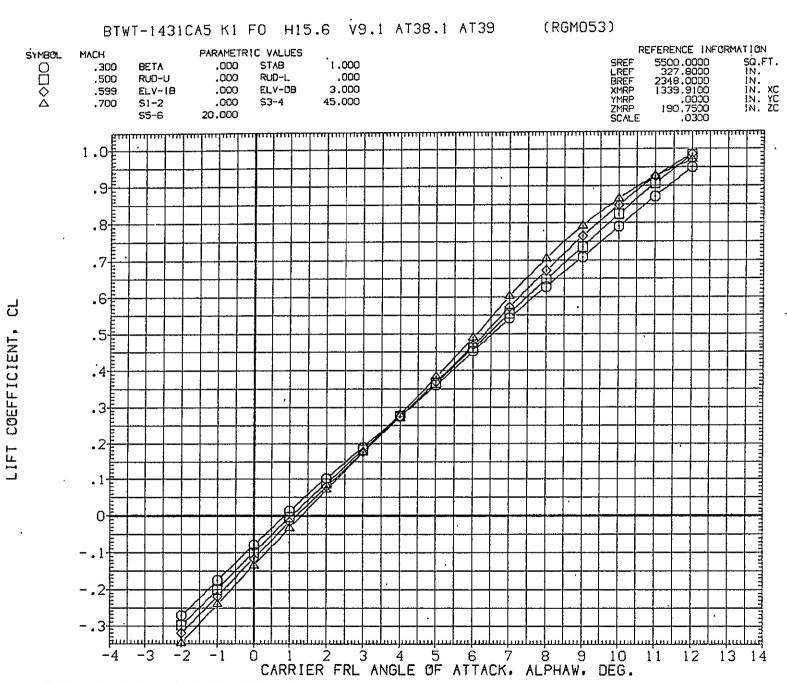


FIG. 8 747 ALONE+TIP FINS+STD SPD BRAKES+ORB SUPPORTS STRUTS(HI)

DRAG COEFFICIENT.CD .16 .18 FIG. 8 747 ALØNE+TIP FINS+STD SPD BRAKES+ØRB SUPPØRTS STRUTS(HI)

.08

.06

.04

.20

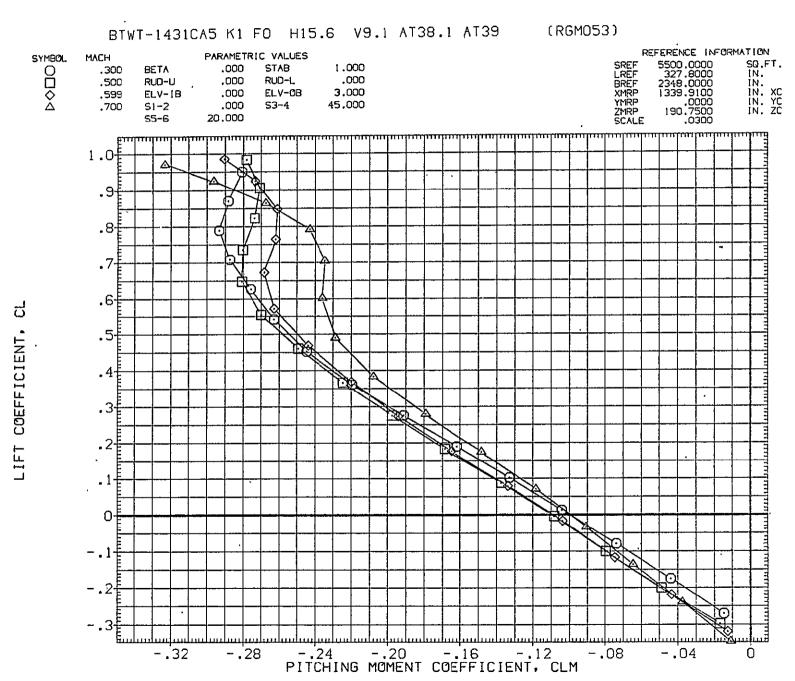


FIG. 8 747 ALONE+TIP FINS+STD SPD BRAKES+ORB SUPPORTS STRUTS(HI)

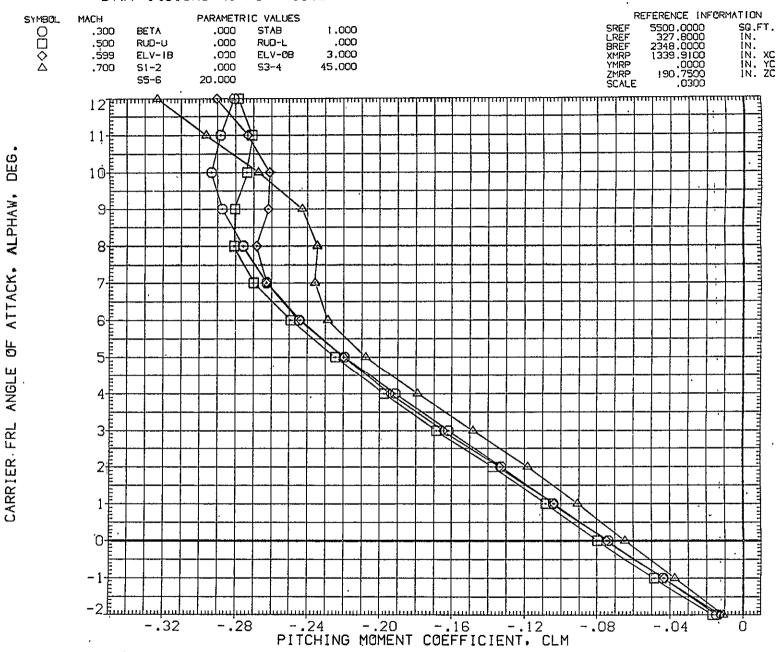


FIG. 8 747 ALONE+TIP FINS+STD SPD BRAKES+ORB SUPPORTS STRUTS(HI)

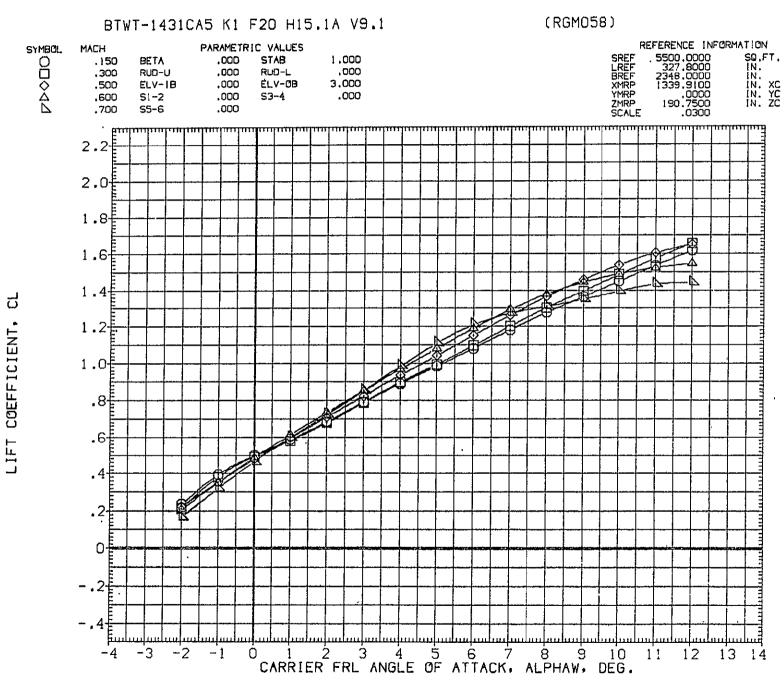


FIG. 9 747 ALONE, BASIC AIRPLANE FLAPS 20, TAIL ON(S=+1)

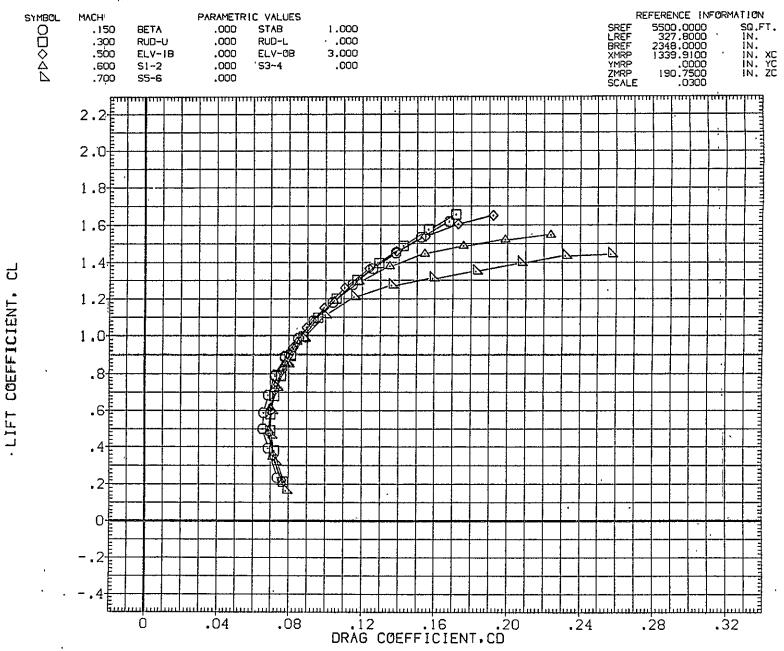


FIG. 9 747 ALONE, BASIC AIRPLANE FLAPS 20, TAIL ON(S=+1)

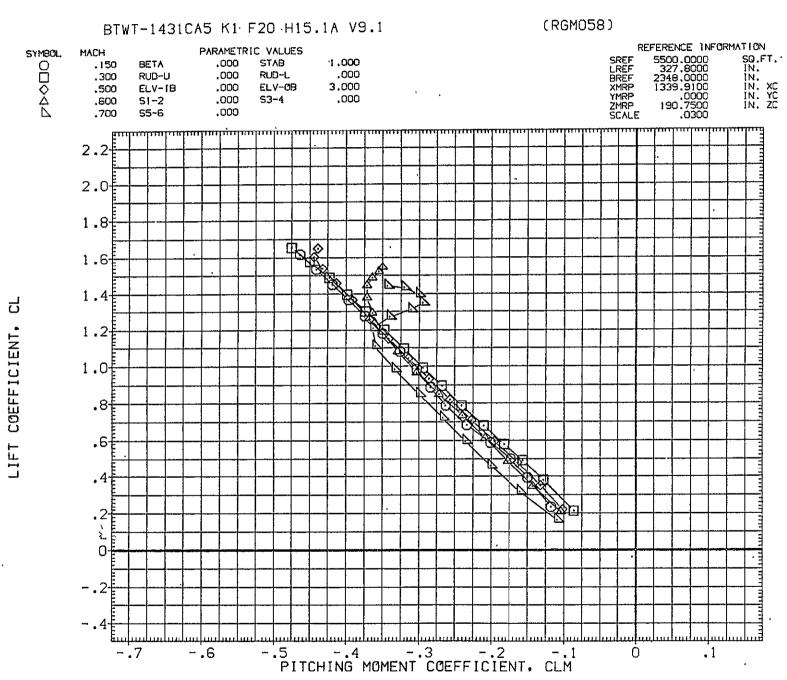


FIG. 9 747 ALONE, BASIC AIRPLANE FLAPS 20, TAIL ON(S=+1)

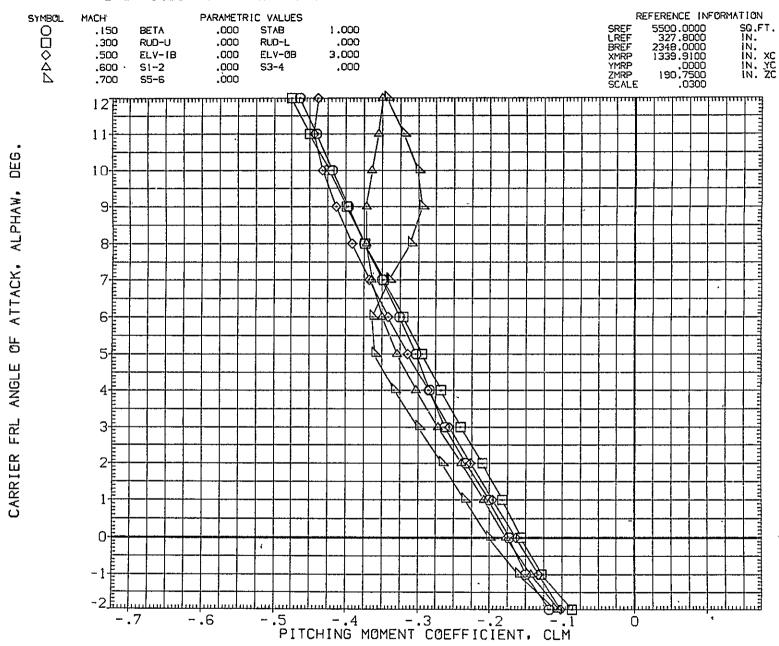


FIG. 9 747 ALONE, BASIC AIRPLANE FLAPS 20, TAIL ON(S=+1).

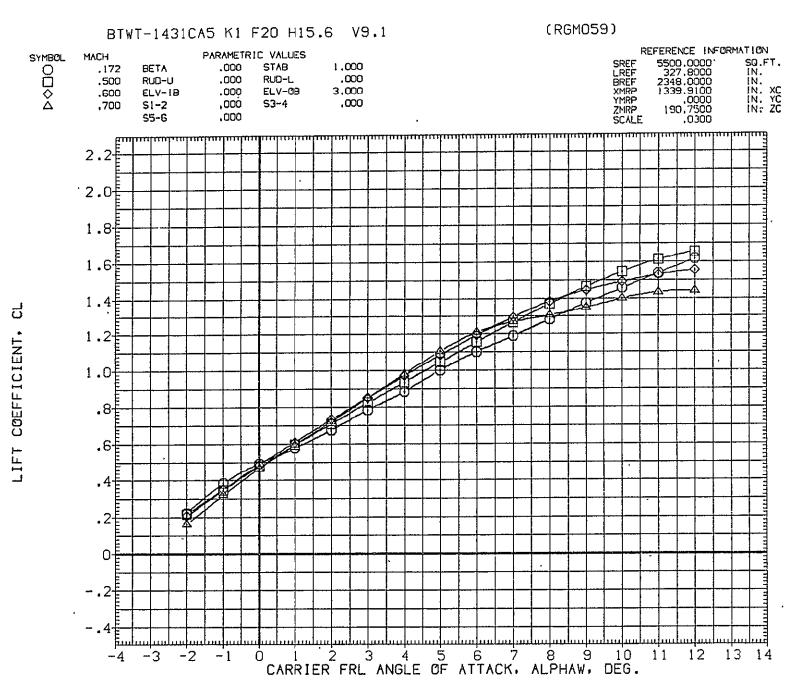


FIG. 10 747 ALONE+TIP FINS FLAPS 20, TAIL ON(S=+1)

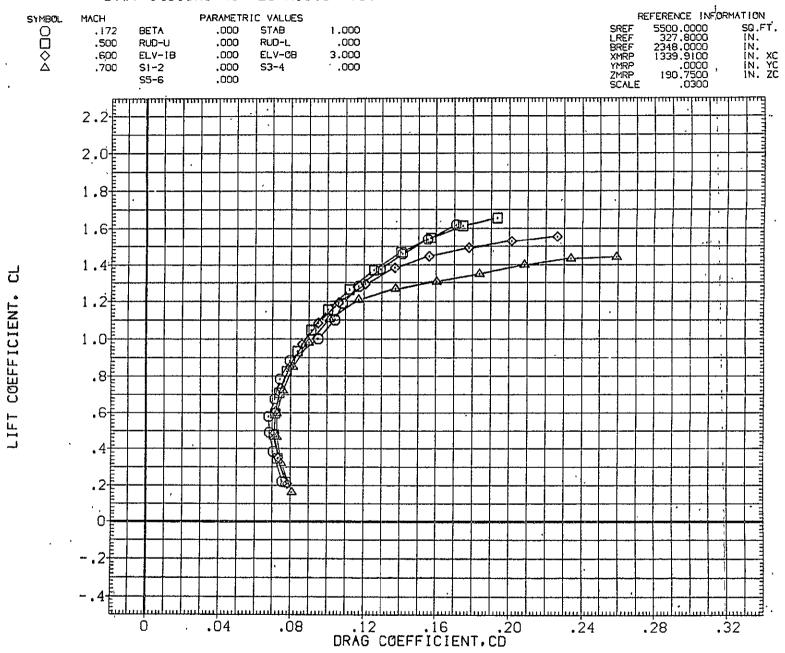


FIG. 10 747 ALONE+TIP FINS FLAPS 20, TAIL ON(S=+1)

FIG. 10 747 ALONE+TIP FINS FLAPS 20, TAIL ON(S=+1)

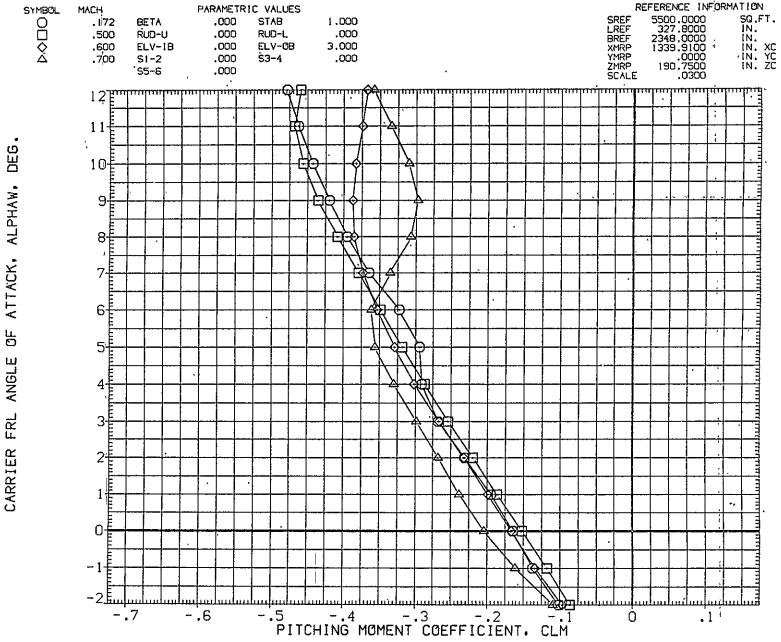


FIG. 10 747 ALONE+TIP FINS FLAPS 20, TAIL ON(S=+1)

CA5 K1 F20 H15.6 V9.1 (PLUS. ORB TC5 AT38.1)(RGMO64)

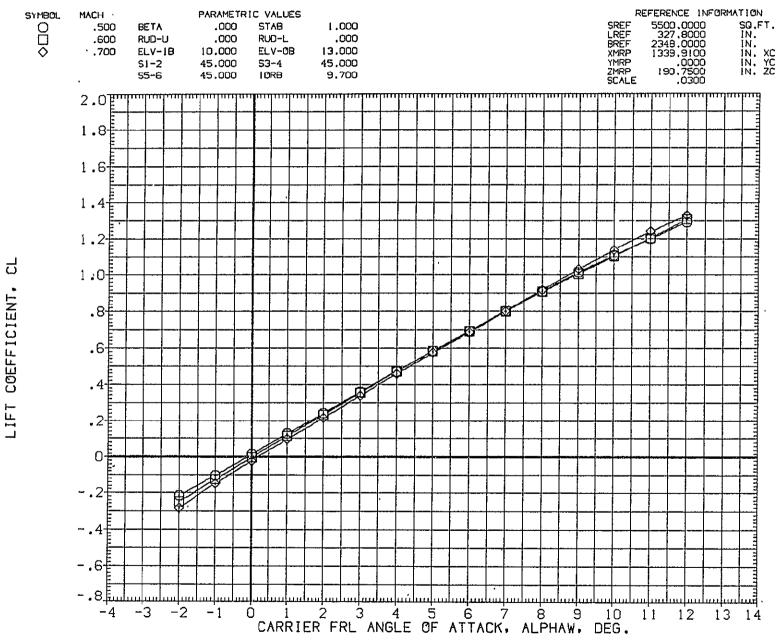


FIG. 11 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, ELEV=10 DEG TOTAL PAGE 41

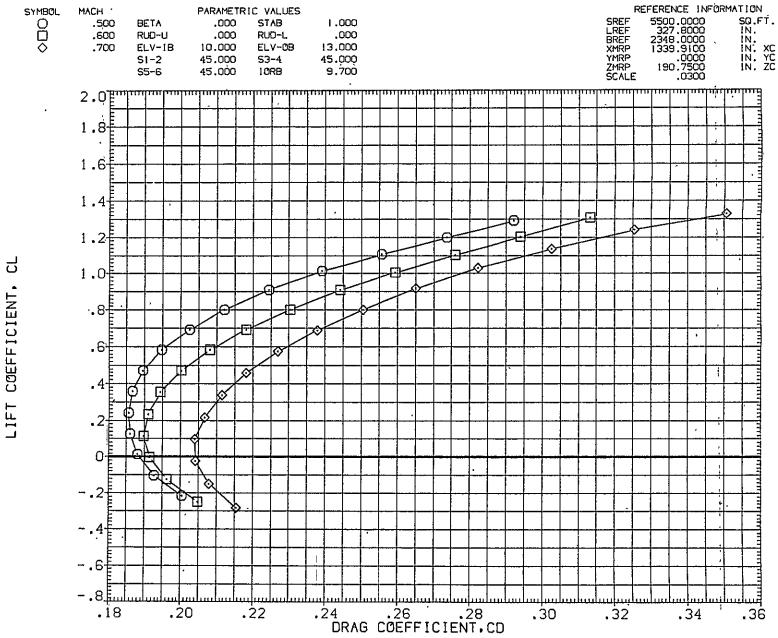


FIG. 11 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, ELEV=10 DEG TOTAL

PAGE 42

CA5 K1 F20 H15.6 V9.1 (PLUS. ORB TC5 AT38.1)(RGM064)

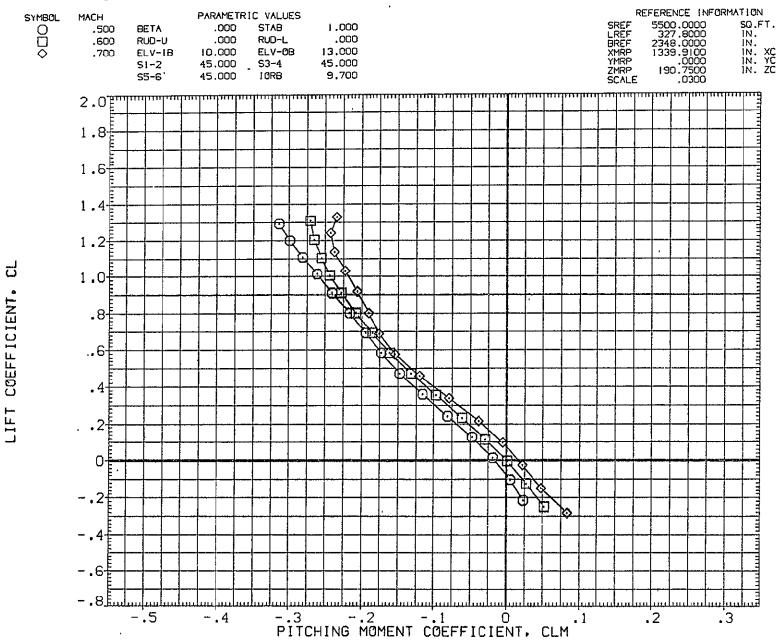
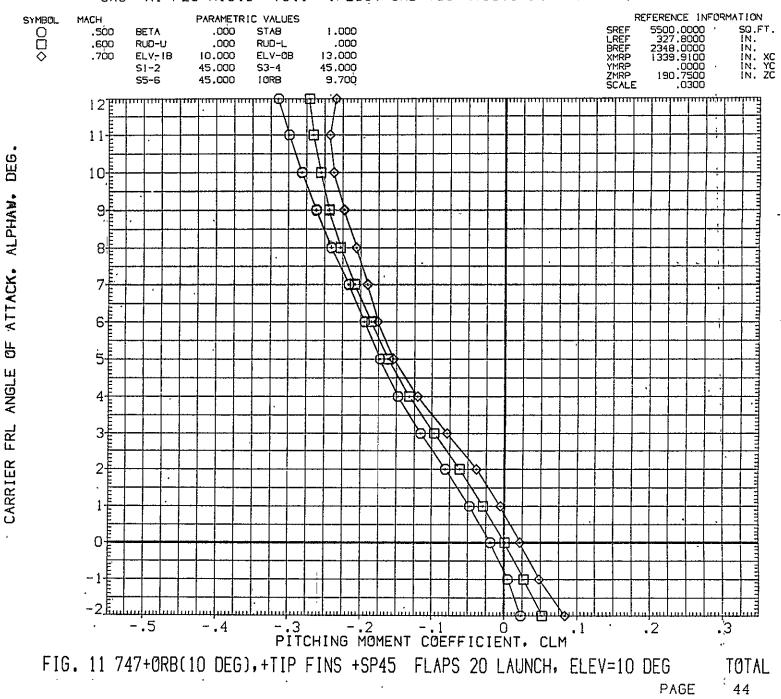
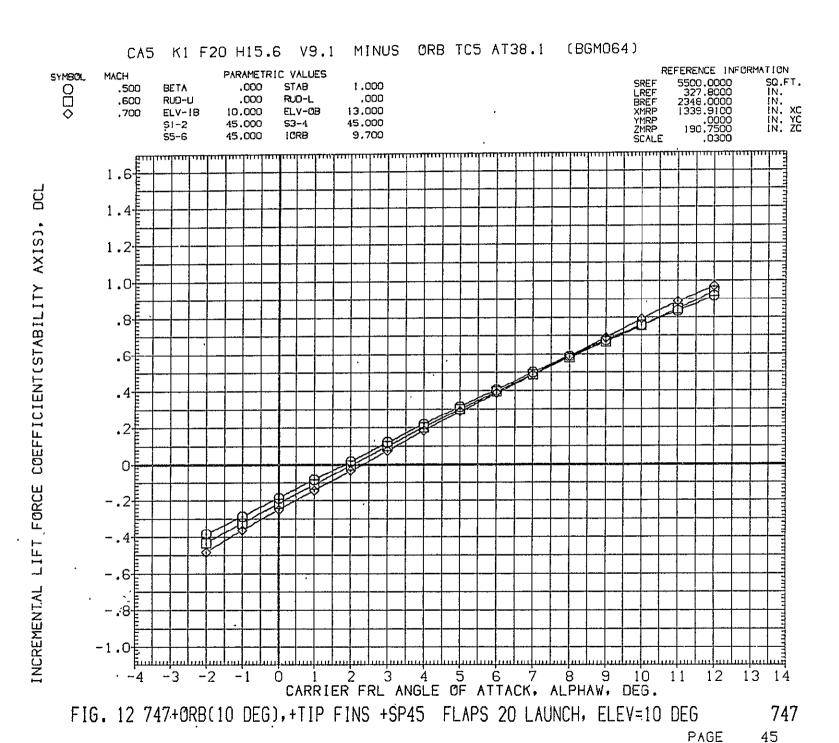


FIG. 11 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, ELEV=10 DEG

TOTAL





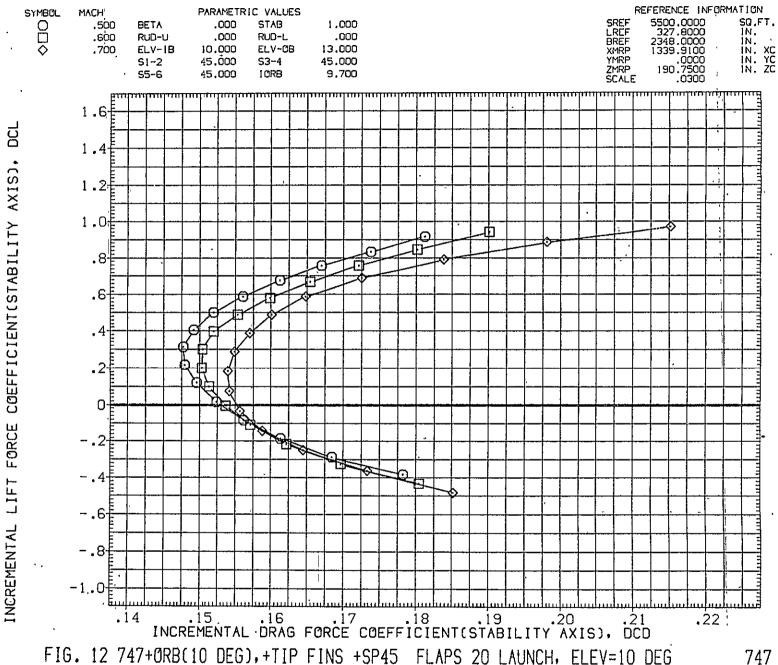
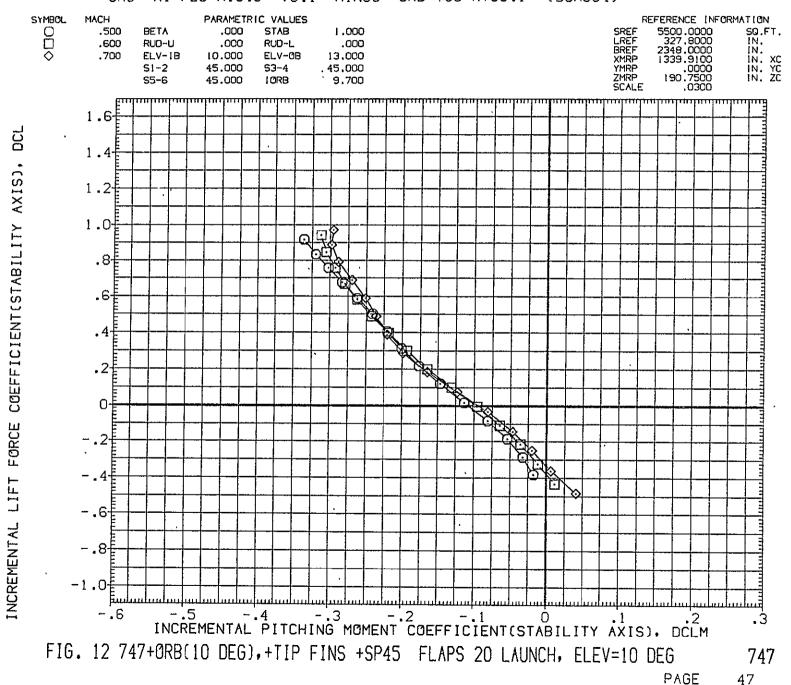


FIG. 12 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, ELEV=10 DEG

D^GE 46

CA5 K1 F20 H15.6 V9.1 MINUS ORB TC5 AT38.1 (BGM064)



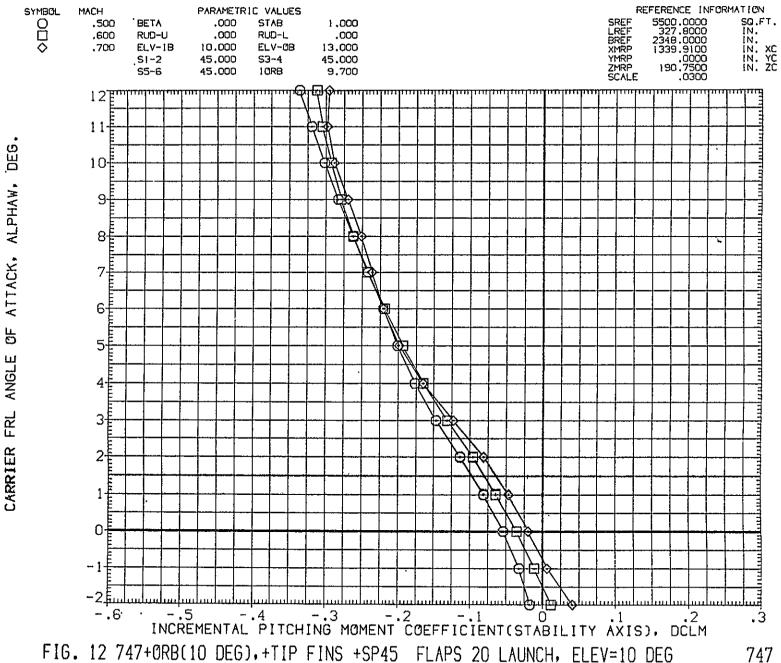


FIG. 12 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, ELEV=10 DEG

PAGE 48

FIG. 13 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, ELEV=10 DEG

PAGE 49

ORB

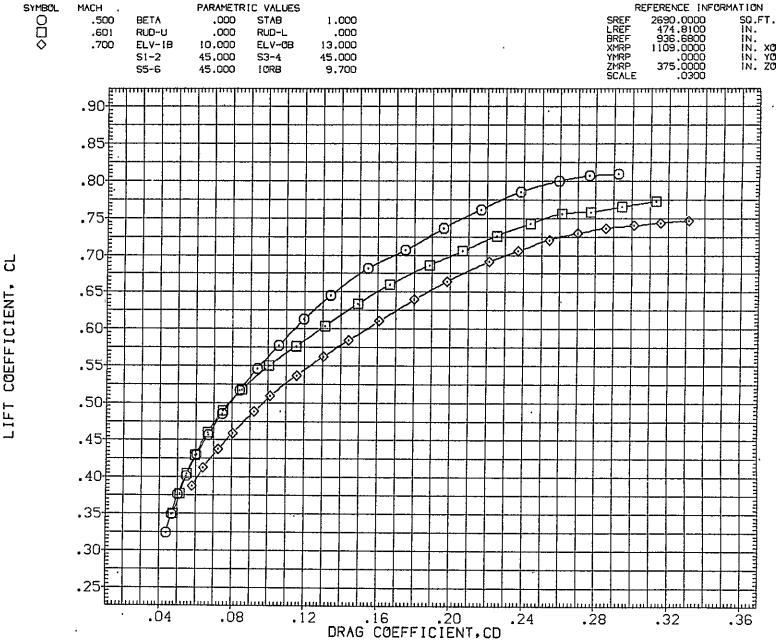


FIG. 13 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, ELEV=10 DEG ORB

CA5 ORB TC5 AT38.1 (PRES. K1 F20 H15.6 V9.1)(YGMO64)

H PARAMETRIC VALUES
500 BETA .000 STAB 1.000 SE

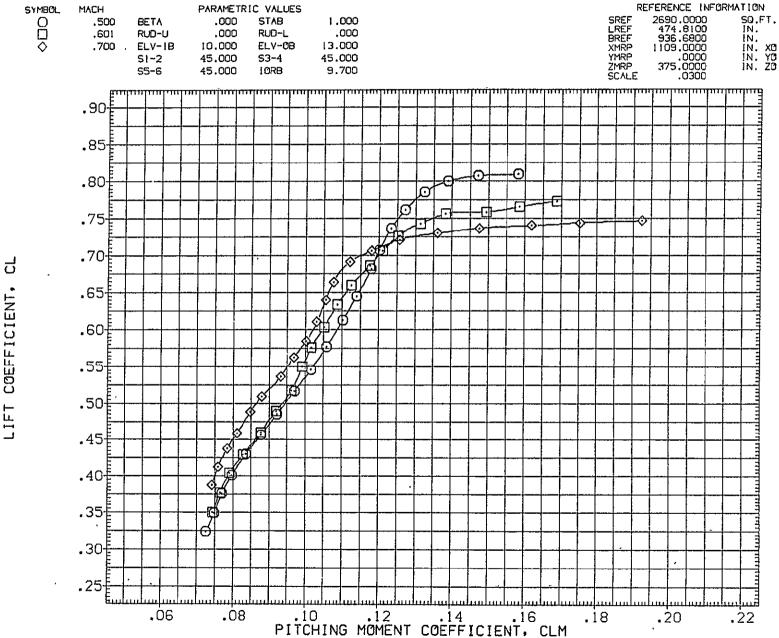


FIG. 13 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, ELEV=10 DEG

ORB

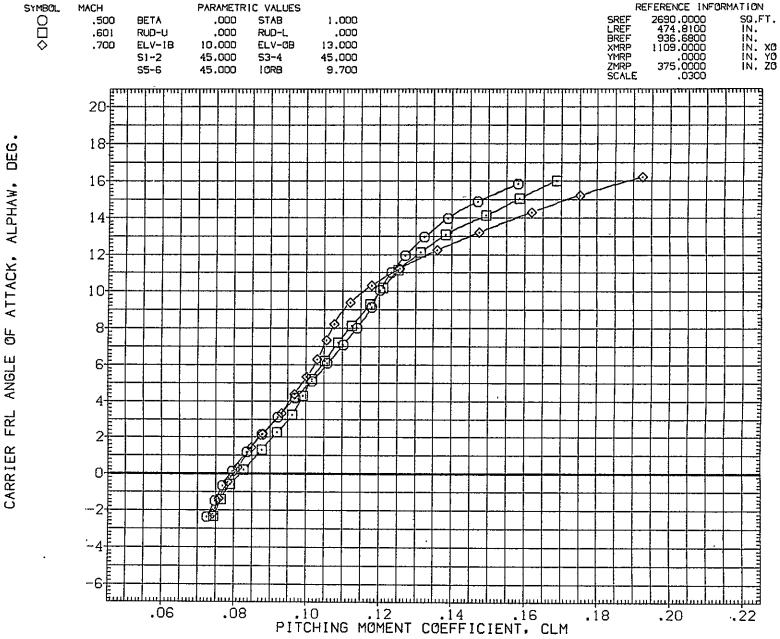


FIG. 13 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, ELEV=10 DEG ORB

CA5 K1 F20 H15.6 V9.1 (PLUS. ORB TC5 AT38.1)(FGM065)

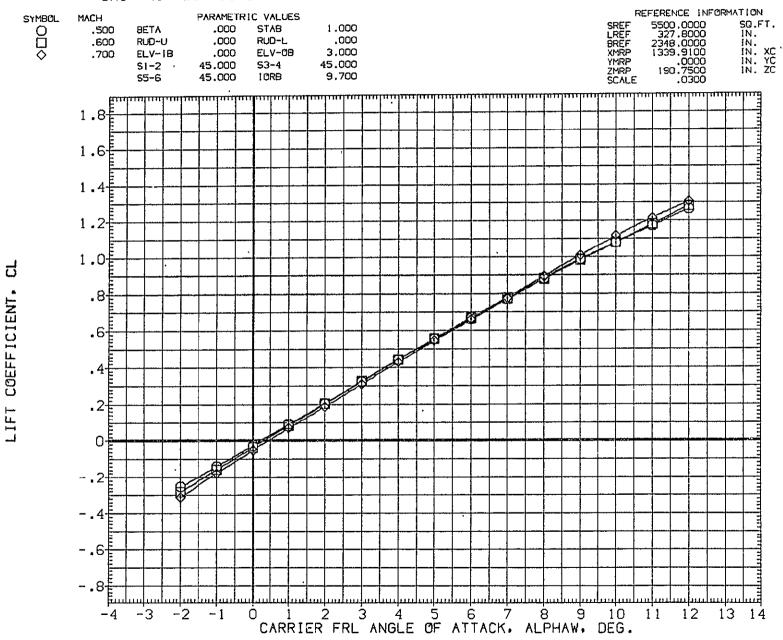


FIG. 14 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, BASELINE(S=+1) TOTAL PAGE 53

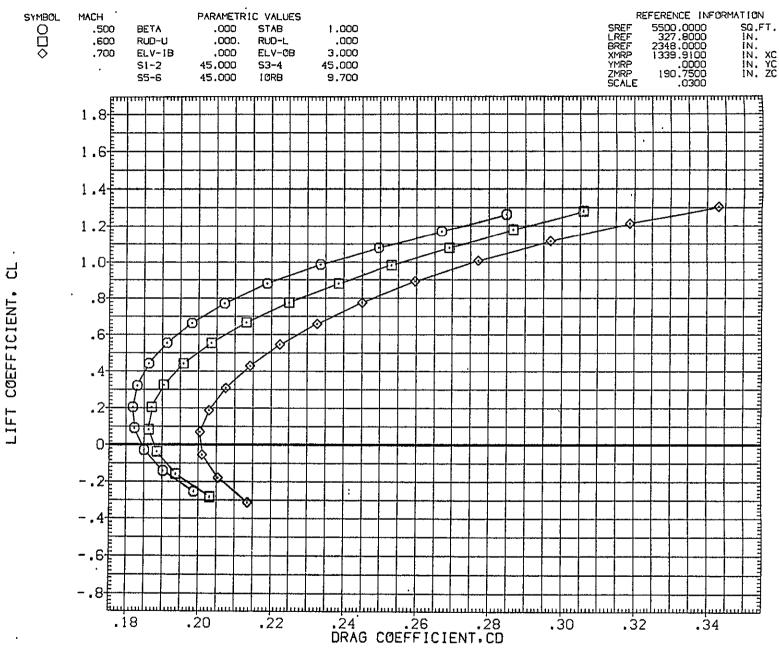


FIG. 14 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, BASELINE(S=+1) TOTAL PAGE 54

THE LOCK THE PARTY OF THE PARTY

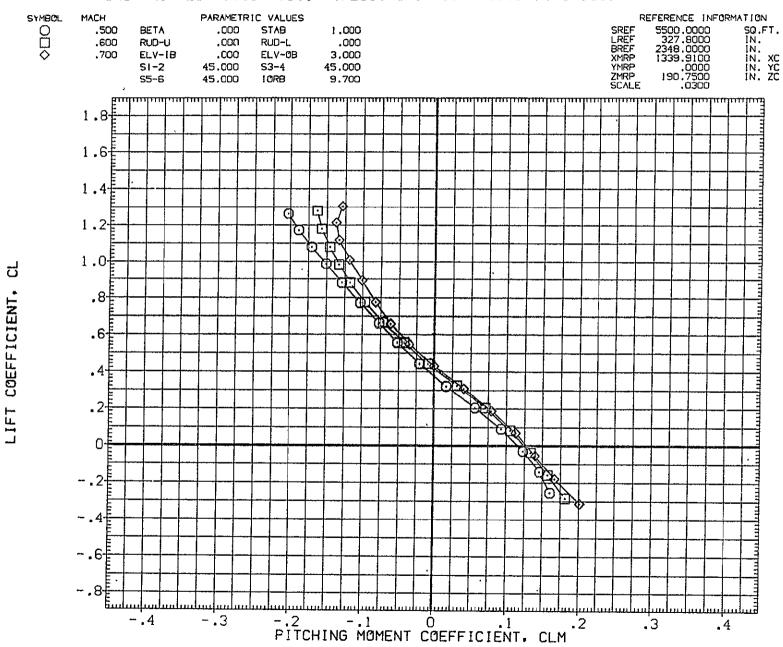


FIG. 14 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, BASELINE(S=+1) TOTAL PAGE 55

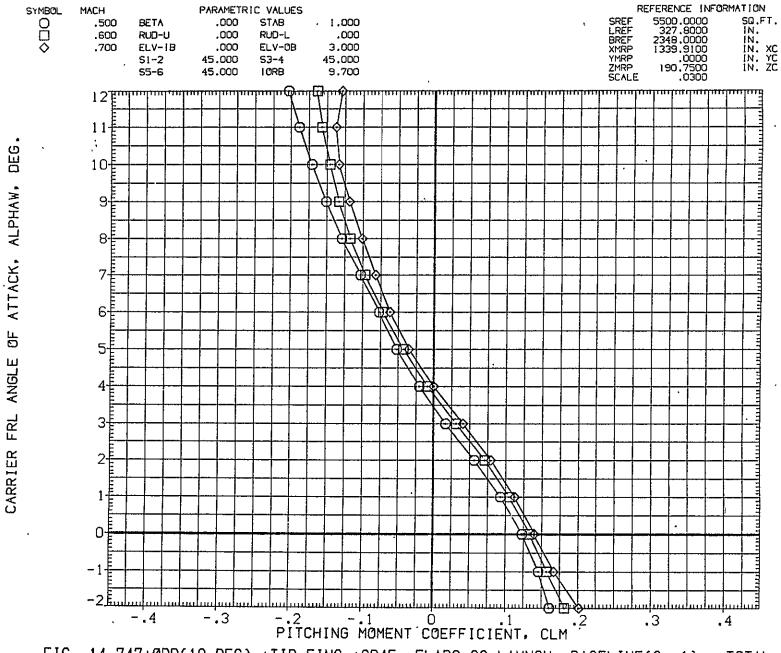


FIG. 14 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, BASELINE(S=+1) TOTAL PAGE 56

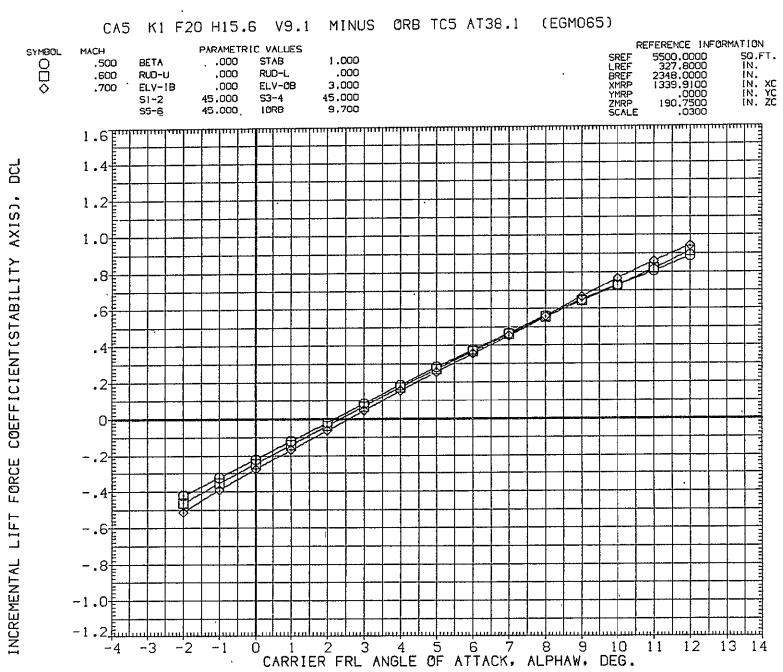


FIG. 15 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, BASELINE(S=+1)

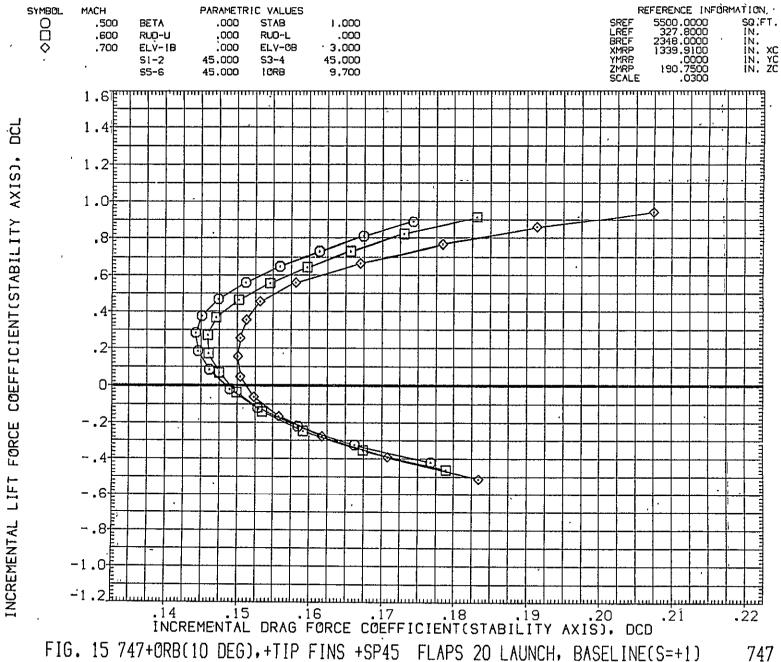
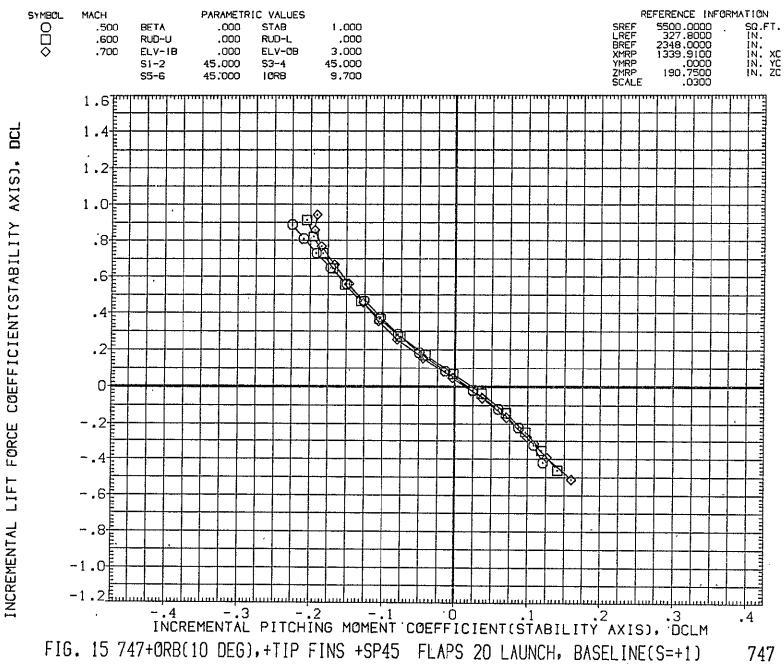
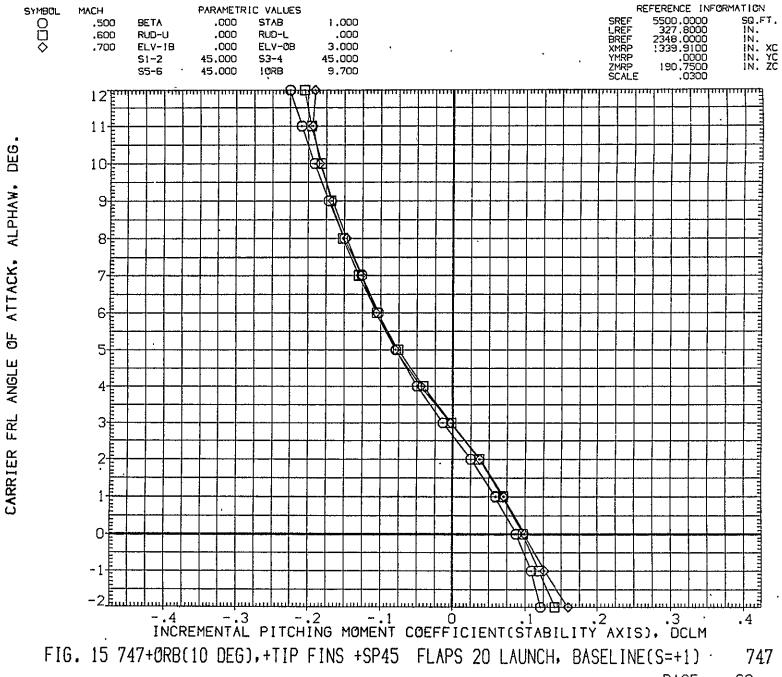


FIG. 15 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, BASELINE(S=+1)

CA5 K1 F20 H15.6 V9.1 MINUS ORB TC5 AT38.1 (EGM065)





60

FIG. 16 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, BASELINE(S=+1) ORB

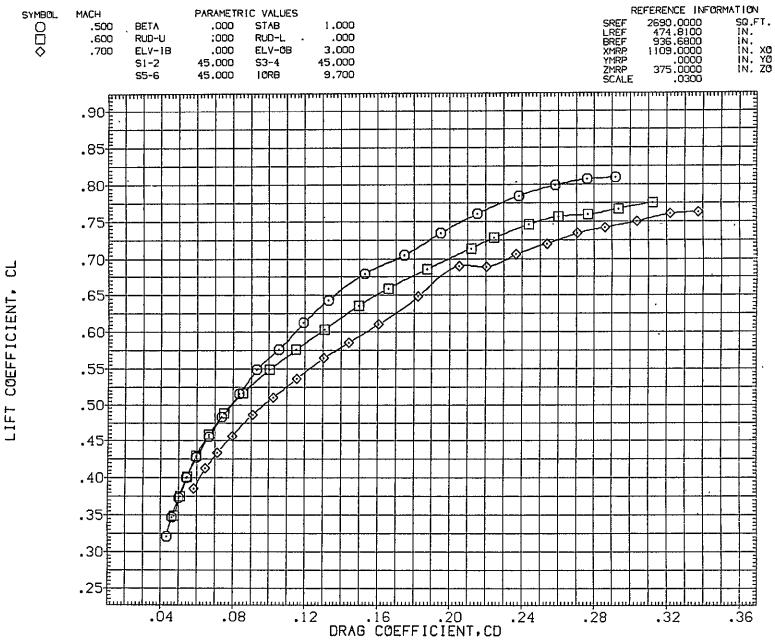
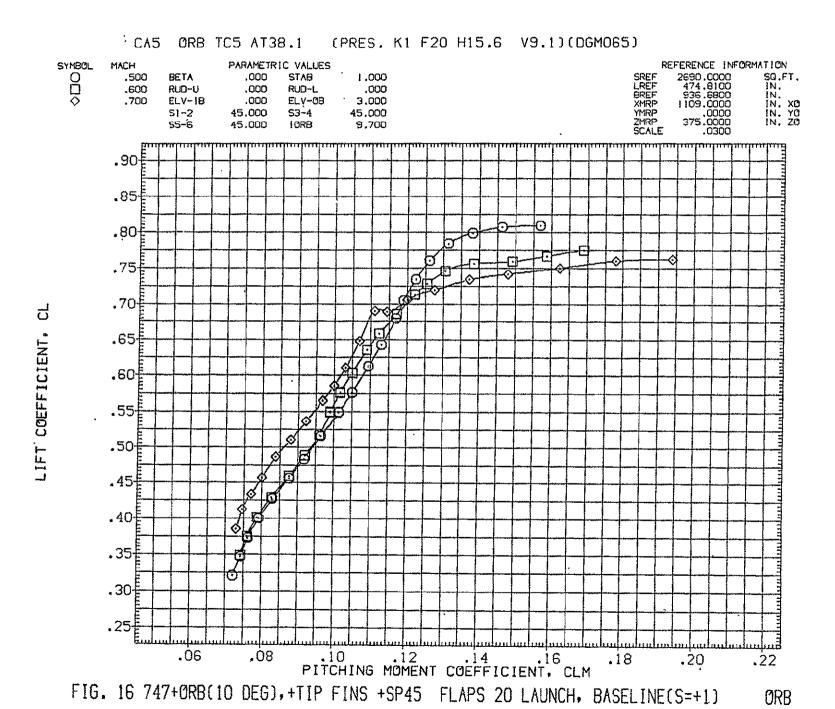


FIG. 16 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, BASELINE(S=+1) ORB
PAGE 62



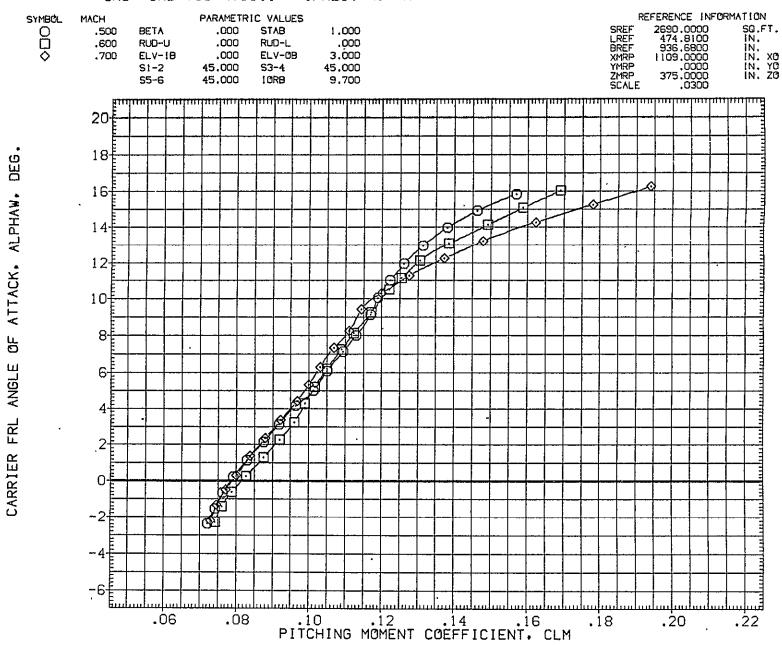


FIG. 16 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, BASELINE(S=+1) ORB



CA5 K1 F20 H15.6 V9.1 (PLUS, ORB TC5 AT38.1)(RGMO68)

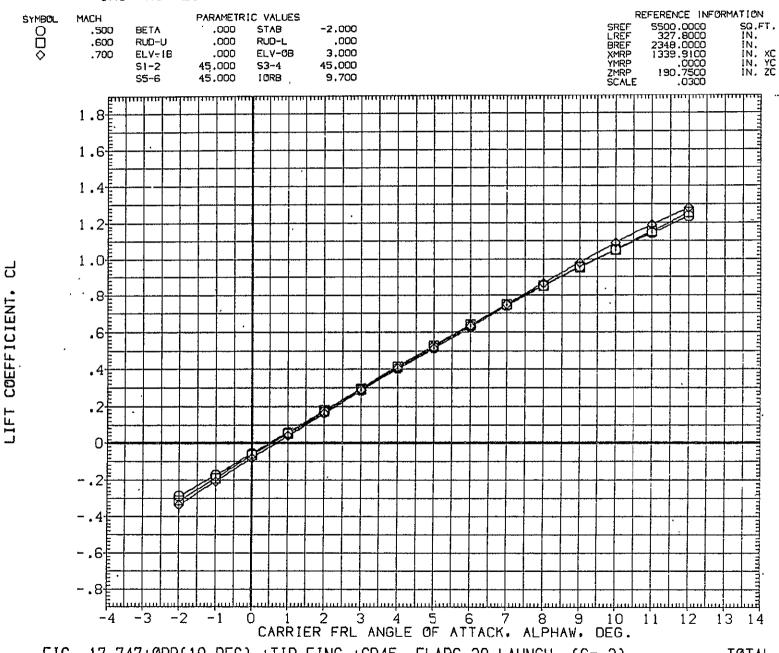


FIG. 17 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, (S=-2)

TOTAL 65

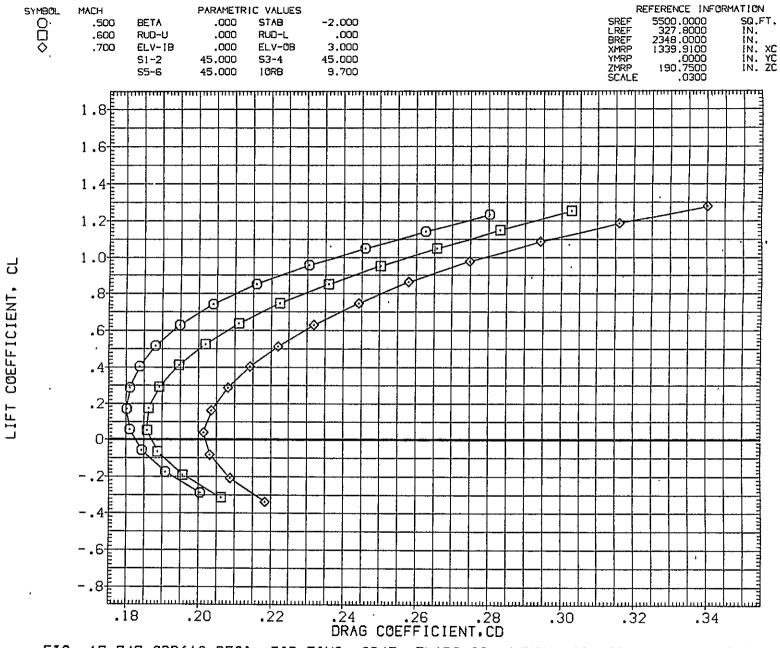


FIG. 17 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, (S=-2)

TOTAL

CA5 K1 F20 H15.6 V9.1 (PLUS. ORB TC5 AT38.1)(RGM068)

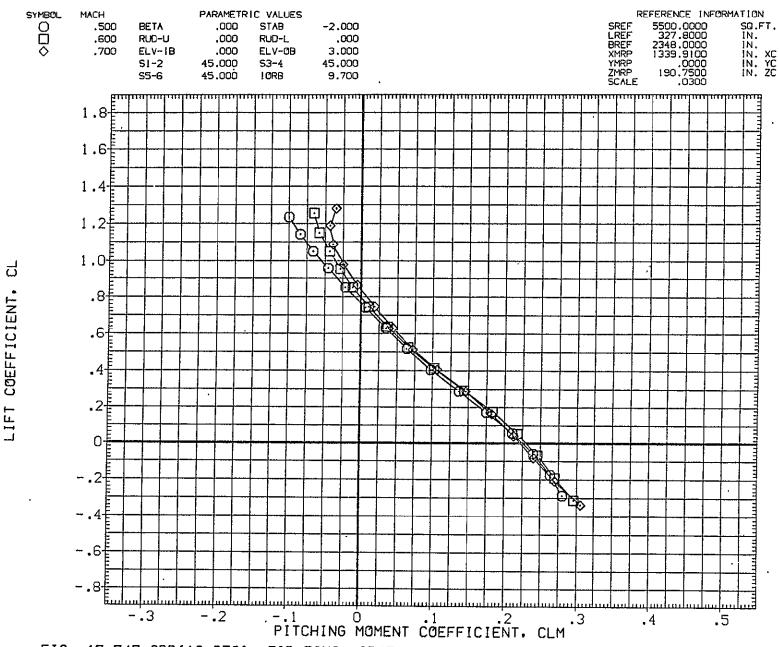


FIG. 17 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, (S=-2)

TOTAL

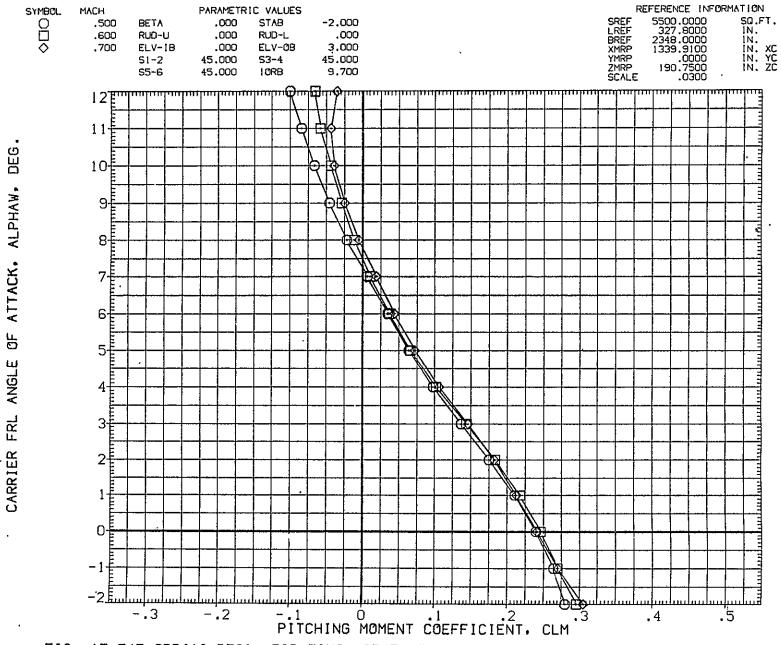


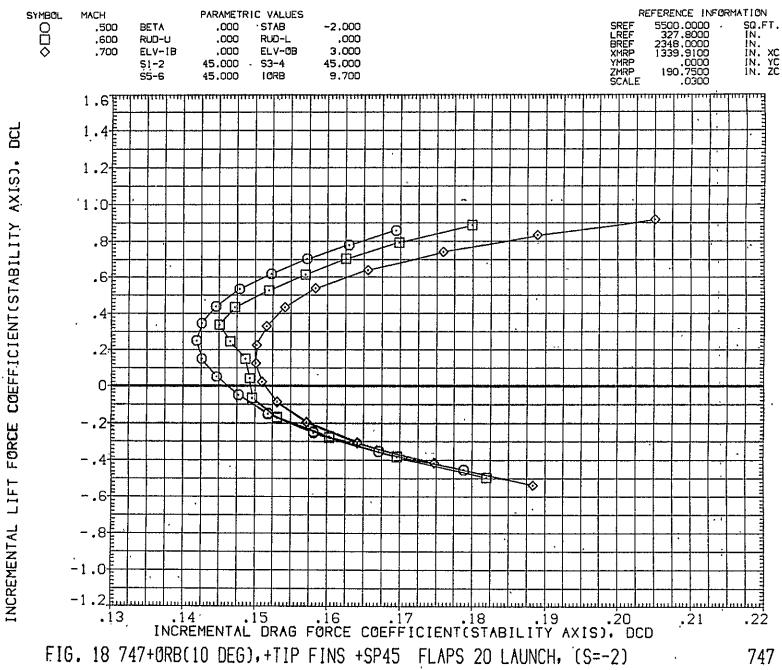
FIG. 17 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, (S=-2)

TOTAL

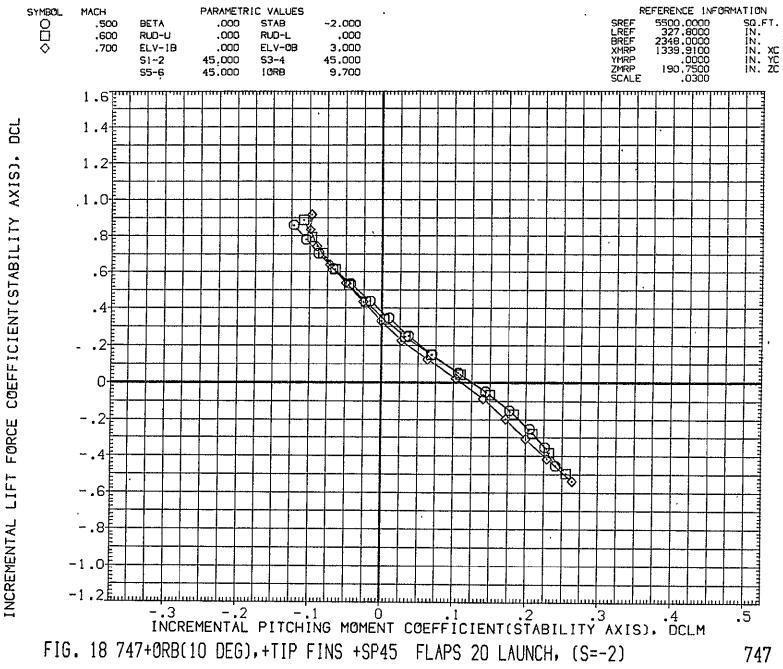
CA5 K1 F20 H15.6 V9.1 MINUS ORB TC5 AT38.1 (BGM068) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH 5500.0000 327.8000 2348.0000 1339.9100 .0000 190.7500 .0300 SQ.FT. 000 BETA .000 STAB -2,000 SREF LREF .500 .000 RUD-L .000 .600 RUD-U IN. XC IN. YC IN. ZC BREF XMRP YMRP .700 ELV-IB .000 ELV-0B 3,000 45.000 S1-2 45.000 53-4 ZMRP SCALE 9.700 S5-6 45.000 LORB 1.6Em ద 1.4卡 FORCE COEFFICIENT(STABILITY AXIS). 1.2 1.0 .8[.6[INCREMENTAL -1.0 1 2 3 4 5 6 7 8 9 10 CARRIER FRL ANGLE OF ATTACK, ALPHAW, DEG. -3 -2 12 13 FIG. 18 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, (S=-2)

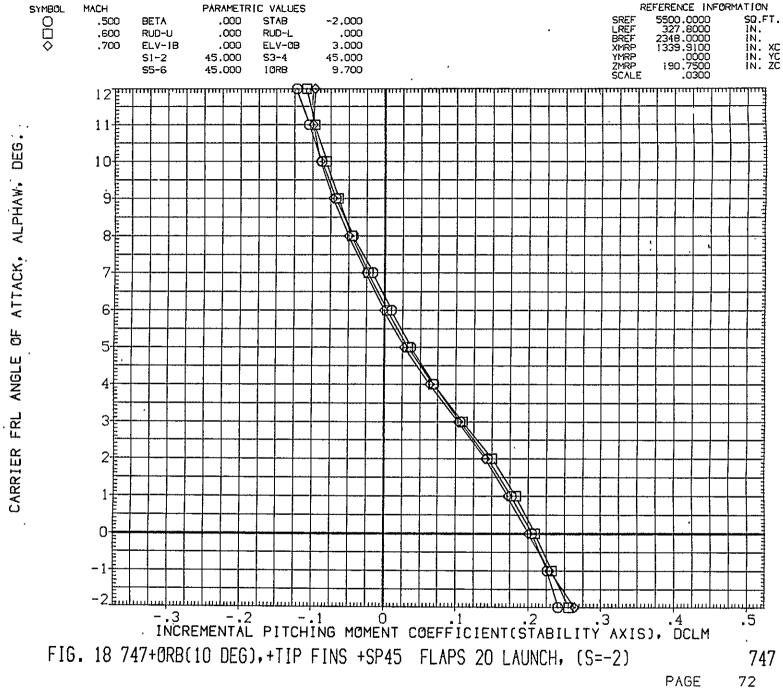
747

69

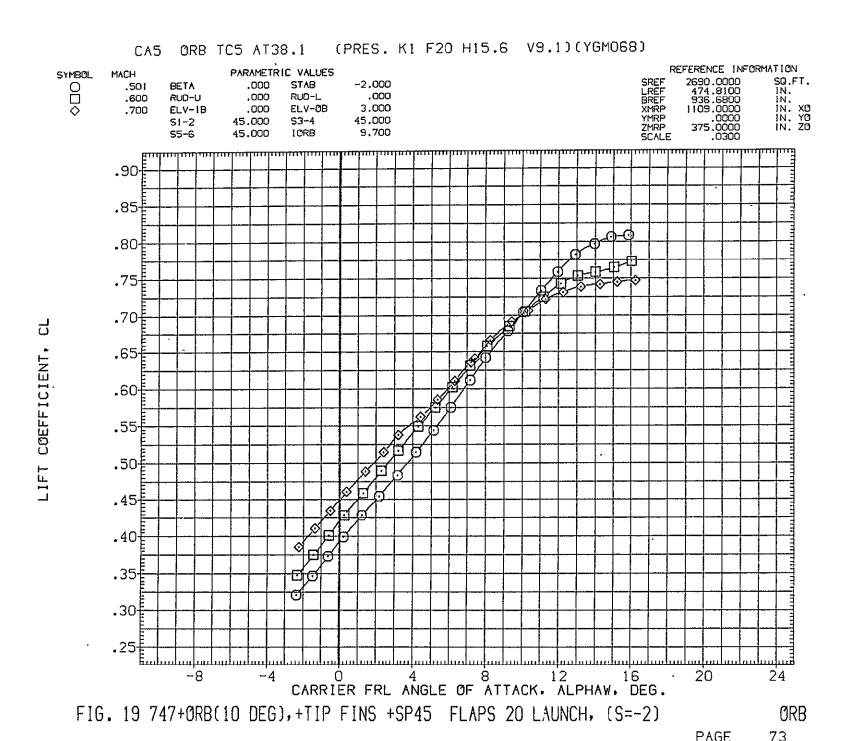


CA5 K1 F20 H15.6 V9.1 MINUS ORB TC5 AT38.1 (BGM068)









CA5 ORB TC5 AT38.1 (PRES. K1 F20 H15.6 V9.1)(YGM068)

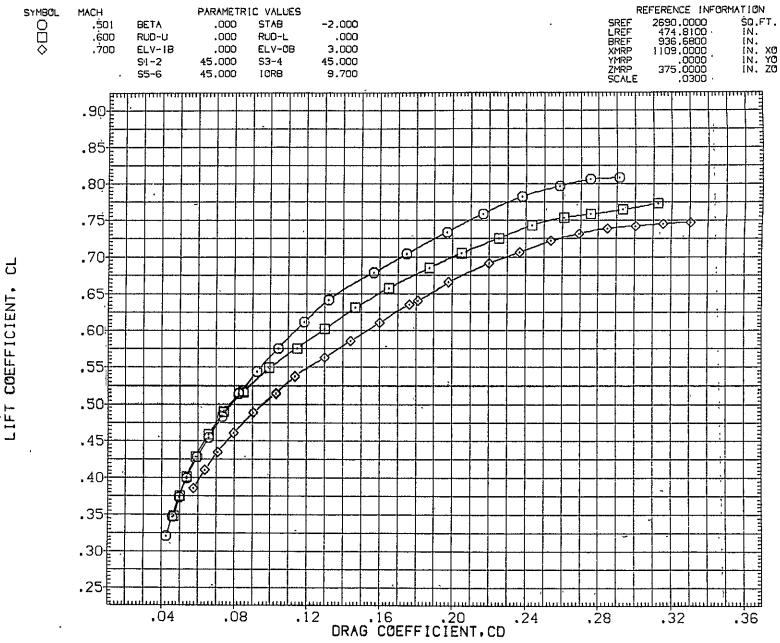


FIG. 19 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, (S=-2)

ORB



FIG. 19 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, (S=-2)

PITCHING MOMENT COEFFICIENT, CLM

75 .

CA5 ORB TC5 AT38.1 (PRES. K1 F20 H15.6 V9.1)(YGM068)

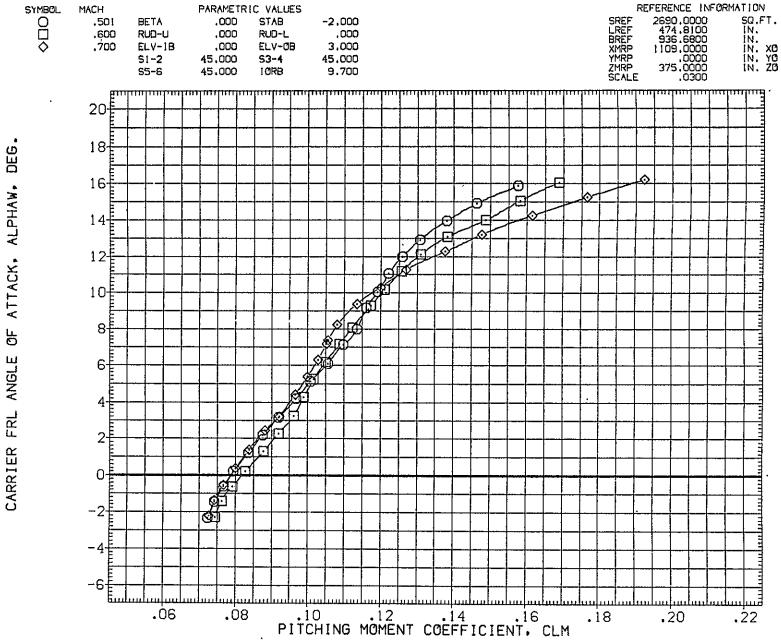


FIG. 19 747+0RB(10 DEG),+TIP FINS +SP45 FLAPS 20 LAUNCH, (S=-2)

ORB

CA5 K1 F20 H15.6 V9.1 (PLUS. ORB TC5 AT38.1)(RGM070) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH 5500,0000 327,8000 2348,0000 1339,9100 ,0000 190,7500 SREF LREF BREF XMRP YMRP ZMRP SCALE 000 .000 STAB 1,000 .500 BETA IN. IN. IN. XC IN. YC IN. ZC RUD-L .000 .600 RUD-U .000 3.000 - .000 ELV-0B .700 -ELV-IB 45.000 S1-2 .000 53-4 20,000 10RB 9,700 **S5-6** 2.0 1.8 1.6[1.4+ 占 COEFFICIENT, 1.0 .8-

FIG. 20 747+0RB(10 DEG),+TIP FINS +STD SPD BRAKES,FLAPS 20 LAUNCH(S=+1)

PAGE 77

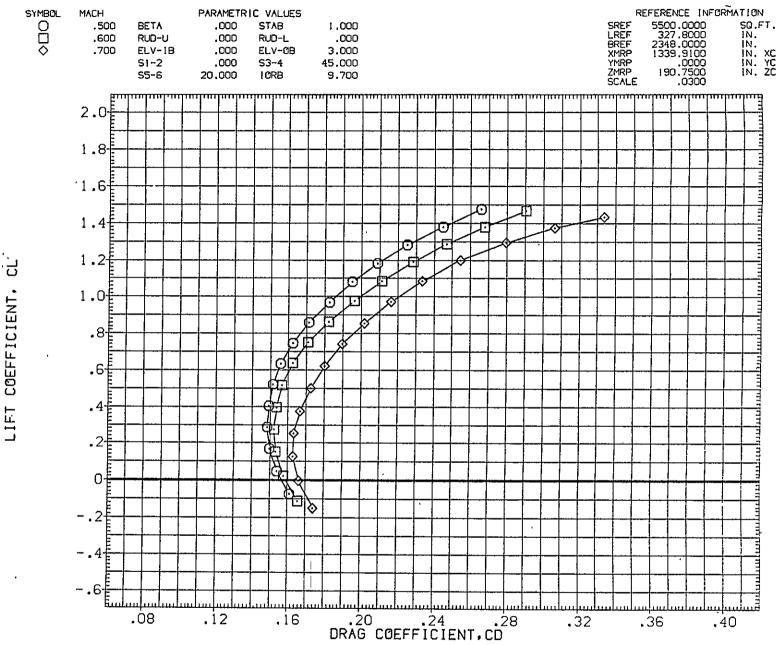


FIG. 20 747+0RB(10 DEG),+TIP FINS +STD SPD BRAKES,FLAPS 20 LAUNCH(S=+1) TOTAL PAGE 78

CA5 K1 F20 H15.6 V9.1 (PLUS. ORB TC5 AT38.1)(RGM070)

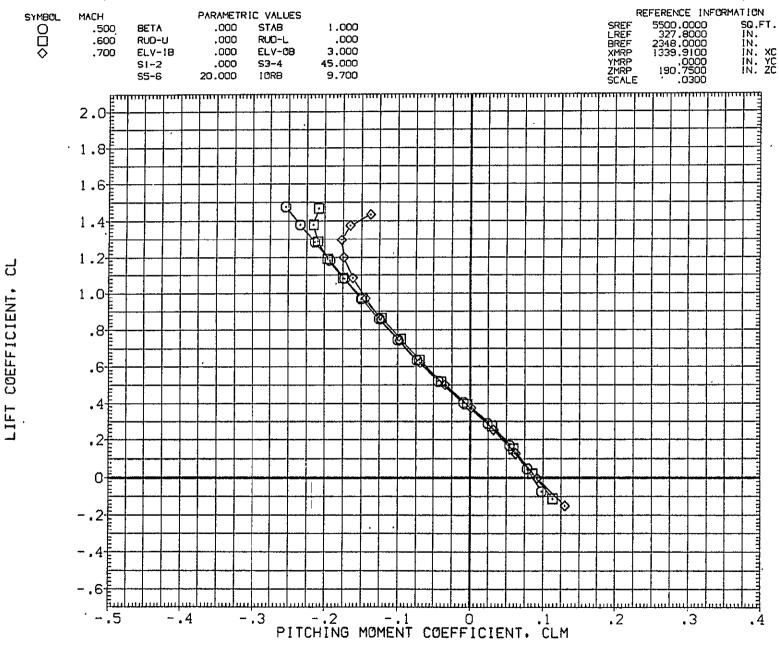


FIG. 20 747+0RB(10 DEG),+TIP FINS +STD SPD BRAKES,FLAPS 20 LAUNCH(S=+1) TOTAL

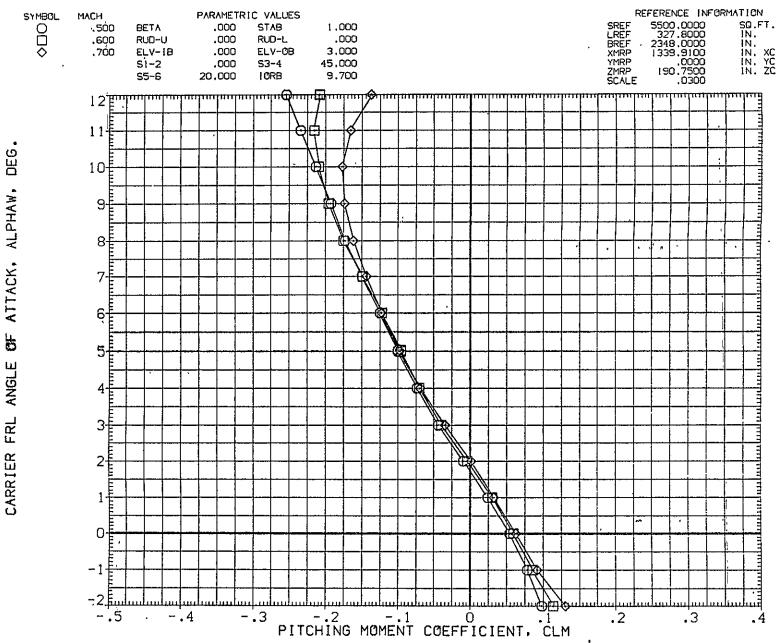


FIG. 20 747+0RB(10 DEG),+TIP FINS +STD SPD BRAKES,FLAPS 20 LAUNCH(S=+1) TOTAL

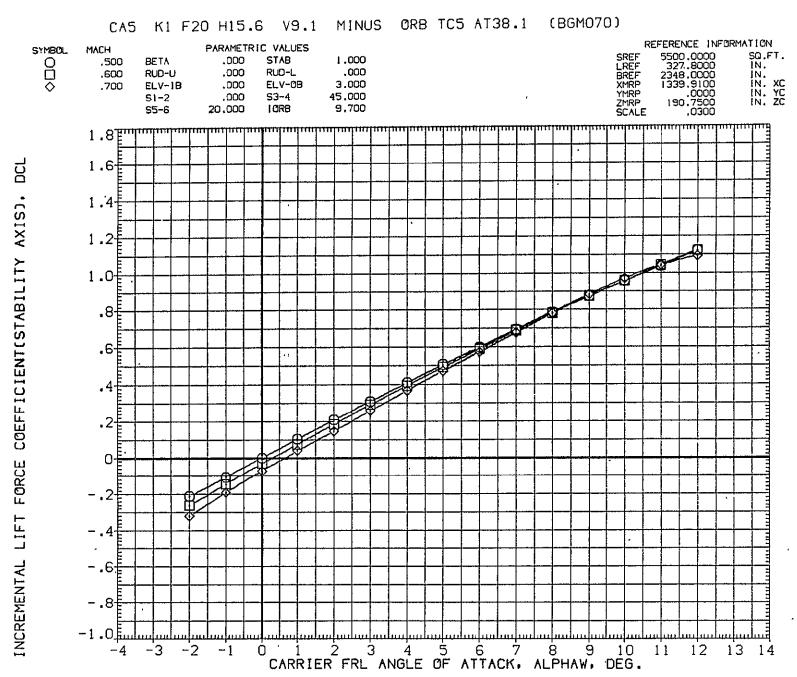


FIG. 21 747+0RB(10 DEG),+TIP FINS +STD SPD BRAKES,FLAPS 20 LAUNCH(S=+1)

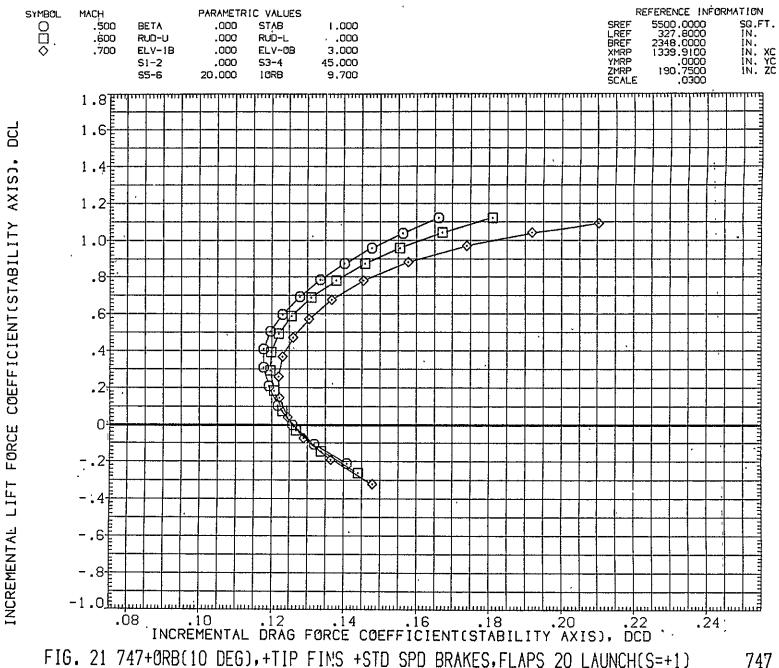
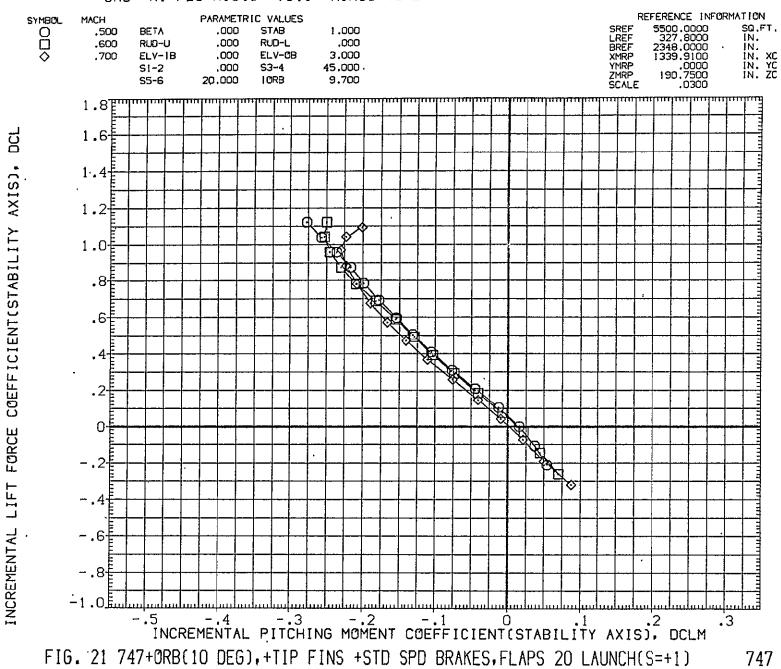


FIG. 21 747+0RB(10 DEG),+TIP FINS +STD SPD BRAKES,FLAPS 20 LAUNCH(S=+1)



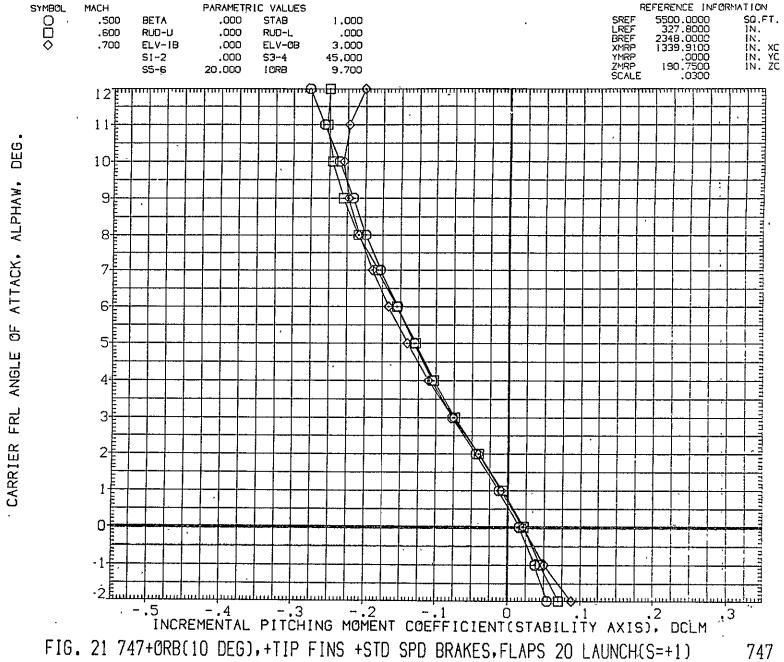


FIG. 21 747+0RB(10 DEG),+TIP FINS +STD SPD BRAKES,FLAPS 20 LAUNCH(S=+1) PAGE 84



FIG. 22 747+0RB(10 DEG),+TIP FINS +STD SPD BRAKES,FLAPS 20 LAUNCH(S=+1)

ØRB

85

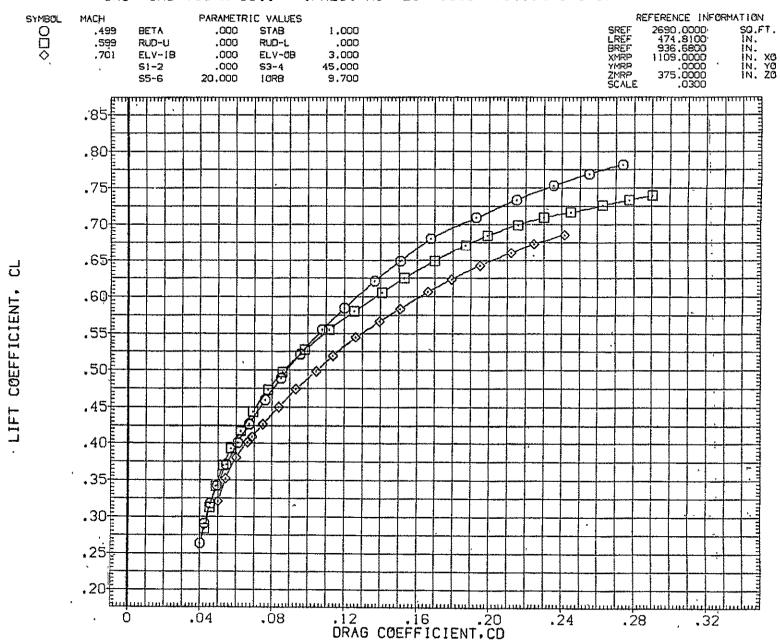
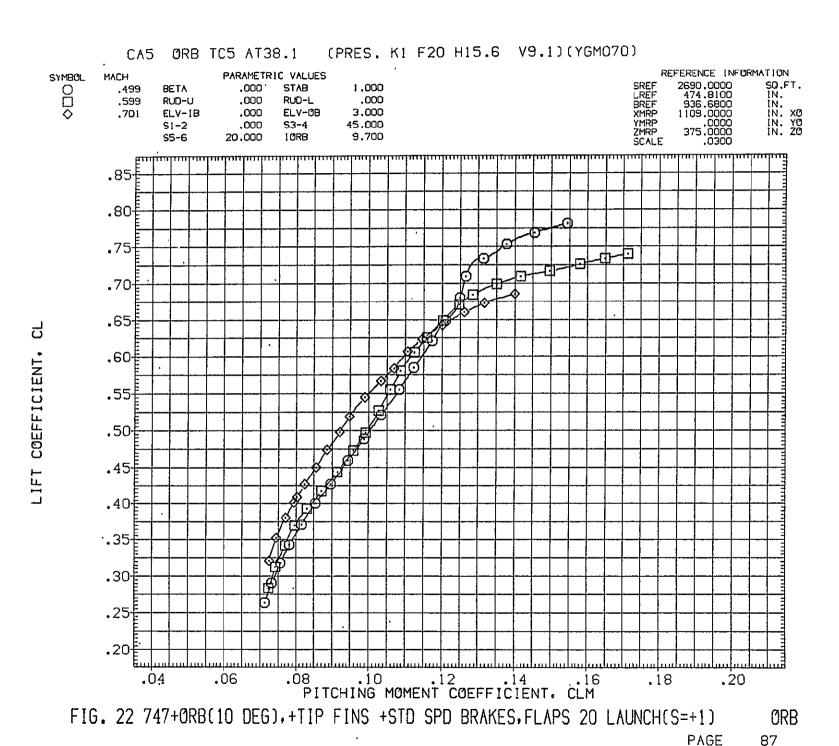


FIG. 22 747+0RB(10 DEG),+TIP FINS +STD SPD BRAKES,FLAPS 20 LAUNCH(S=+1) ORB





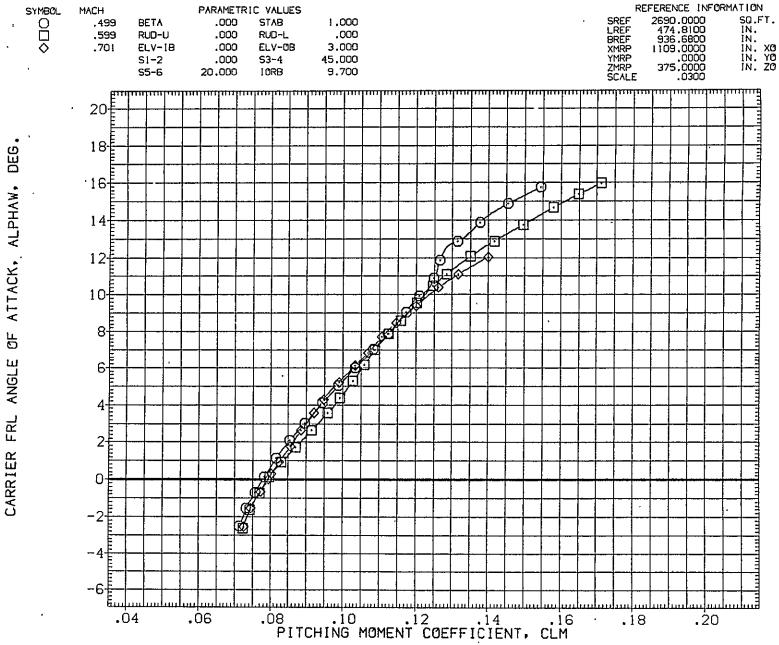


FIG. 22 747+0RB(10 DEG),+TIP FINS +STD SPD BRAKES,FLAPS 20 LAUNCH(S=+1)

ØRB

CA5 K1 F20 H15.6 V9.1 (PLUS. ORB TC5 AT38.1)(RGM071)

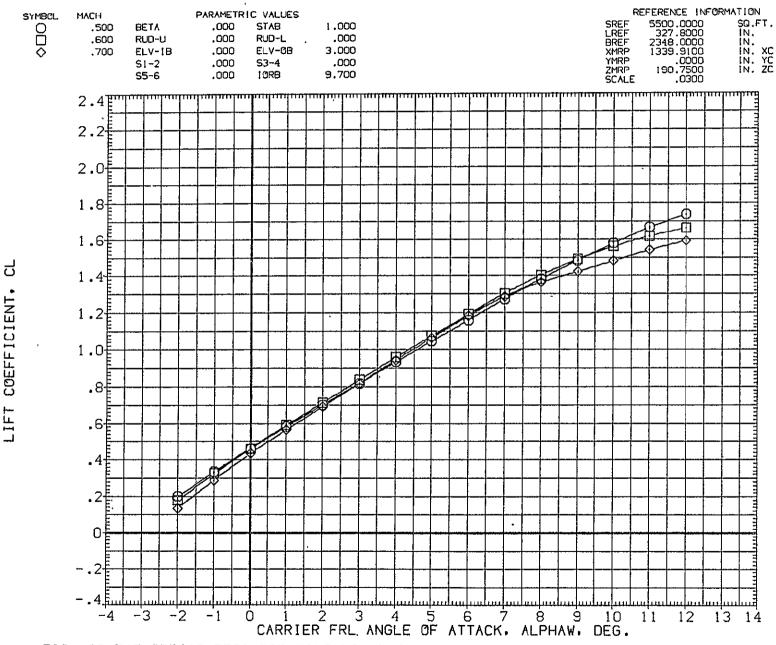


FIG. 23 747+0RB(10 DEG)+TIP FINS, NO SPOILERS, FLAPS 20 LAUNCH(S=+1)

TOTAL

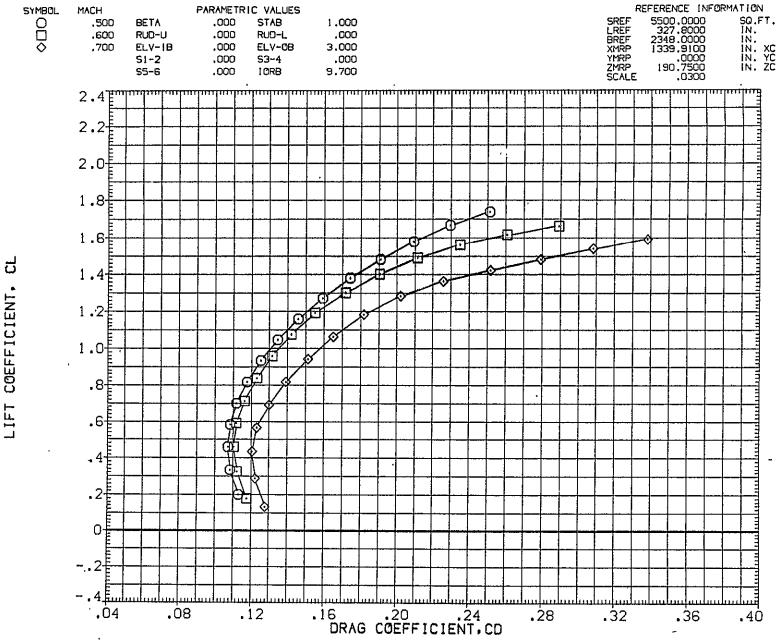


FIG. 23 747+0RB(10 DEG)+TIP FINS,NO SPOILERS,FLAPS 20 LAUNCH(S=+1)
PAGE 90

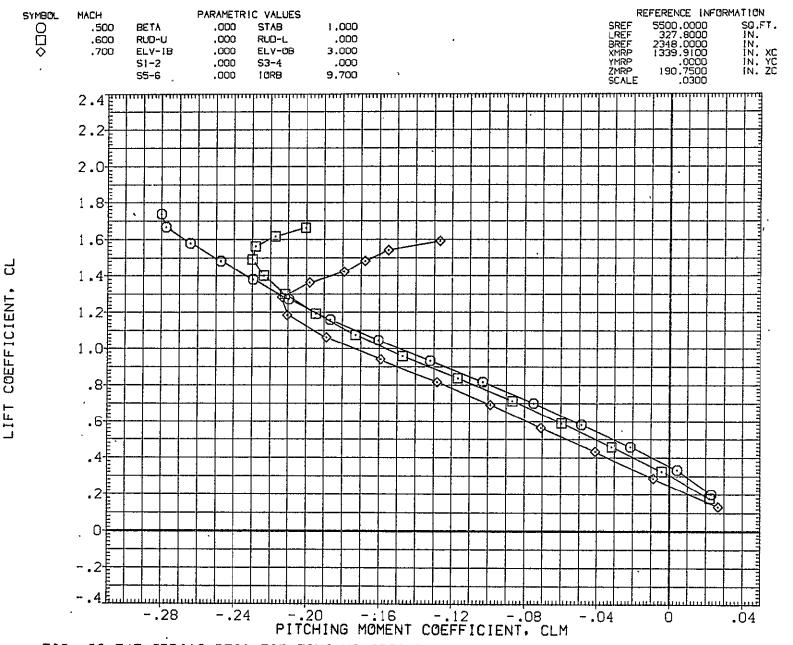


FIG. 23 747+0RB(10 DEG)+TIP FINS.NO SPOILERS.FLAPS 20 LAUNCH(S=+1)

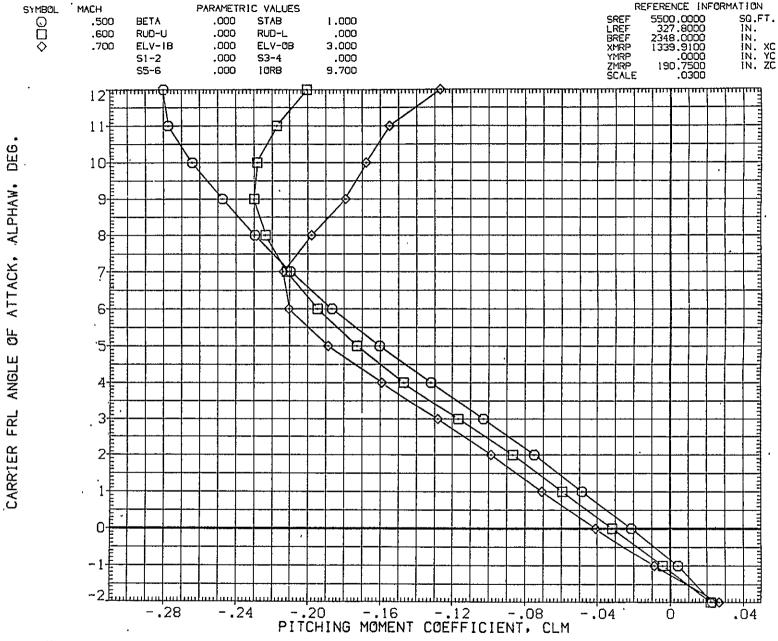


FIG. 23 747+0RB(10 DEG)+TIP FINS,NO SPOILERS,FLAPS 20 LAUNCH(S=+1)

FIG. 24 747+0RB(10 DEG)+TIP FINS.NO SPOILERS.FLAPS 20 LAUNCH(S=+1)

CARRIER FRL ANGLE OF ATTACK, ALPHAW, DEG.

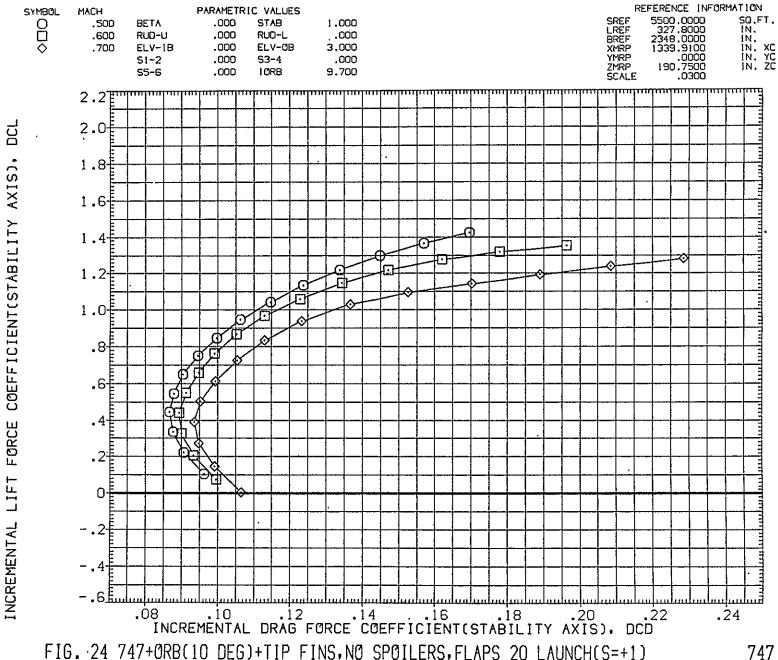
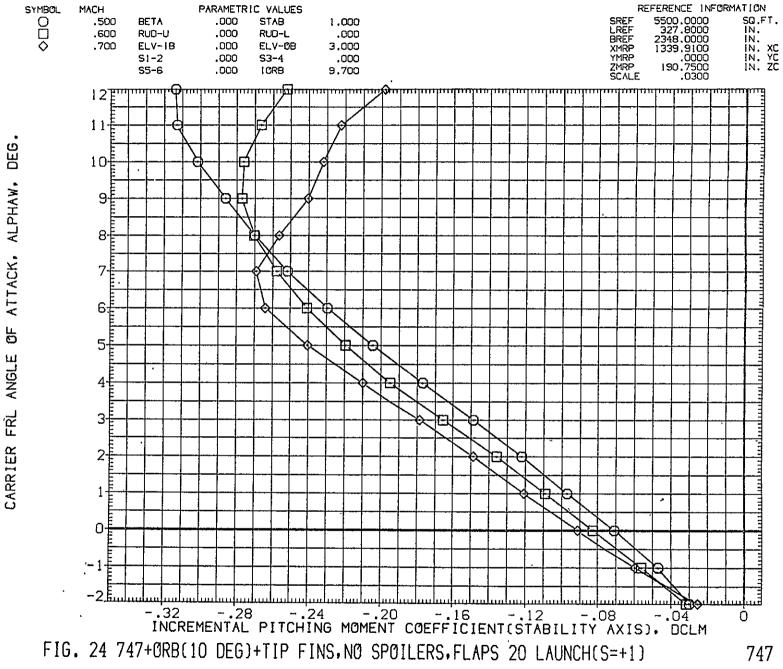


FIG. 24 747+0RB(10 DEG)+TIP FINS,NO SPOILERS,FLAPS 20 LAUNCH(S=+1)

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CA5 K1 F20 H15.6 V9.1 MINUS ORB TC5 AT38.1 (BGM071) PARAMETRIC VALUES REFERENCE INFORMATION SYMBOL MACH SREF LREF 5500.0000 000 .000 STAB .500 BETA 1.000 ĬŇ. 327.8000 .000 RUD-L .600 RUD-U .000 327,9000 2348,0000 1339,9100 .0000 190,7500 BREF XMRP IN. .700 ELV-1B .000 ELV-OB 3.000 IN. XC IN. YC IN. ZC YMRP ZMRP SCALE S1-2 .000 S3-4 .000 S5-6 10RB 9.700 .000 DC 2.0 AXIS), 1.8 1.6 COEFFICIENT(STABILITY 0 1.4 1.2 1.0 .8-.6‡ O. FORCE .4+ LIFT 0-INCREMENTAL -.20 -.24 -.28 -.16 -.12 ~.08 INCREMENTAL PITCHING MOMENT COEFFICIENT(STABILITY AXIS), DCLM FIG. 24 747+0RB(10 DEG)+TIP FINS, NO SPOILERS, FLAPS 20 LAUNCH(S=+1)



ORB

97

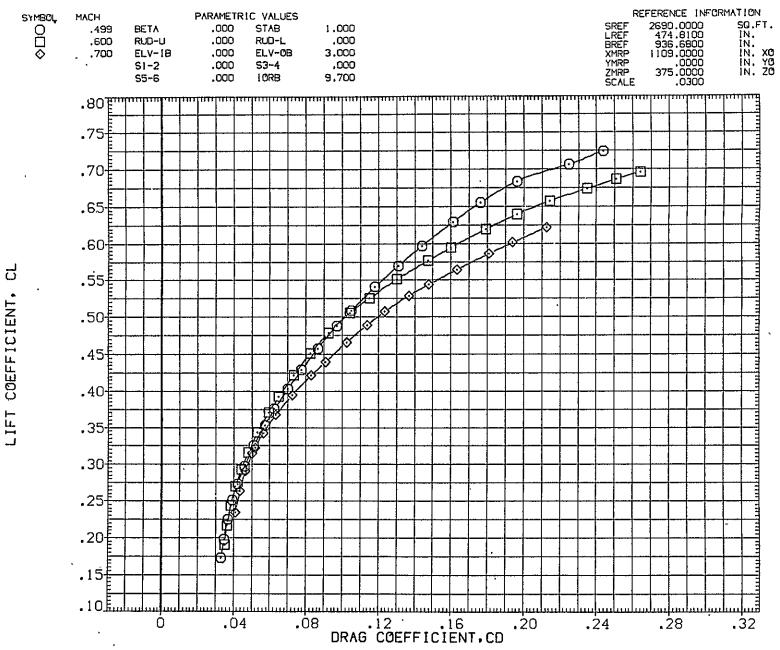


FIG. 25 747+0RB(10 DEG)+TIP FINS,NØ SPØILERS,FLAPS 20 LAUNCH(S=+1)

PAGE

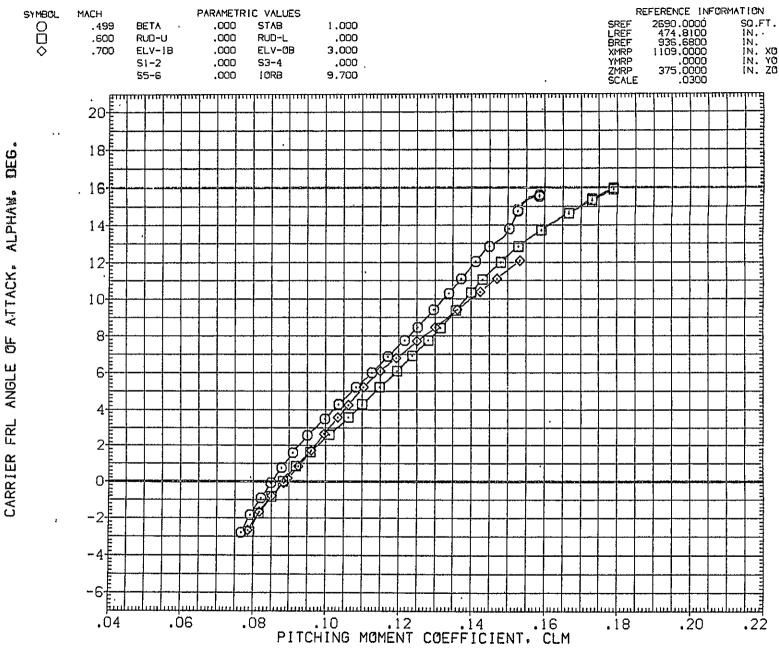


FIG. 25 747+0RB(10 DEG)+TIP FINS.NO SPOILERS.FLAPS 20 LAUNCH(S=+1)

· ORB

CA5 K1 F20 H15.1A V9.1 (PLUS. ORB TC5 AT38.1)(RGM074) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH SREF LREF BRCF 5500.0000 327.8000 SQ.FT. 000 BETA .000 STAB 1,000 .500 IN. XC IN. YC IN. ZC RUD-L ,000 RUD-U .000 .600 ELV-OB 3.000 .700 ELV-18 ,000 XMRP YMRP ZMRP SCALE S1-2 .000 S3~4 .000 IORB 9,700 .000 S5-6 2.4 2.2 2.0 1.8 1.6 겁 1.4 COEFFICIENT. 1.2 1.0 .8[.6+ .2 3 5 6 Ż Ò 9 CARRIER FRL ANGLE OF ATTACK, ALPHAW, DEG.

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PAGE

FIG. 26 747+0RB(10 DEG), BASIC TAIL, FLAPS 20 LAUNCH(S=+1)

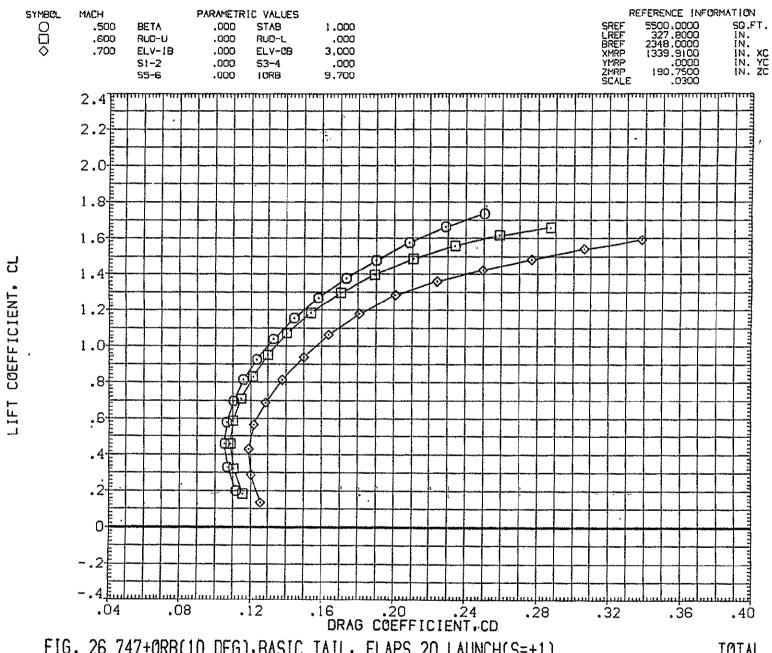


FIG. 26 747+0RB(10 DEG), BASIC TAIL, FLAPS 20 LAUNCH(S=+1)

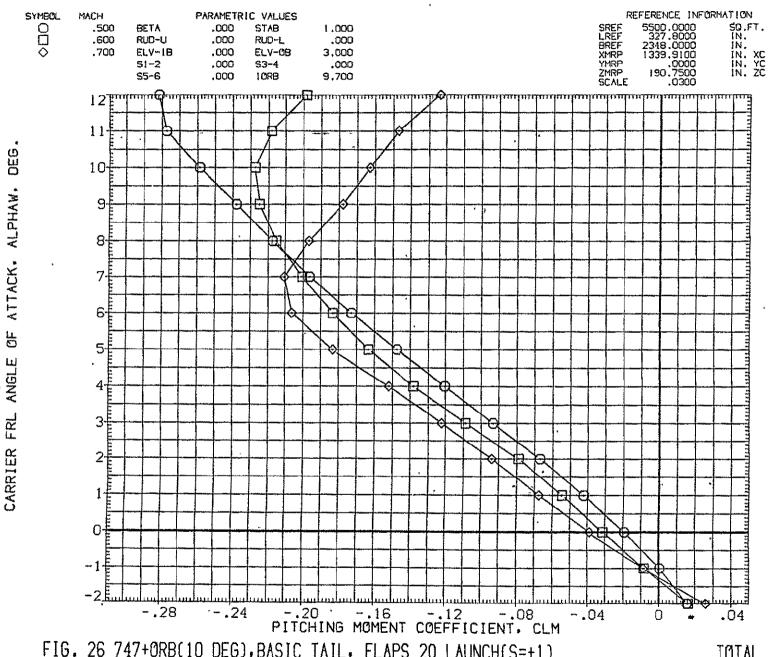


FIG. 26 747+0RB(10 DEG), BASIC TAIL, FLAPS 20 LAUNCH(S=+1)

CA5 K1 F20 H15.1A V9.1 MINUS ORB TC5 AT38.1 (BGM074) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH 5500,0000 327,8000 2348,0000 1339,9100 .0000 190,7500 .0300 SO FT. IN IN. IN. XC IN. YC IN. ZC 0□◊ .000 STAB 1.000 BETA .500 .600 RUD-U .000 RUD-L .000 .700 ELV-1B .000 ELV-08 3.000 XMRP YMRP ZMRP SCALE S1-2 .000 S3-4 .000 S5-6 .000 IORB 9.700 2.2 DCL 2.0 COEFFICIENT(STABILITY AXIS). 1.8 1.6[1.4- 1.0 .8 .6 FORCE 0 = INCREMENTAL 1 2 3 4 5 6 7 8 9 10 CARRIER FRL ANGLE OF ATTACK, ALPHAW, DEG. 12 13 FIG. 27 747+0RB(10 DEG), BASIC TAIL, FLAPS 20 LAUNCH(S=+1) 747

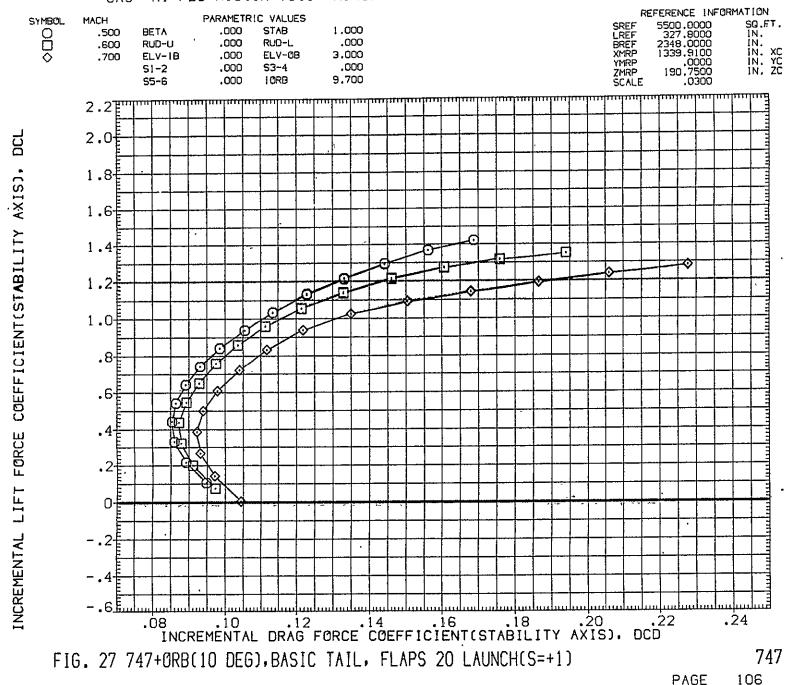


FIG. 27 747+0RB(10 DEG), BASIC TAIL, FLAPS 20 LAUNCH(S=+1)

DCL

AXIS3,

FORCE COEFFICIENT(STABILITY

LIFT

INCREMENTAL

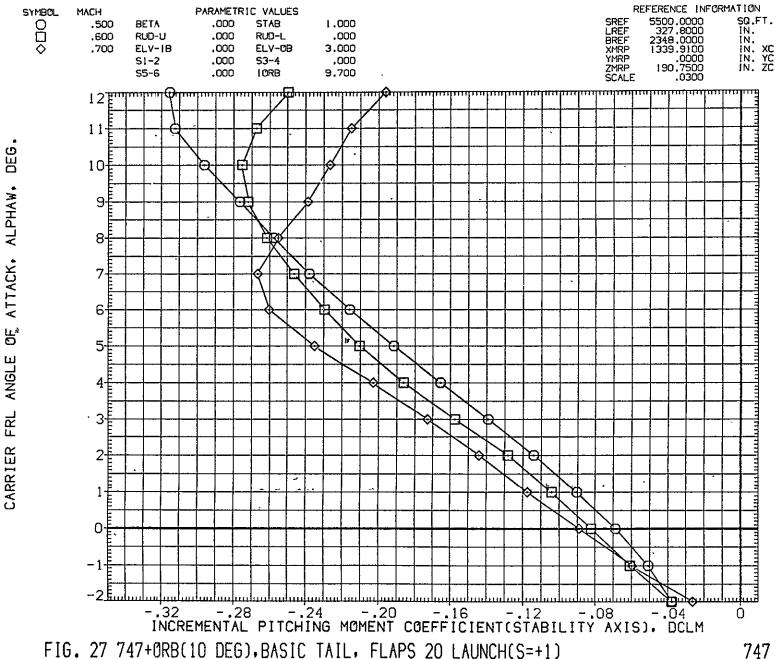
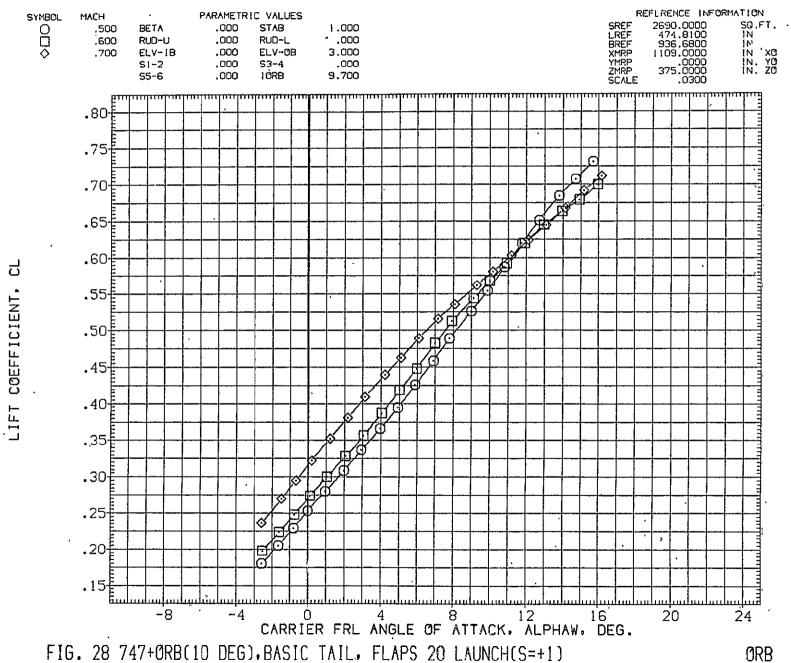


FIG. 27 747+0RB(10 DEG), BASIC TAIL, FLAPS 20 LAUNCH(S=+1)

CA5 ORB TC5 AT38.1 (PRES. K1 F20 H15.1A V9.1)(YGM074)



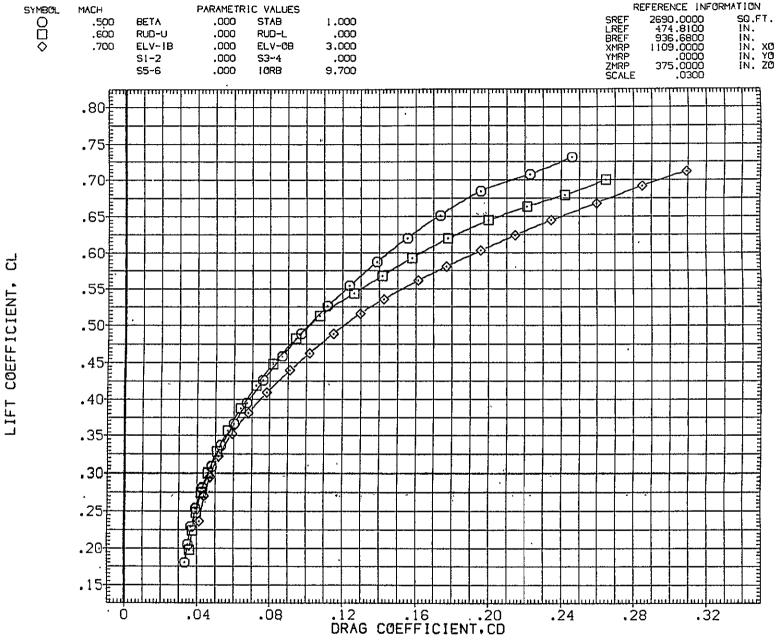
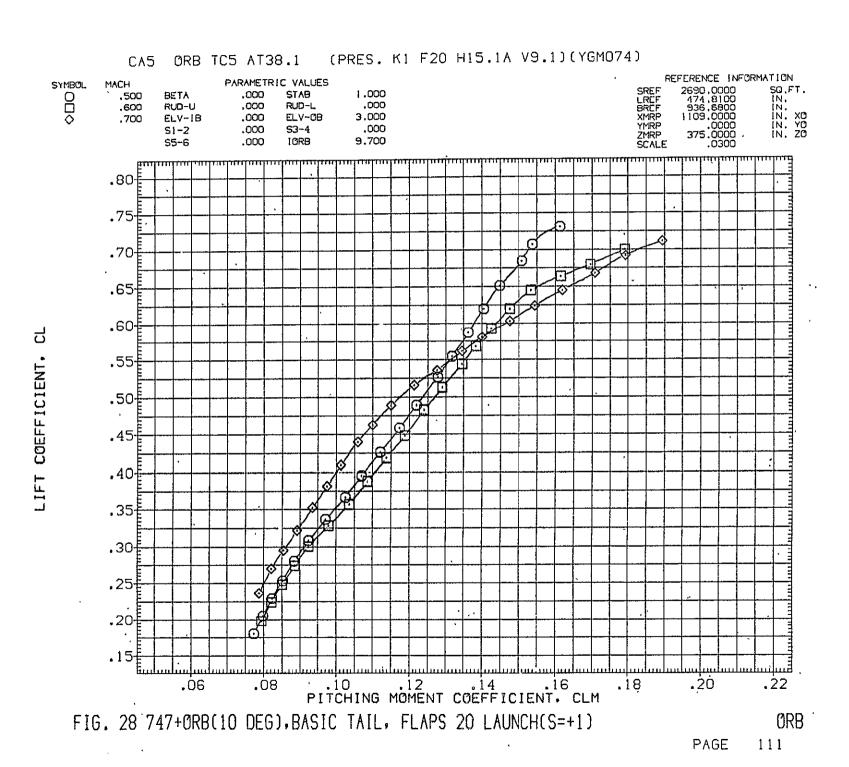
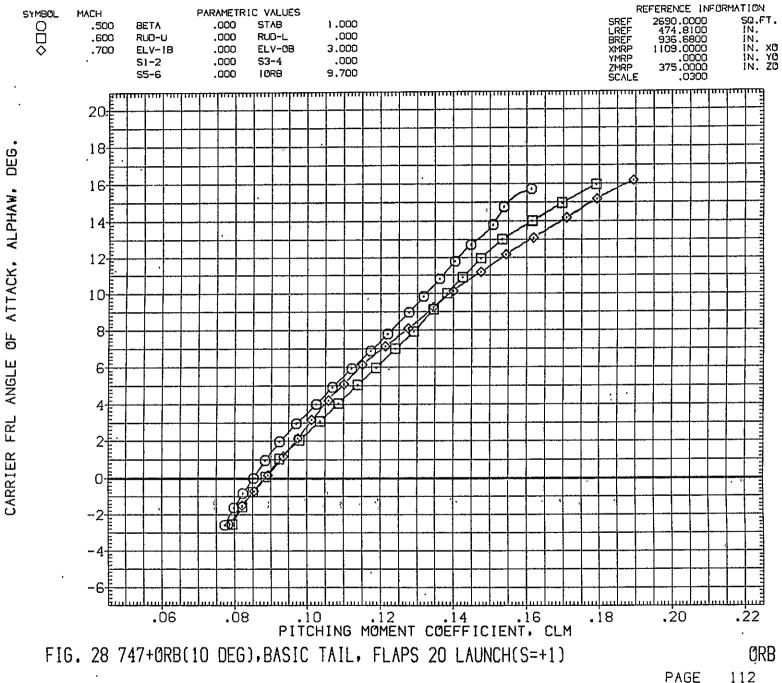


FIG. 28 747+0RB(10 DEG), BASIC TAIL, FLAPS 20 LAUNCH(S=+1)

ORB



CA5 ORB TC5 AT38.1 (PRES. K1 F20 H15.1A V9.1)(YGM074)



CA5 K1 F20 H15.1A V9.1 (PLUS. ORB TC5 AT38.1)(RGM075)

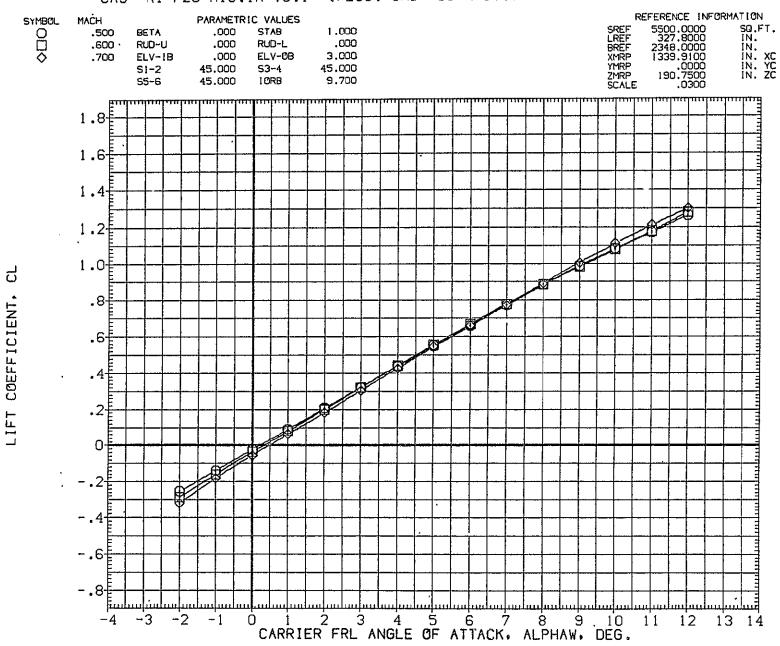


FIG. 29 747+0RB(10 DEG)+SP45,BASIC TAIL, FLAPS 20 LAUNCH(S=+1)

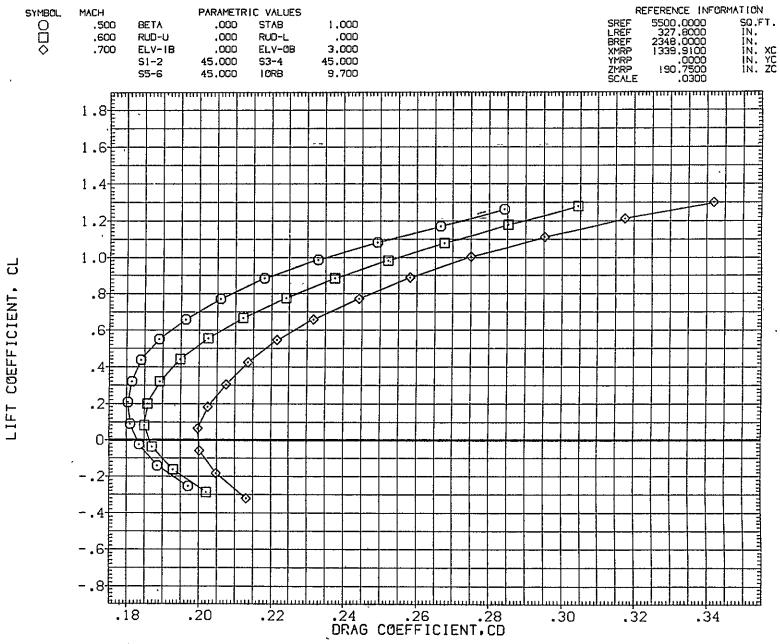


FIG. 29 747+ORB(10 DEG)+SP45, BASIC TAIL, FLAPS 20 LAUNCH(S=+1)

CA5 K1 F20 H15.1A V9.1 (PLUS. ORB TC5 AT38.1)(RGM075) REFERENCE INFORMATION PARAMETRIC VALUES 5500.0000 327.8000 2348.0000 1339.9100 .0000 190.7500 .0300 SYMBOL MACH SREF LREF BREF XMRP YMRP ZMRP SCALE SO.FT. IN. IN. XC IN. YC IN. ZC 1.000 000 .000 STAB .500 BETA .000 .000 RUD-L .600 RUD-U 3.000 ELV-1B .000 ELV-08 45.000 45.000 S3-4 S1-2 9,700 TORB S5-6 45.000 1.8 1.6-1.4 믺 Q 1.2 1.0 김 COEFFICIENT. .8 .6[LIFT 0 60 -.6[.3 -.2 -.1 0 .1 PITCHING MOMENT COEFFICIENT, CLM . 4 - .3 TOTAL FIG. 29 747+ORB(10 DEG)+SP45, BASIC TAIL, FLAPS 20 LAUNCH(S=+1)

CA5 K1 F20 H15.1A V9.1 (PLUS. ORB TC5 AT38.1)(RGMU75)

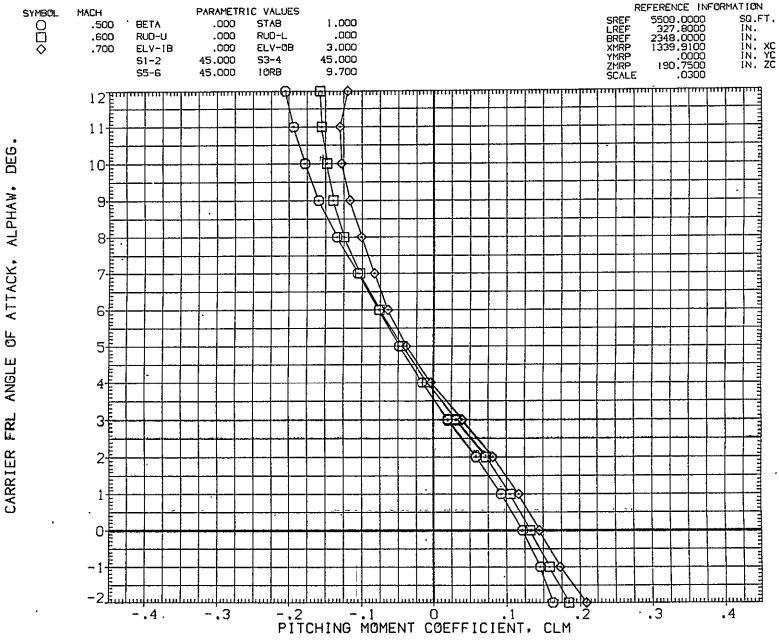


FIG. 29 747+ORB(10 DEG)+SP45, BASIC TAIL, FLAPS 20 LAUNCH(S=+1)

TOTAL

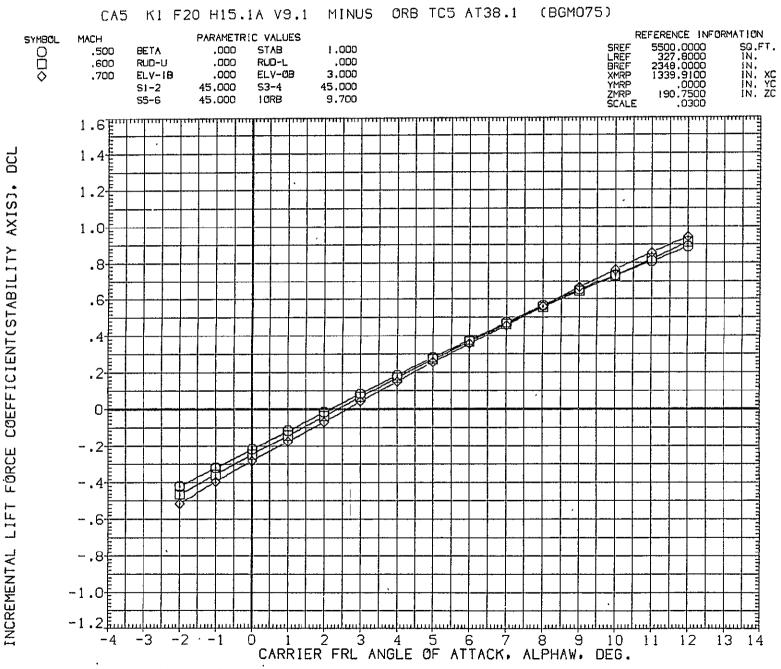
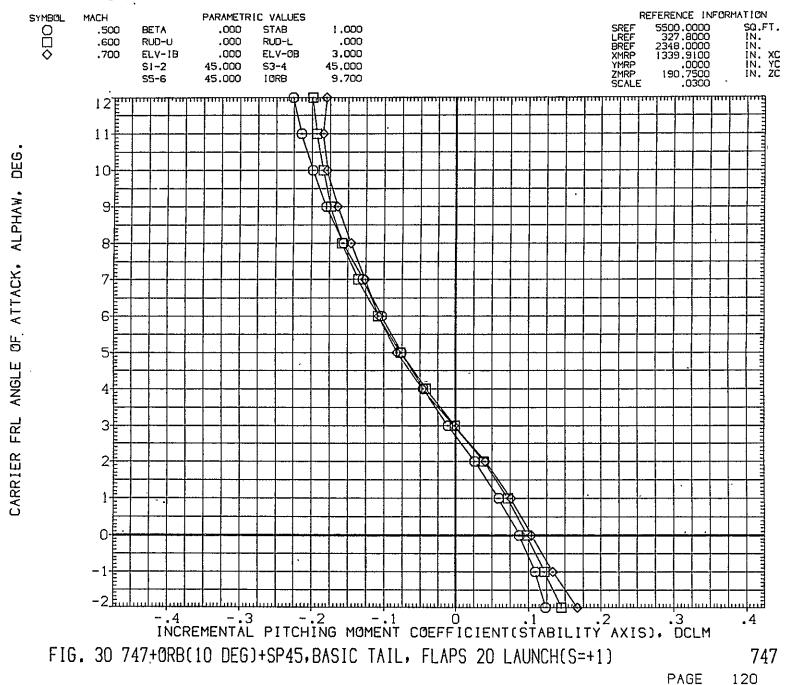
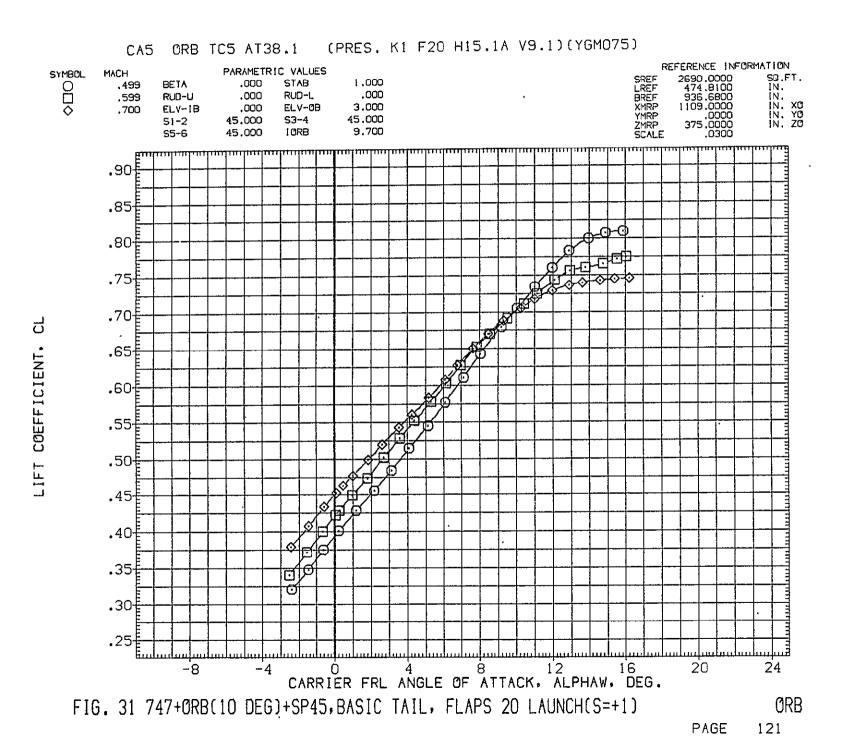


FIG. 30 747+0RB(10 DEG)+SP45, BASIC TAIL, FLAPS 20 LAUNCH(S=+1)

FIG. 30 747+0RB(10 DEG)+SP45, BASIC TAIL, FLAPS 20 LAUNCH(S=+1)





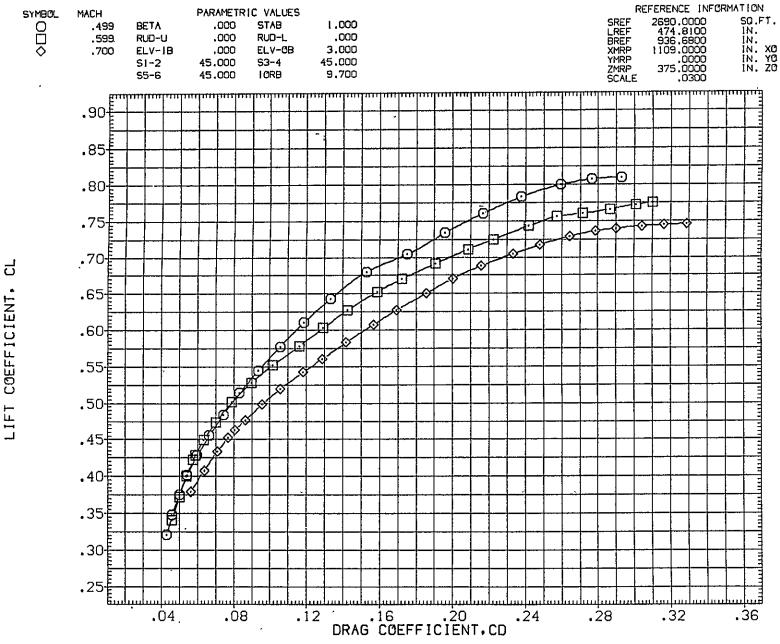
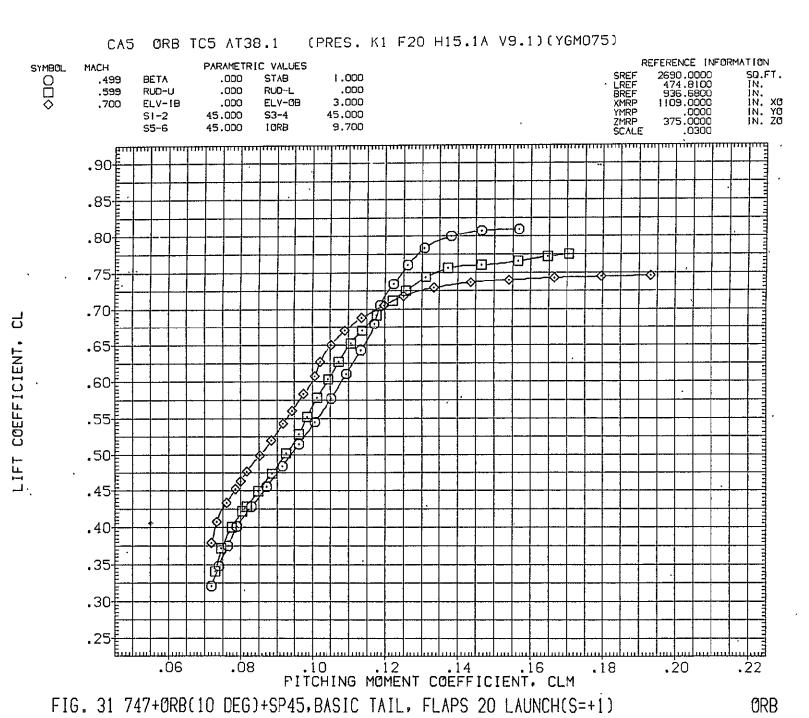


FIG. 31 747+0RB(10 DEG)+SP45, BASIC TAIL, FLAPS 20 LAUNCH(S=+1)

ORB



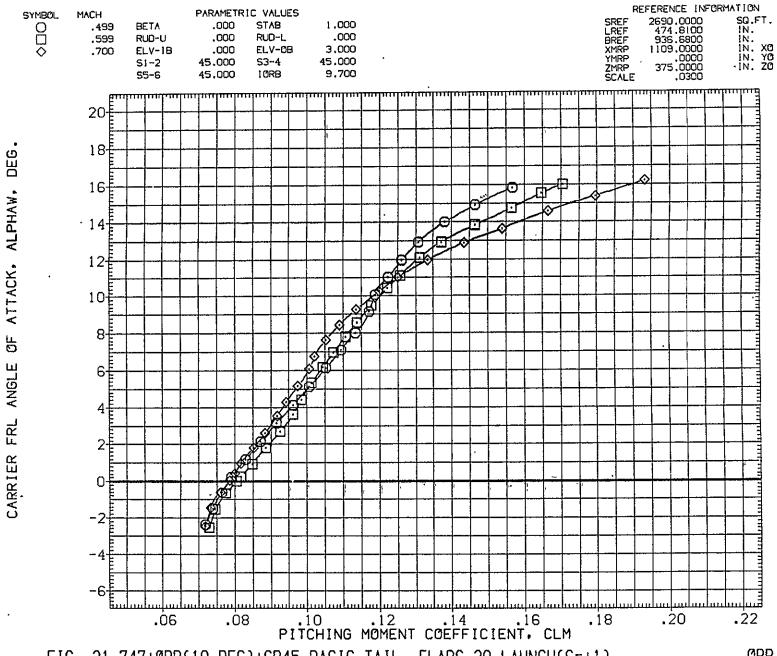


FIG. 31 747+0RB(10 DEG)+SP45, BASIC TAIL, FLAPS 20 LAUNCH(S=+1)

ORB

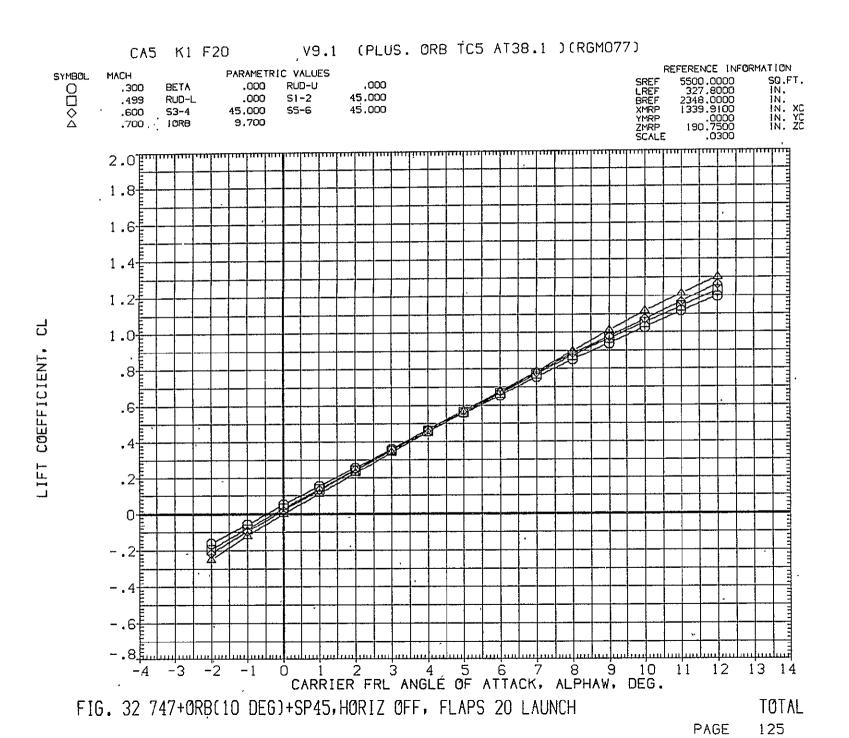


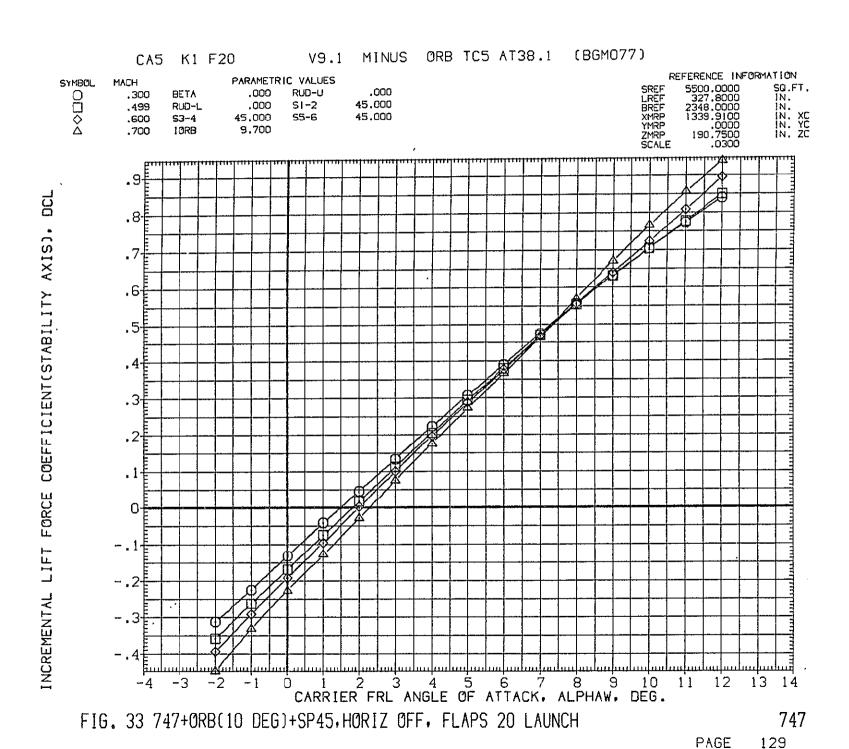
FIG. 32 747+0RB(10 DEG)+SP45, HORIZ OFF, FLAPS 20 LAUNCH

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PITCHING MOMENT COEFFICIENT, CLM

FIG. 32 747+0RB(10 DEG)+SP45, HORIZ OFF, FLAPS 20 LAUNCH

TOTAL



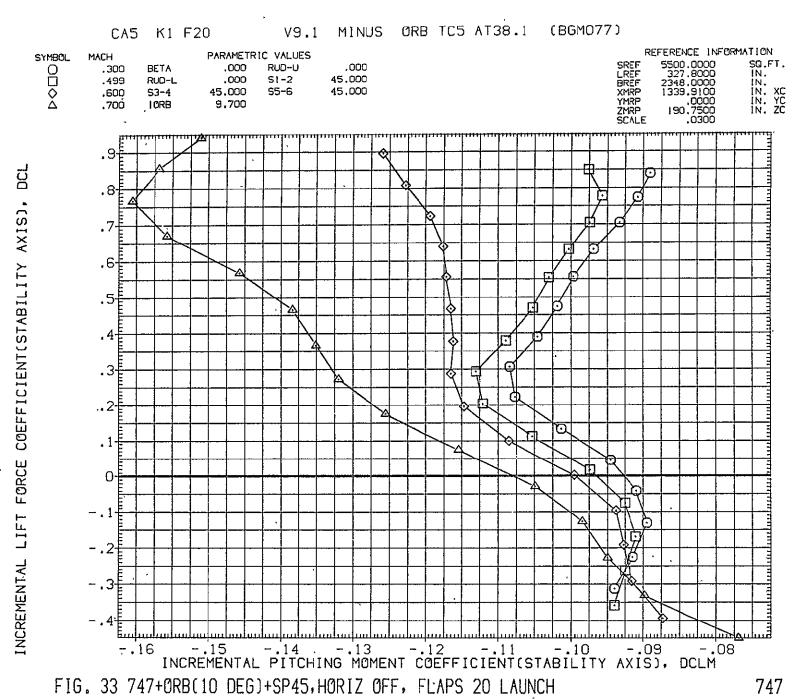


FIG. 33 747+0RB(10 DEG)+SP45, HORIZ OFF, FLAPS 20 LAUNCH

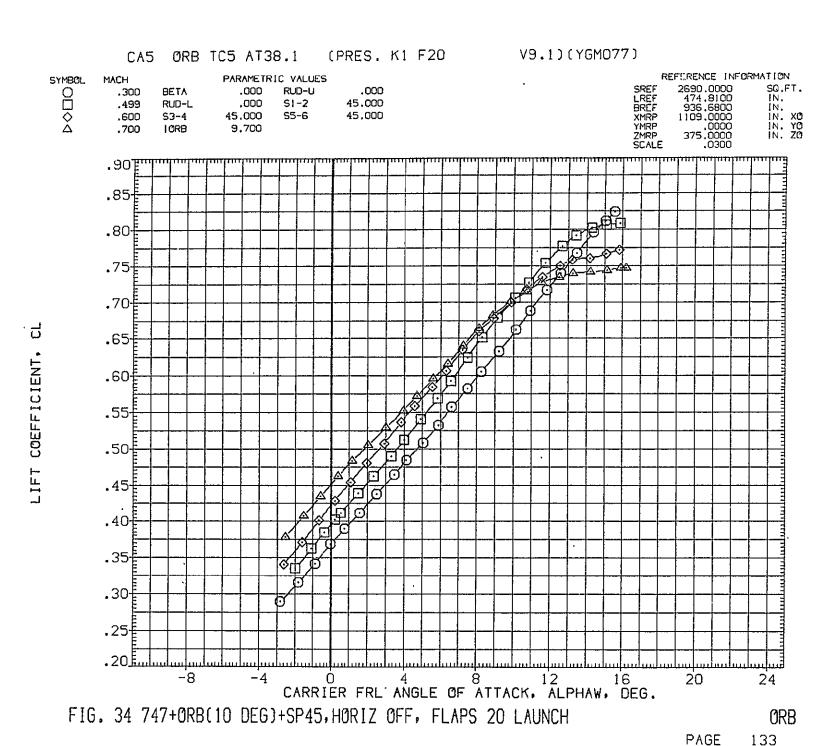


FIG. 34 747+0RB(10 DEG)+SP45, HORIZ OFF, FLAPS 20 LAUNCH

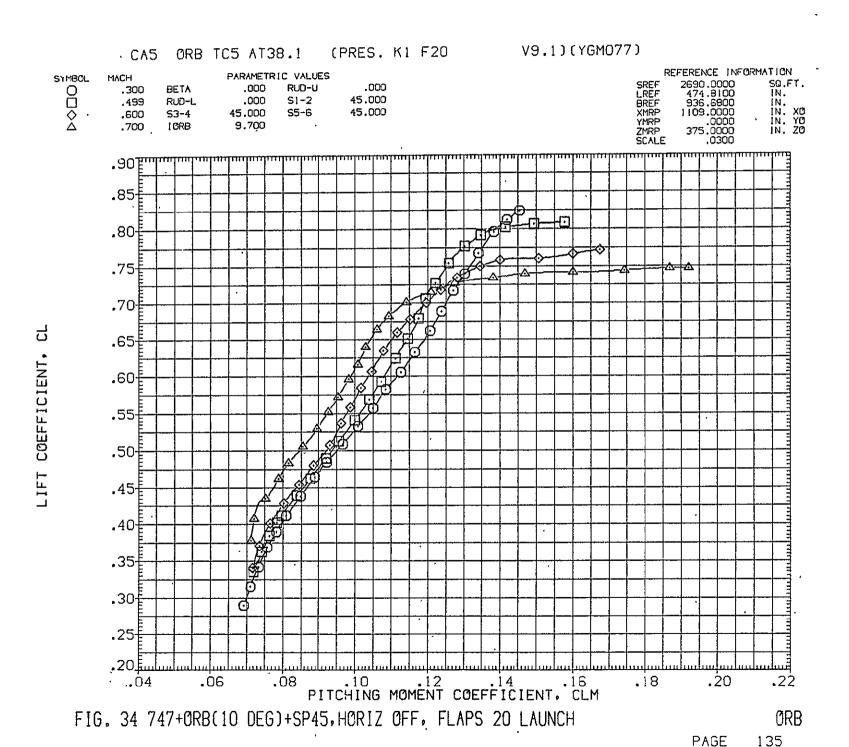


FIG. 34 747+0RB(10 DEG)+SP45, HORIZ OFF, FLAPS 20 LAUNCH

CA5 K1 F20 H15.6 V9.1 (PLUS. ORB TC5 AT38.2)(RGMO80) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH 5500.0000 327.8000 2348.0000 1339.9100 .0000 190.7500 SREF LREF BREF XMRP YMRP ZMRP SCALE SQ.FT. IN. IN. XC IN. YC IN. ZC 000 BETA ,000 STAB 1.000 .500 RUD-L .000 .600 RUD-U .000 ELV-0B 3.000 .700 ELV-1B .000 45.000 7.050 \$3-4 45.000 S1-2 S5-6 45,000 LORB 1.8 1.6[1.4 1.2 1.0 ŭ LIFT COEFFICIENT, .8 .6 1 2 3 4 5 6 7 8 9 10 CARRIER FRL ANGLE OF ATTACK, ALPHAW, DEG. -2

FIG. 35 747+0RB(8 DEG)+TIP FINS+SP45, FLAPS UP LAUNCH(S=+1)

CA5 K1 F20 H15.6 V9.1 (PLUS. ORB TC5 AT38.2)(RGMO80)

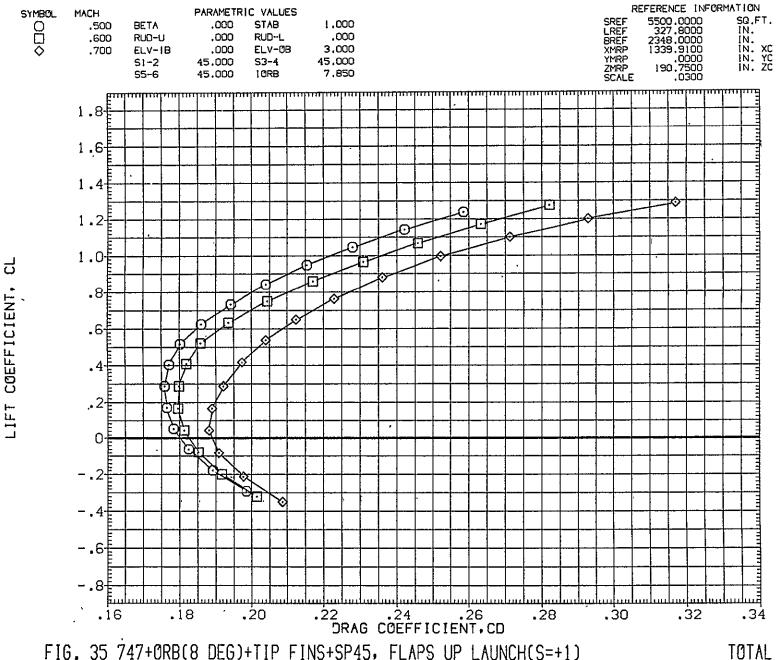


FIG. 35 747+ORB(8 DEG)+TIP FINS+SP45, FLAPS UP LAUNCH(S=+1)

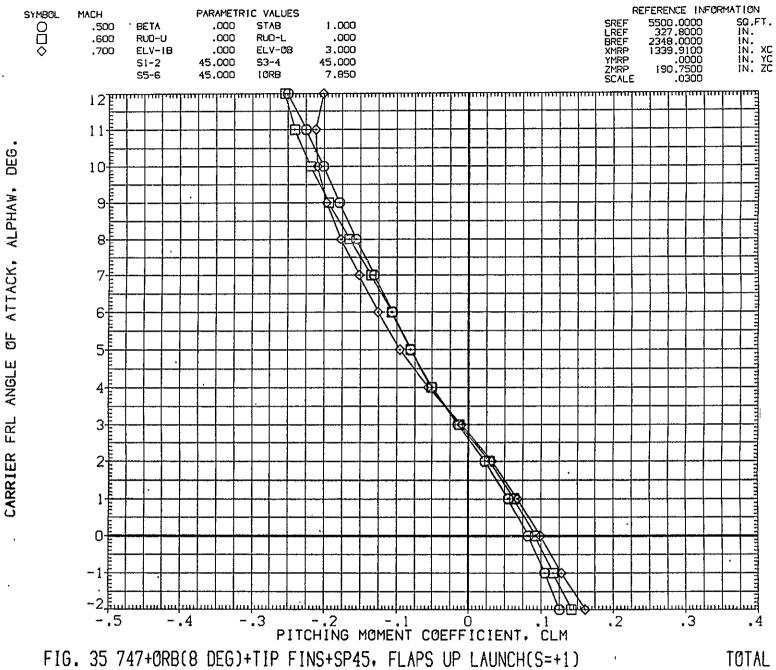
CA5 K1 F20 H15.6 V9.1 (PLUS. ORB TC5 AT38.2)(RGMO80) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL. MACH SQ.FT. IN. IN. XC IN. YC IN. ZC 000 1,000 .000 STAB .500 BETA 327,8000 2348,0000 1339,9100 ,0000 190,7500 .000 RUD-L .000 .600 RUD-U .700 .000 ELV-08 3,000 ELV-IB 45,000 \$1-2 45,000 S3-4 **IORB** 7,850 S5-6 45.000 .8-1.6 1.4 1.2 1.0 .8 .6 -.2 -.1 0 .1 PITCHING MOMENT COEFFICIENT, CLM .3

PAGE

FIG. 35 747+ORB(8 DEG)+TIP FINS+SP45, FLAPS UP LAUNCH(S=+1)

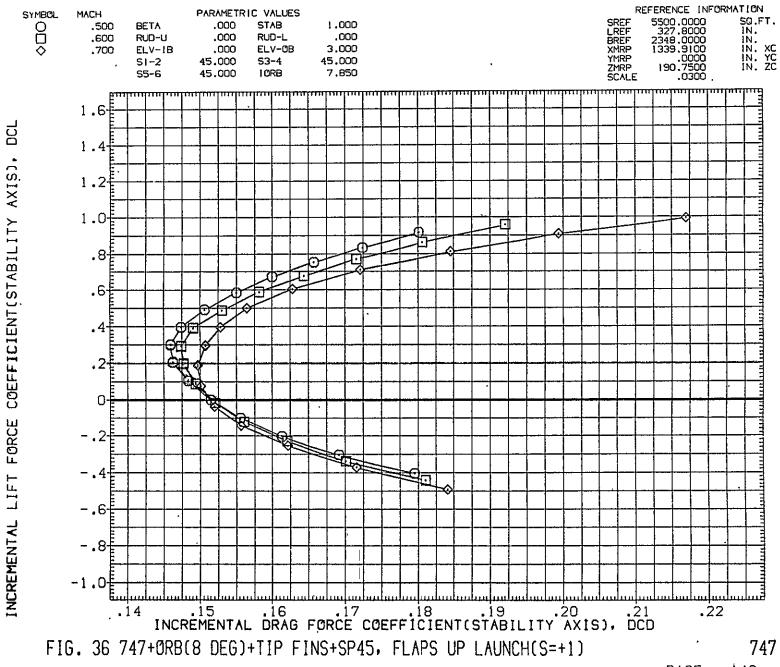
 Image: Control of the control of the

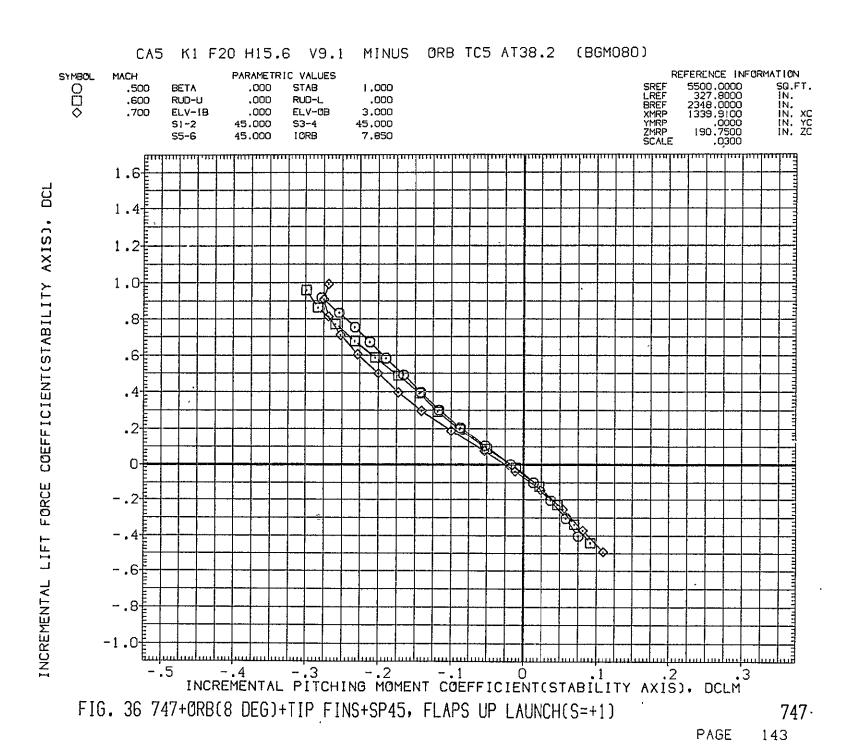
LIFT COEFFICIENT,



CA5 K1 F20 H15.6 V9.1 MINUS ORB TC5 AT38.2 (BGM080) REFERENCE INFORMATION SYMBOL MACH PARAMETRIC VALUES SQ.FT. IN. IN. IN. YC IN. YC 5500.0000 327.8000 000 SREF LREF .000 STAB 1.000 :500 BETA .000 .600 RUD-U .000 RUD-L 2348.0000 1339.9100 .0000 190.7500 .0300 .000 ELV-0B 3.000 .700 ELV-!B YMRP ZMRP SCALE 45.000 53-4 45.000 S1-2 S5-6 45.000 IORB 7.850 1.6 DCL 1.4 INCREMENTAL LIFT FORCE COEFFICIENT(STABILITY AXIS), 1.2 1.0 .8 .6[0 1 2 3 4 5 6 7 8 9 10 CARRIER FRL ANGLE OF ATTACK, ALPHAW, DEG. 5 FIG. 36 747+0RB(8 DEG)+TIP FINS+SP45, FLAPS UP LAUNCH(S=+1) 747

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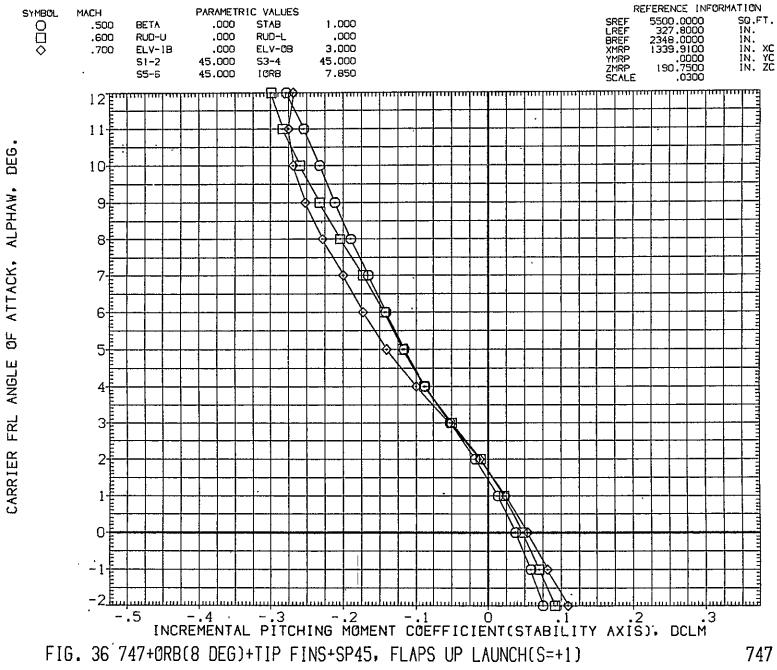


FIG. 37 747+0RB(8 DEG)+TIP FINS+SP45, FLAPS UP LAUNCH(S=+1)

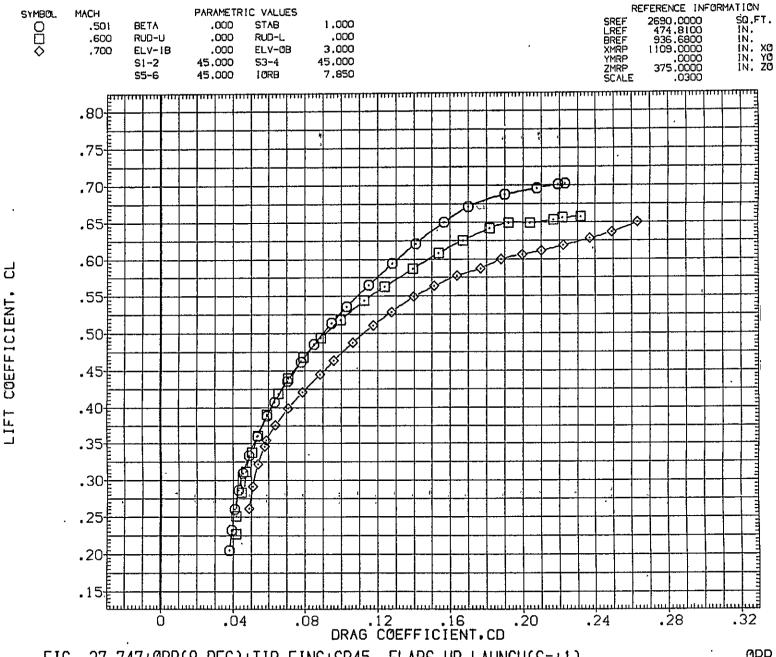
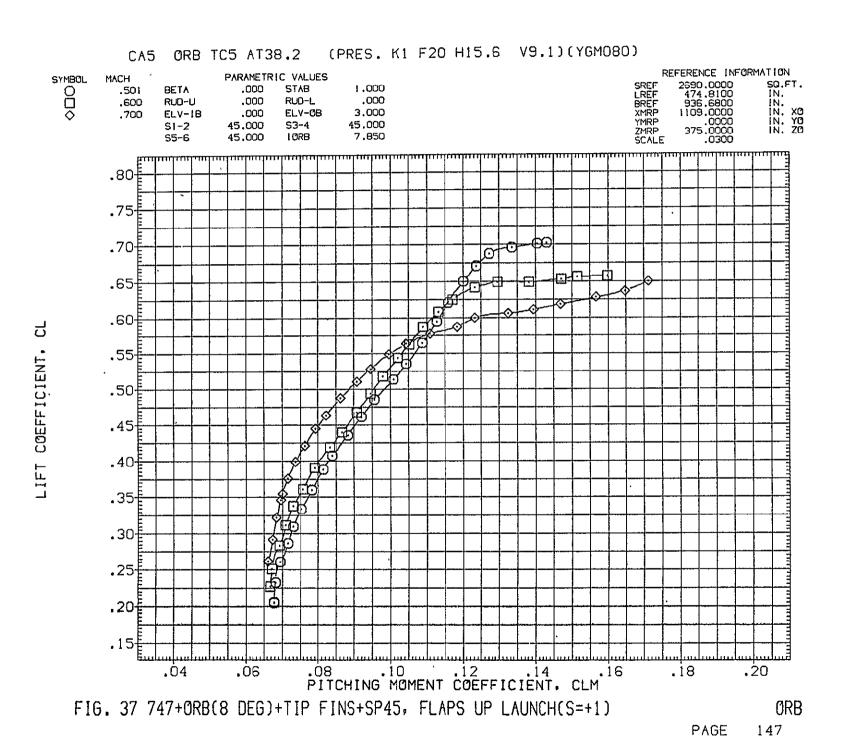


FIG. 37 747+ORB(8 DEG)+TIP FINS+SP45, FLAPS UP LAUNCH(S=+1)

ORB



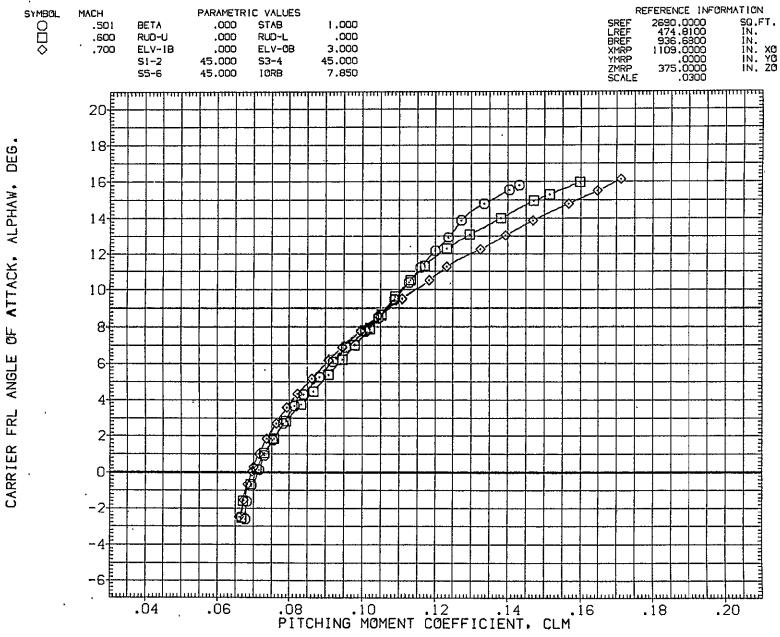


FIG. 37 747+0RB(8 DEG)+TIP FINS+SP45, FLAPS UP LAUNCH(S=+1)

ORB

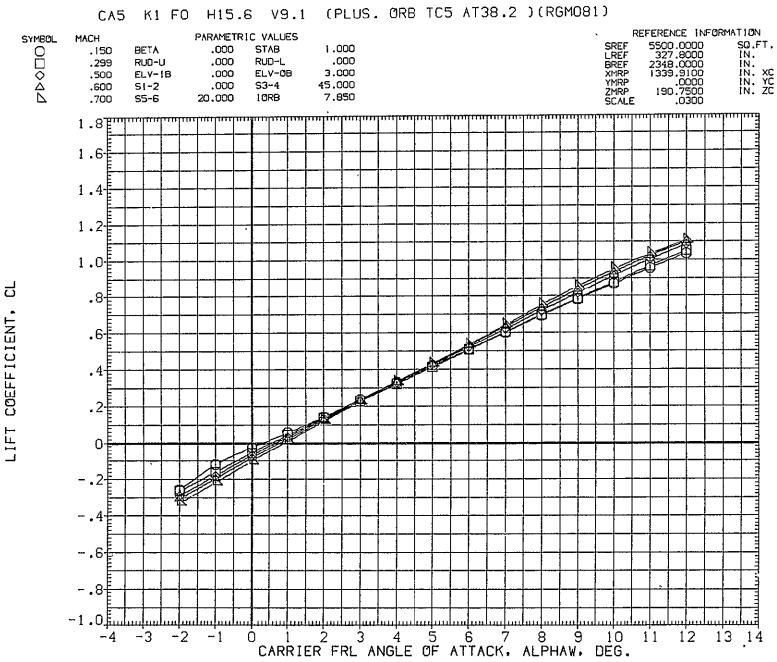


FIG. 38 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, BASELINE(S=1) TOTAL
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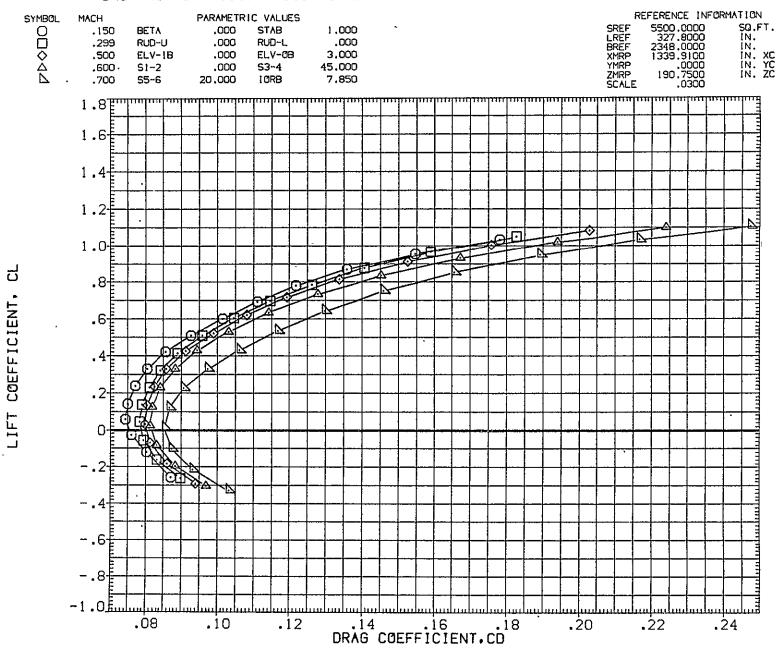


FIG. 38 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, BASELINE(S=1) TOTAL PAGE 150

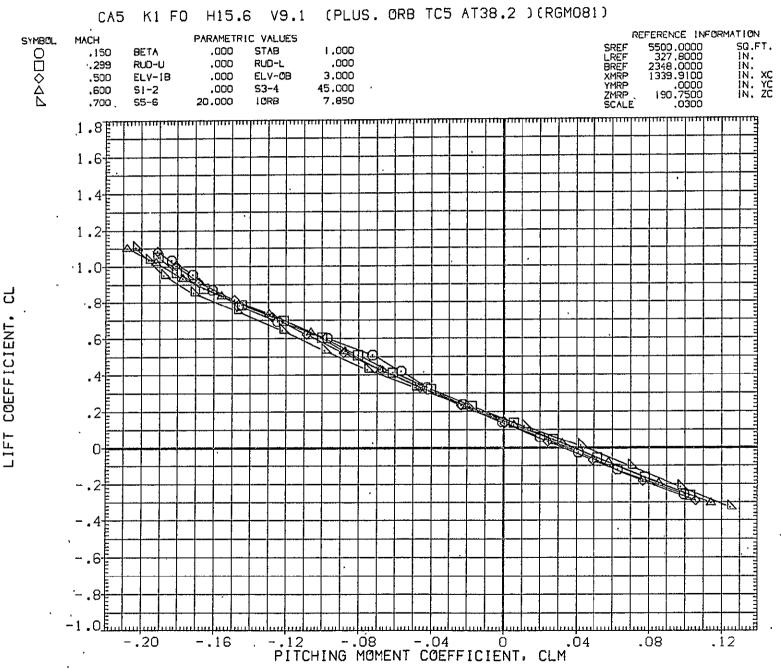


FIG. 38 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, BASELINE(S=1) TOTAL

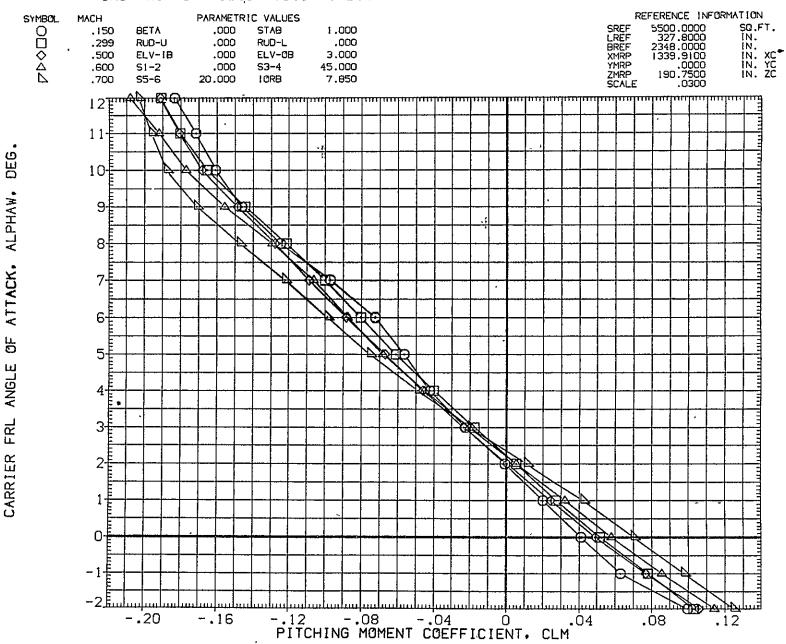


FIG. 38 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP. LAUNCH, BASELINE(S=1) TOTAL
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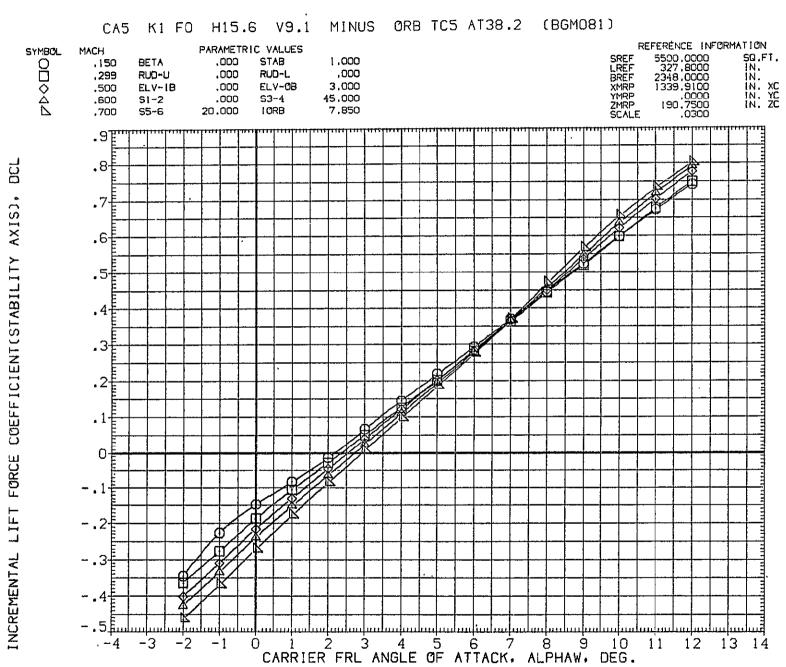


FIG. 39 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, BASELINE(S=1) 747
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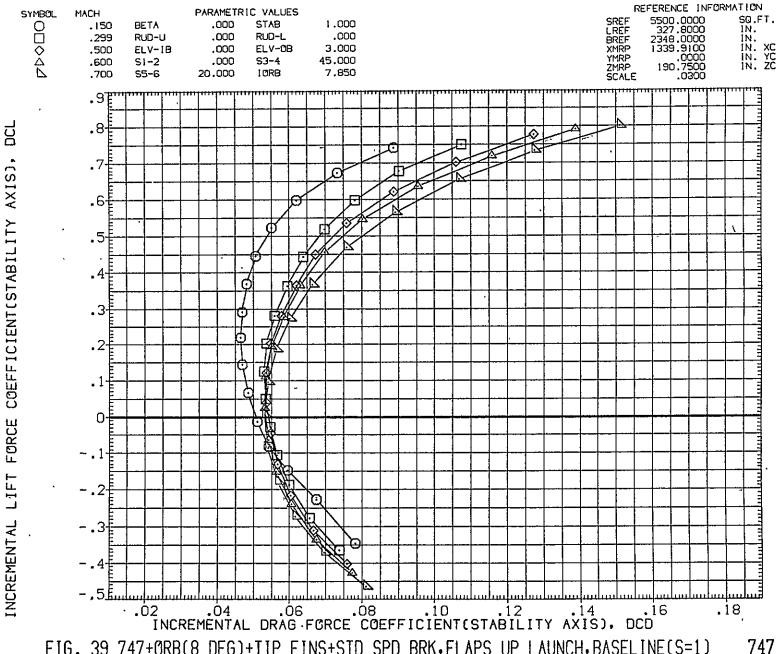
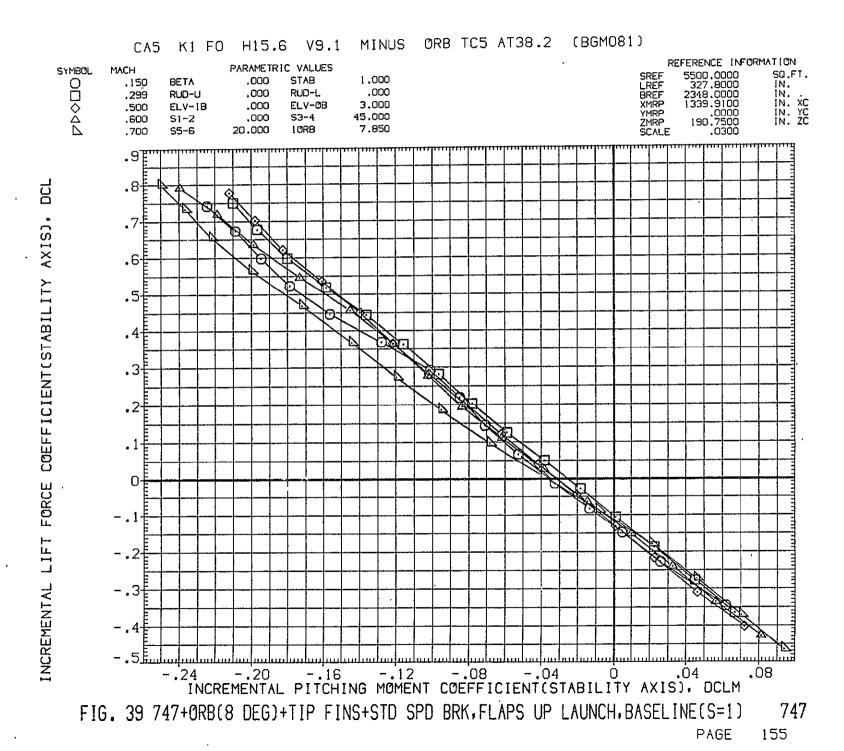


FIG. 39 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, BASELINE(S=1) 74



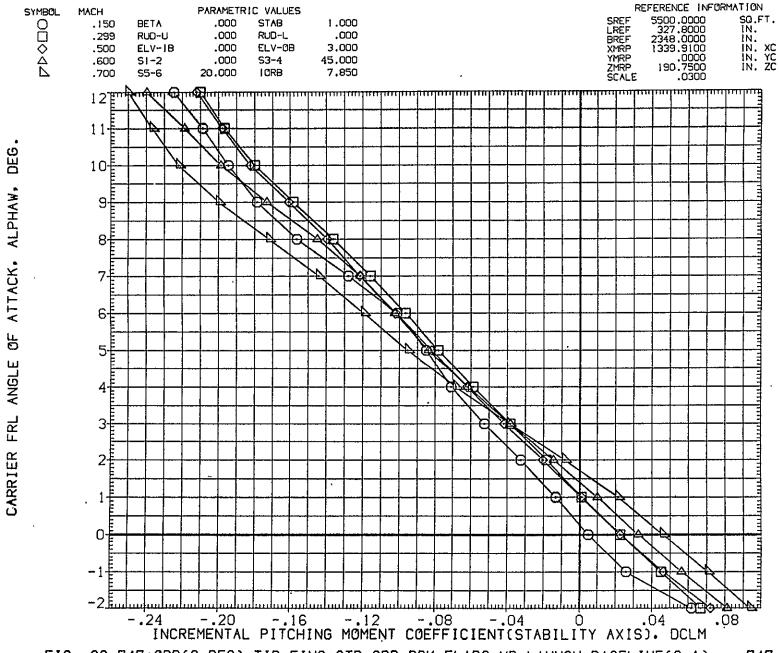
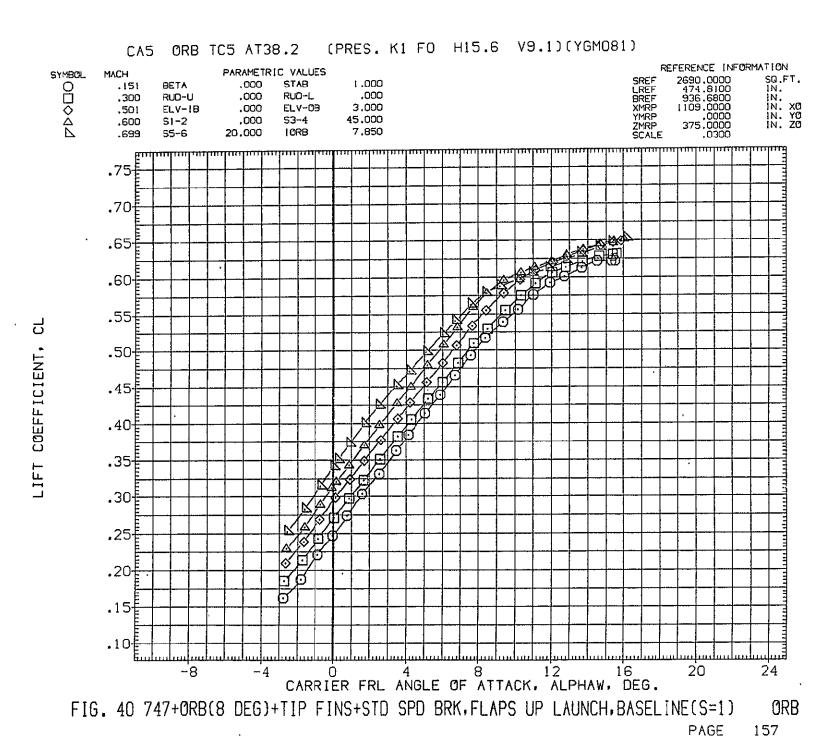


FIG. 39 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, BASELINE(S=1) 747
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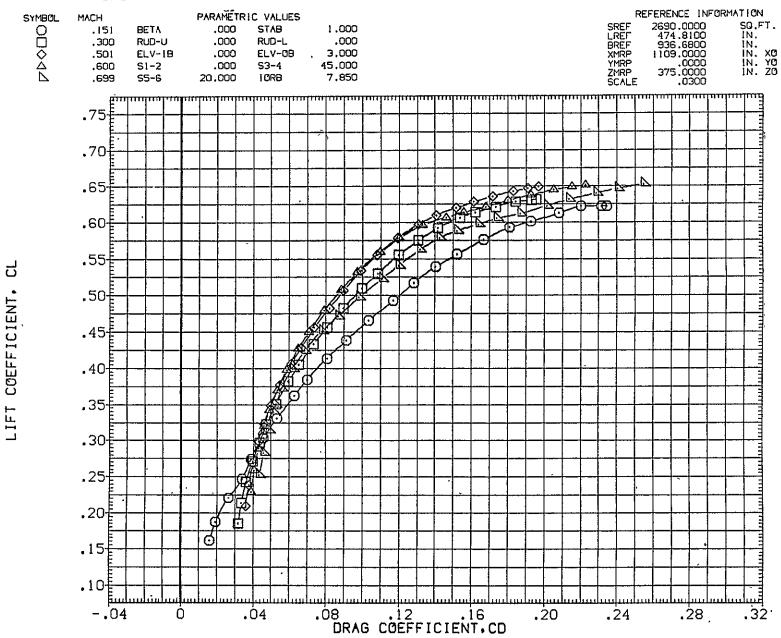


FIG. 40 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, BASELINE(S=1) ORB

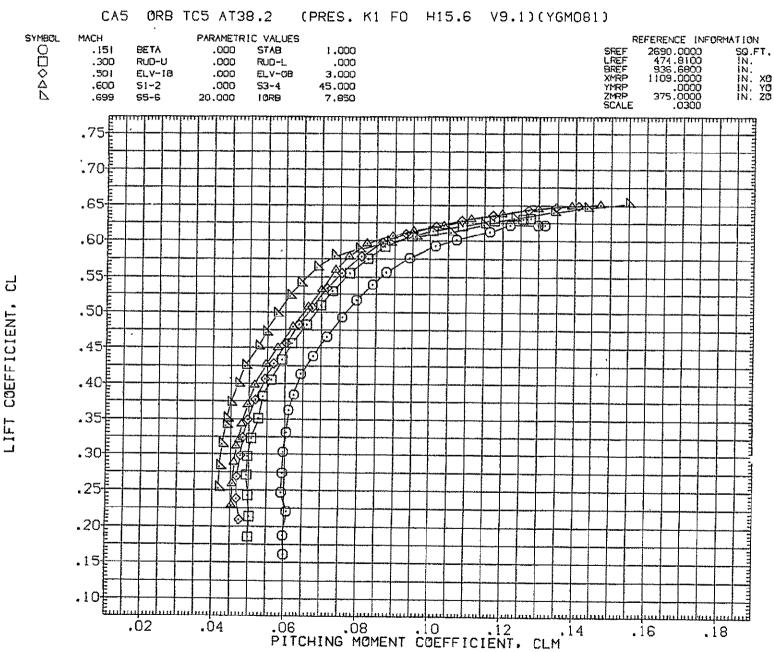


FIG. 40 747+0RB(8 DEG)+TIP FINS+STD SPD BRK,FLAPS UP LAUNCH,BASELINE(S=1) ORB

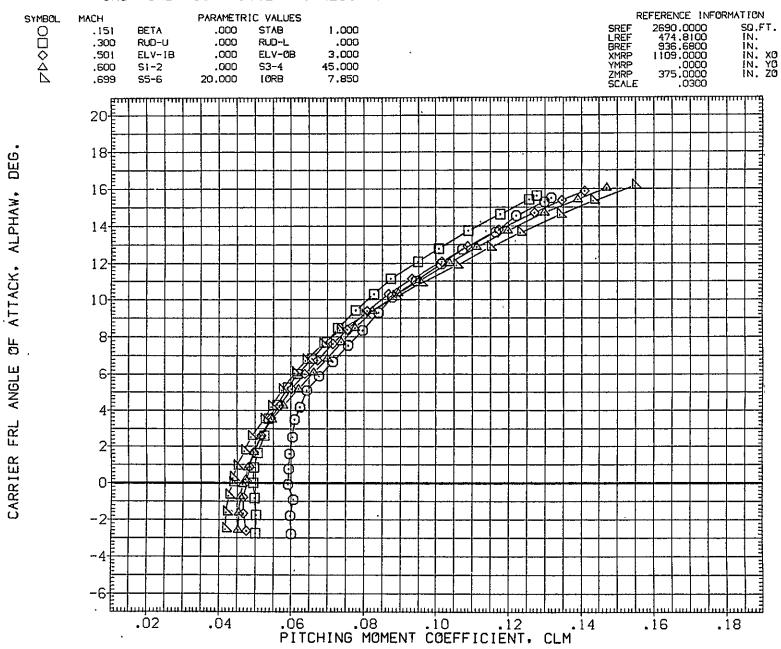


FIG. 40 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, BASELINE(S=1) ORB

FIG. 41 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, (S=-2)

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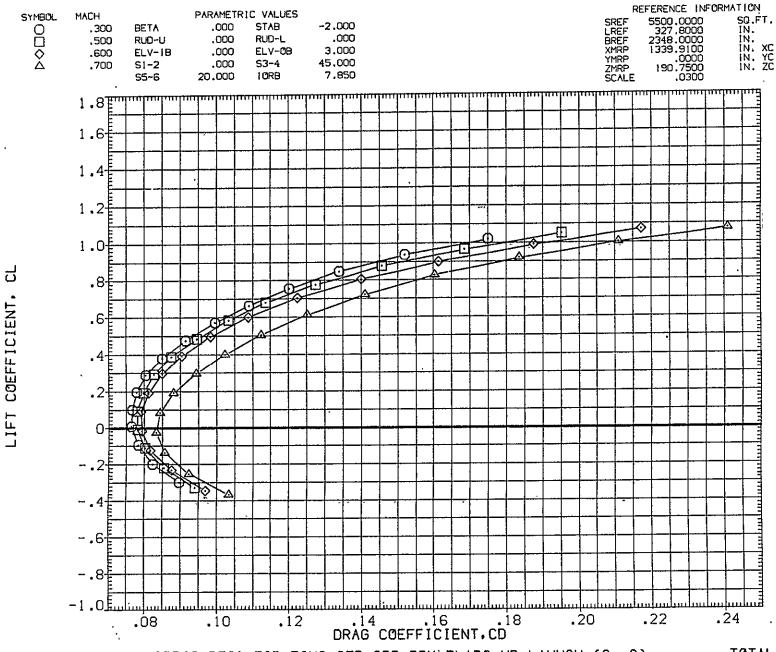


FIG. 41 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, (S=-2)

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CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.2)(RGMO84)

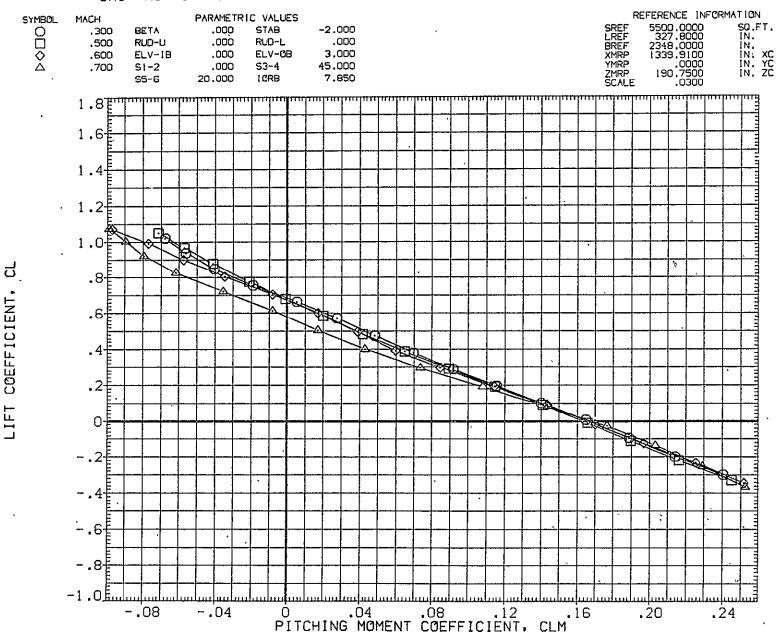


FIG. 41 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, (S=-2)

TOTAL

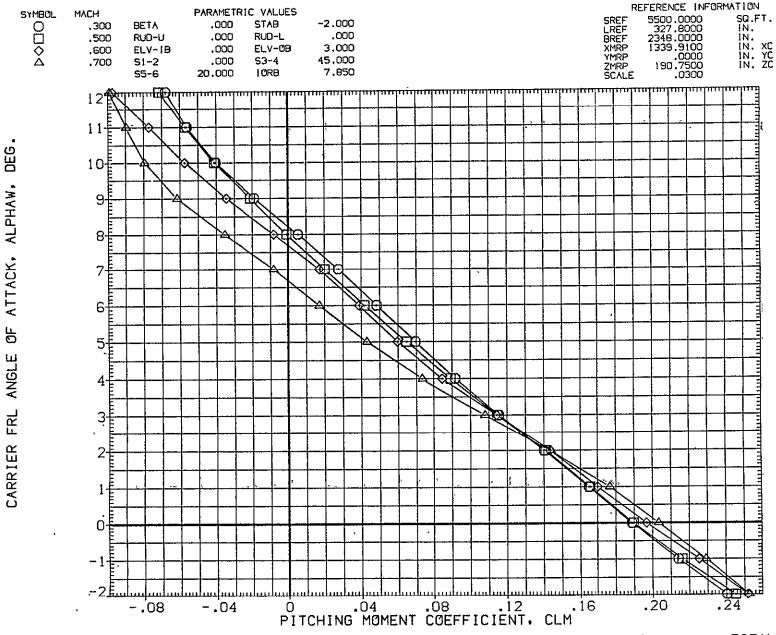


FIG. 41 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, (S=-2)

TOTAL

ORB TC5 AT38.2 (BGM084) . CA5 K1 FO H15.6 V9.1 MINUS REFERENCE INFORMATION SYMBOL O O O PARAMETRIC VALUES MACH SQ.F 5500.0000 327.8000 .000 STAB -2,000 BETA .300 ,000 IN. XC IN. YC IN. ZC .000 RUD-L .500 RUD-U 2348,0000 1339.9100 .0000 190.7500 .0300 ELV-08 3.000 .600 ELV-1B .000 YMRP ZMRP SCALE 53-4 45.000 .000 .700 S1-2 7.850 20.000 LORB S5-6 AXIS) .6 FORCE COEFFICIENT(STABILITY ·4E .3[.1[INCREMENTAL 1 2 3 4 5 6 7 8 9 10 CARRIER FRL ANGLE OF ATTACK. ALPHAW, DEG. 10 11 FIG. 42 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, (S=-2) 747

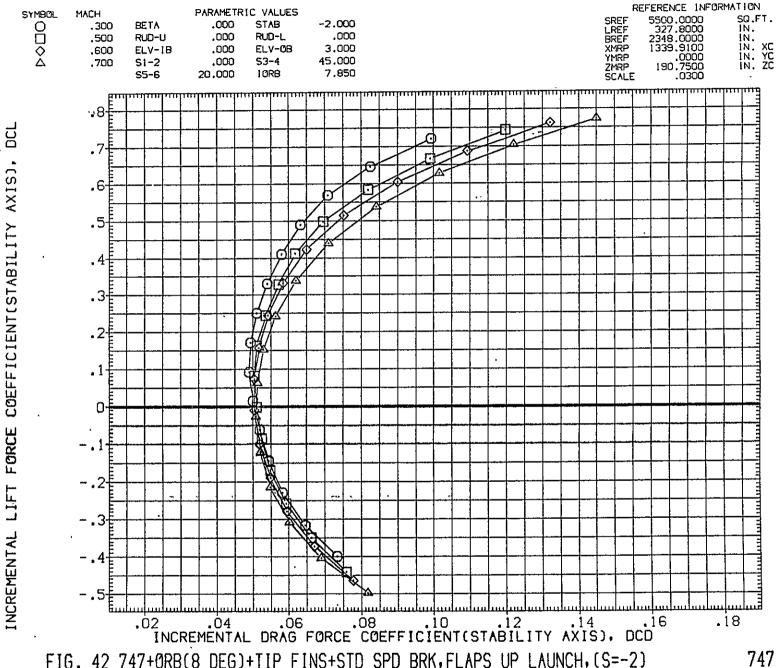
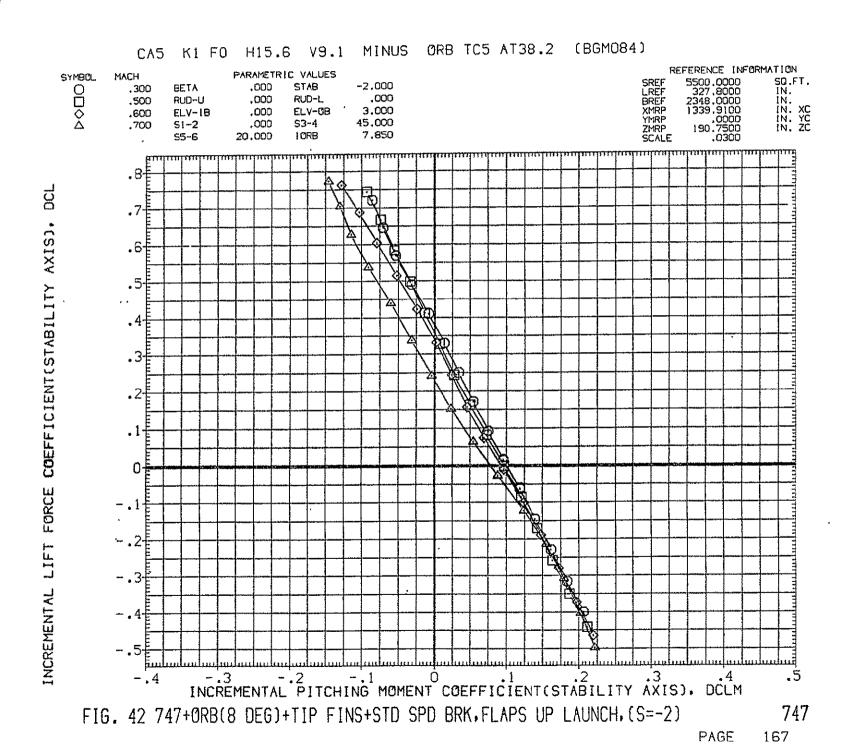
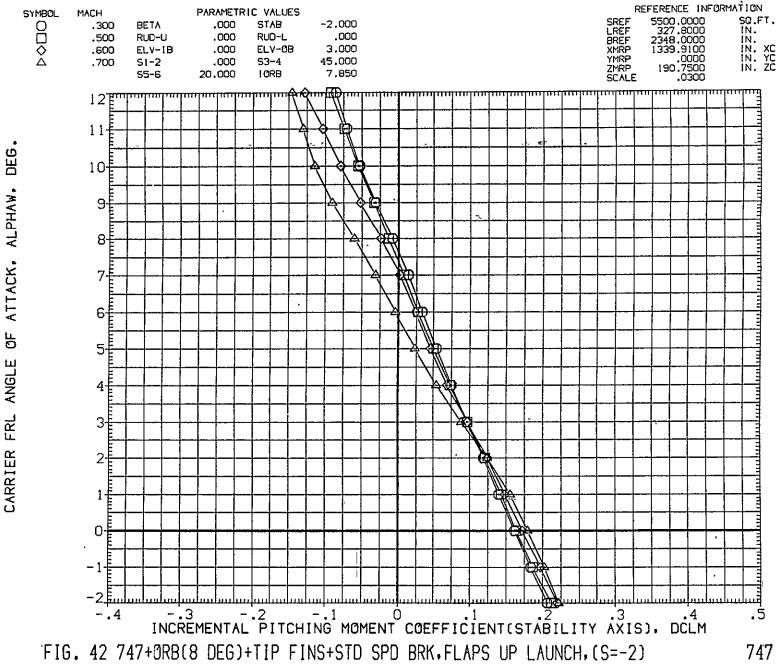


FIG. 42 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, (S=-2)





CA5 ORB TC5 AT38.2 (PRES. K1 FO H15.6 V9.1)(YGM084)

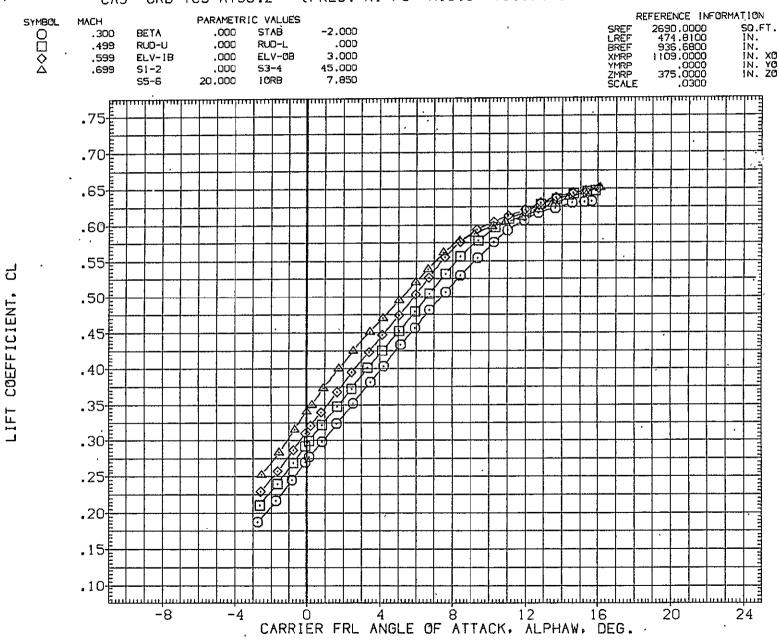
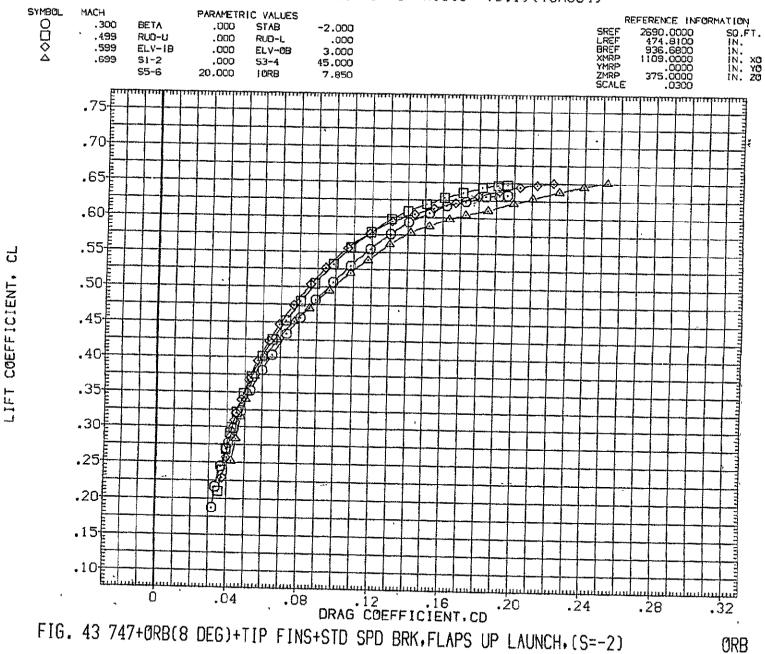
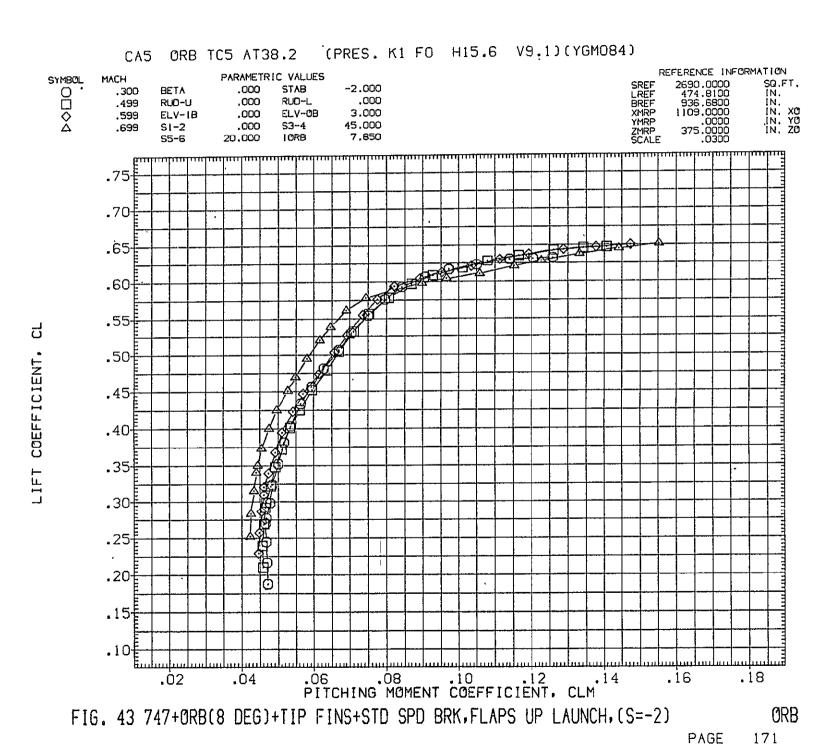


FIG. 43 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, (S=-2)

ORB

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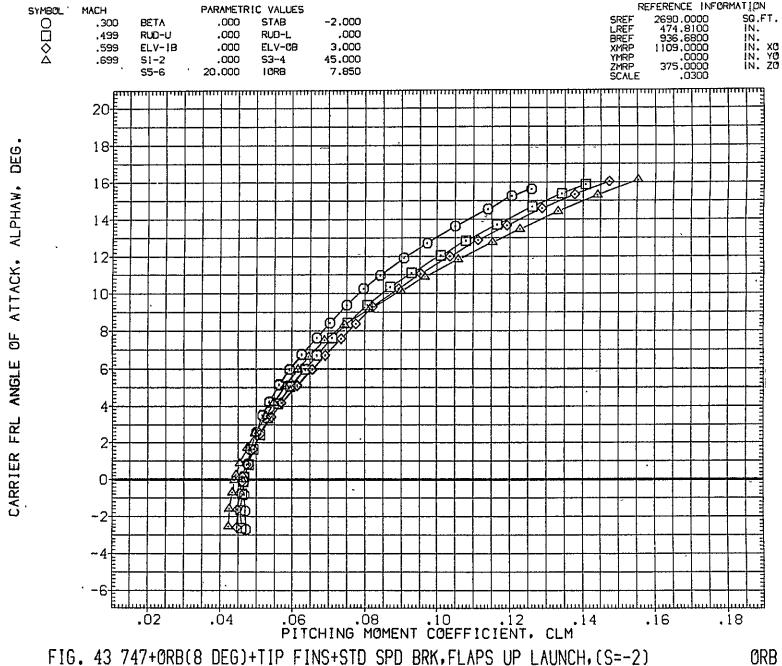
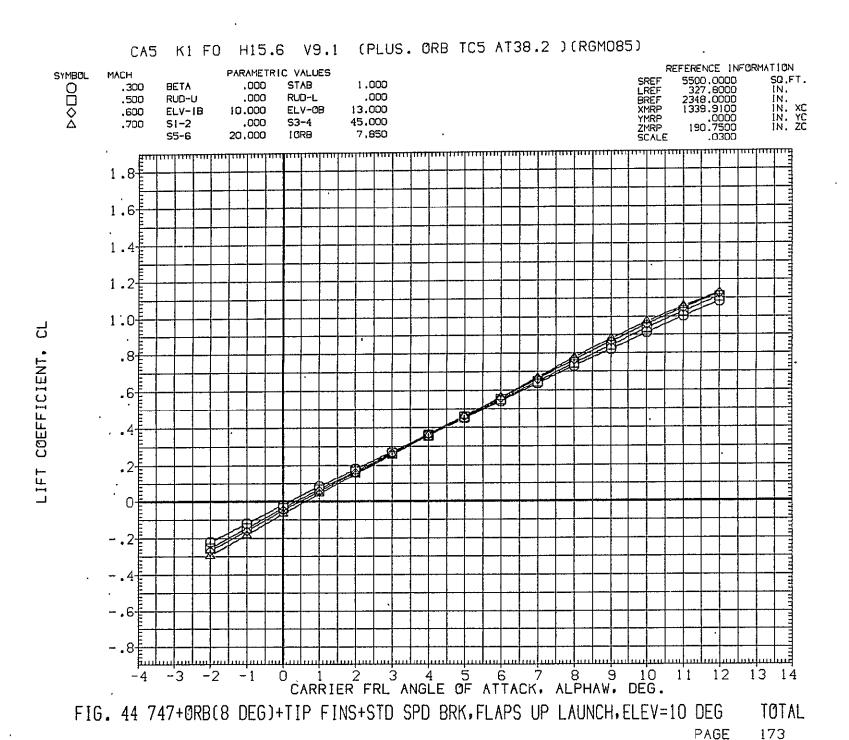


FIG. 43 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, (S=-2)



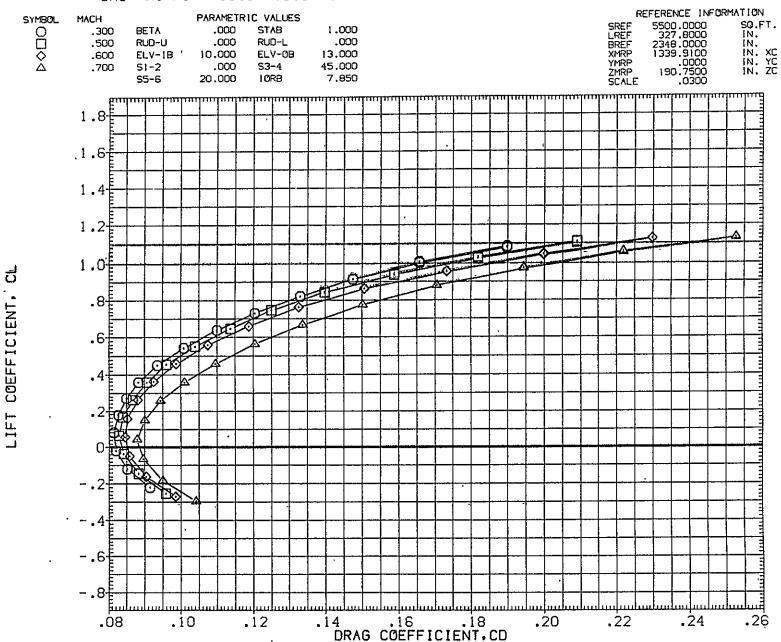


FIG. 44 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ELEV=10 DEG TOTAL

CA5 K1 F0 H15.6 V9.1 (PLUS. ORB TC5 AT38.2) (RGMO85)

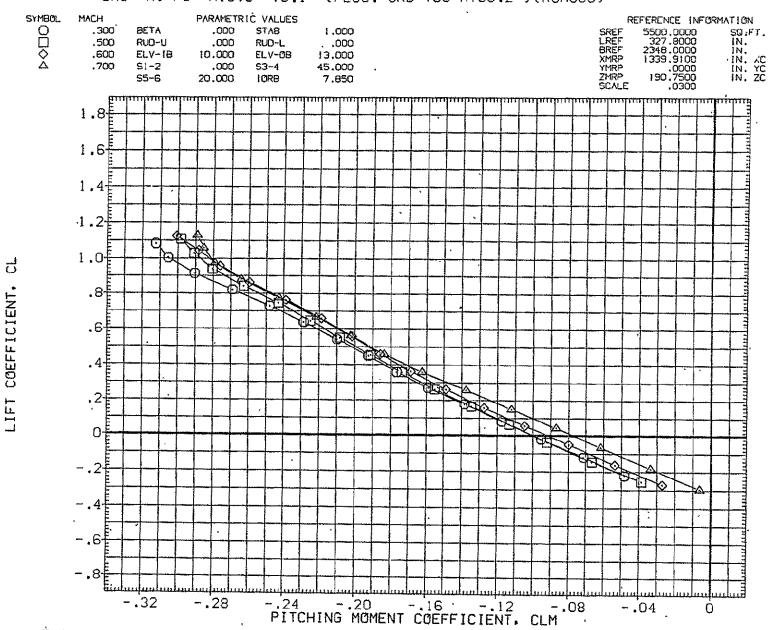


FIG. 44 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ELEV=10 DEG TOTAL

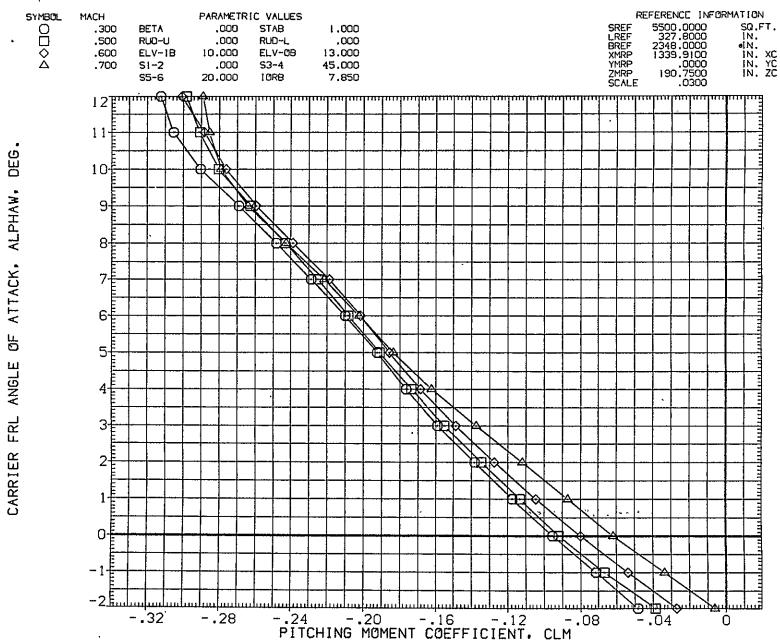
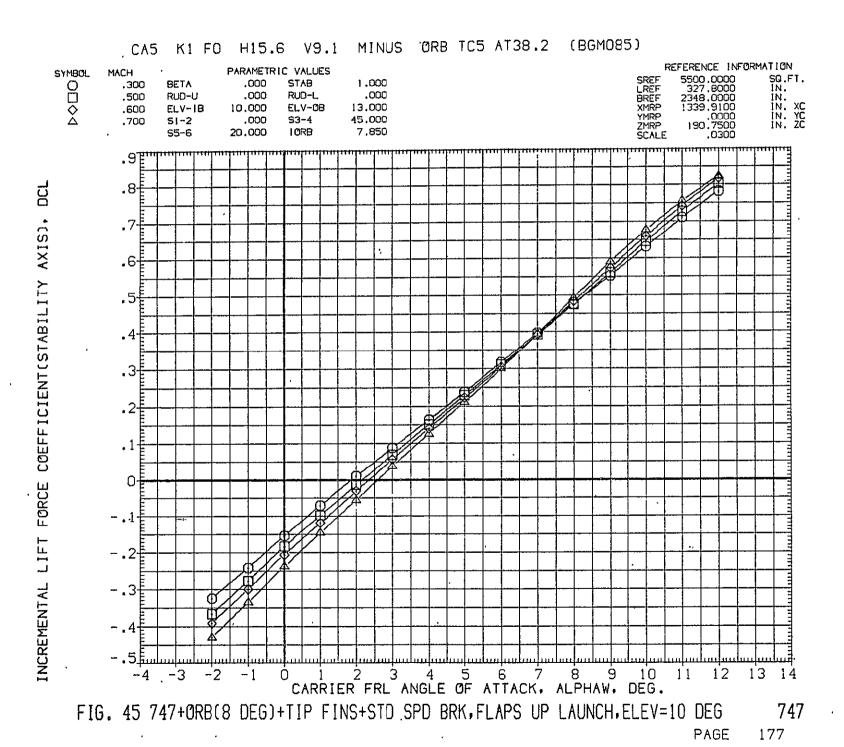


FIG. 44 747+ORB(8 DEG)+TIP FINS+STD SPD BRK,FLAPS UP LAUNCH,ELEV=10 DEG TOTAL PAGE 176



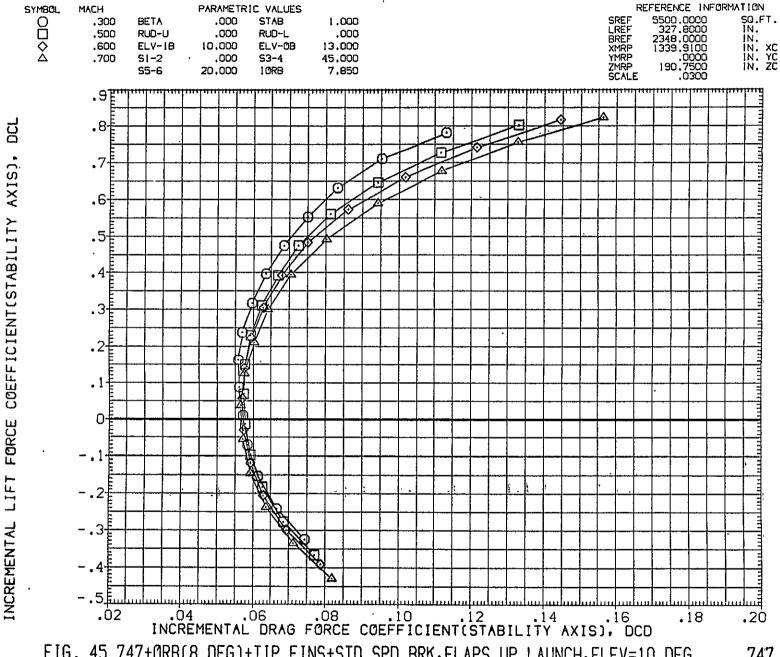
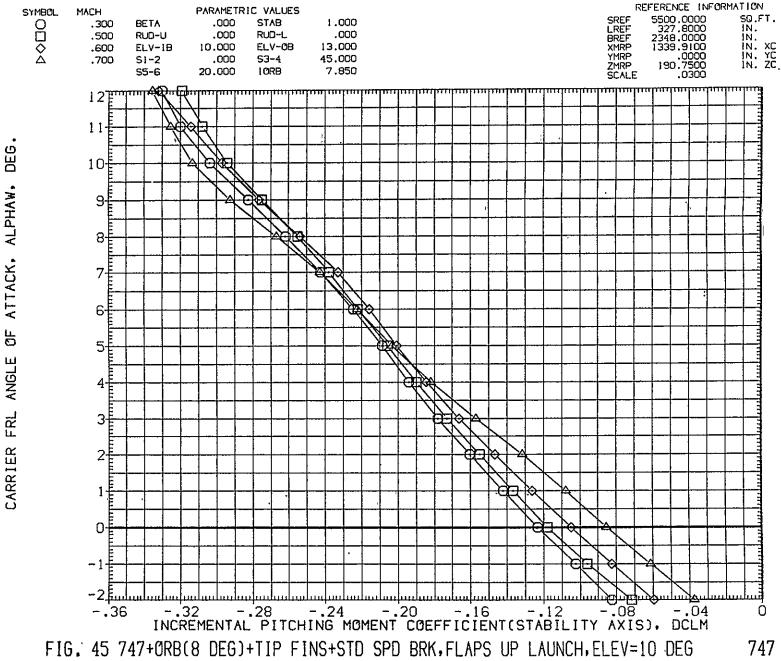


FIG. 45 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ELEV=10 DEG
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ORB TC5 AT38.2 (BGM085) CA5 K1 FO H15.6 V9.1 MINUS REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL SQ.FT. IN. IN. IN. XC IN. YC IN. ZC 5500.0000 327.8000 2348.0000 000.1 BETA STAB .300 RUD-L BREF XMRP YMRP ZMRP SCALE .500 RUD-U .000 1339.9100 .0000 190.7500 .0300 ELV-0B 13.000 10.000 .600 ELV-IB 45.000 .000 S3-4 .700 S1-2 7.850 S5-6 20.000 LORB .9[.8 디디 COEFFICIENT(STABILITY AXIS), .6[.1卡 LIFT FORCE INCREMENTAL -.32 -.28 -.24 -.20 -.16 -.12 -.08 -.04 INCREMENTAL PITCHING MOMENT COEFFICIENT(STABILITY AXIS), DCLM FIG. 45 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ELEV=10 DEG 747



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ORB TC5 AT38.2 (PRES. K1 FO H15.6 V9.1)(YGMO85) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH 2690,0000 474,8100 936,6800 1109,0000 ,0000 375,0000 .0300 SQ.FT. 0000 1,000 STAB BETA .000 .300 ĬÑ. RUD-L .000 .500 RUD-U .000 IN. IN. XO IN. YO IN. ZO ELV-0B 13.000 .601 ELV-IB 10,000 45.000 .000 53-4 .700 \$1-2 7.850 S5-6 20.000 LORB .75 .70 .65[.60[.55 .50 COEFFICIENT. .45 .40[.35 .30₽ .25€ .20= .15[-10 20 24 -8 CARRIER FRL ANGLE OF ATTACK, ALPHAW; DEG.

FIG. 46 747+ORB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ELEV=10 DEG

ORB

181

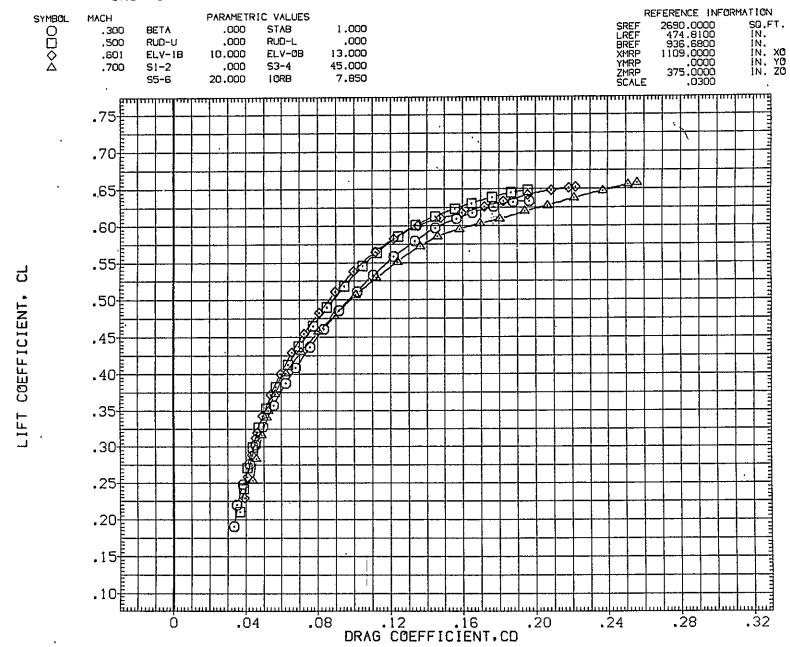
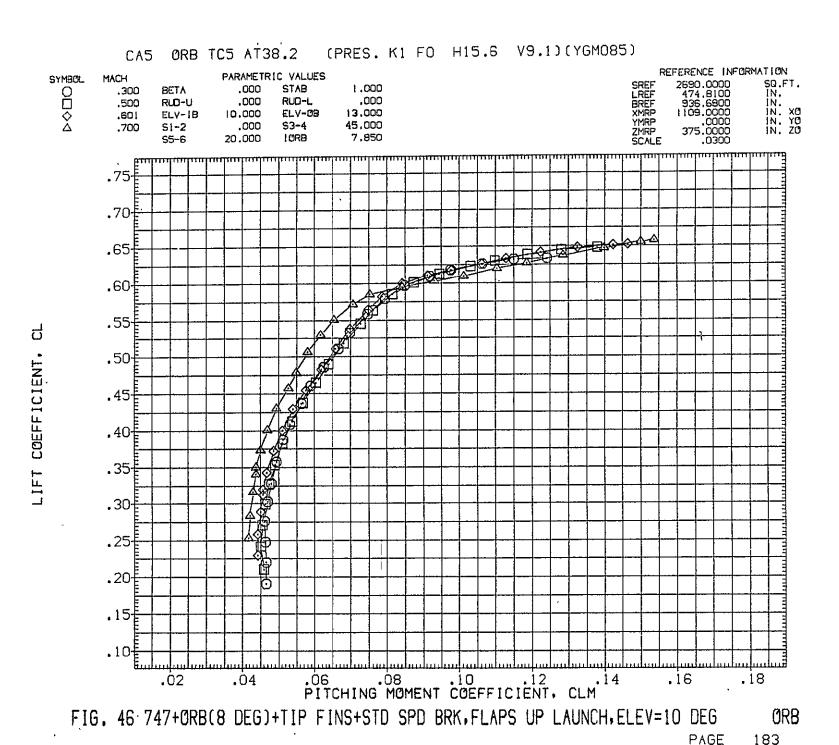


FIG. 46 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ELEV=10 DEG ORB



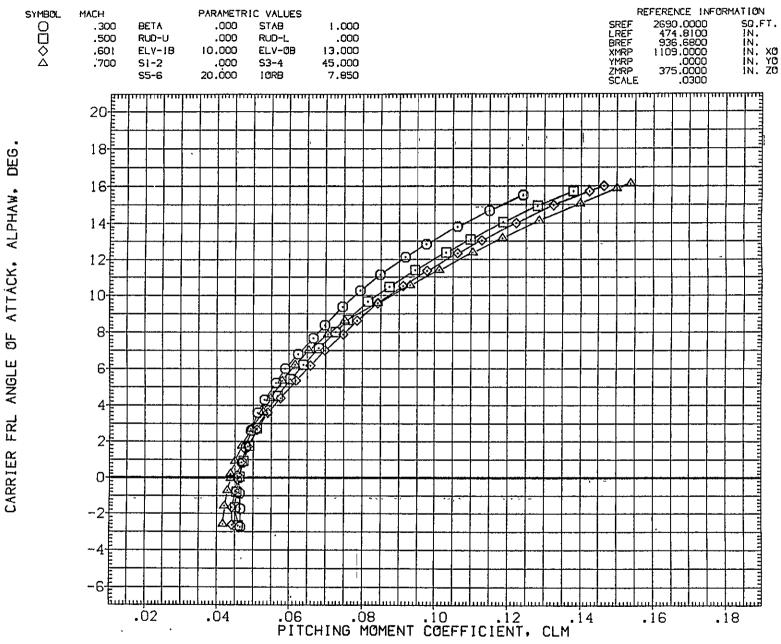


FIG. 46 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ELEV=10 DEG ORB

CA5 K1 FO H15.6 V9.1 (PLUS. ØRB TC5 AT38.2)(RGMO87) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL 5500.0000 327.8000 2348.0000 1339.9100 .0000 190.7500 .0300 SQ.FT. STAB 0 1.000 .000 .500 BETA ELV-1B .000 .600 RUD-U .000 IN. IN. XC IN. YC IN. ZC 51-2 .000 ELV-08 3.000 YMRP ZMRP SCALE 45.000 SS-6 20.000 \$3-4 IORB 7.850 ORBELV 5.000 .1 .8-1.6 1.4 1.2 1.0 ರ .8[LIFT COEFFICIENT, .6- 0 [-.6[1 2 3 4 5 6 7 8 9 10 CARRIER FRL ANGLE OF ATTACK, ALPHAW, DEG.

FIG. 47 747+0RB(8 DEG)+TIP FINS, FLAPS UP LAUNCH, ORB ELEVON=5 DEG

TOTAL

185

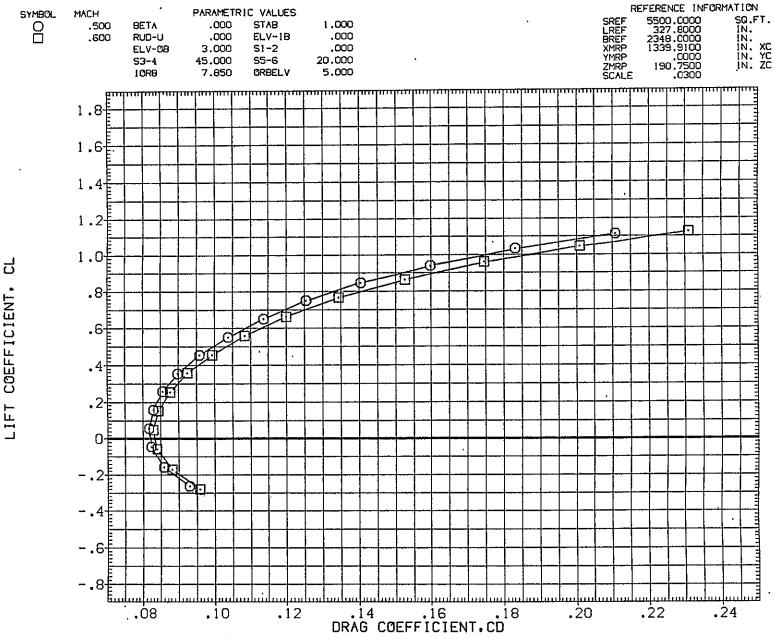


FIG. 47 747+0RB(8 DEG)+TIP FINS, FLAPS UP LAUNCH, ORB ELEVON=5 DEG TOTAL

FIG. 47 747+0RB(8 DEG)+TIP FINS, FLAPS UP LAUNCH, ORB ELEVON=5 DEG

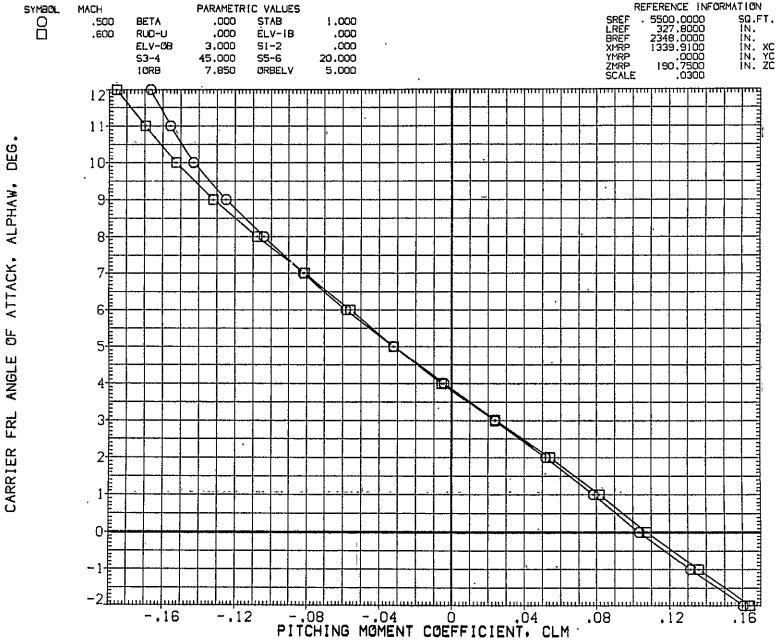


FIG. 47 747+0RB(8 DEG)+TIP FINS, FLAPS UP LAUNCH, ORB ELEVON=5 DEG

TOTAL

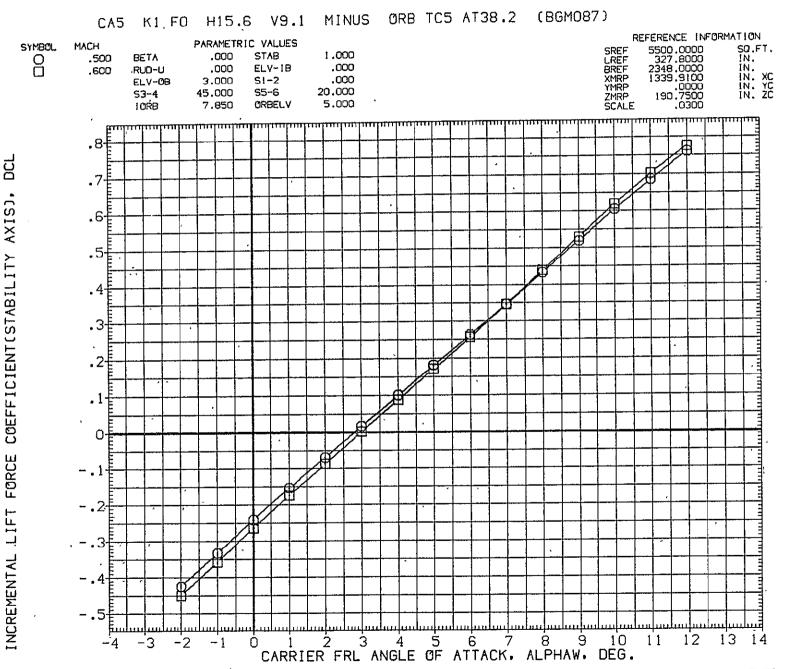


FIG. 48 747+0RB(8 DEG)+TIP FINS, FLAPS UP LAUNCH, ORB ELEVON=5 DEG

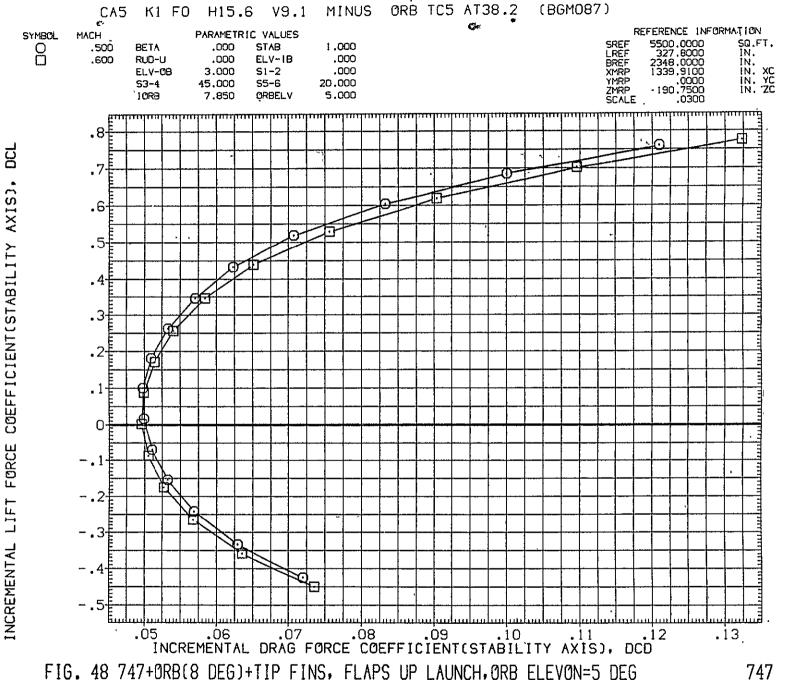
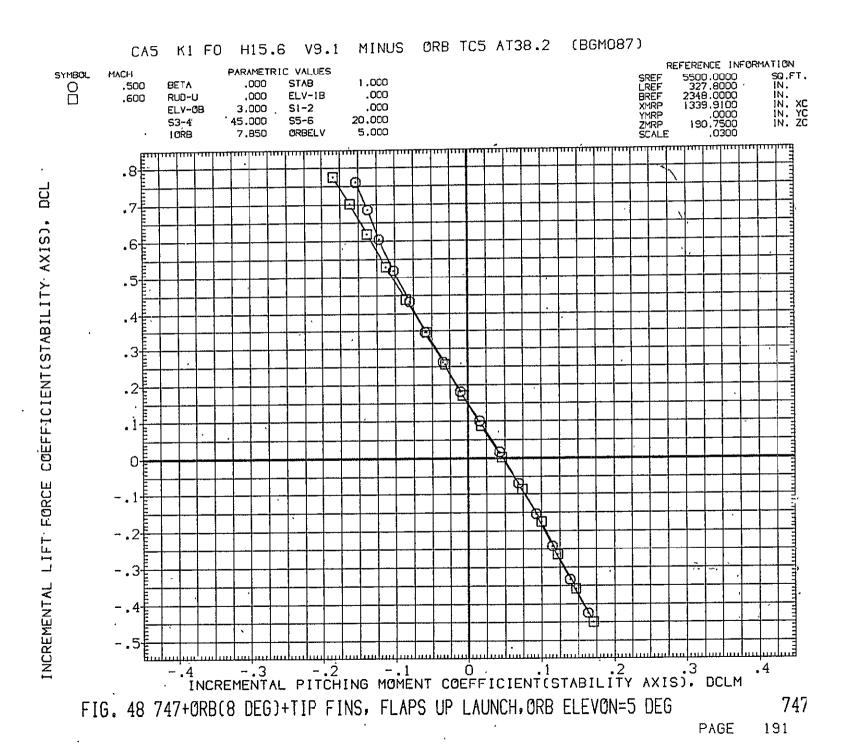
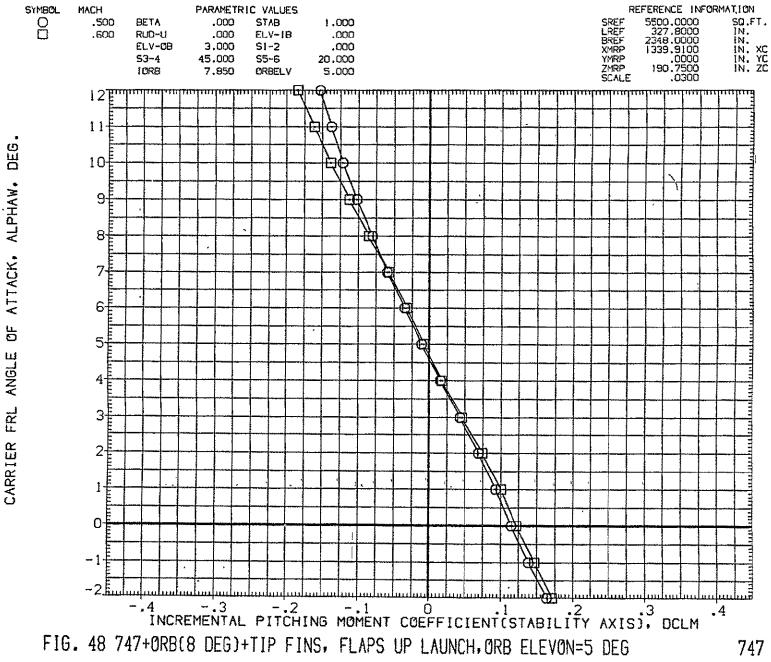
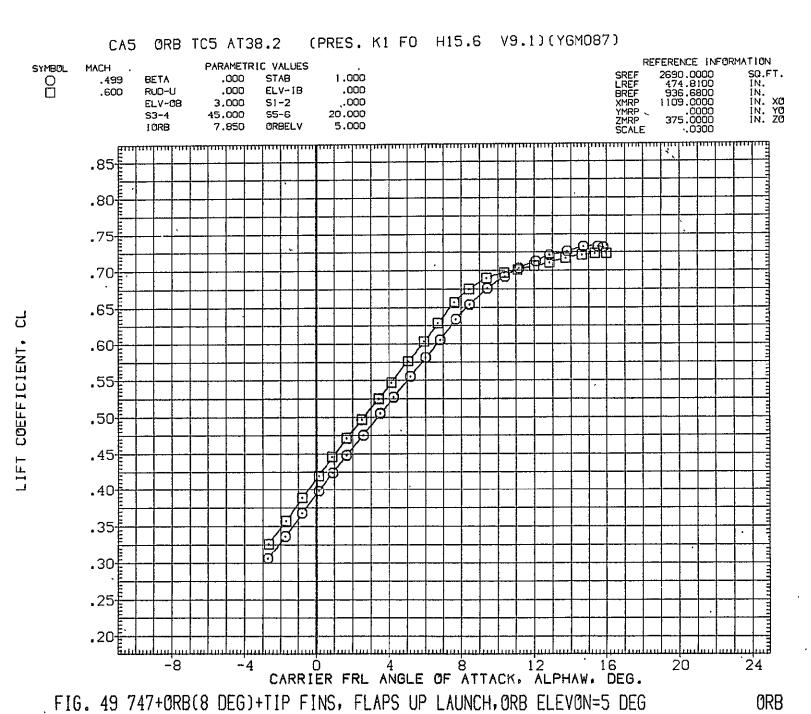


FIG. 48 747+0RB(8 DEG)+TIP FINS, FLAPS UP LAUNCH, ORB ELEVON=5 DEG





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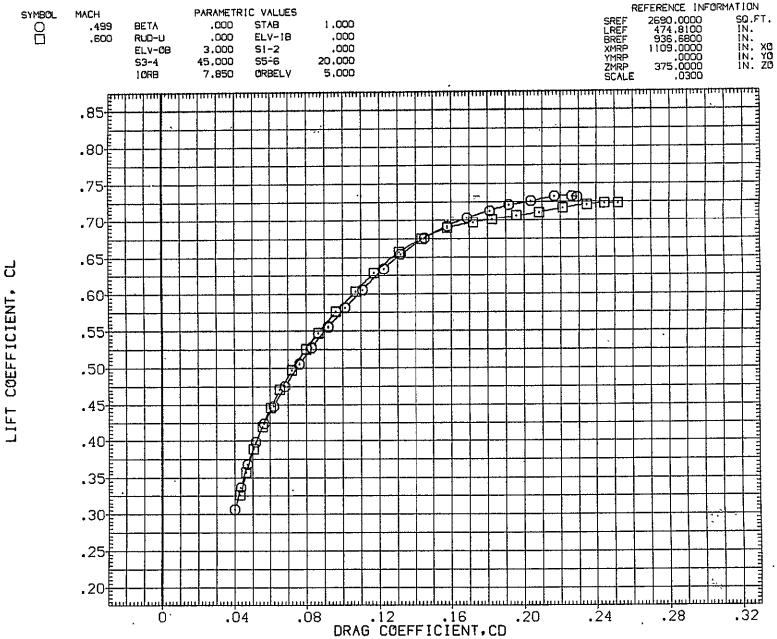
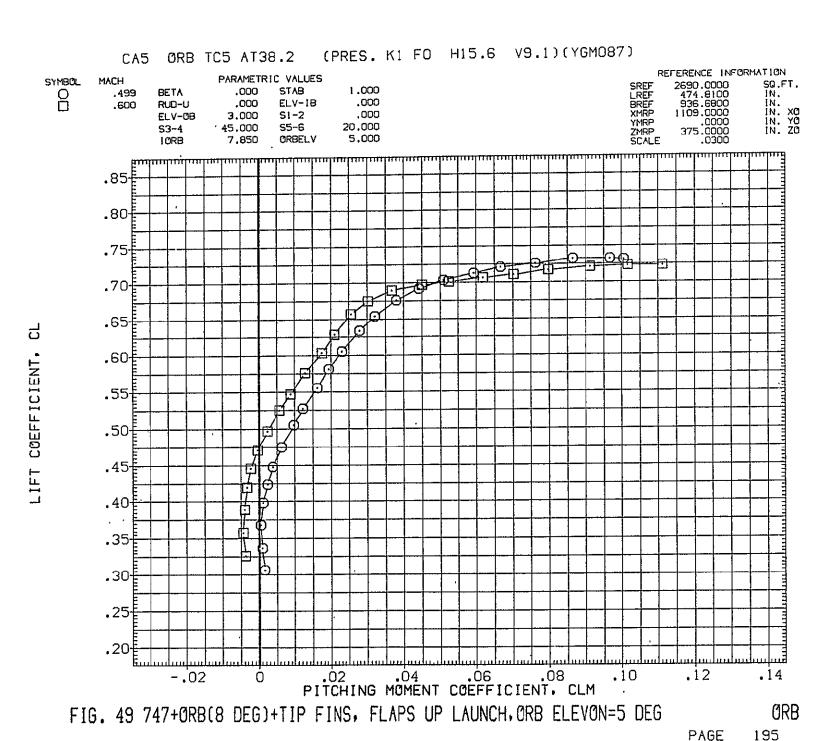


FIG. 49 747+0RB(8 DEG)+TIP FINS, FLAPS UP LAUNCH, ORB ELEVON=5 DEG

`ORB



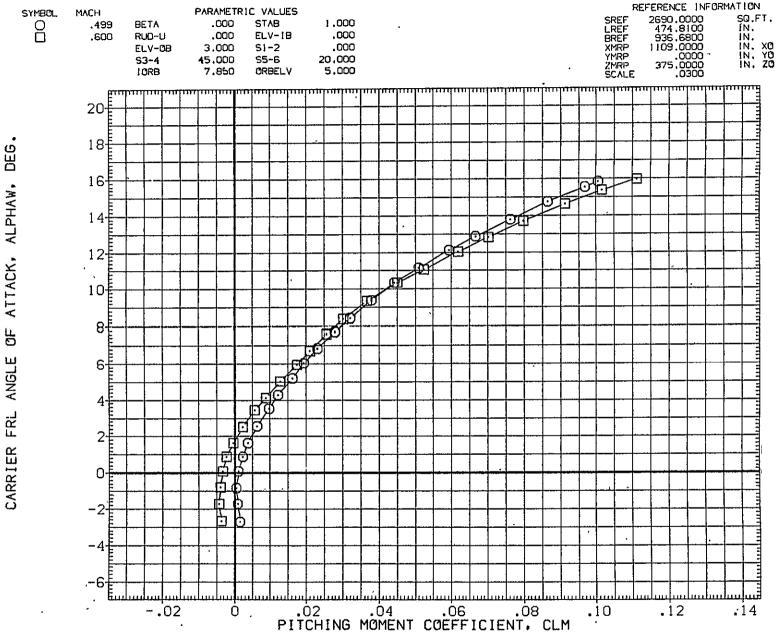


FIG. 49 747+0RB(8 DEG)+TIP FINS, FLAPS UP LAUNCH, ORB ELEVON=5 DEG

ORB

FIG. 50 747+0RB(8 DEG)+TIP FINS, FLAPS UP LAUNCH, NO SPEED BRAKES

TOTAL

197

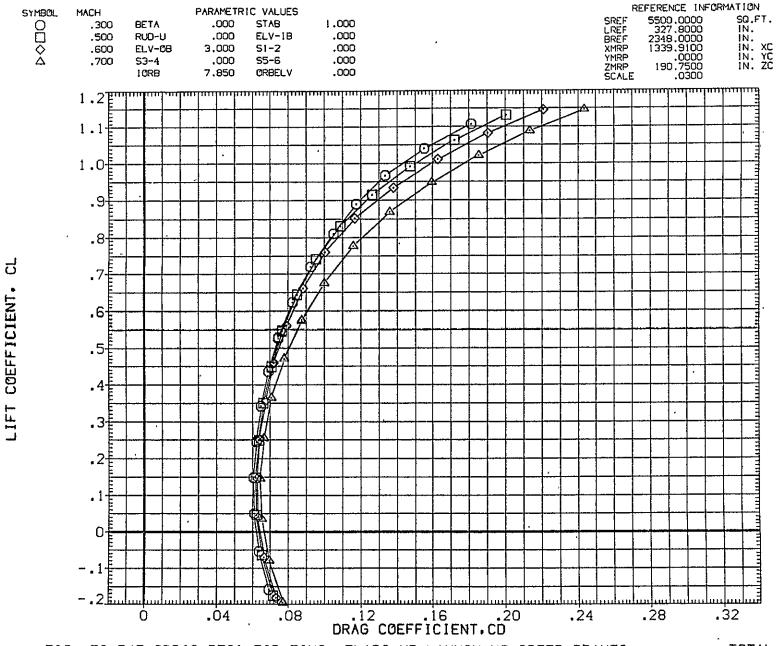


FIG. 50 747+0RB(8 DEG)+TIP FINS, FLAPS UP LAUNCH, NO SPEED BRAKES

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CA5 K1 FO H15.6 V9.1 (PLUS. ØRB TC5 AT38.2)(RGM090) REFERENCE INFORMATION SYMBOL MACH PARAMETRIC VALUES 5500.0000 327.8000 SO.FT. .300 BETA .000 STAB 1,000 .000 ELV-IB .000 .500 RUD-U 2348.0000 1339.9100 .0000 190.7500 IN. XC IN. XC IN. YC IN. ZC S1-2 .000 .600 ELV-0B 3,000 YMRP .700 .000 S3-4 .000 S5-6 ZMRP SCALE 7.850 CRBELV .000 1.1卡 1.0 ·6| LIFT 16 -.12 -.08 -.04 O PITCHING MOMENT COEFFICIENT, CLM .04 .Ó8

FIG. 50 747+0RB(8 DEG)+TIP FINS, FLAPS UP LAUNCH, NO SPEED BRAKES

TOTAL

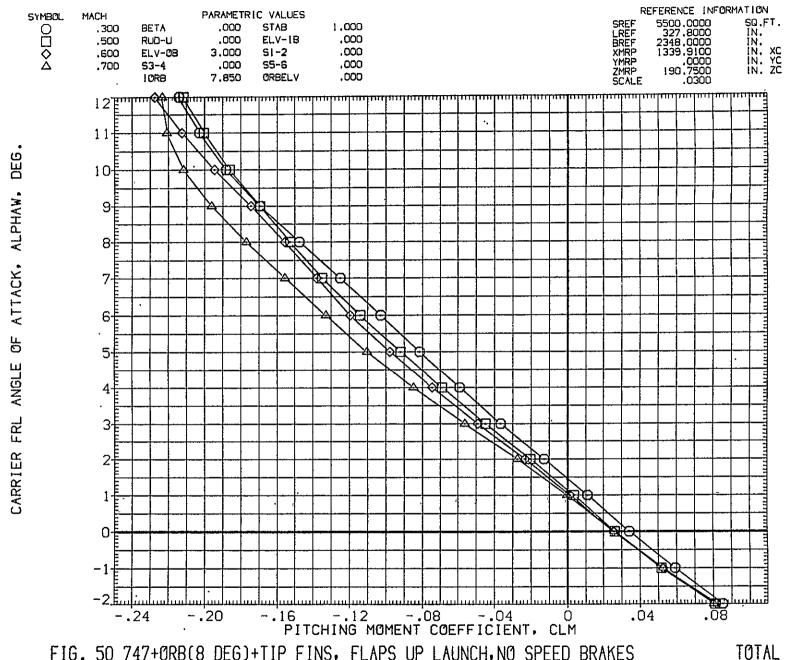


FIG. 50 747+0RB(8 DEG)+TIP FINS, FLAPS UP LAUNCH, NO SPEED BRAKES





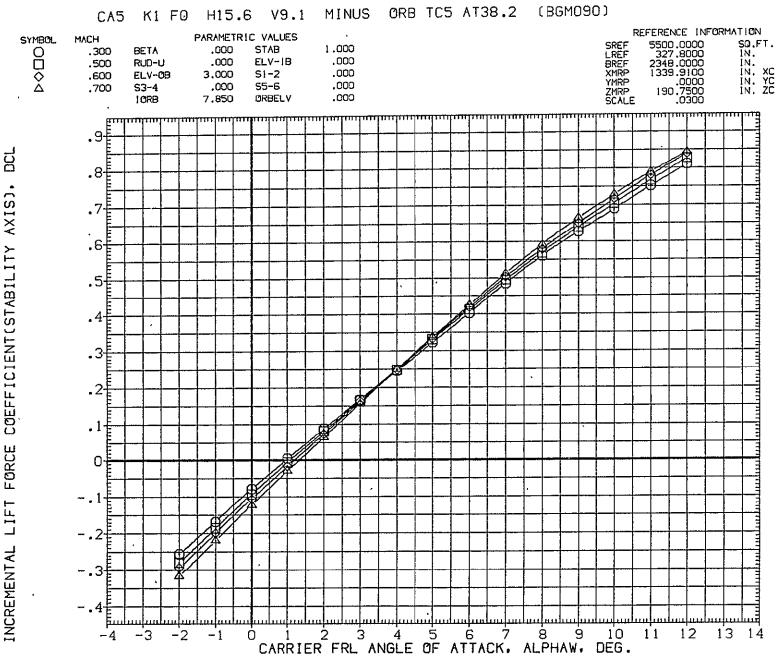


FIG. 51 747+0RB(8 DEG)+TIP FINS, FLAPS UP LAUNCH, NO SPEED BRAKES

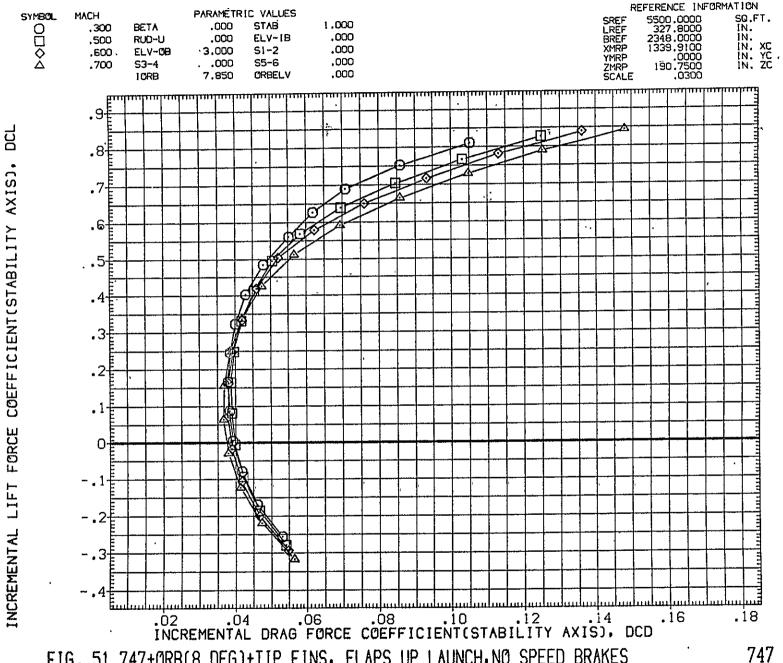


FIG. 51 747+ORB(8 DEG)+TIP FINS, FLAPS UP LAUNCH, NO SPEED BRAKES

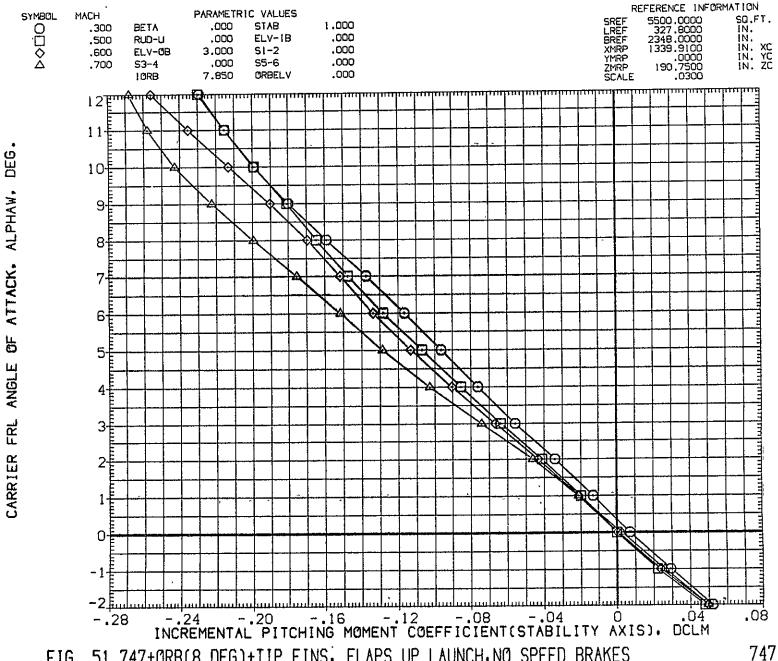
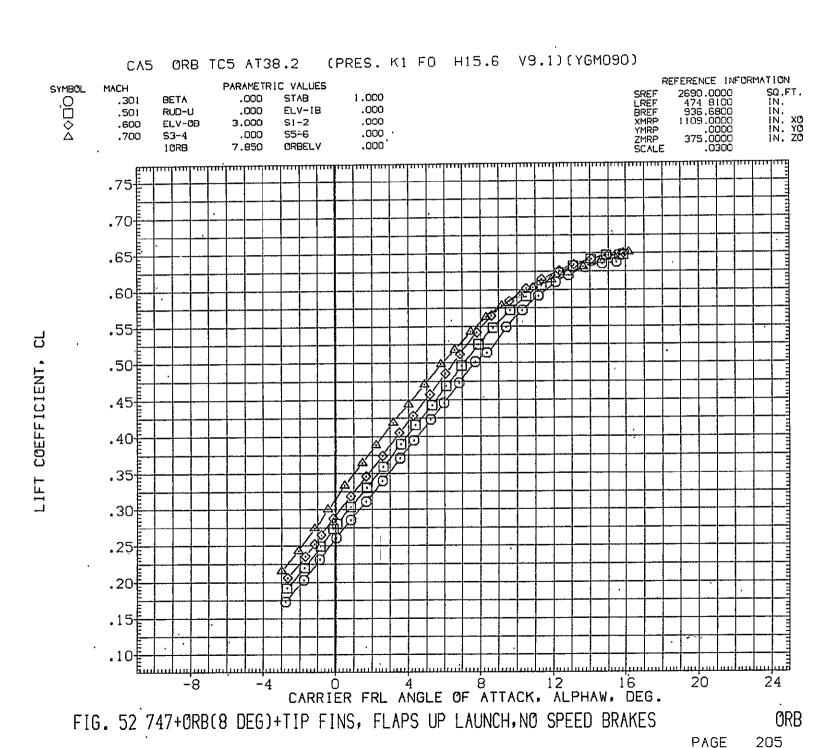


FIG. 51 747+0RB(8 DEG)+TIP FINS, FLAPS UP LAUNCH, NO SPEED BRAKES



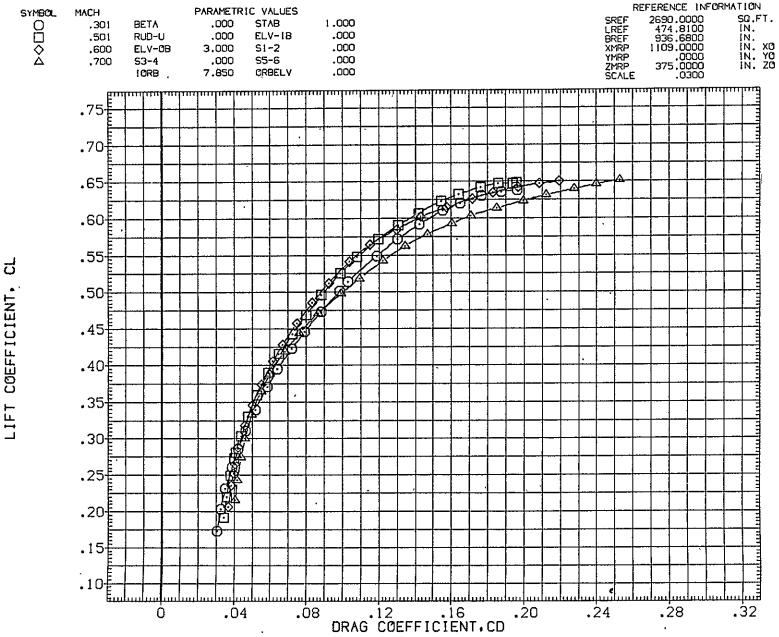
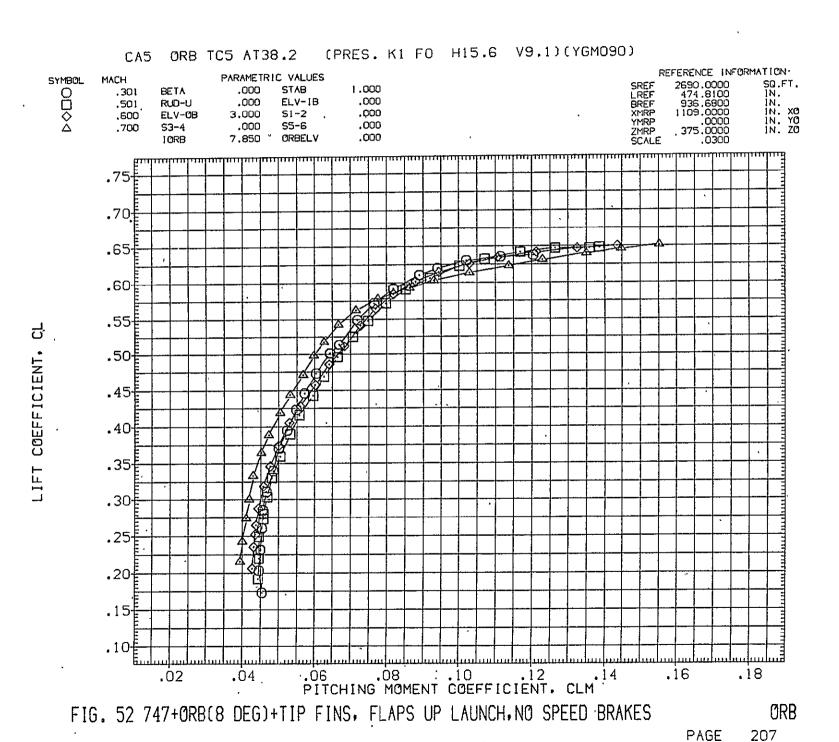


FIG. 52 747+0RB(8 DEG)+TIP FINS, FLAPS UP LAUNCH, NO SPEED BRAKES

ORB



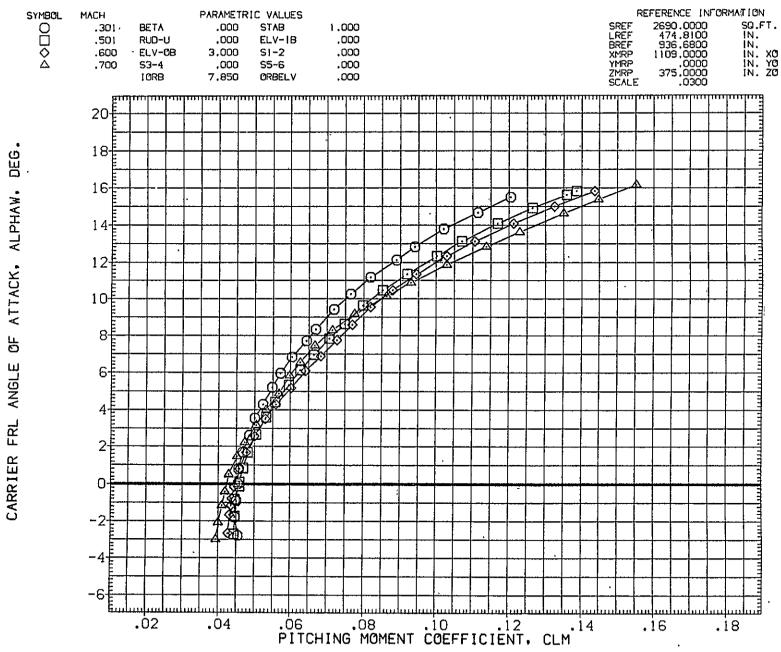


FIG. 52 747+0RB(8 DEG)+TIP FINS, FLAPS UP LAUNCH, NO SPEED BRAKES

· ORB

AMPOUNTABLES

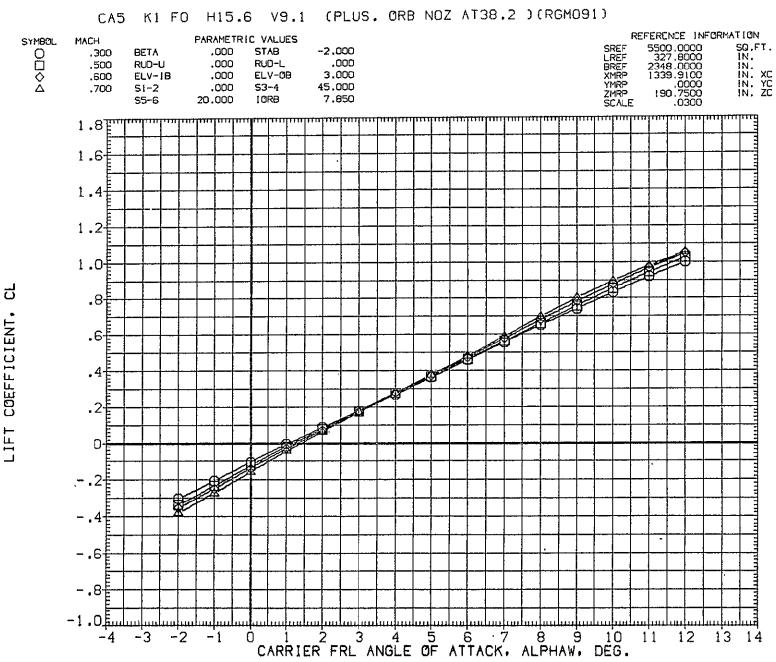


FIG. 53 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP(S=-2) TAIL CONE OFF TOTAL PAGE 209

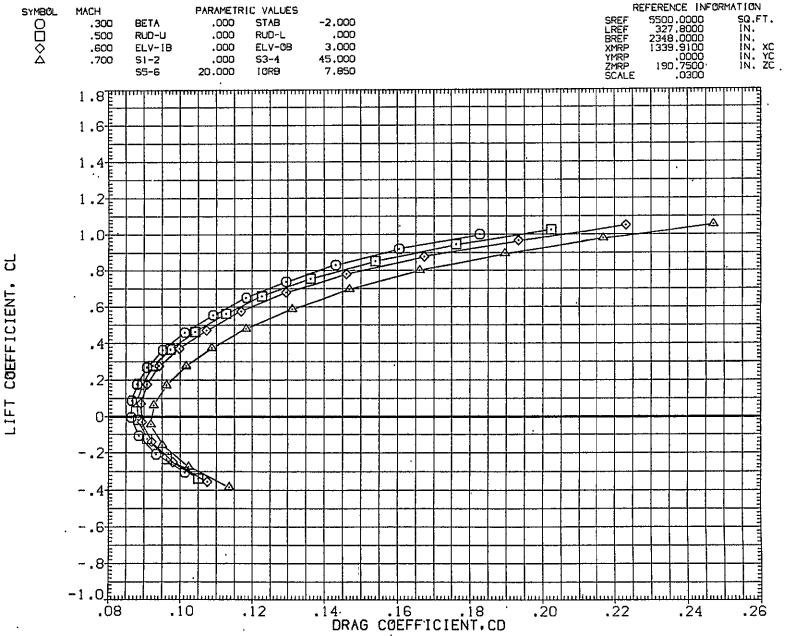


FIG. 53 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP(S=-2) TAIL CONE OFF TOTAL
PAGE 210

CA5 K1 FO H15.6 V9.1 (PLUS. ORB NOZ AT38.2)(RGMO91) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL 5500.0000 327.8000 2348.0000 1339.9100 .0000 190.7500 SQ.FT. 0000 BETA .000 STAB -2.000 .000 .000 RUD-L ,500 RUD-U IN. XC IN. YC IN. ZC ,000 ELV-ØB 3,000 .600 ELV-IB YMRP ZMRP SCALE 45.000 .700 S1-2 .000 S3-4 7.850 S5-6 20,000 1.8 [1.6 1.4 1.2 1.0 .8[-.6 -.8[.04 .08 .12 .16 PITCHING MOMENT COEFFICIENT, CLM .28 24 -.04 .20 0

FIG. 53 747+0RB(8 DEG)+TIP FINS+STD SPD BRK,FLAPS UP(S=-2) TAIL CONE OFF TOTA

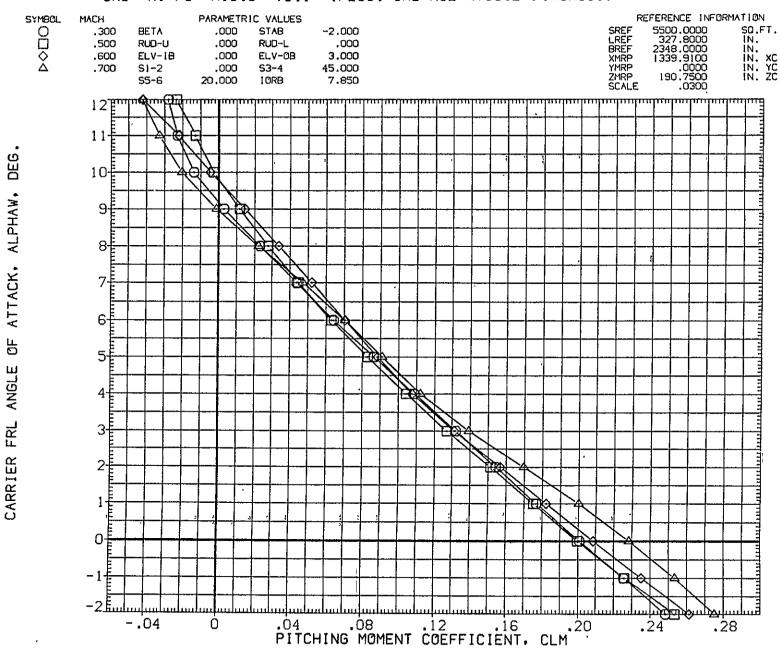
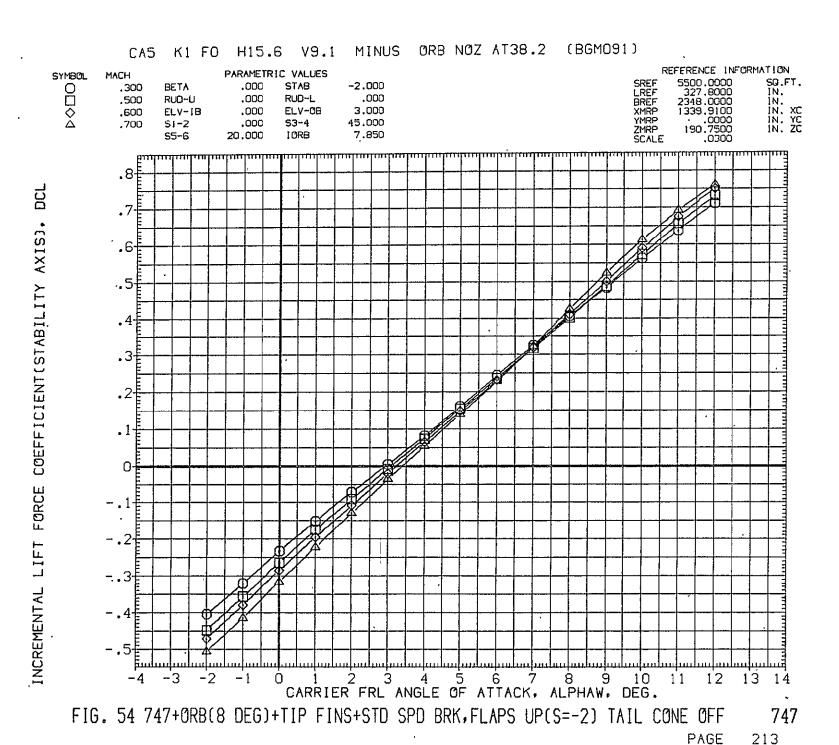


FIG. 53 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP(S=-2) TAIL CONE OFF TOTAL PAGE 212



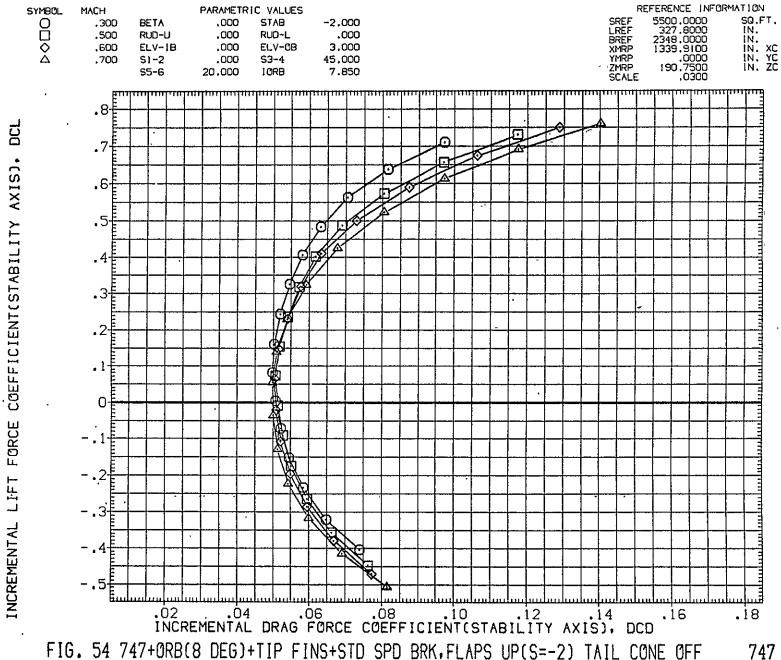
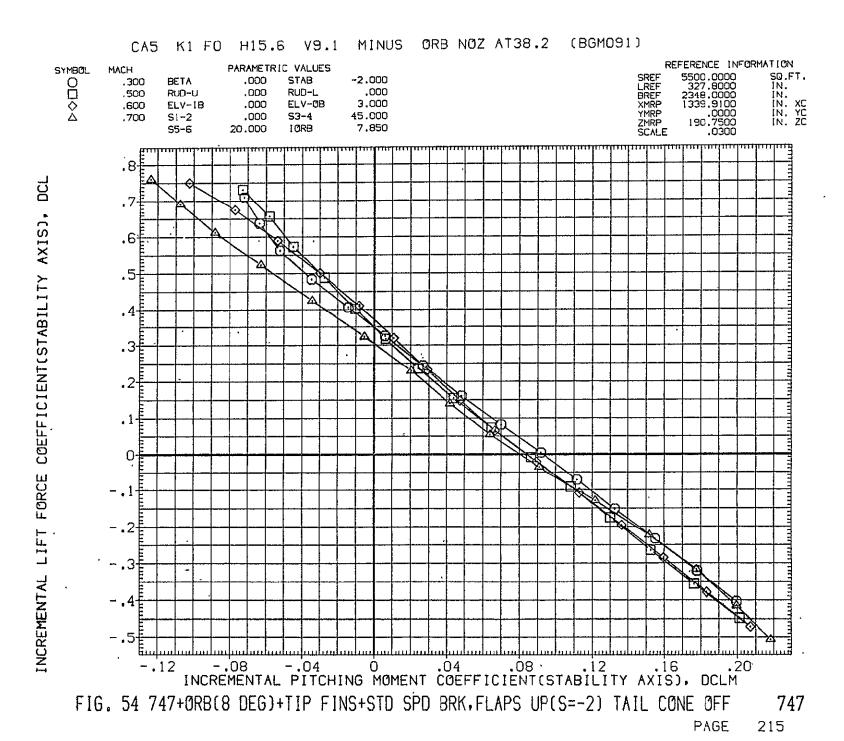


FIG. 54 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP(S=-2) TAIL CONE OFF 214 PAGE

parameter survey and the



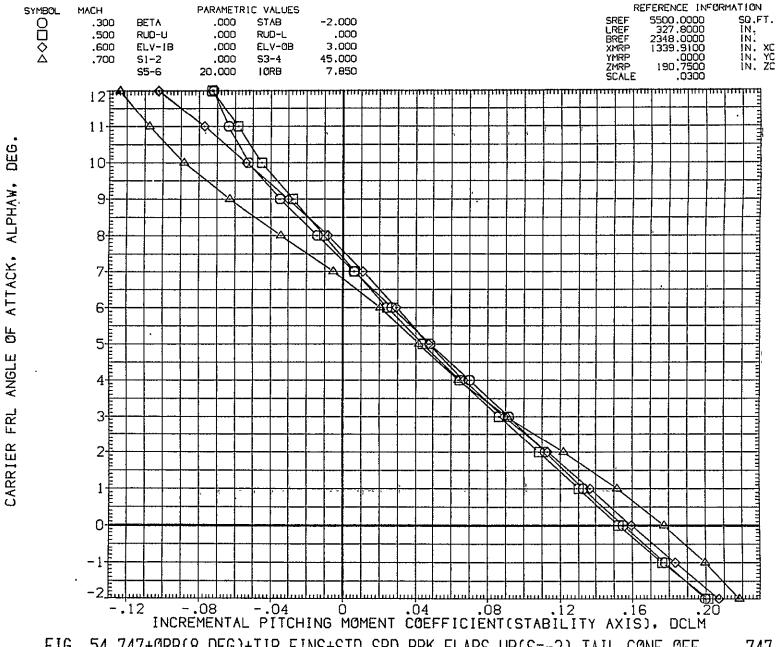


FIG. 54 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP(S=-2) TAIL CONE OFF
PAGE 216

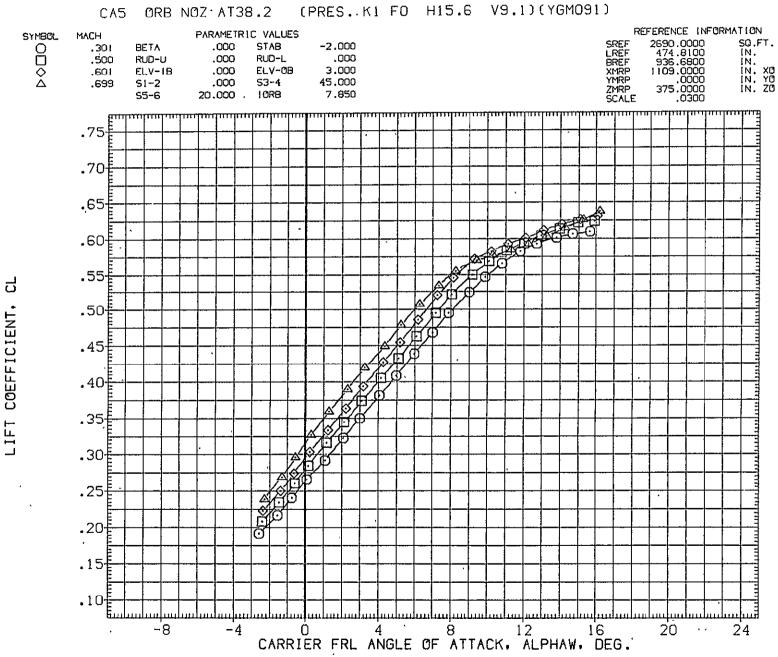


FIG. 55 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP(S=-2) TAIL CONE OFF ORB

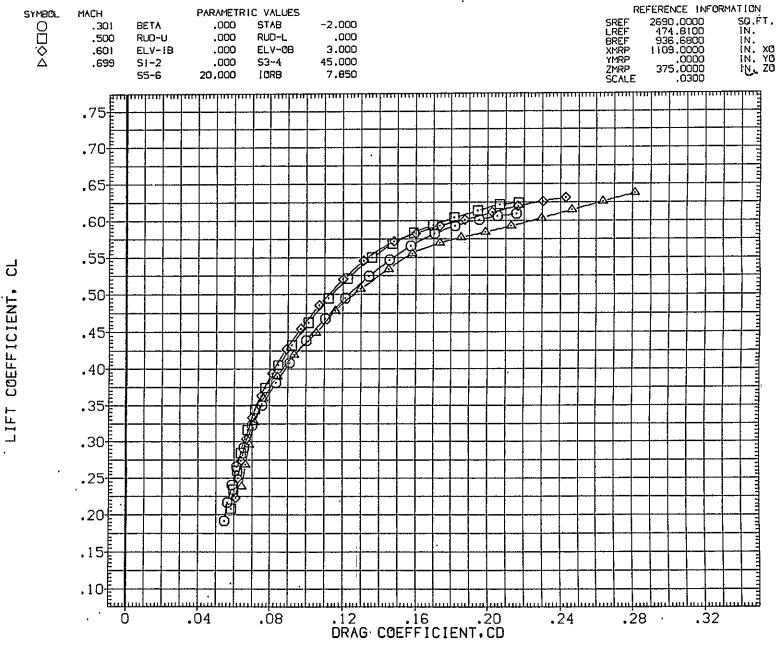


FIG. 55 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP(S=-2) TAIL CONE OFF ORB

(PRES. K1 FO H15.6 V9.1)(YGMO91) CAS ORB NOZ AT38.2 REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH 2690.0000 474.8100 936.6800 1109.0000 .0000 375.0000 SQ.FT. .000 STAB -2,000 .301 BETA LREF .IN. .000 .500 RUD-U .000 RUD-L IN. X0 IN. X0 IN. Y0 IN. Z0 3.000 ELV-0B .601 ELV-1B .000 XMRP YMRP ZMRP SCALE ,699 .000 \$3-4 45,000 S1-2 LORB 7.850 20,000 S5-6 .75E .70 .65₽ .60[.55[占 COEFFICIENT, .50 .45[.40 .35[.30[.25 .20 .15[.10 .08 .10 .12 .14 PITCHING MOMENT COEFFICIENT, CLM .04 .06 .i6 .18 .20

FIG. 55 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP(S=-2) TAIL CONE OFF ORB

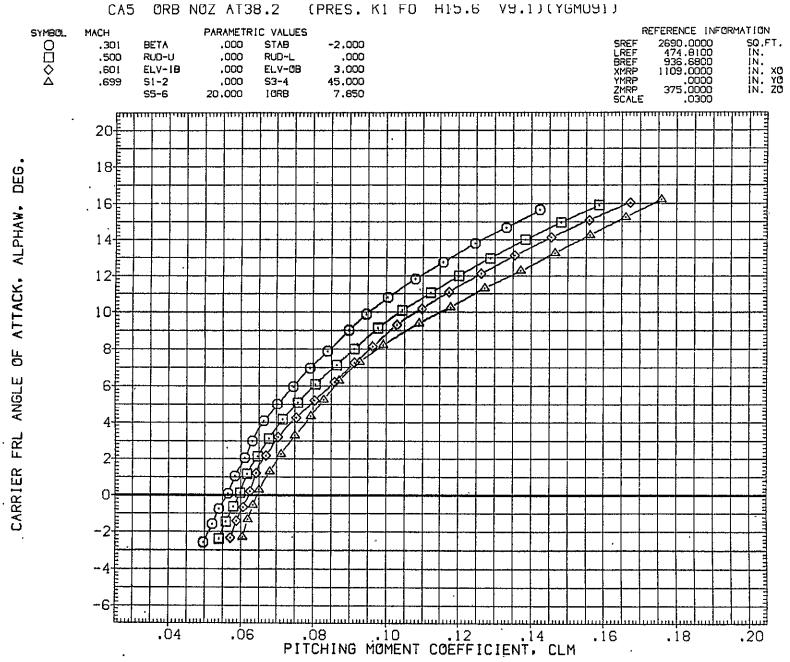


FIG. 55 747+0RB(8 DEG)+TIP FINS+STD SPD BRK.FLAPS UP(S=-2) TAIL CONE OFF ORB

FIG. 56 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH (S=+1)

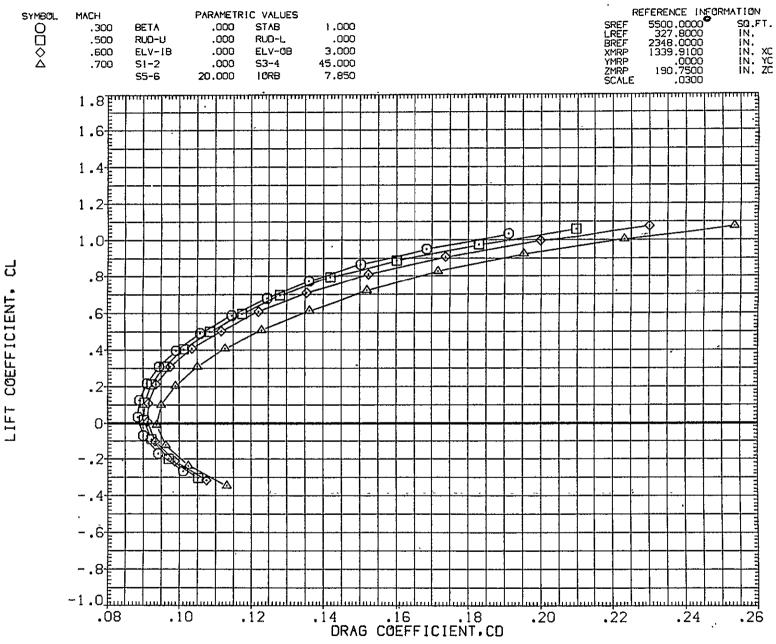


FIG. 56 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH (S=+1)
PAGE 222

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COEFFICIENT.

FIG. 56 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH (S=+1)

PITCHING MOMENT COEFFICIENT, CLM

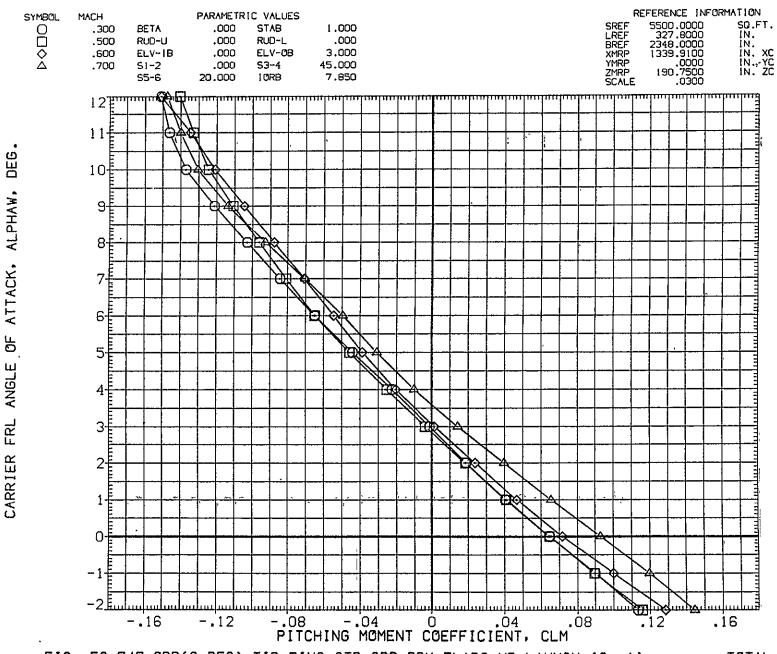
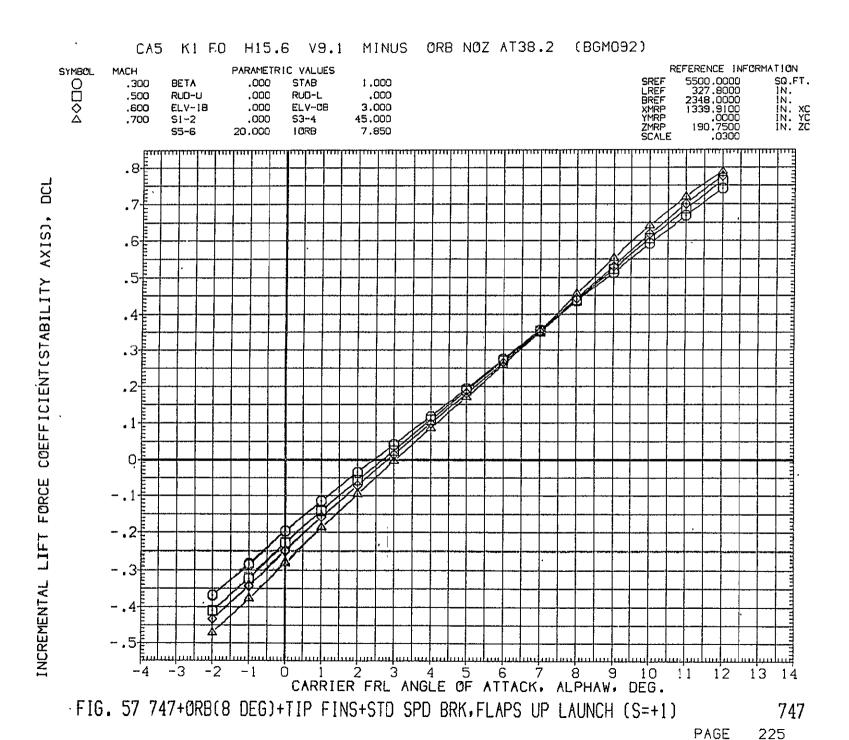


FIG. 56 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH (S=+1)

PAGE 224



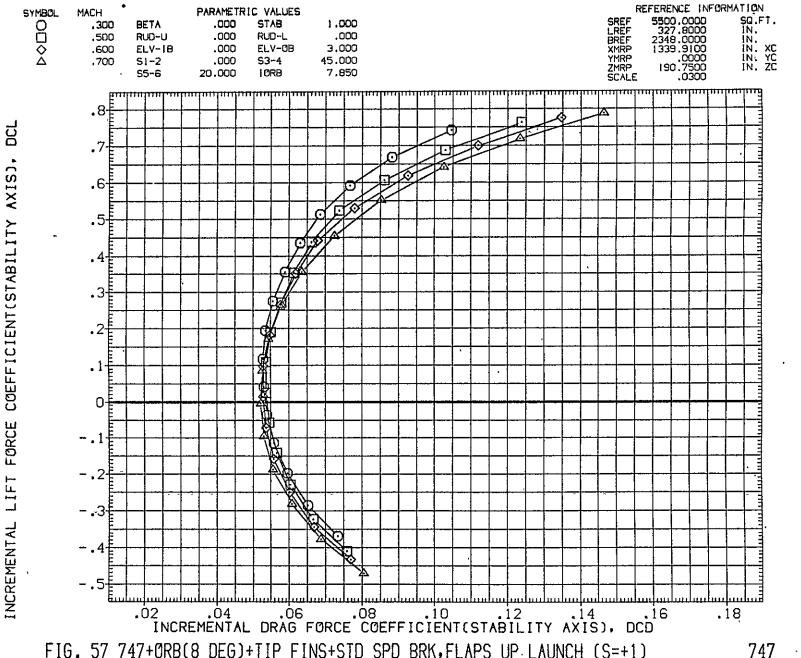


FIG. 57 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH (S=+1)

CA5 K1 FO H15.6 V9.1 MINUS ORB NOZ AT38.2 (BGM092) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH 5500.0000 327.8000 2348.0000 1339.9100 .0000 190.7500 .0300 SQ.FT. IN. IN. XC IN. XC IN. YC 0000 STAB 1.000 .000 ,300 BETA LREF BREF XMRP .000 RUD-U .000 RUD-L .500 3.000 .000 ELV-0B .600 ELV-1B YMRP ZMRP SCALE 45.000 .000 S3-4 ,700 S1-2 20.000 LORB 7.850 S5-6 덤 AXIS) .6‡ COEFFICIENTCSTABILITY ·4= .1| 0 FORCE INCREMENTAL -.20 -.16 -.12 -.08 -.04 0 .04 .0 INCREMENTAL PITCHING MOMENT COEFFICIENT(STABILITY AXIS). DCLM FIG. 57 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH (S=+1) 747

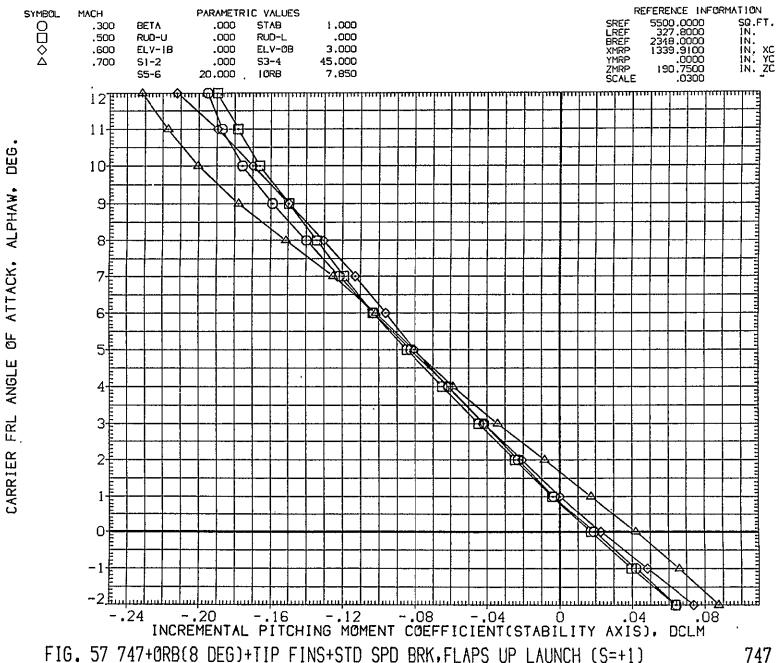


FIG. 57 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH (S=+1)

12

CARRIER FRL ANGLE OF ATTACK, ALPHAW, DEG.

16

0000

김

COEFFICIENT,

LIFT

FIG. 58 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH (S=+1)

-8

ORB

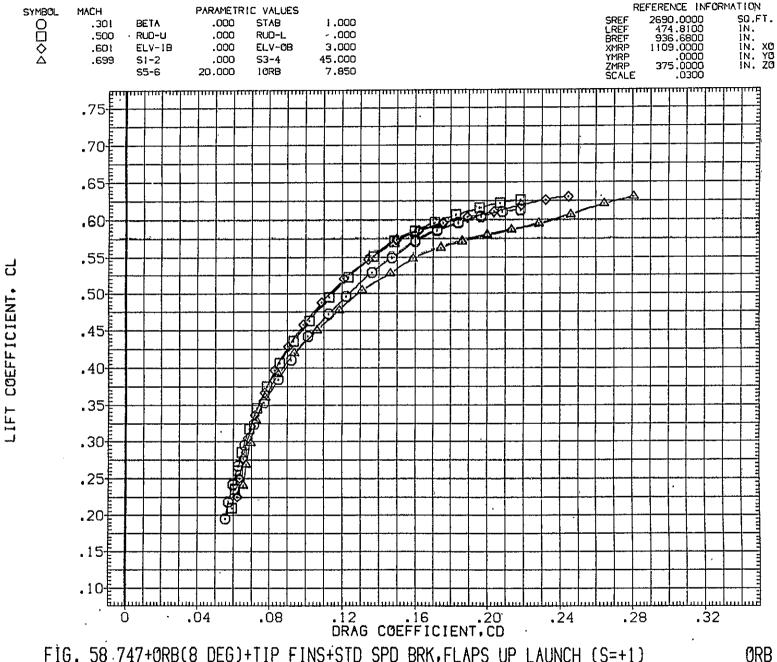
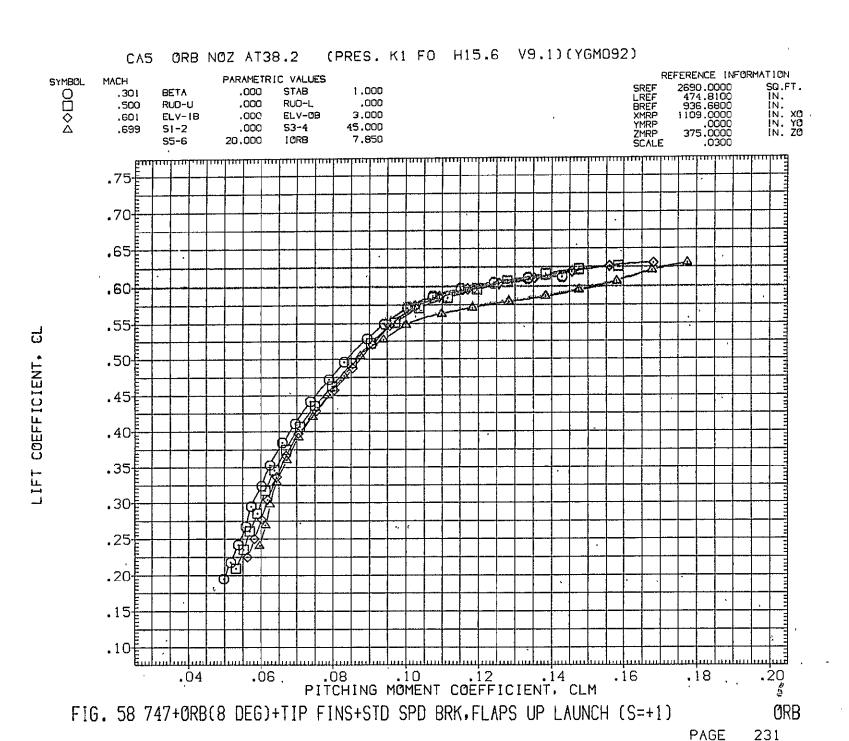


FIG. 58 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH (S=+1)



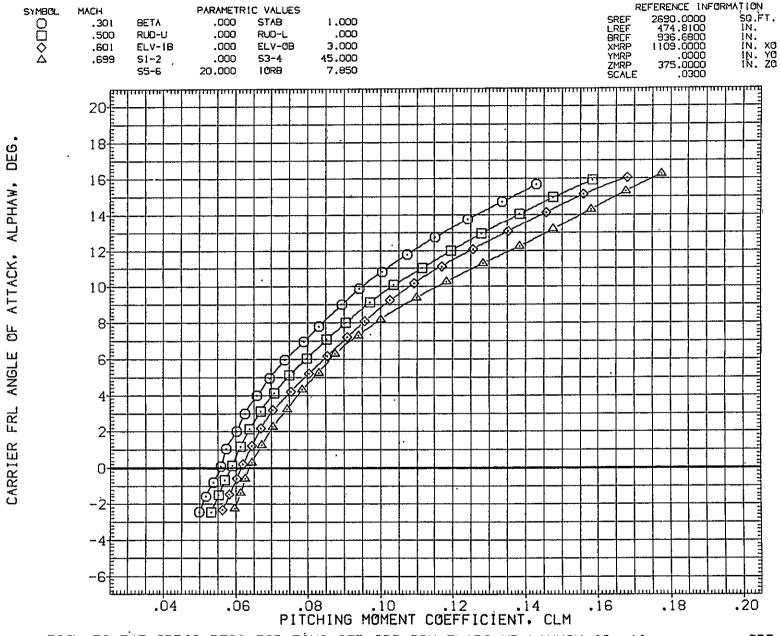


FIG. 58 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH (S=+1)

ØRB

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CA5 K1 FO H15.6 V9.1 (PLUS. ØRB NØZ AT38.2)(RGM095)

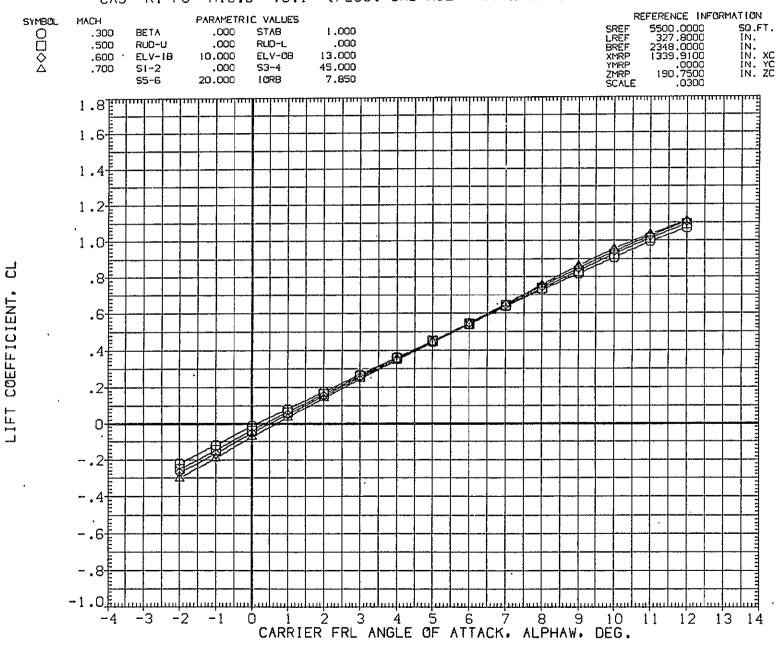


FIG. 59 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP, TAIL CONE OFF, DE=10 TOTAL PAGE 233

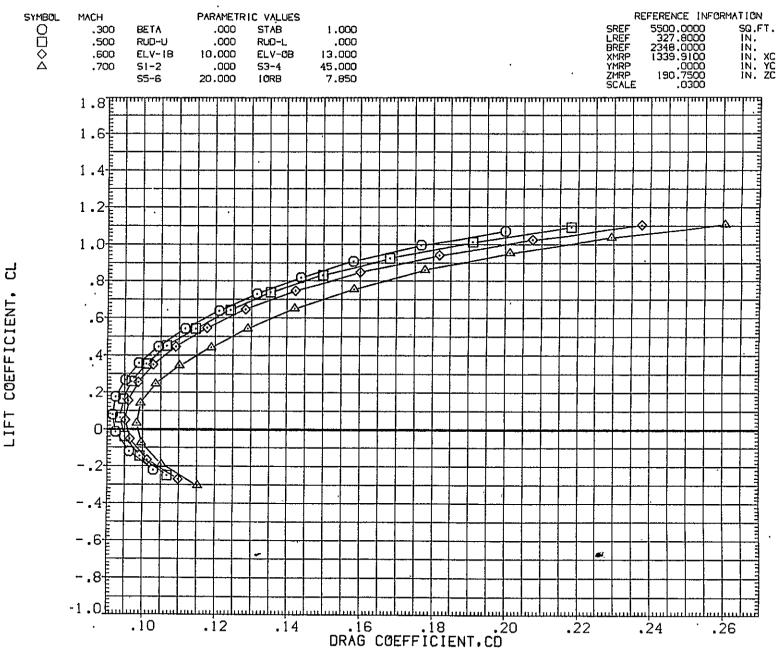


FIG. 59 747+0RB(8 DEG)+TIP FINS+STD SPD BRK,FLAPS UP.TAIL CONE OFF.DE=10 TOTAL PAGE 234

FIG. 59 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP, TAIL CONE OFF, DE=10 TOTAL PAGE 235

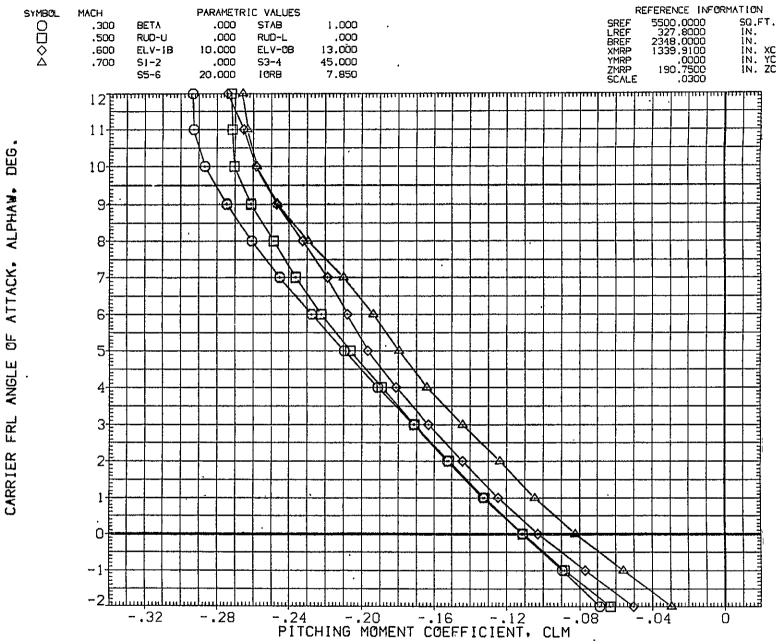


FIG. 59 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP, TAIL CONE OFF, DE=10 TOTAL

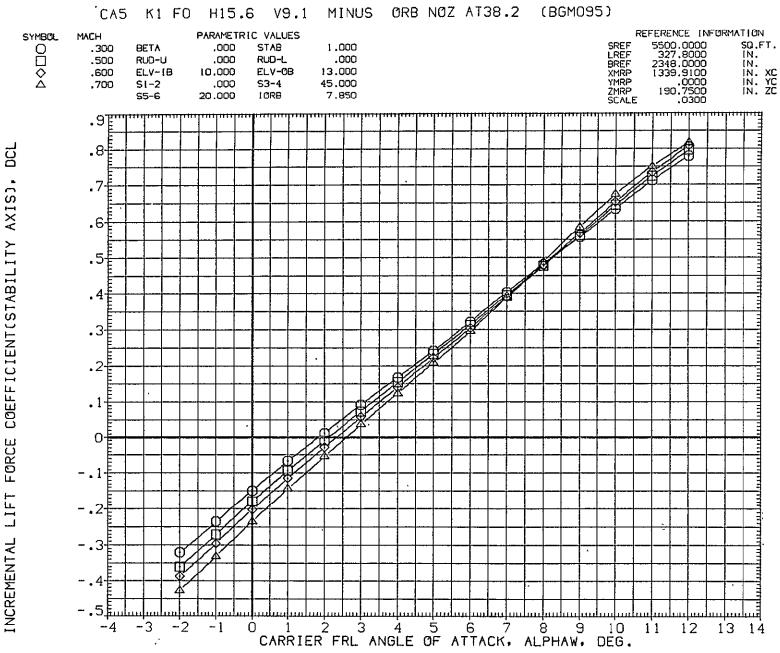


FIG. 60 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP, TAIL CONE OFF, DE=10

PAGE 237

747

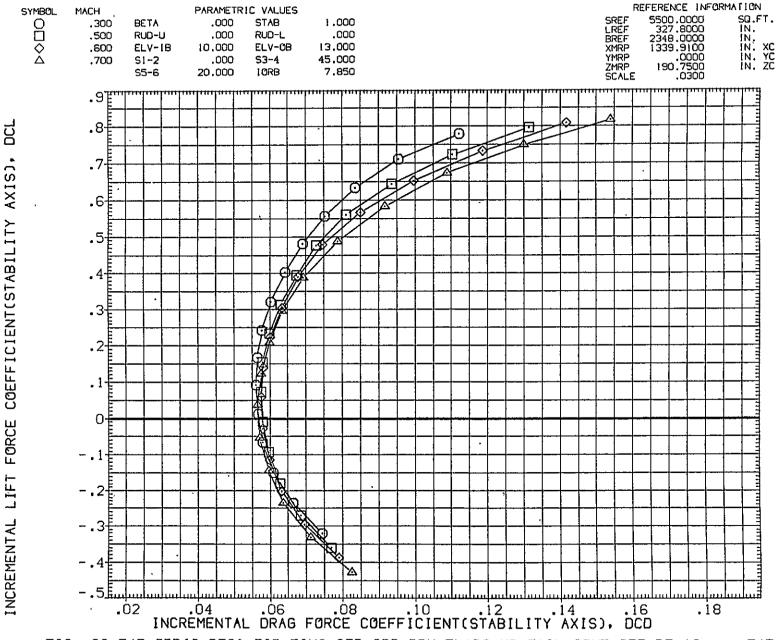
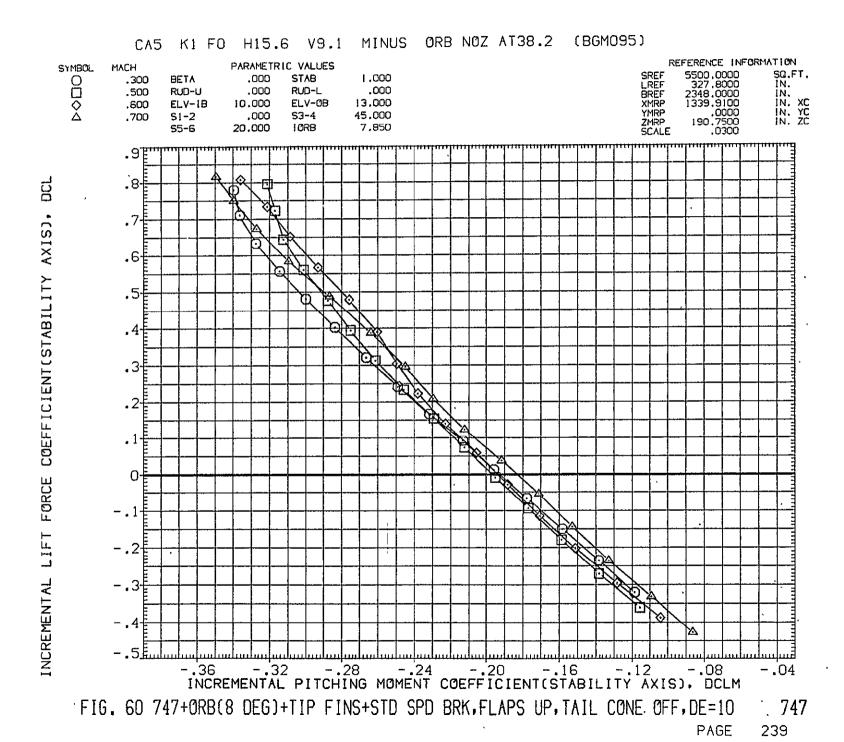


FIG. 60 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP, TAIL CONE OFF, DE=10 747



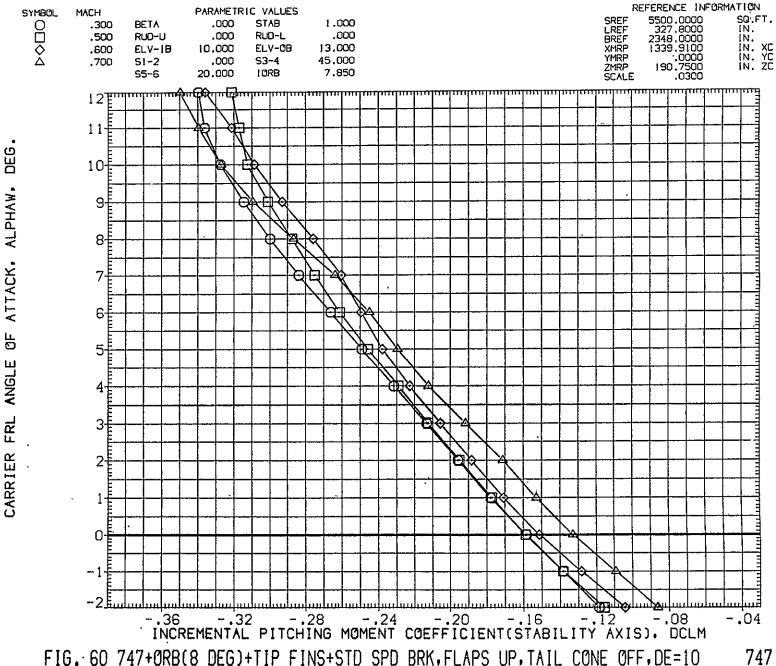


FIG. 60 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP, TAIL CONE OFF, DE=10 PAGE 240

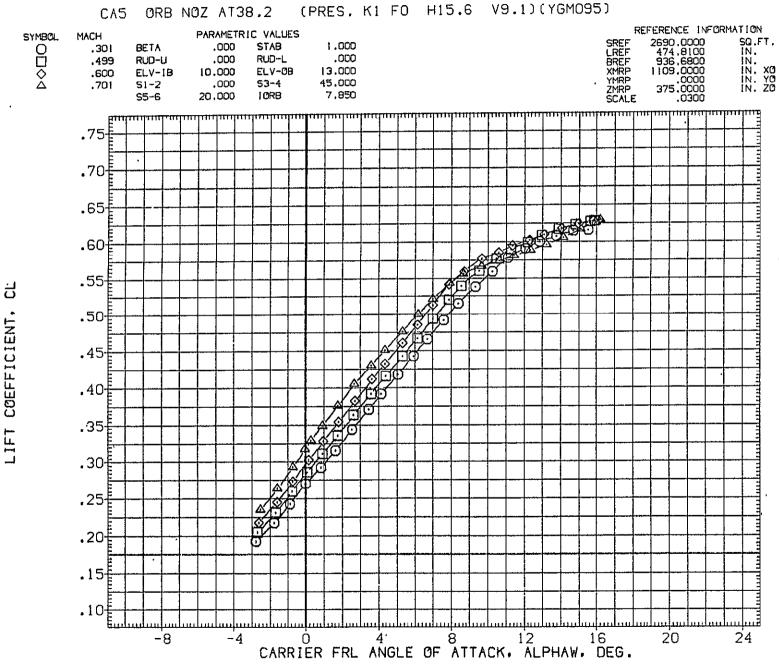


FIG. 61 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP, TAIL CONE OFF, DE=10
PAGE 241

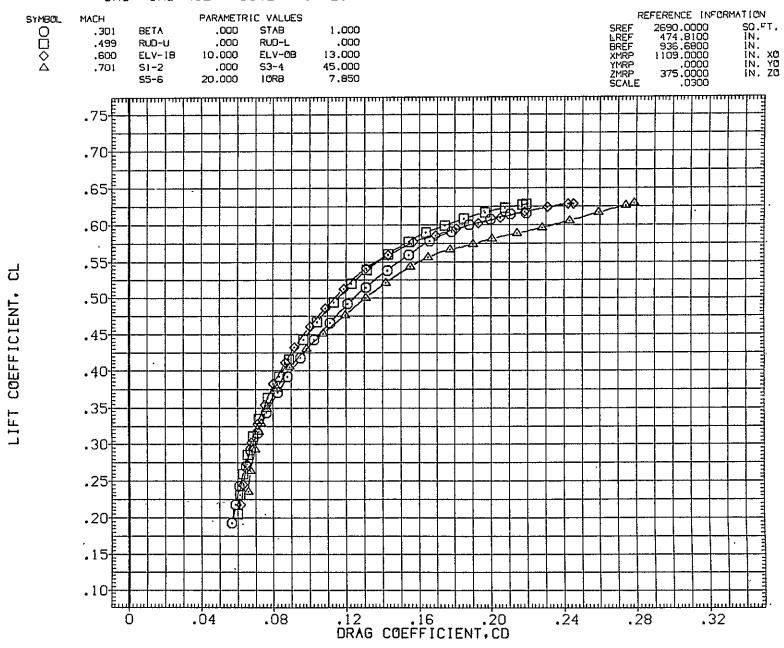


FIG. 61 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP, TAIL CONE OFF, DE=10 ORB

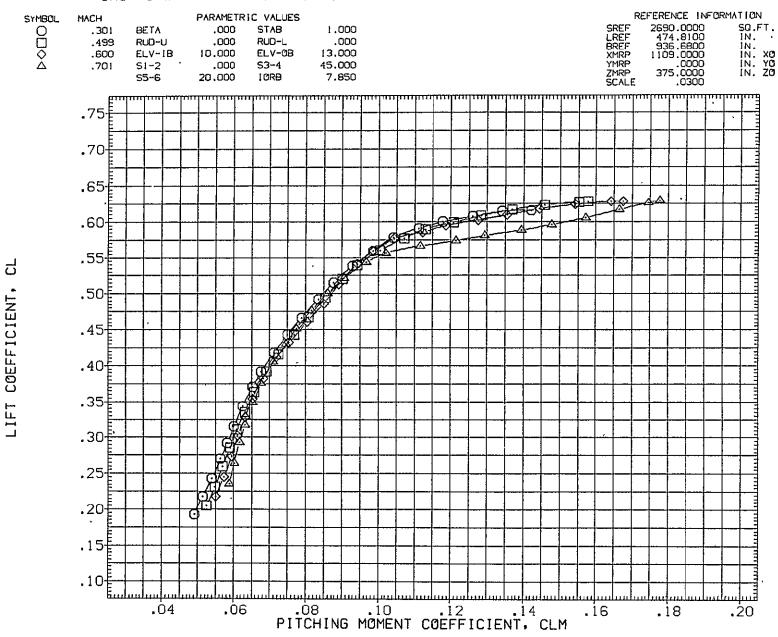


FIG. 61 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP, TAIL CONE OFF, DE=10 ORB

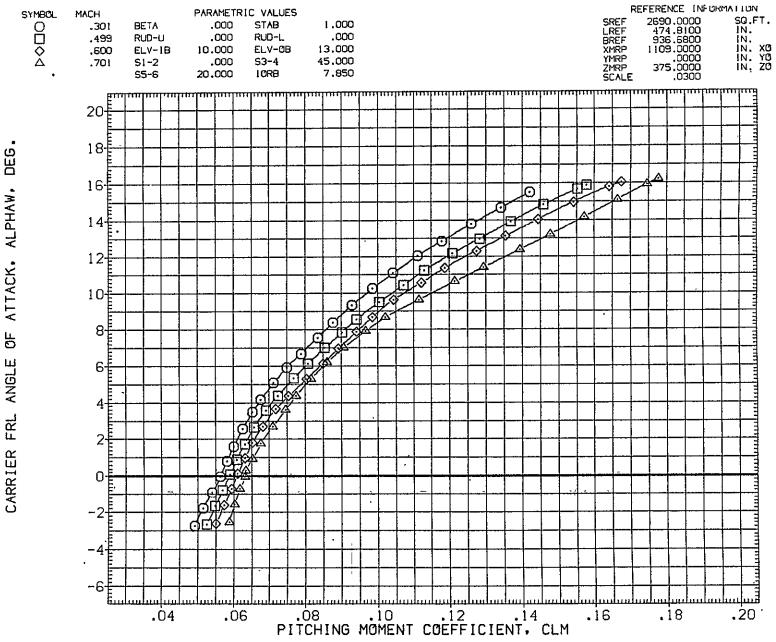
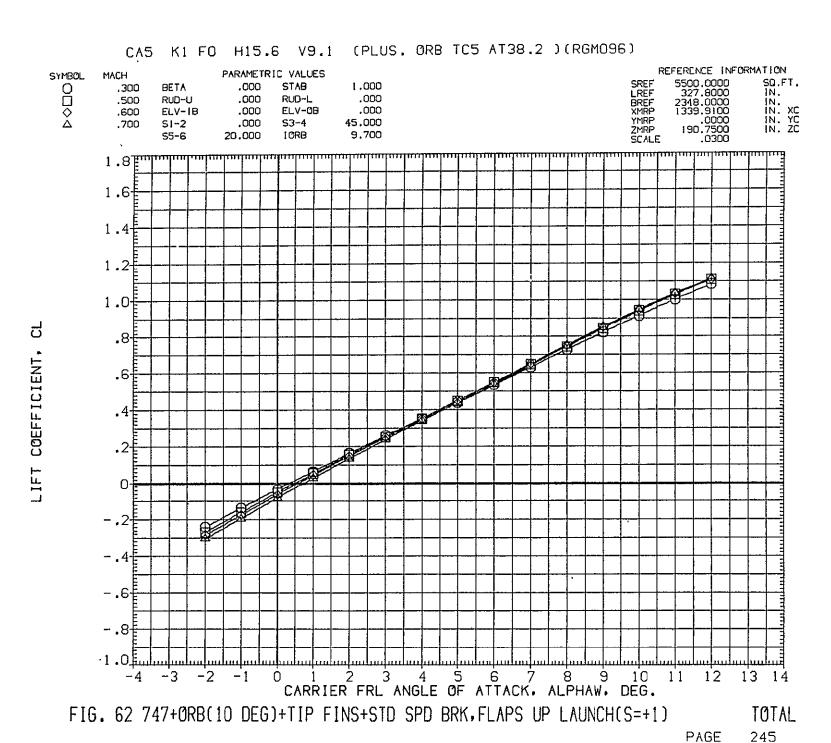


FIG. 61 747+0RB(8 DEG)+TIP FINS+STD SPD BRK,FLAPS UP,TAIL CONE OFF,DE=10 ORB



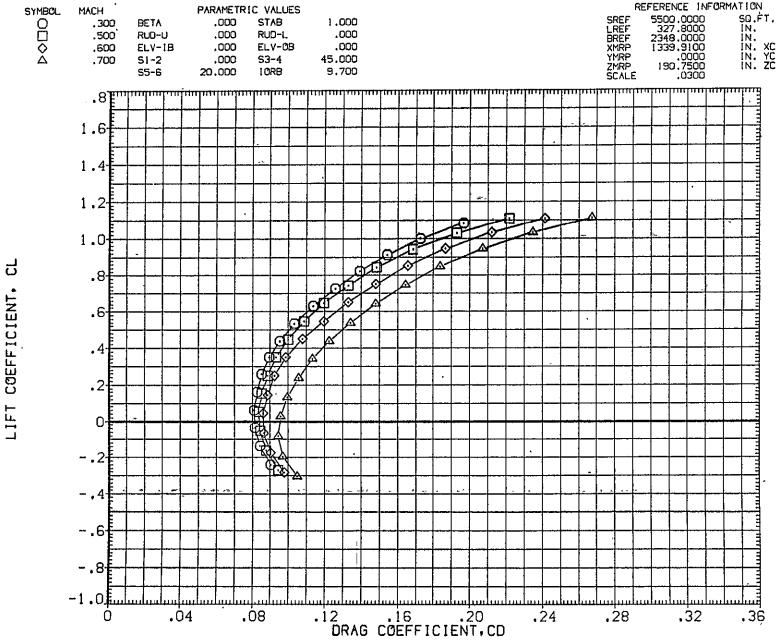


FIG. 62 747+0RB(10 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH(S=+1)

PAGE 246

SYMBOL

COEFFICIENT,

-.8[

-1.0<u>E</u>

FIG. 62 747+0RB(10 DEG)+TIP FINS+STD SPD BRK,FLAPS UP LAUNCH(S=+1)

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.Ò4

-.04

, - .08

TOTAL

.20

.16

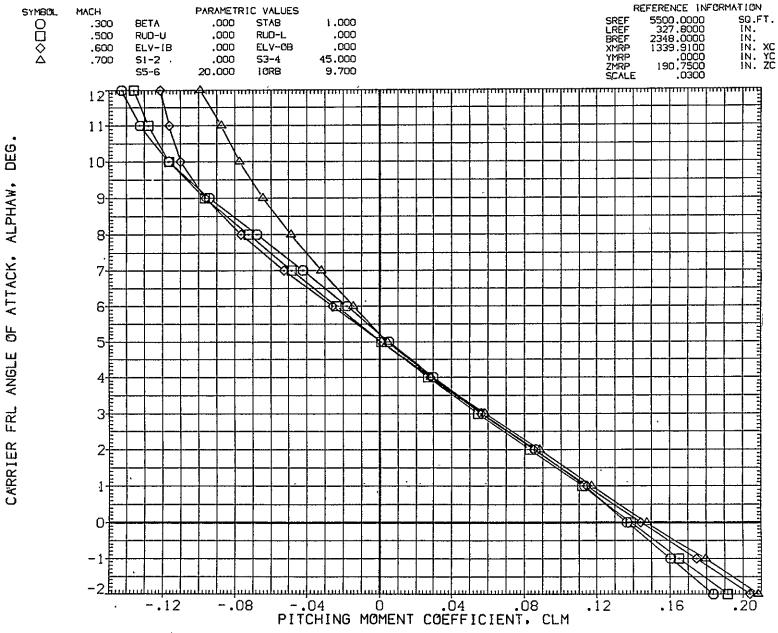


FIG. 62 747+0RB(10 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH(S=+1)

TOTAL



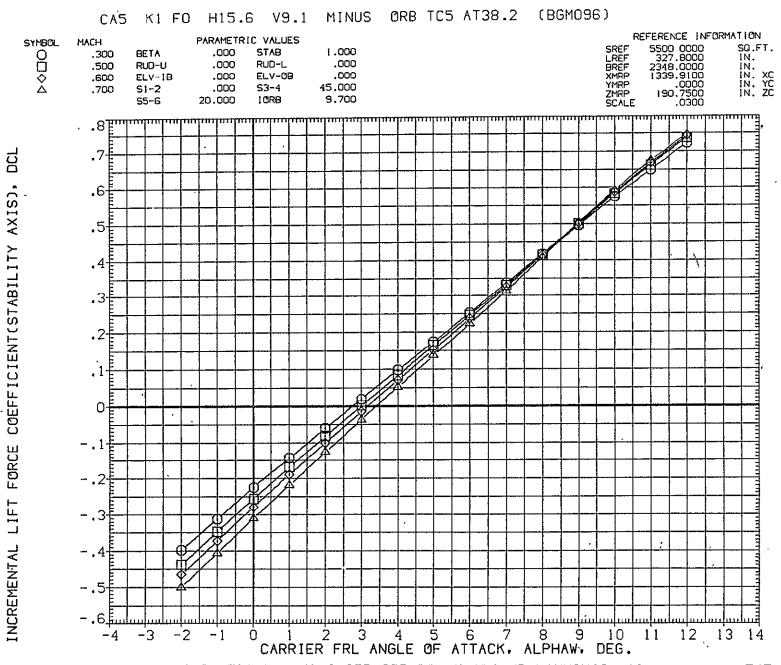
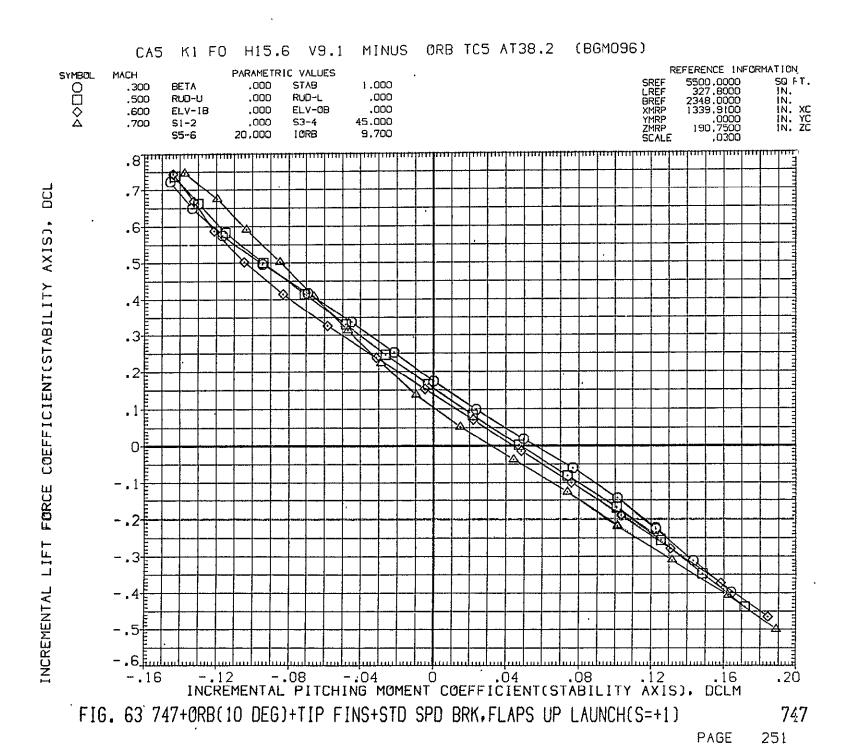


FIG. 63 747+0RB(10 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH(S=+1)

FIG. 63 747+0RB(10 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH(S=+1)

747



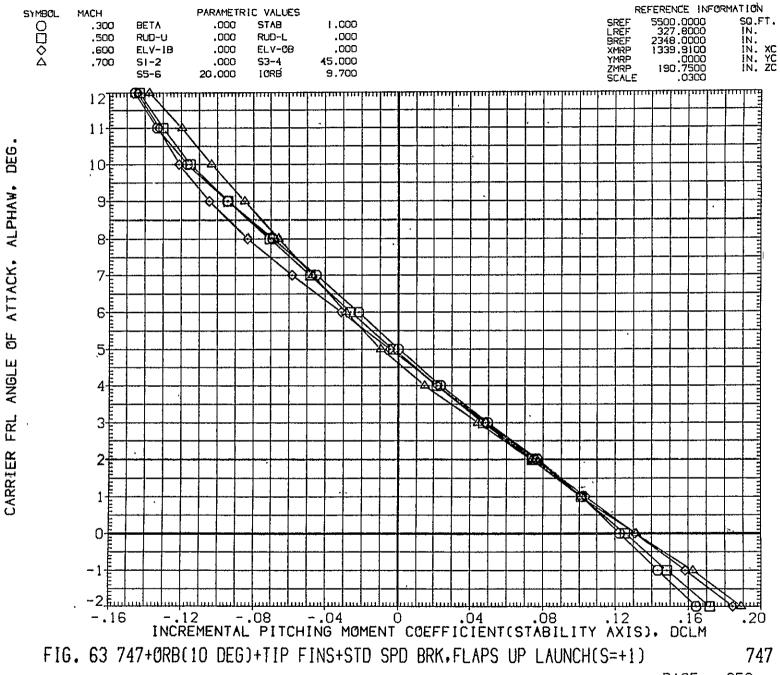


FIG. 64 747+0RB(10 DEG)+TIP FINS+STD SPD BRK.FLAPS UP LAUNCH(S=+1)

CARRIER FRL ANGLE OF ATTACK, ALPHAW, DEG.

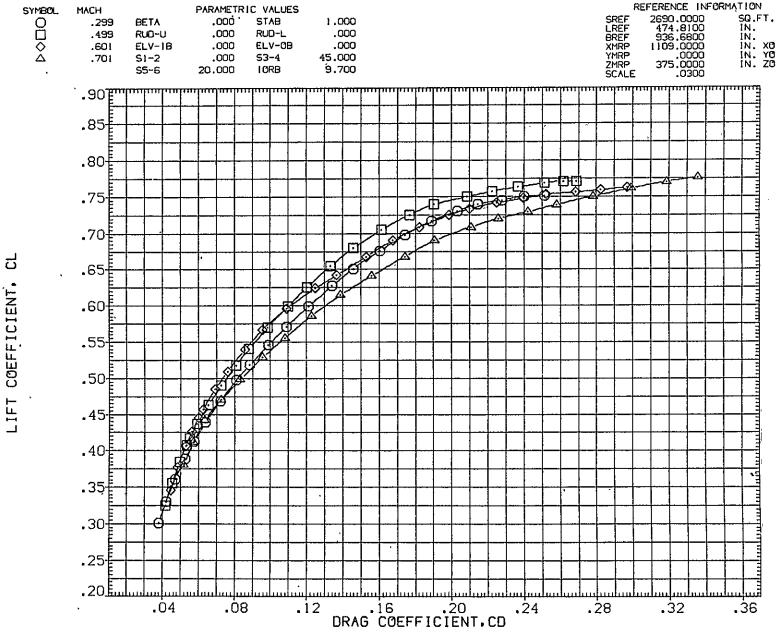


FIG. 64 747+0RB(10 DEG)+TIP FINS+STD SPD BRK,FLAPS UP LAUNCH(S=+1)

(PRES. K1 FO H15.6 V9.1)(YGM096) CA5 ORB TC5 AT38.2 REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH SREF LREF BREF XMRP YMRP ZMRP SCALE SQ.FT. 0000 1.000 .000 STAB ,299 BETA 474.8100 936.6800 1109.0000 .0000 375.0000 IN. .000 RUD-U .000 RUD-L .499 IN. XO IN. YO IN. ZO .000 .000 ELV-08 .601 ELV-IB 45.000 .000 53-4 .701 S1-2 20,000 LORB 9.700 S5-6 .907 .85 .80[.75 .70[<u>م</u> .65‡ COEFFICIENT. .60[.55 .50[.45 .40 .35‡ .30ŧ .25[· 20 E... .O2 .Ò4 .Ó6 06 .08 .10 .12 .14 PITCHING MOMENT COEFFICIENT, CLM .ie .18

FIG. 64 747+0RB(10 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH(S=+1)

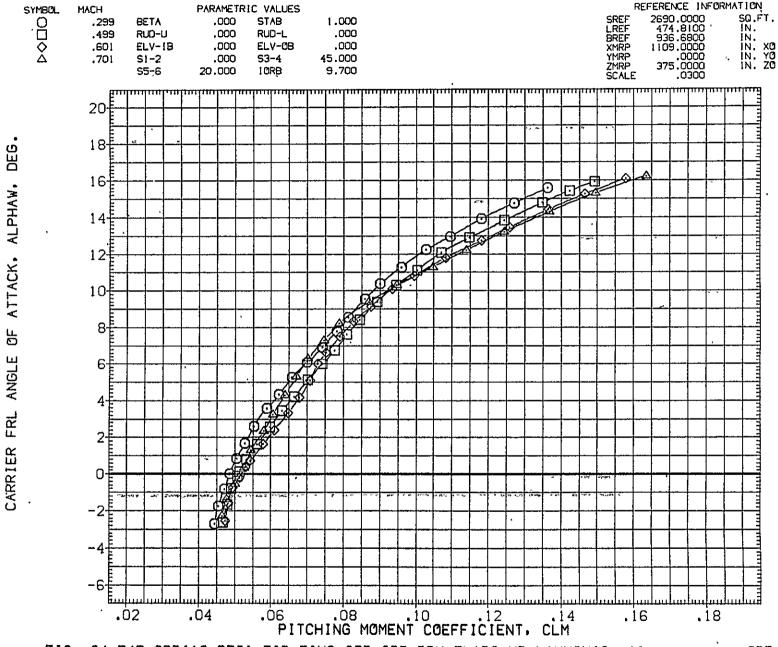
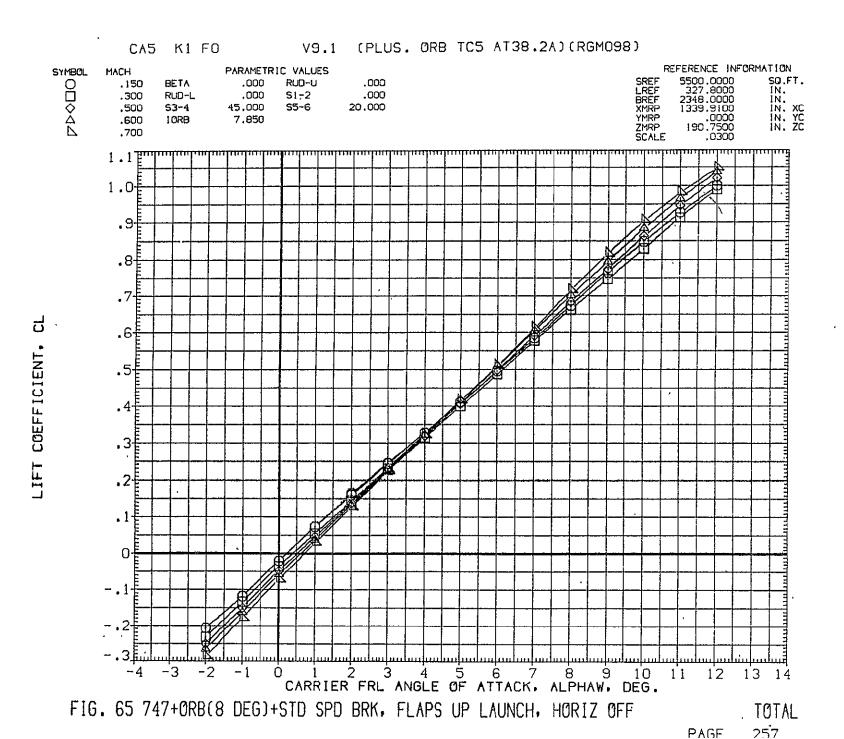


FIG. 64 747+0RB(10 DEG)+TIP FINS+STD SPD BRK.FLAPS UP LAUNCH(S=+1)



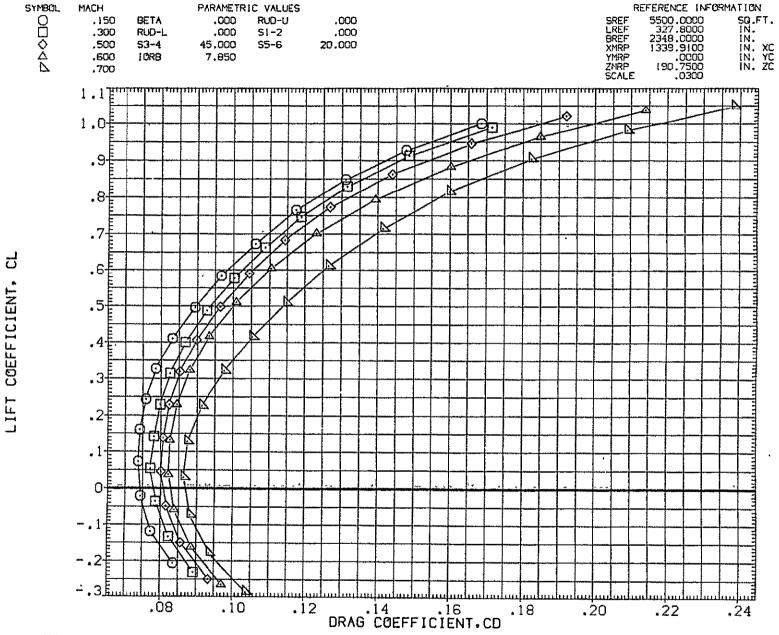


FIG. 65 747+0RB(8 DEG)+STD SPD BRK, FLAPS UP LAUNCH, HORIZ OFF

TOTAL

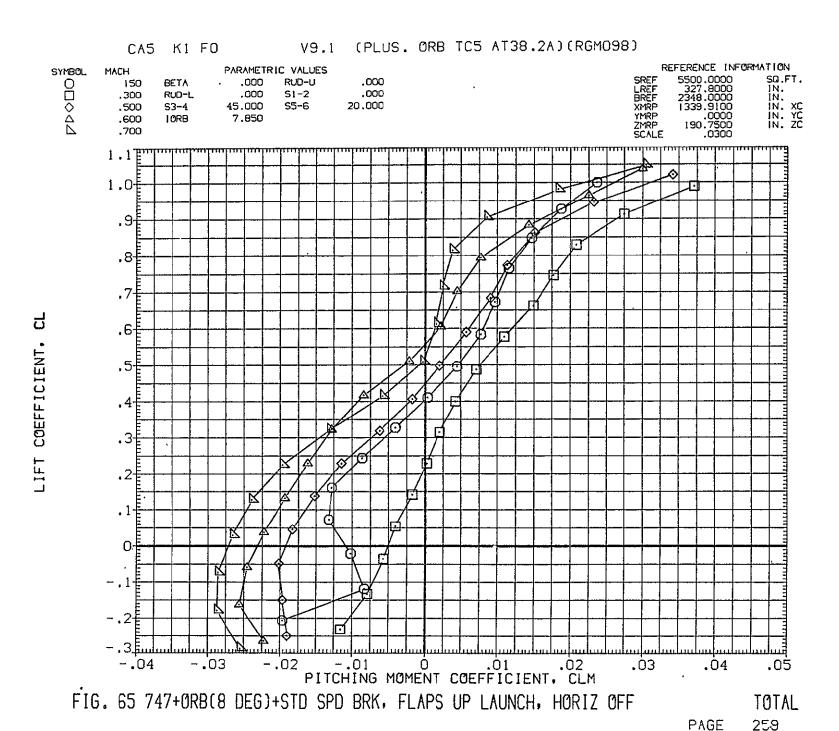
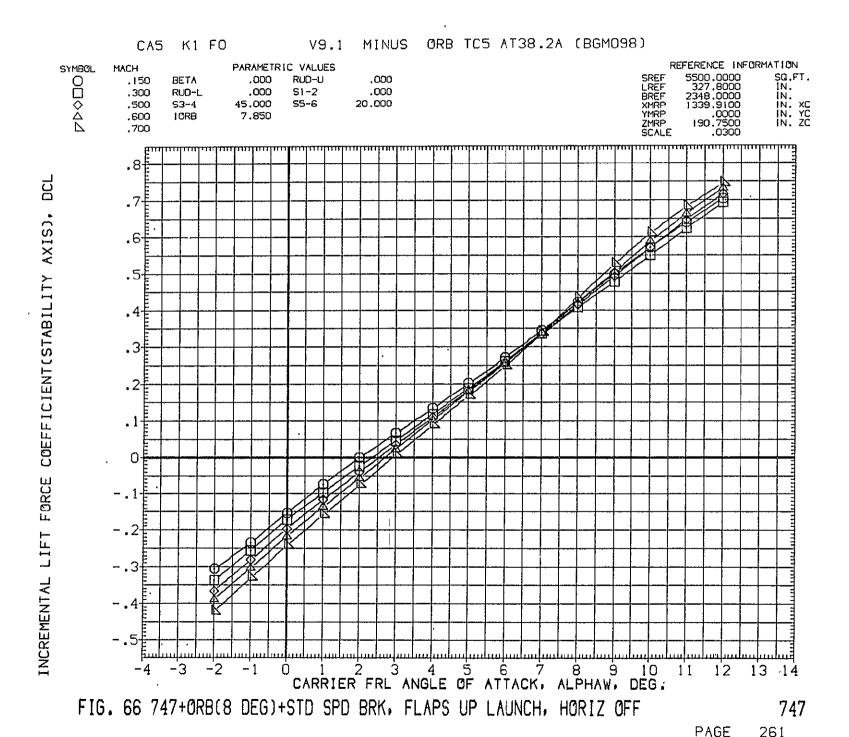


FIG. 65 747+0RB(8 DEG)+STD SPD BRK, FLAPS UP LAUNCH, HORIZ OFF

TOTAL





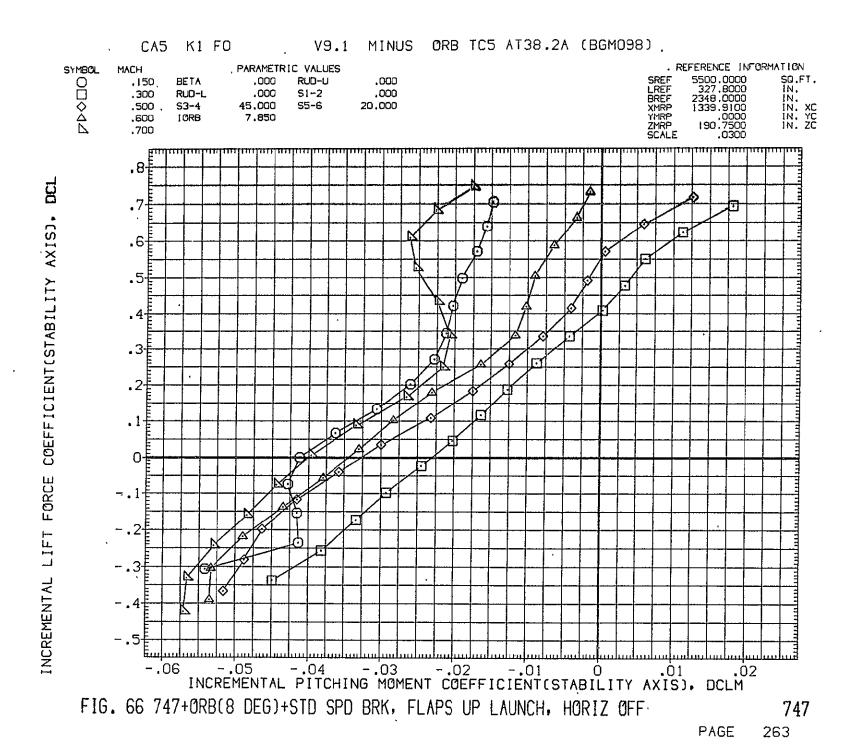
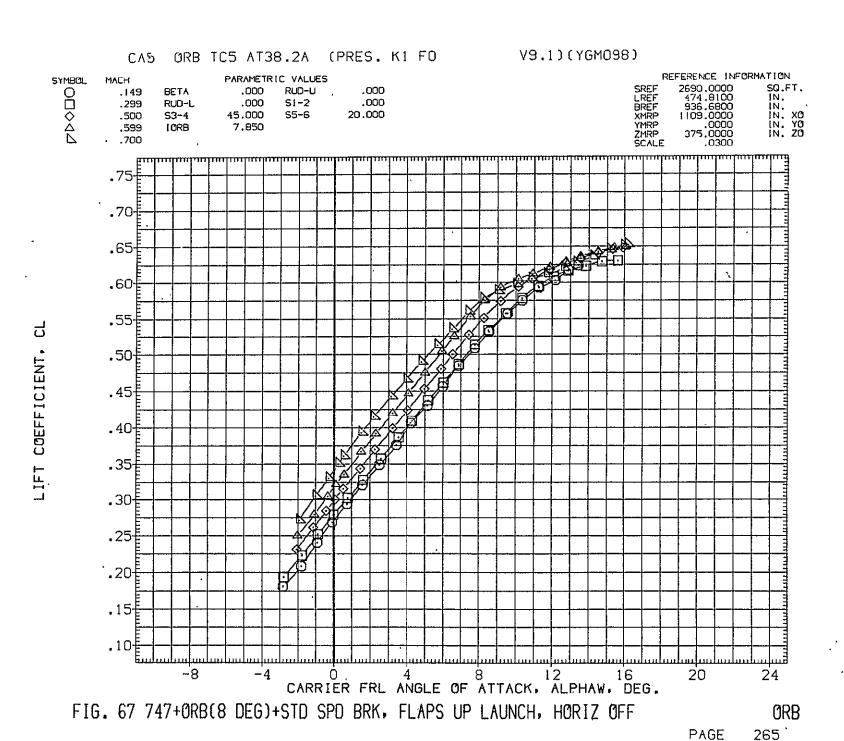


FIG. 66 747+0RB(8 DEG)+STD SPD BRK, FLAPS UP LAUNCH, HORIZ OFF

PAGE 264

.747



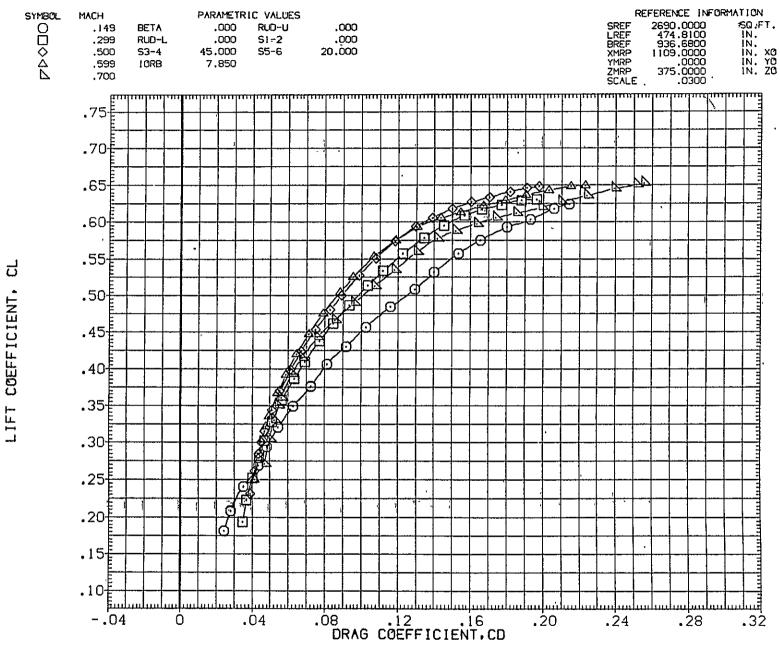


FIG. 67 747+0RB(8 DEG)+STD SPD BRK, FLAPS UP LAUNCH, HORIZ OFF

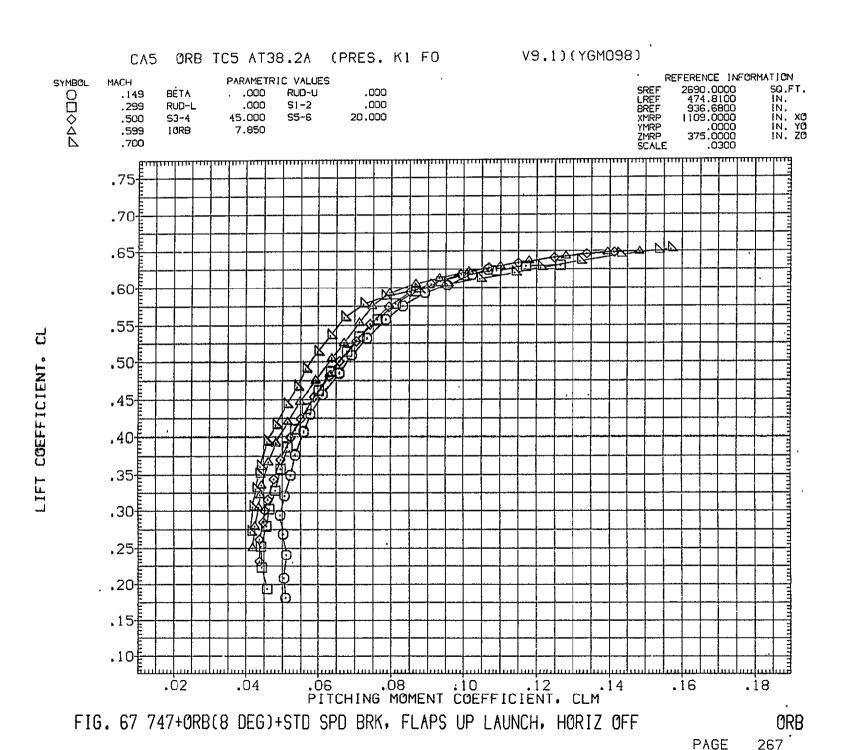


FIG. 67 747+0RB(8 DEG)+STD SPD BRK, FLAPS UP LAUNCH, HORIZ OFF

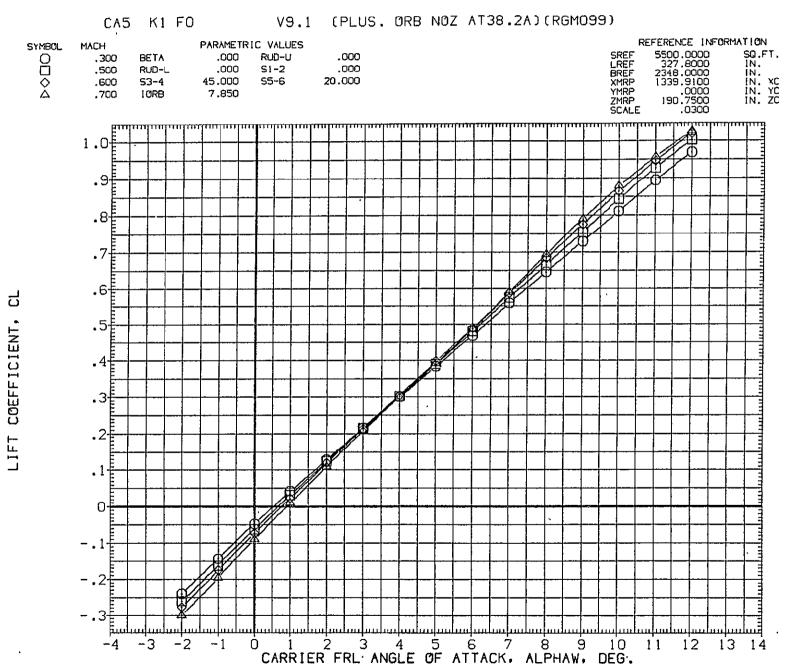


FIG. 68 747+0RB(8 DEG)+STD SPD BRK, FLAPS UP LAUNCH, HOR OFF, TAIL CONE OFF TOTAL
PAGE 269

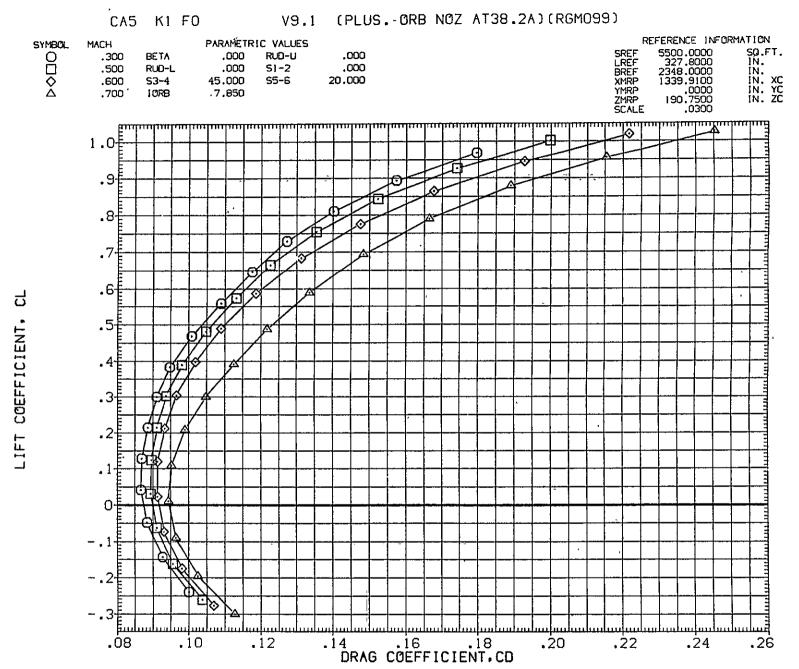


FIG. 68 747+0RB(8 DEG)+STD SPD BRK, FLAPS UP LAUNCH, HOR OFF, TAIL CONE OFF TOTAL

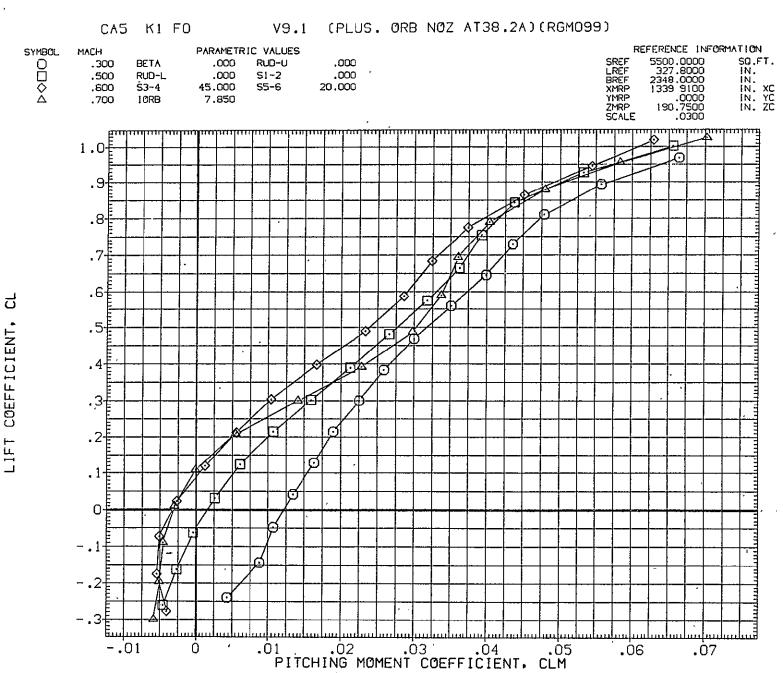


FIG. 68 747+0RB(8 DEG)+STD SPD BRK, FLAPS UP LAUNCH, HOR OFF, TAIL CONE OFF TOTAL
PAGE 271

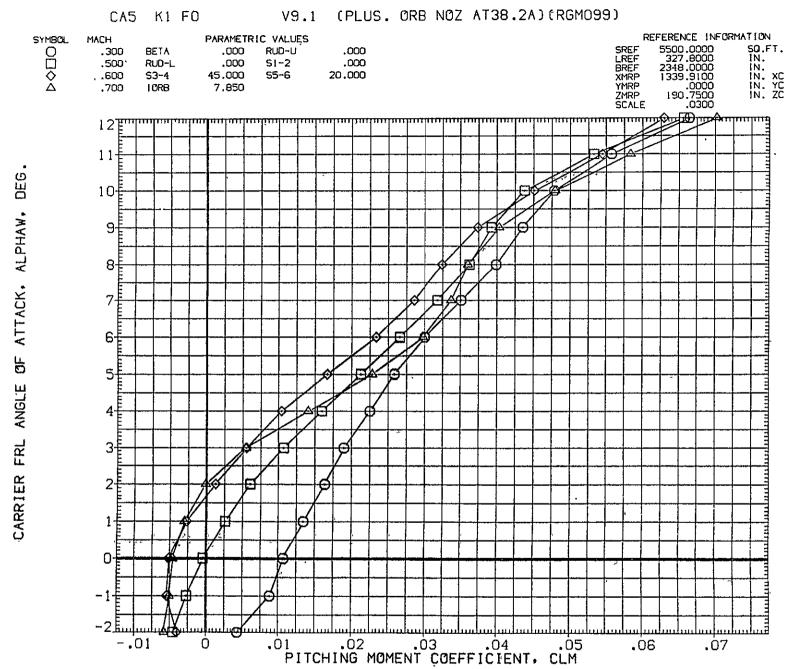


FIG. 68 747+0RB(8 DEG)+STD SPD BRK, FLAPS UP LAUNCH, HOR OFF, TAIL CONE OFF TOTAL PAGE 272

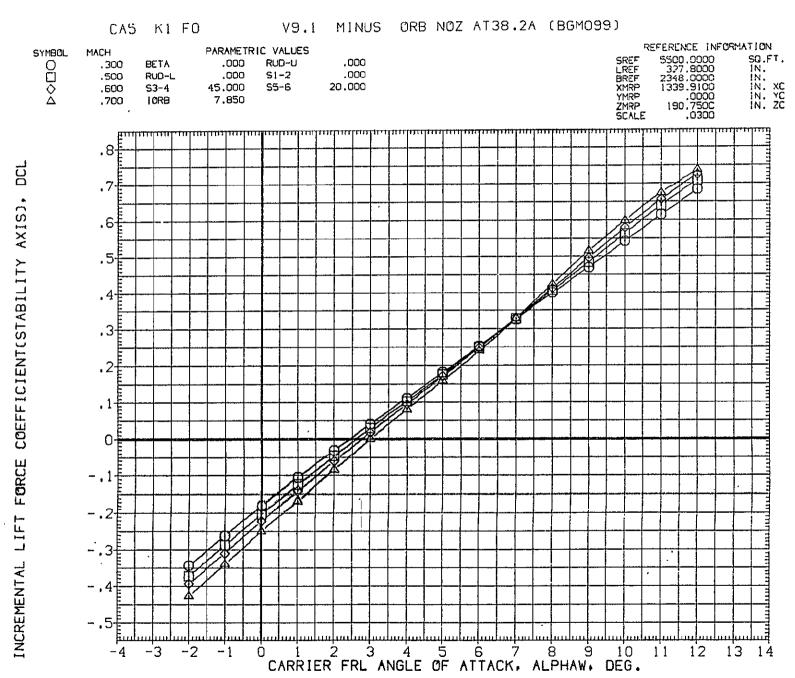
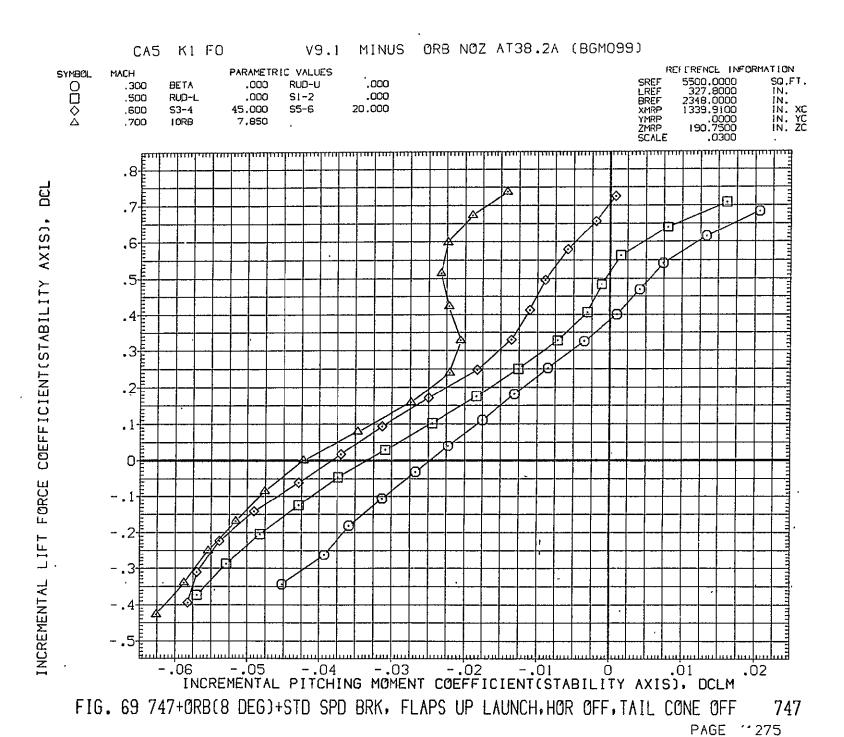
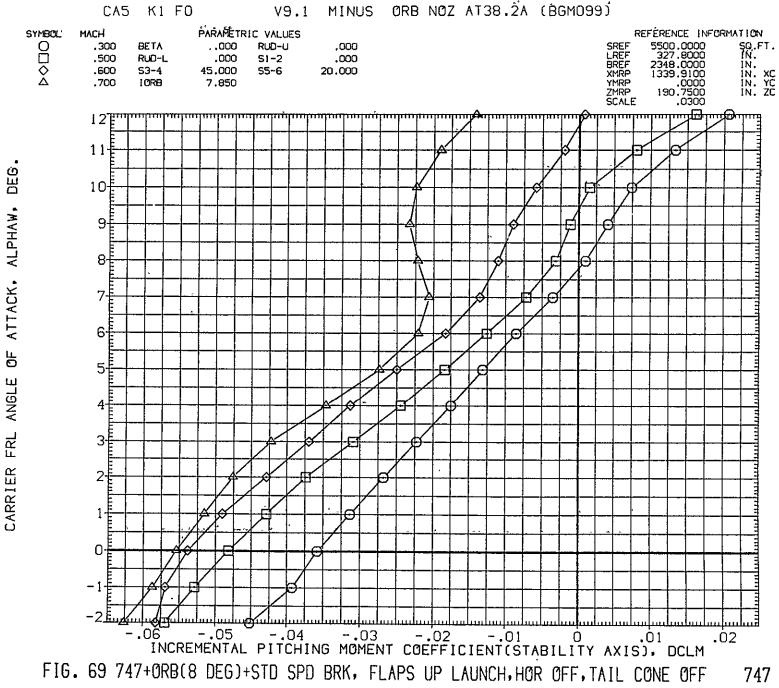


FIG. 69 747+0RB(8 DEG)+STD SPD BRK, FLAPS UP LAUNCH, HOR OFF, TAIL CONE OFF
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FIG. 70 747+ORB(8 DEG)+STD SPD BRK, FLAPS UP LAUNCH, HOR OFF, TAIL CONE OFF
PAGE 278

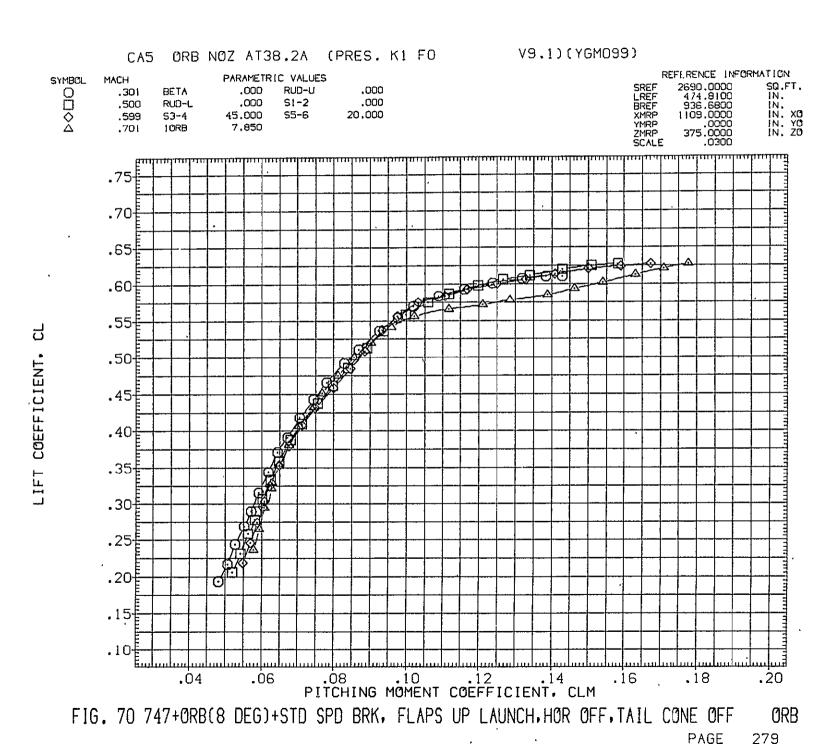


FIG. 70 747+0RB(8 DEG)+STD SPD BRK, FLAPS UP LAUNCH, HOR OFF, TAIL CONE OFF
PAGE 280

PITCHING MOMENT COEFFICIENT, CLM

.12

.16

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.20

.10

.04

.06

.08

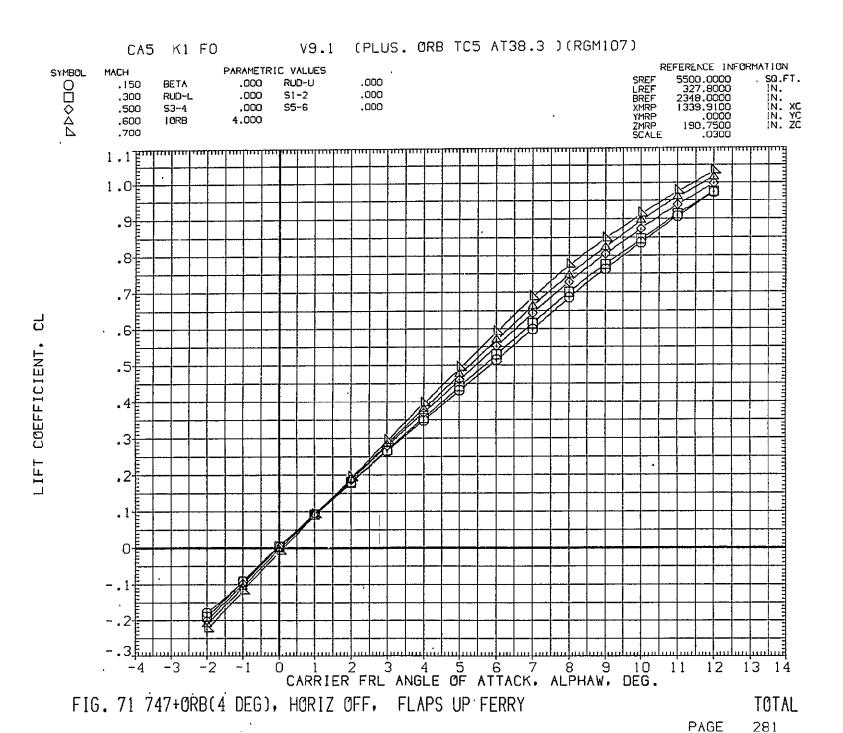


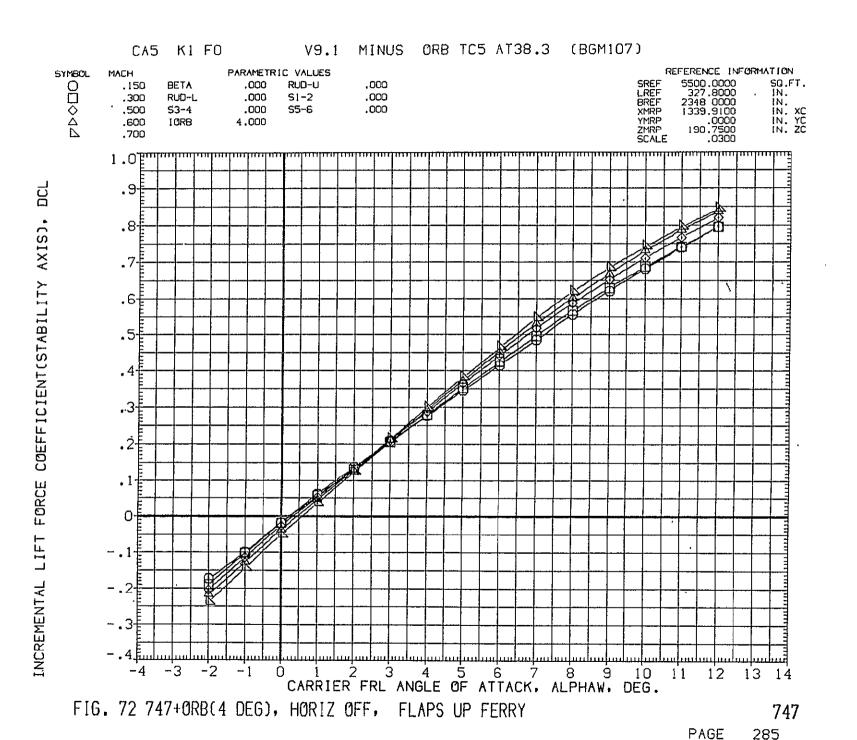
FIG. 71 747+0RB(4 DEG), HORIZ OFF, FLAPS UP FERRY

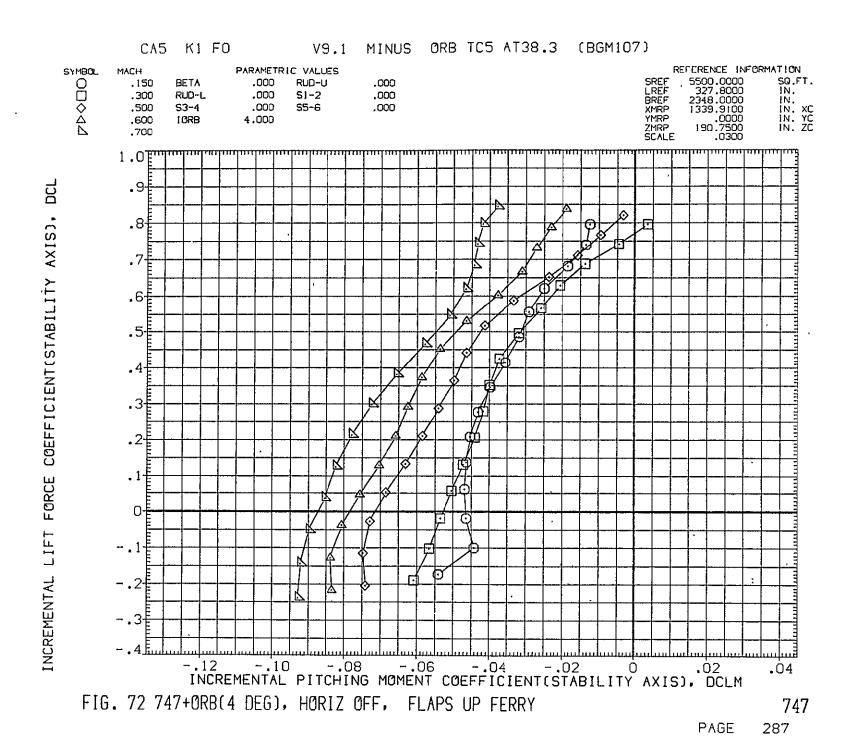
TOTAL

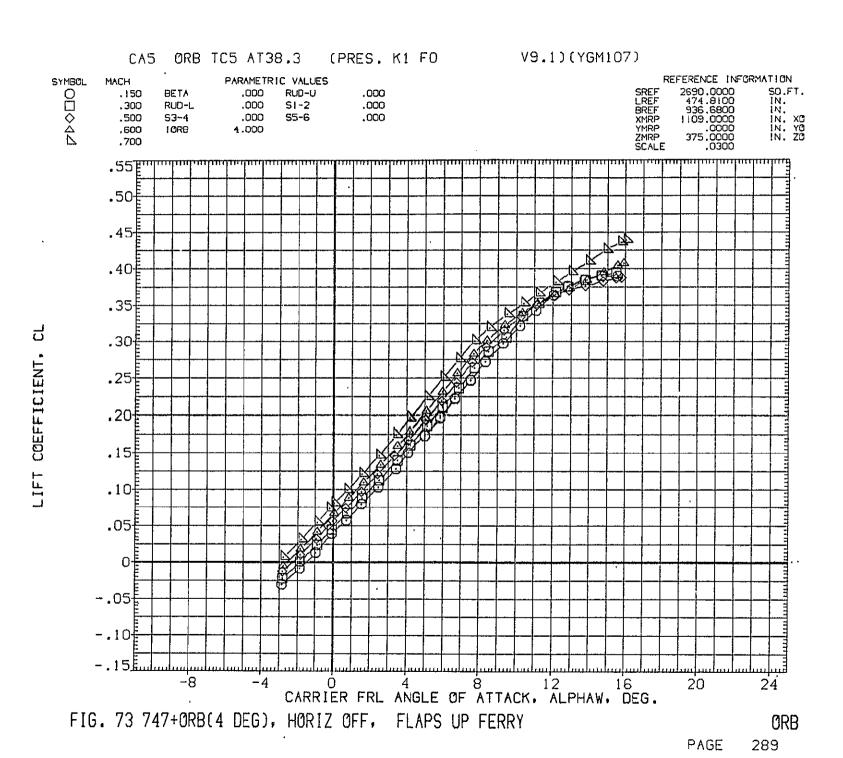
283

FIG. 71 747+ORB(4 DEG), HORIZ OFF, FLAPS UP FERRY

TOTAL







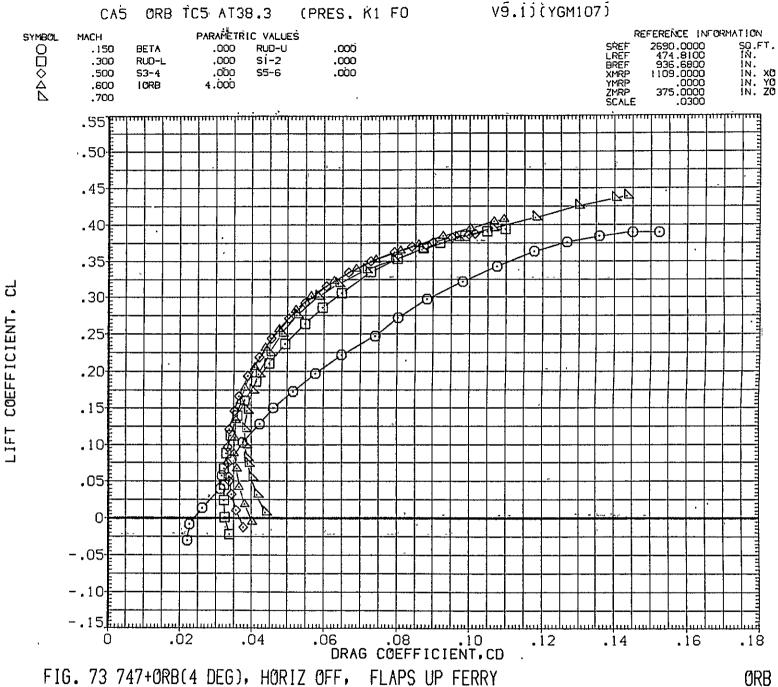
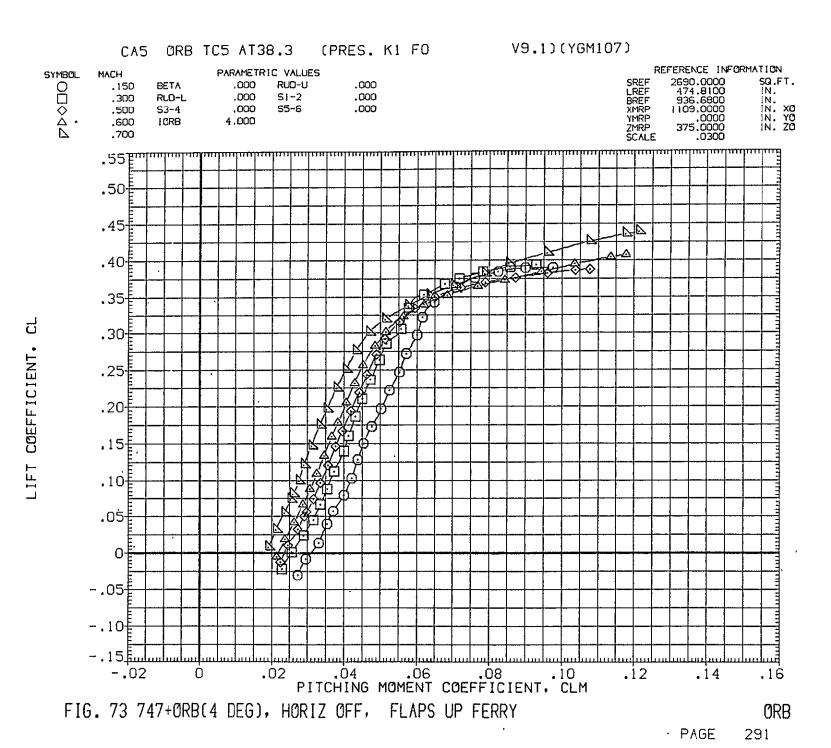
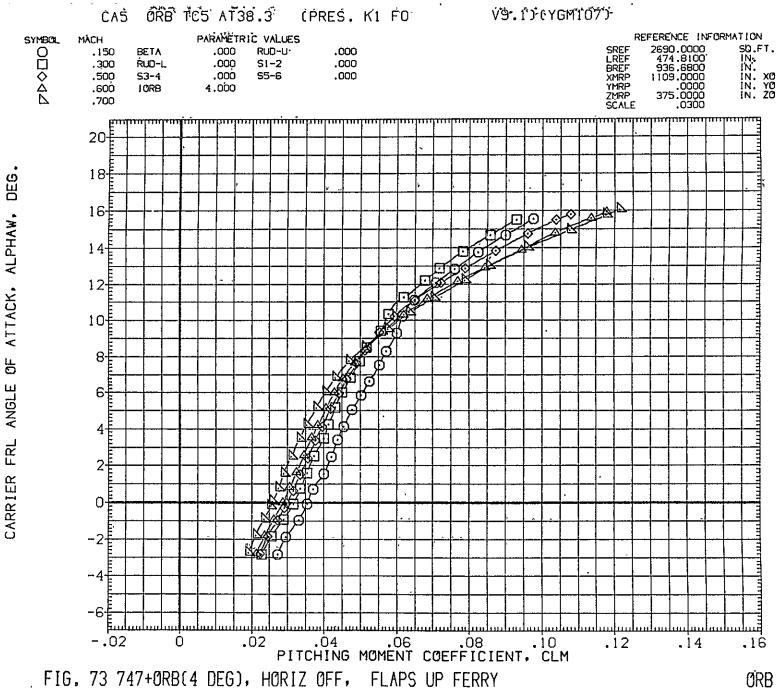


FIG. 73 747+ORB(4 DEG), HORIZ OFF, FLAPS UP FERRY







CA5 K1 FO H15.1A V9.1 (PLUS. ORB TC5 AT38.3)(RGM108)

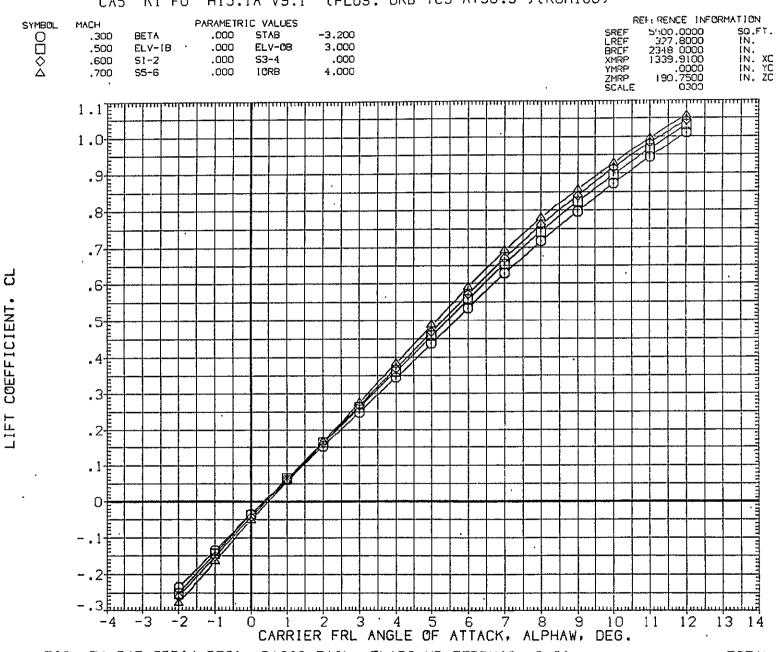


FIG. 74 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY(S=-3.2)

CA5 KI FO HI5.1A V9.1 (PLUS. ORB TC5 AT38.3)(RGM108)

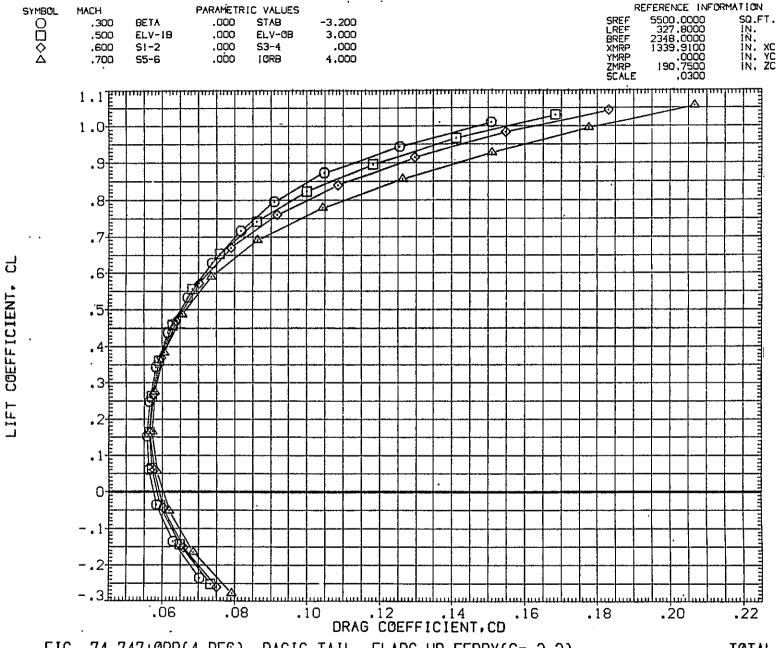


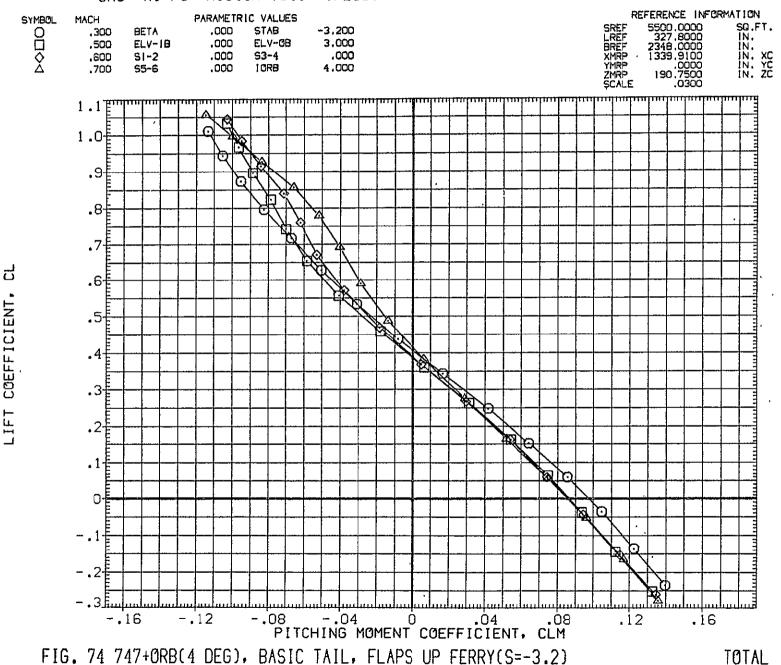
FIG. 74 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY(S=-3.2)

TOTAL



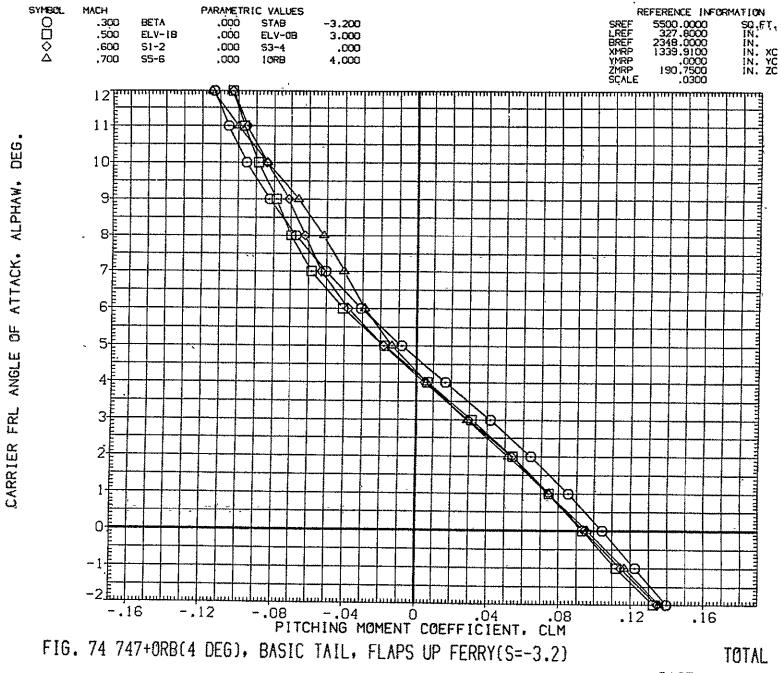
المستقدمة المستقدمة والمستقدمة والمستقدمة والمستقدمة والمستقدمة والمستقدمة والمستقدمة والمستقدمة والمستقدمة و والمستقدمة والمستقدم والمستقدمة والمستقدمة والمستقدمة والمستقدمة والمستقدمة والمستقدم وال

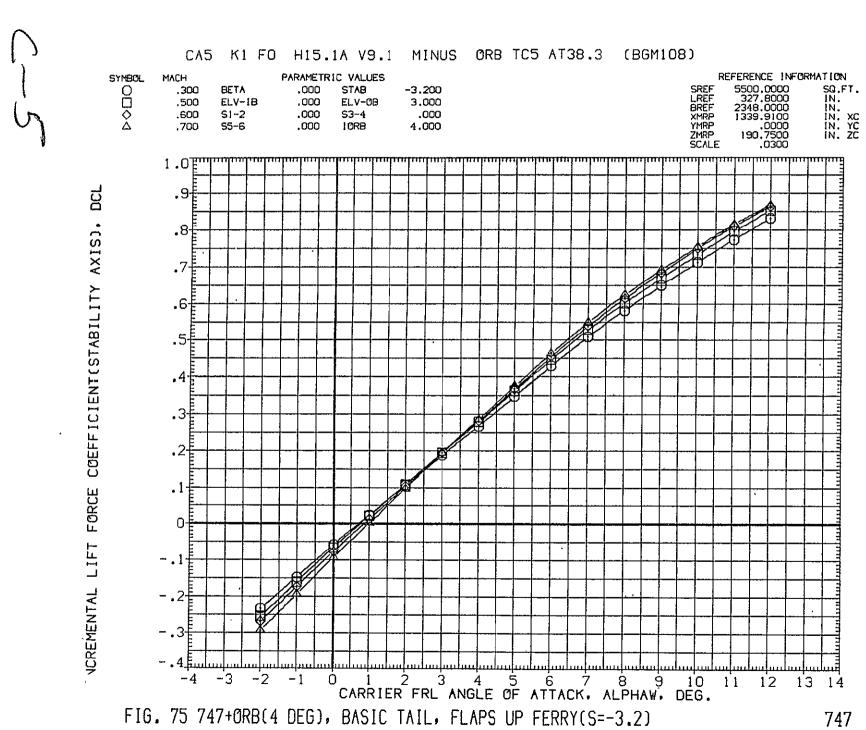
CA5 K1 FO H15.1A V9.1 (PLUS. ORB TC5 AT38.3)(RGM108)

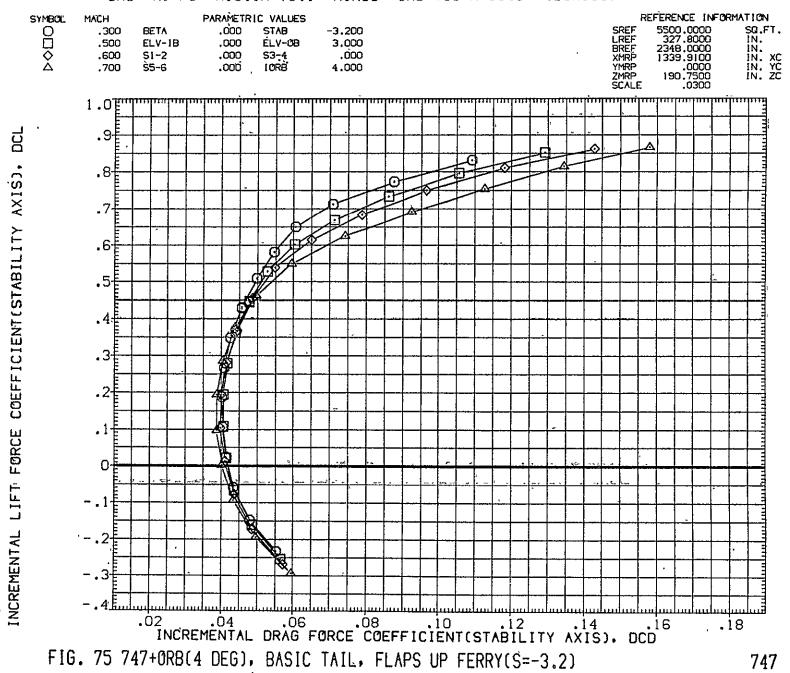


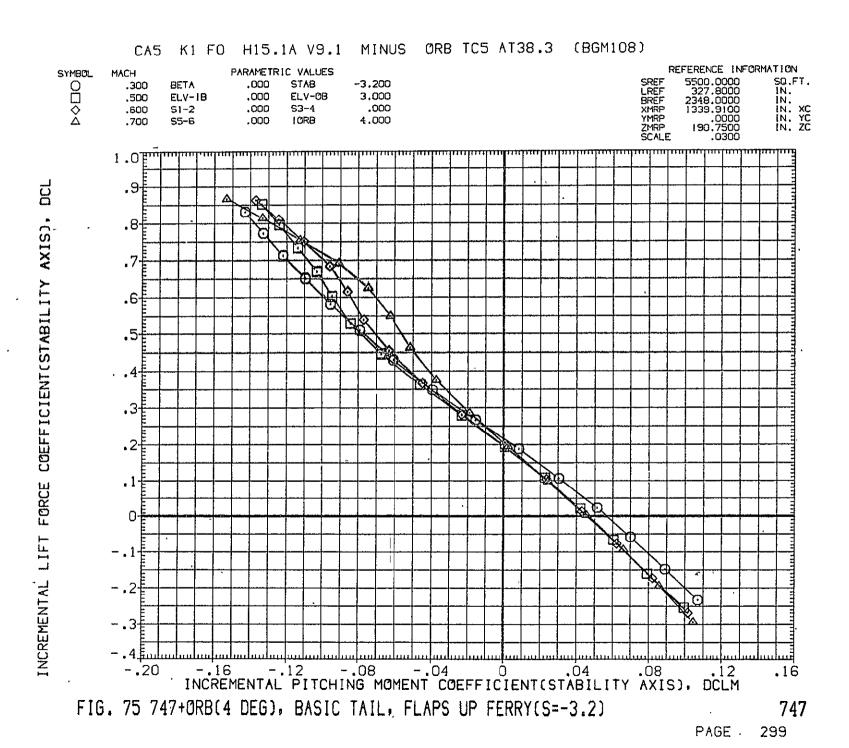
295,

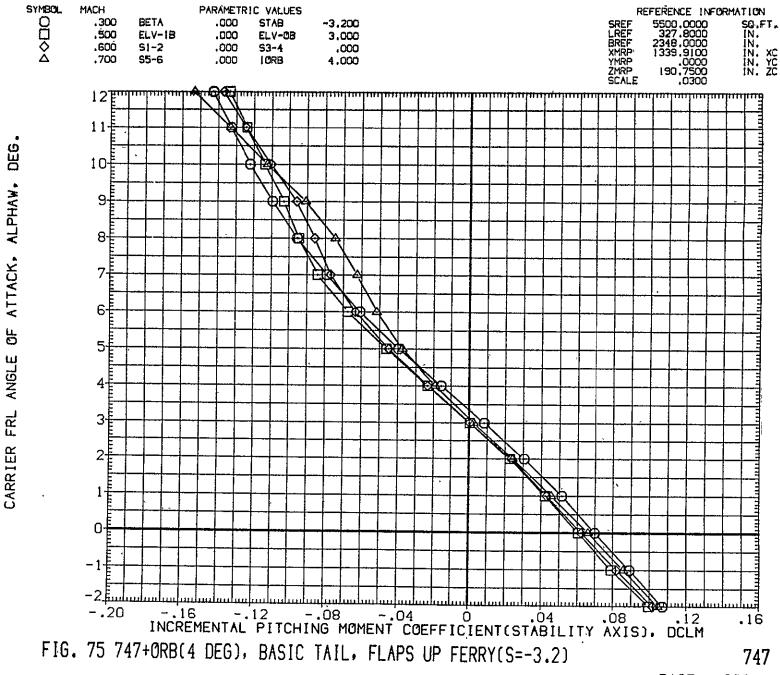
CA5 K1 FO H15.1A V9.1 (PLUS. ORB TC5 AT38.3)(RGM108)











CA5 ORB TC5 AT38.3 (PRES. K1 FO H15.1A V9.1)(YGM108)

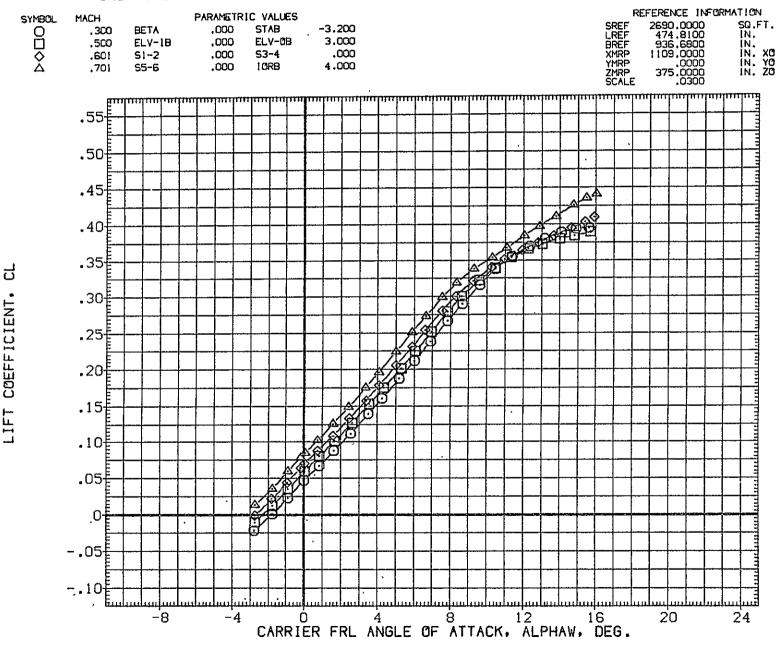
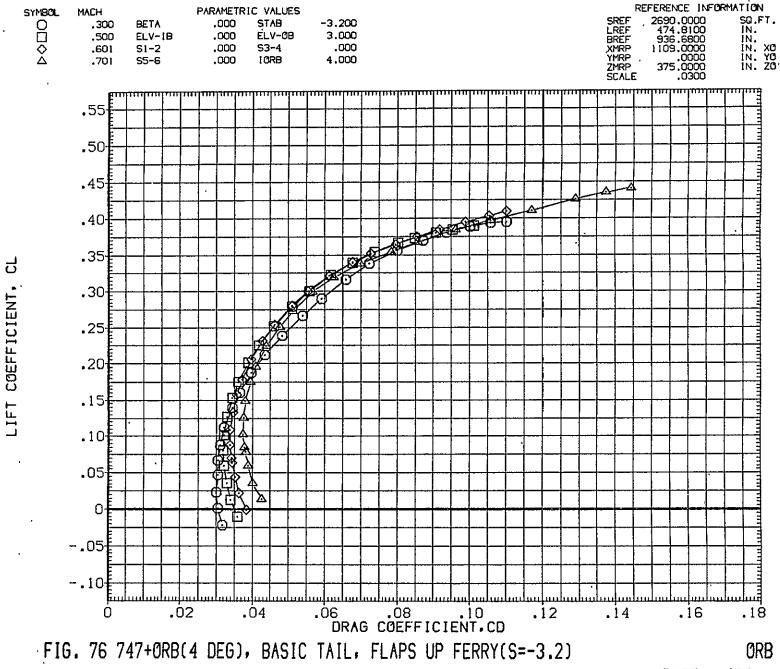


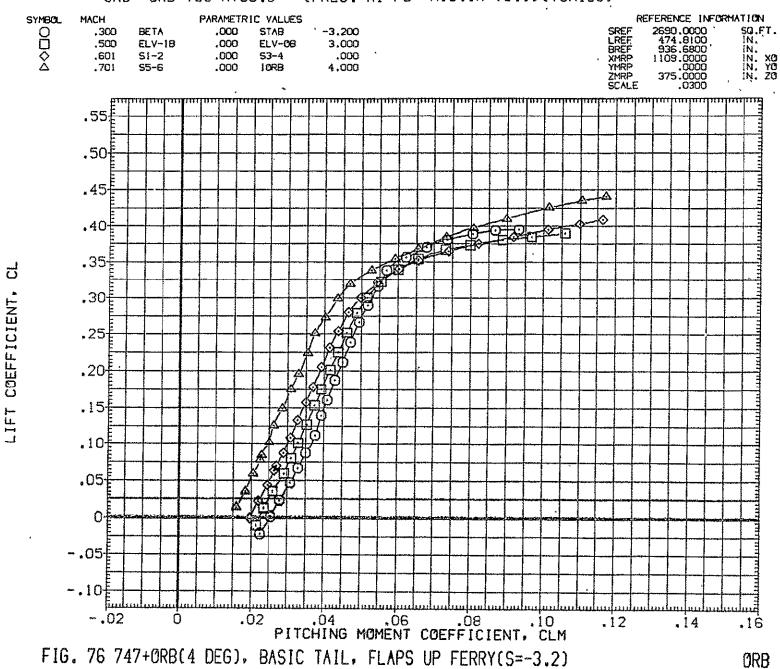
FIG. 76 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY(S=-3.2)

ORB

CA5 ORB TC5 AT38.3 (PRES. K1 FO H15.1A V9.1)(YGM108)



CA5 ORB TC5 AT38.3 (PRES. K1 FO H15.1A V9.1)(YGM108)



303

ORB TC5 AT38.3 (PRES. K1 FO H15.1A V9.1)(YGM108) CA5 REFERENCE INFORMATION PARAMETRIC VALUES " SYMBOL MACH 2690.0000 474.8100 936.6800 1109.0000 .0000 375.0000 .0300 SQ.FT. IN. IN. IN. XO IN. YO IN. ZO SREF LREF BREF XMRP YMRP ZMRP SCALE 0000 -3.200 .300 BETA .000 STAB 3.000 .000 ELV-0B .500 ELV-IB .000 51-2 .000 53-4 .601 4.000 LORB .701 S5-6 .000 18- 16‡ ALPHAW. ATTACK. 10- 8 면 ANGLE 4 CARRIER FRL -6+ .02 PITCHING MOMENT COEFFICIENT, CLM ,i2 . 14 .16 Ó FIG. 76 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY(S=-3.2) ORB

K1 F0 H15.6 'V9.1 (PLUS. ORB TC5 AT38.3)(RGM112) CA5 REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL. MACH SQ.FT. IN. IN. XC IN. YC IN. ZC 5500.0000 327.8000 2348.0000 STAB -3,200 .000 .150 · BETA 0000 .000 RUD-L BREF XMRP YMRP ZMRP SCALE RUD-U .000 .300 1339,9100 .0000 190,7500 .0300 ELV-0B 3.000 ELV-1B .000 .500 .000 .000 S3-4 .600 S1-2 LORB 4.000 .700 S5~6 .000 1.1 Em 1.0 COEFFICIENT, 1 2 3 4 5 6 7 8 9 10 CARRIER FRL ANGLE OF ATTACK, ALPHAW, DEG. 12 13 14 11

TOTAL 305

PAGE

FIG. 77 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY(S=-3.2)

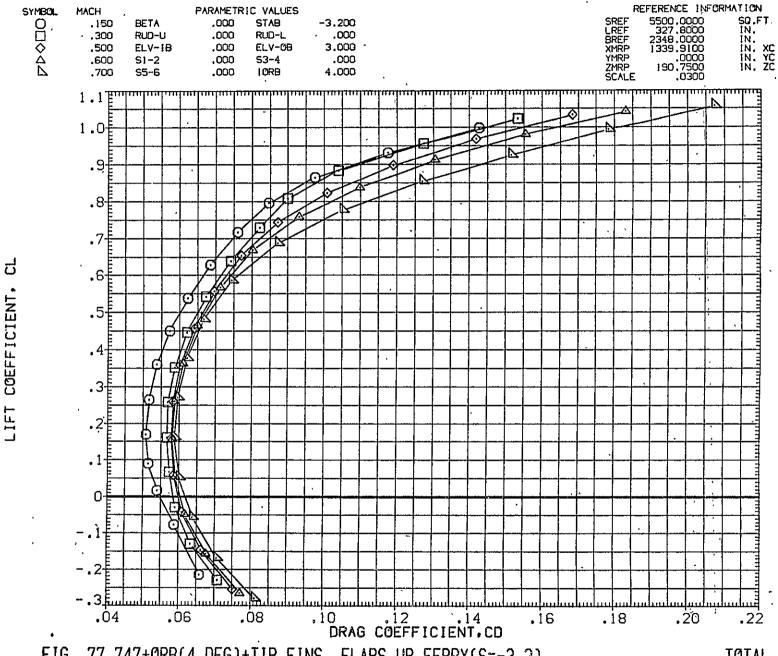
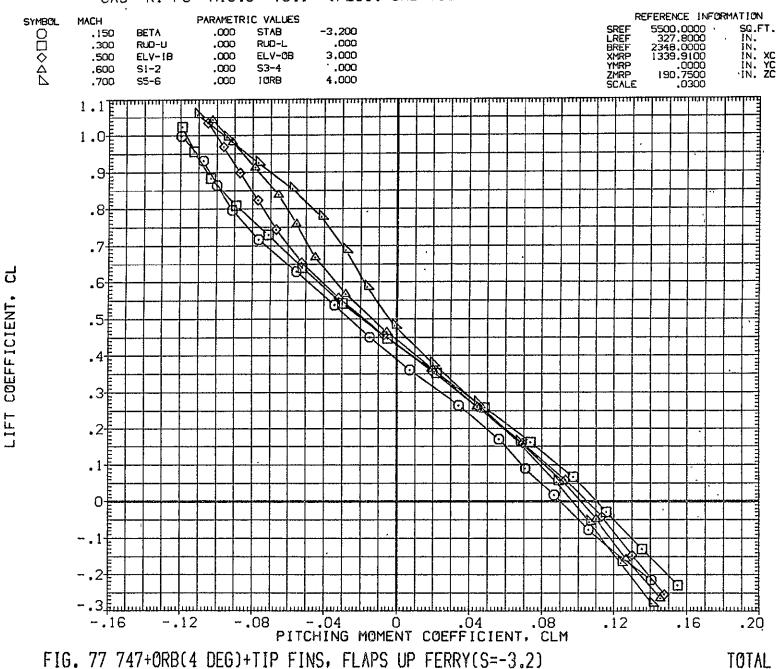


FIG. 77 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY(S=-3.2)

CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.3)(RGM112)



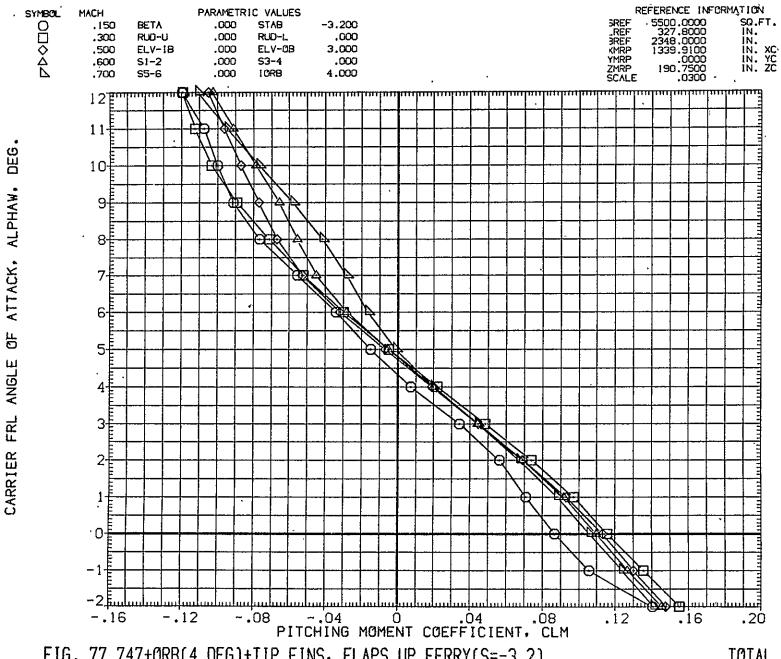


FIG. 77 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY(S=-3.2)

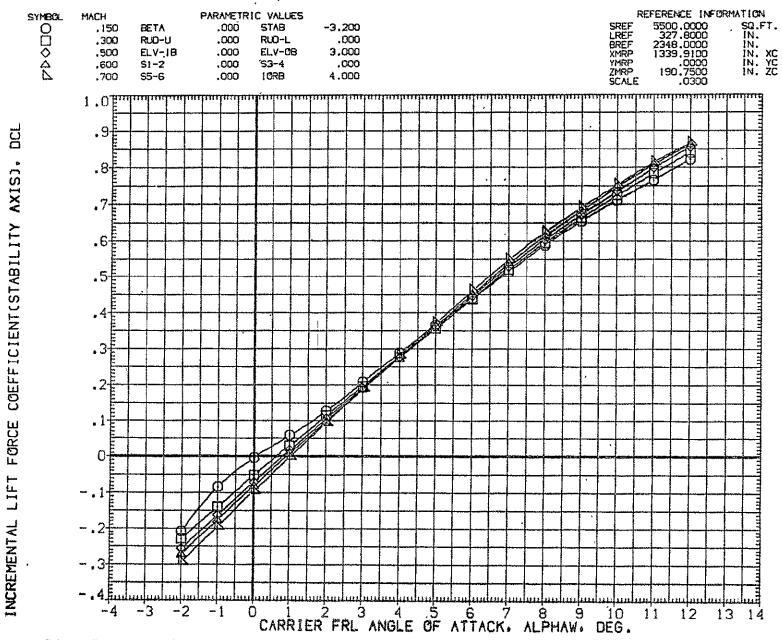
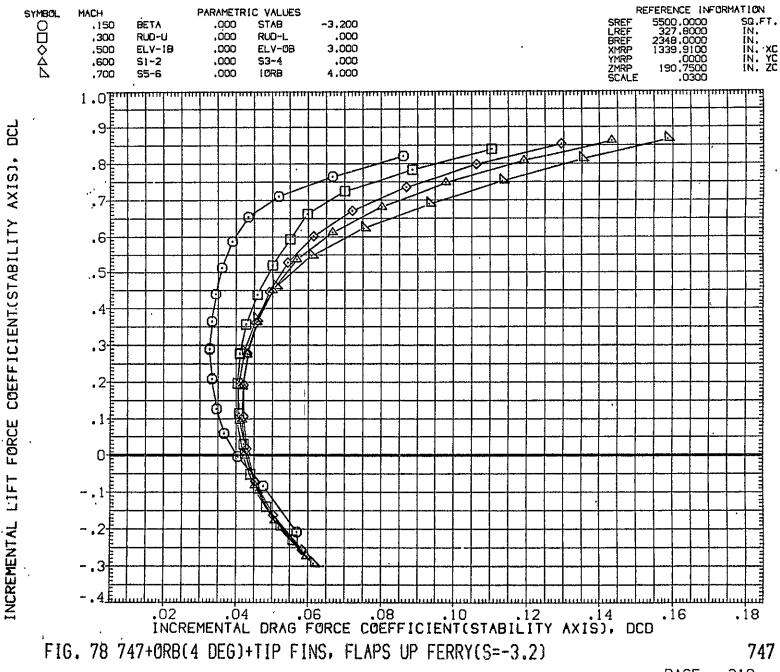
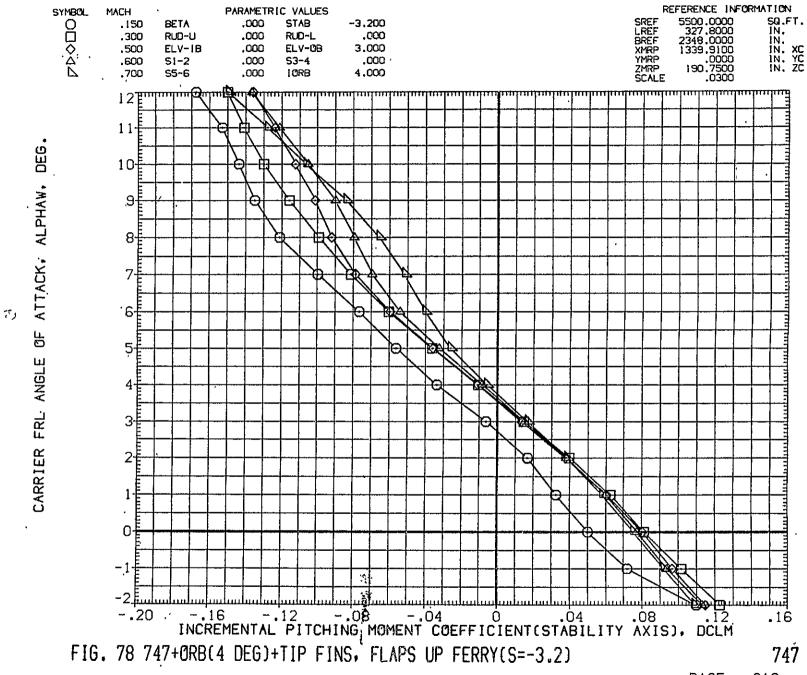


FIG. 78 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY(S=-3.2)

CA5 K1 FO H15.6 V9.1 MINUS ORB TC5 AT38.3 (BGM112)



MINUS ORB TC5 AT38.3 CA5 K1 F0 H15.6 V9.1 (BGM112) REFERENCE INFORMATION SYMBOL MACH PARAMETRIC VALUES SREF LREF BREF XMRP YMRP ZMRP SCALE 5500.0000 327.8000 2348.0000 1339.9100 .0000 190.7500 .0300 SO.FT. IN. IN. IN. XC IN. YC IN. ZC 000044 ~3.200 BETA .000 STAB .150 . RUD-L .000 .300 RUD-U .000 ELV-1B .000 ELV-08 3.000 .000 S1-2 .000 53-4 .600 4.000 .700 S5-6 .000 10RB 1.0 [임 .8 AXIS) COEFFICIENT(STABILITY .6[\mathfrak{O} FORCE LIFT INCREMENTAL -.16 -.12 -.08 -.04 0 .04 .08 .12 INCREMENTAL PITCHING MOMENT COEFFICIENT(STABILITY AXIS). DCLM -.20 .16 FIG. 78 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY(S=-3.2) 747



ORB TC5 AT38.3 (PRES. K1 FO H15.6 V9.1)(YGM112) CA5 REFERENCE INFORMATION PARAMETRIC VALUES MACH SYMBOL. SQ.FT. IN. IN. XO IN. YO IN. ZO SREF LREF BREF XMRP YMRP ZMRP SCALE 2690.0000 474.8100 936.6800 STAB -3.200 0000 .150 BETA .000 RUD-L .000 .000 .300 RUD-U 3,000 ELV-IB .000 ELV-0B .499 .000 51-2 .000 53-4 4.000 S5-6 .000 **LORB** .700 .50[.45 .40 .35€ .30# .25 .20 .15 .10 .05[-.05[-.10[-.15 E... -8 12 20 24

CARRIER FRL ANGLE OF ATTACK, ALPHAW, DEG.

FIG. 79 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY(S=-3.2)

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COEFFICIENT,

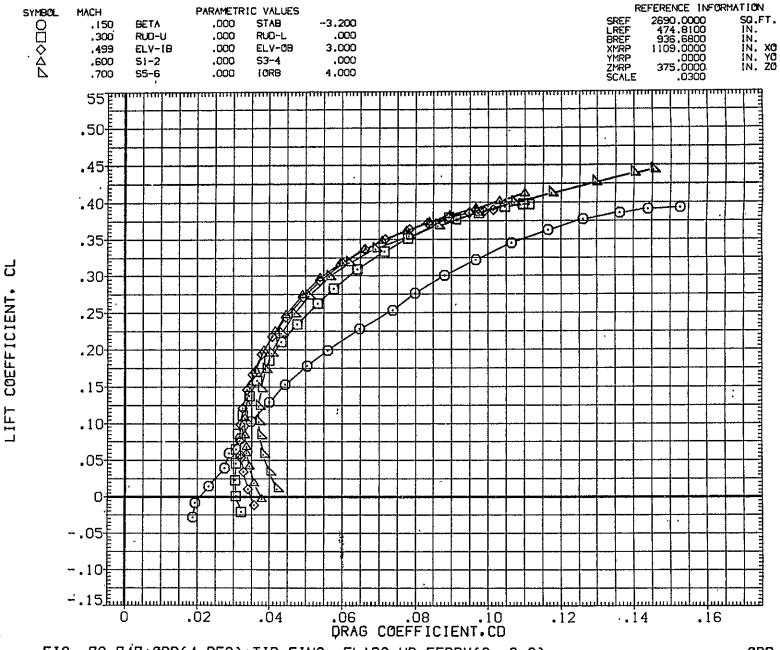


FIG. 79 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY(S=-3.2)

ORB

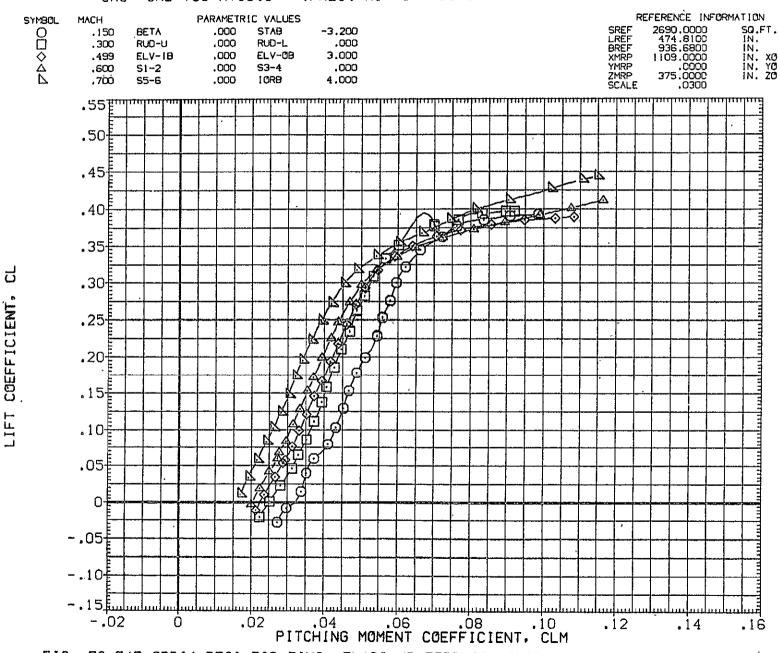
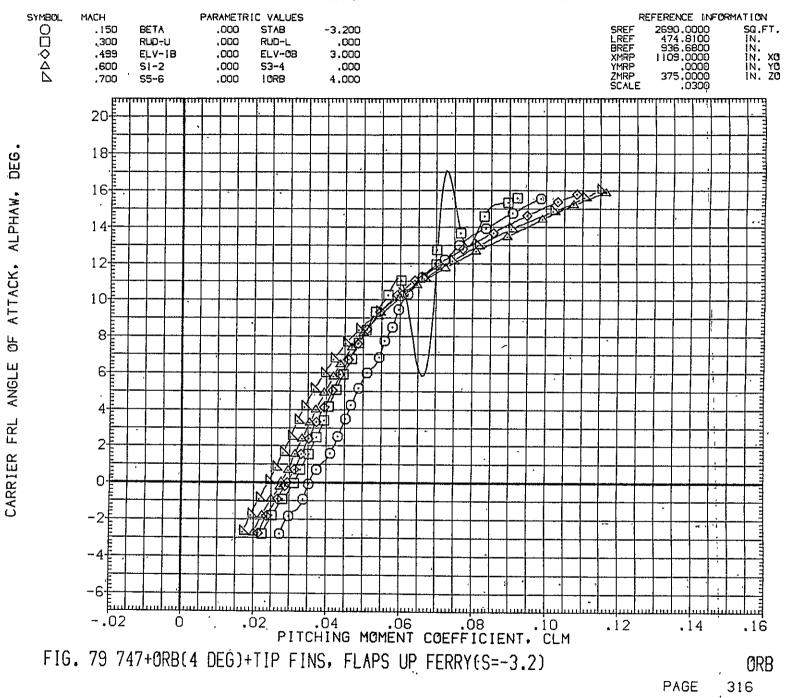


FIG. 79 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY(S=-3.2)

CA5 ORB TC5 AT38.3 (PRES. K1 FO H15.6 V9.1)(YGM112)



CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.3)(RGM116) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH 5500,0000 327,8000 2348,0000 1339,9100 .0000 190,7500 SQ.FT. IN. IN. IN. XC IN. YC IN. ZC 000 1,000 .000 STAB BETA .300 .000 .500 RUD-U .000 RUO-L BREF XMRP YMRP ZMRP 3,000 ELV-IB .000 ELV-08 .600 .000 .700 S1-2 .000 S3-4 4.000 .000 TORB S5-6 1.1€ 1.0 .8[占 COEFFICIENT, LIFT 1 2 3 4 5 6 7 8 9 10 CARRIER FRL ANGLE OF ATTACK, ALPHAW, DEG. 13 14

FIG. 80 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY(S=+1)

CA5 K1 FO H15.6 V9.1 (PLUS. ØRB TC5 AT38.3)(RGM116)

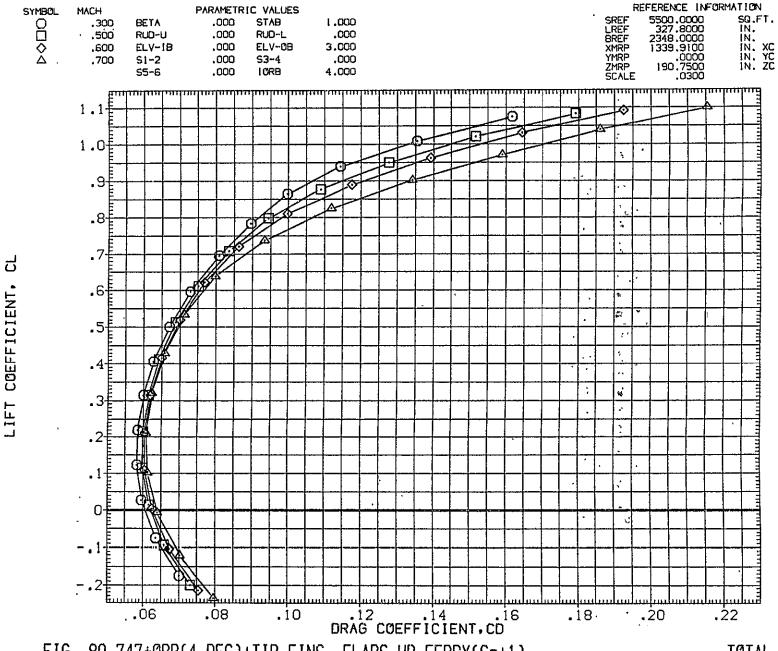
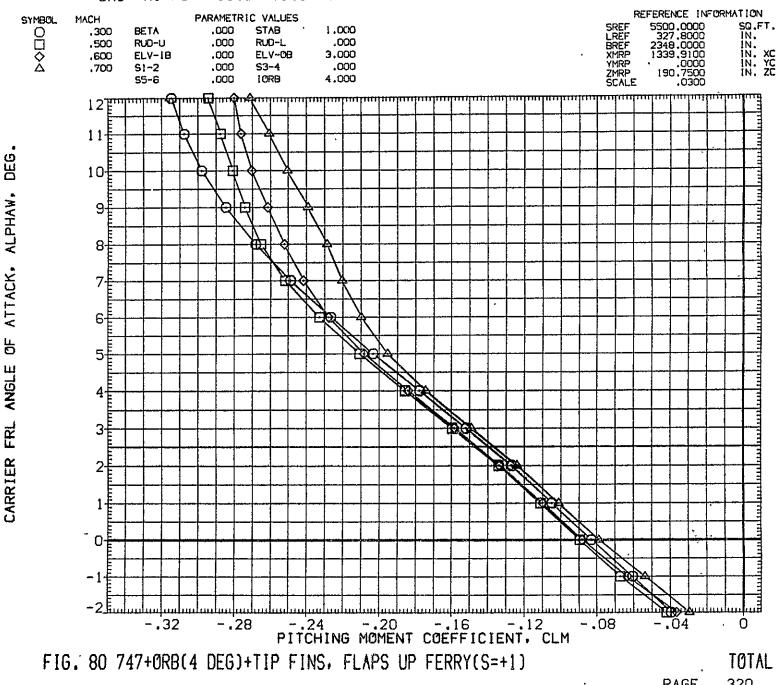


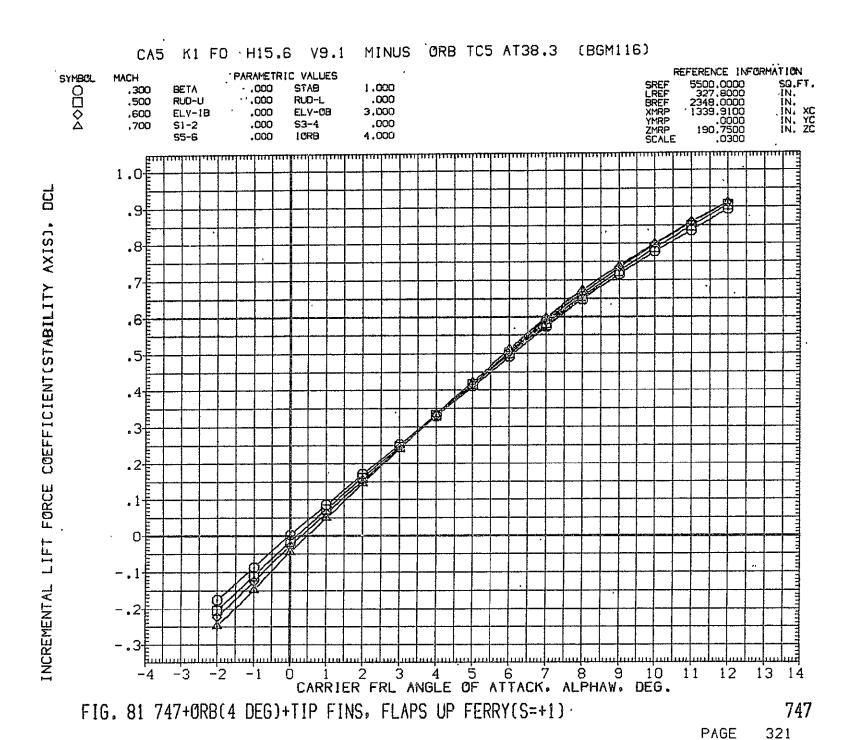
FIG. 80 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY(S=+1)

FIG. 80 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY(S=+1)

CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.3) (RGM116)







CA5 K1 FO H15.6 V9.1 MINUS ORB TC5 AT38.3 (BGM116)

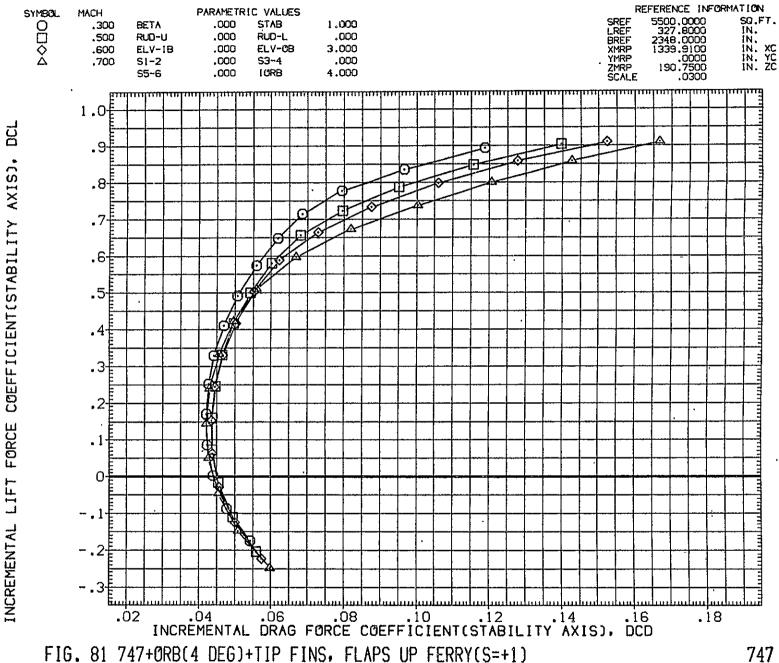
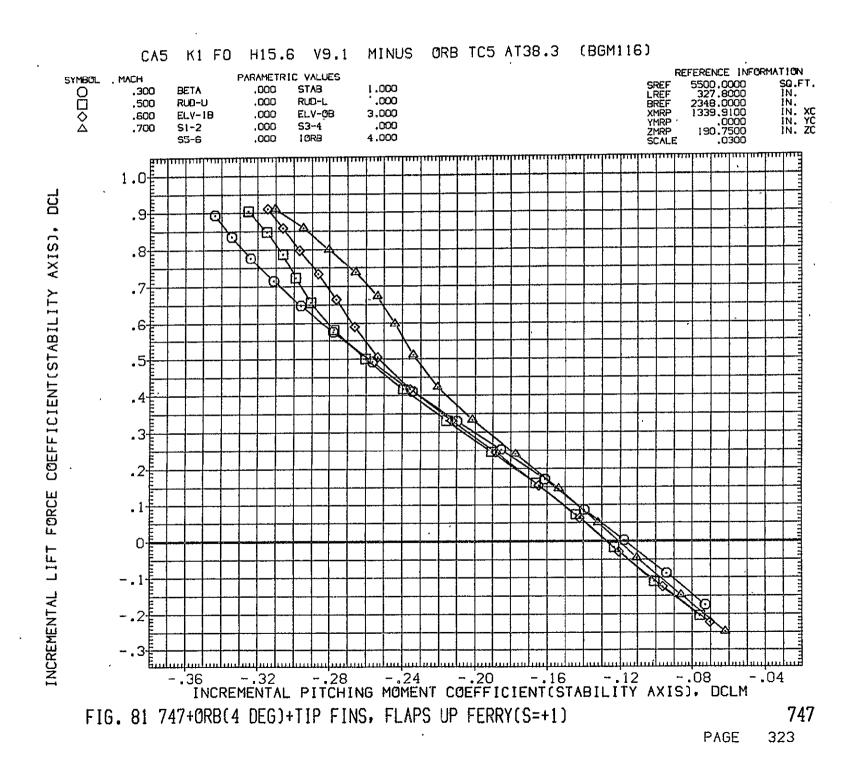
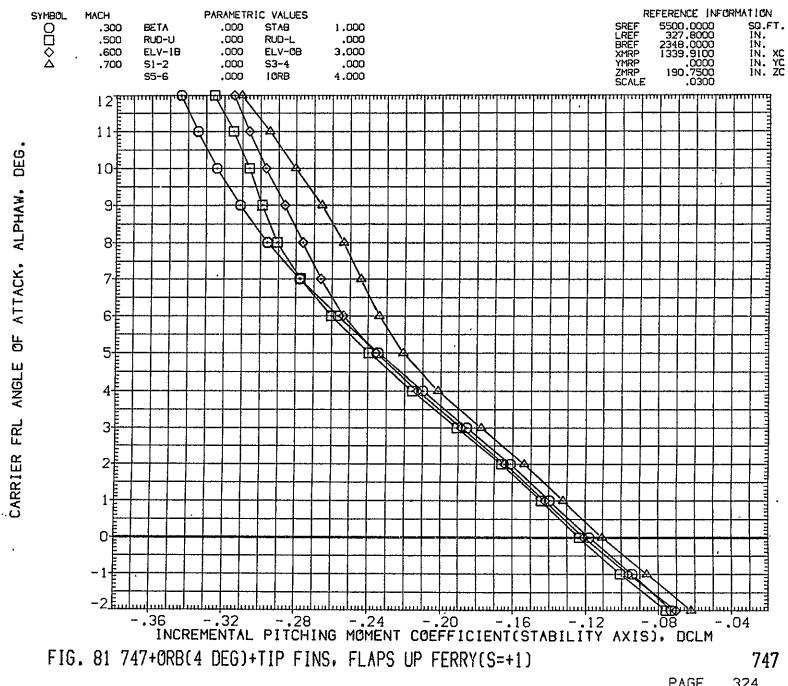


FIG. 81 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY(S=+1)



CA5 K1 FO H15.6 V9.1 MINUS ORB TC5 AT38.3 (BGM116)



CA5 ORB TC5 AT38.3 (PRES. K1 FO H15.6 V9.1) (YGM116)

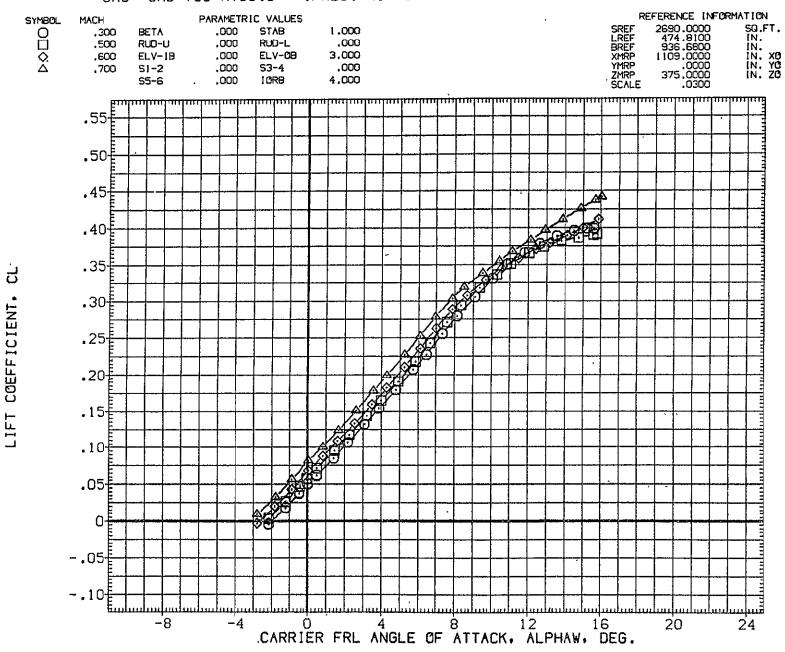


FIG. 82 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY(S=+1)

CA5 ORB TC5 AT38.3 (PRES. K1 FO H15.6 V9.1)(YGM116)

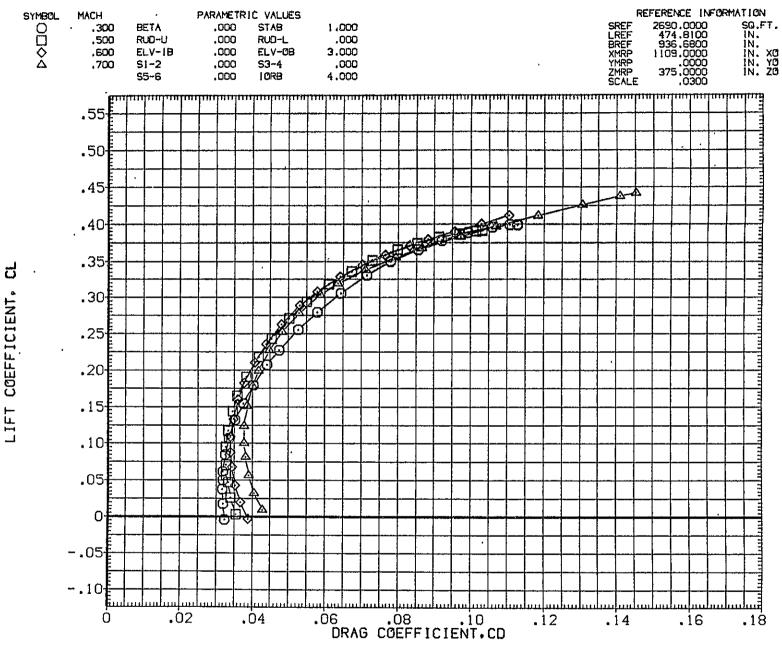


FIG. 82 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY(S=+1)

ORB



FIG. 82 747+ORB(4 DEG)+TIP FINS, FLAPS UP FERRY(S=+1)

ORB

327

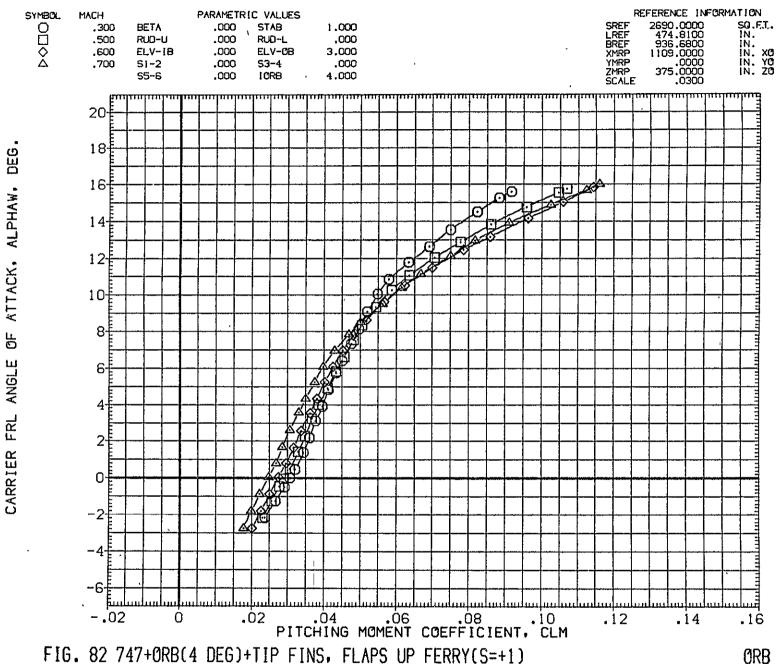


FIG. 82 747+ORB(4 DEG)+TIP FINS, FLAPS UP FERRY(S=+1)

CA5 .K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.3)(RGM117)

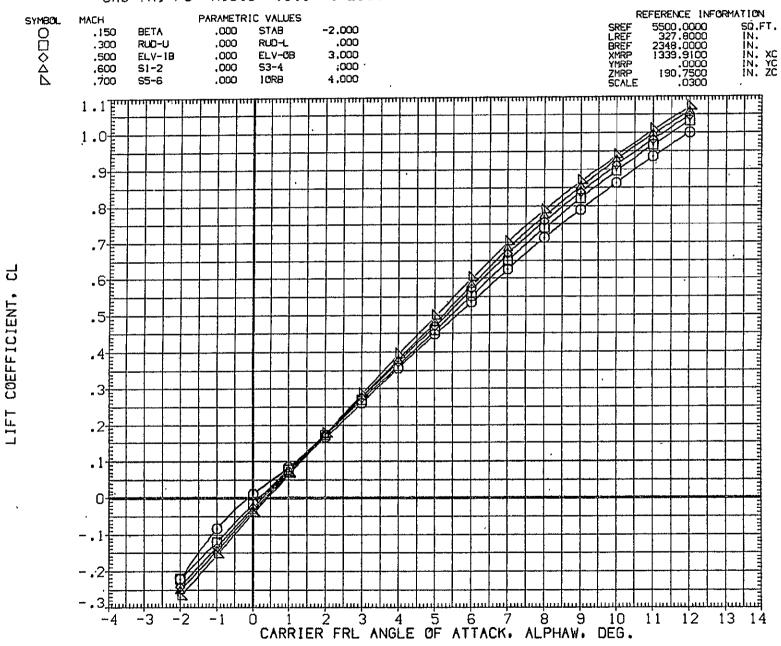


FIG. 83 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, BASELINE, (S=-2)

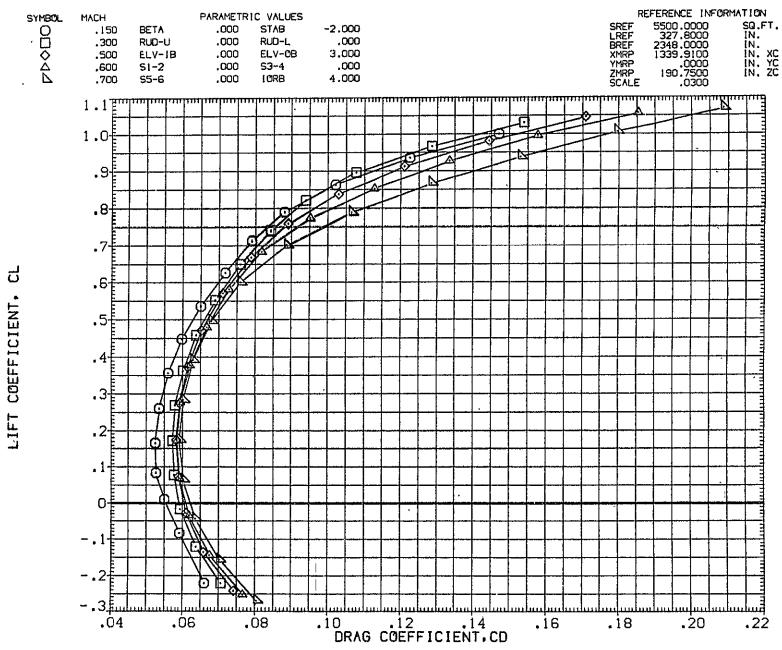


FIG. 83 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, BASELINE, (S=-2)



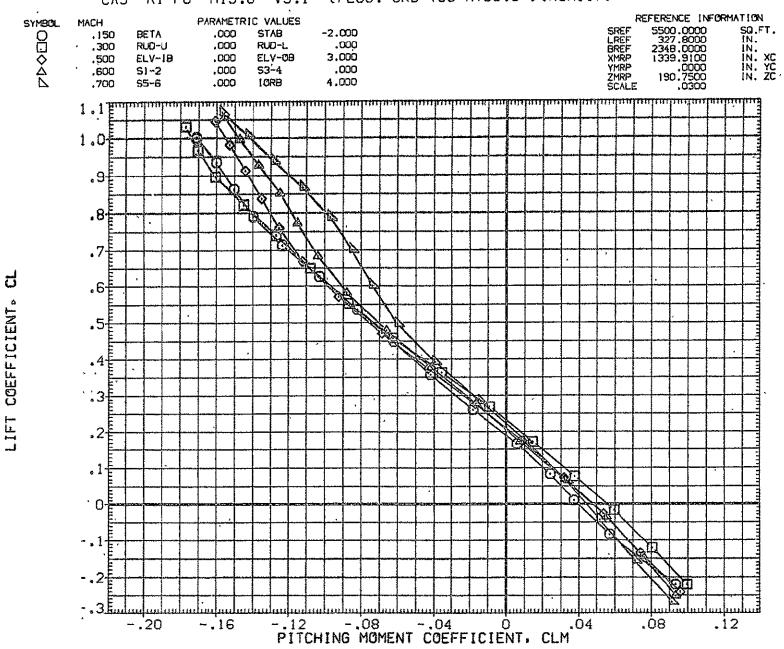


FIG. 83 747+ORB(4 DEG)+TIP FINS, FLAPS UP FERRY, BASELINE, (S=-2)

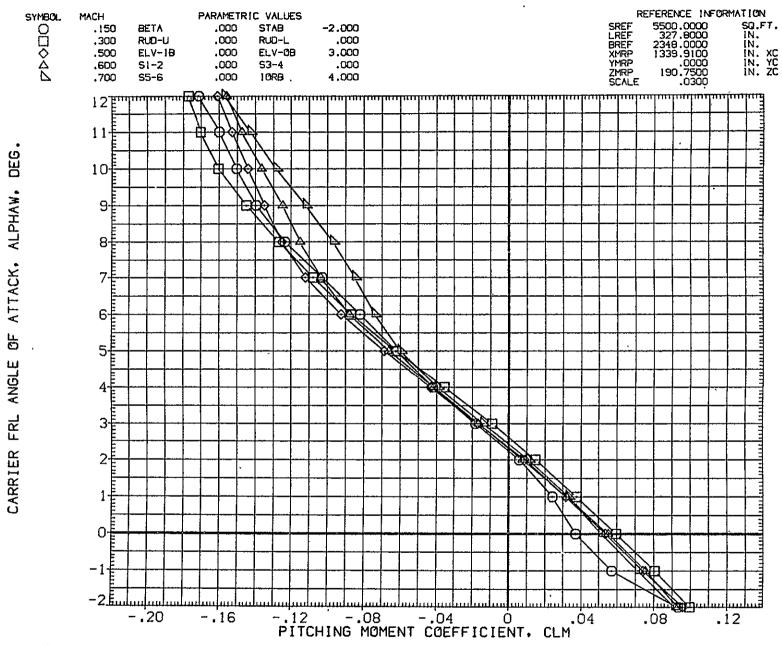


FIG. 83 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, BASELINE, (S=-2)

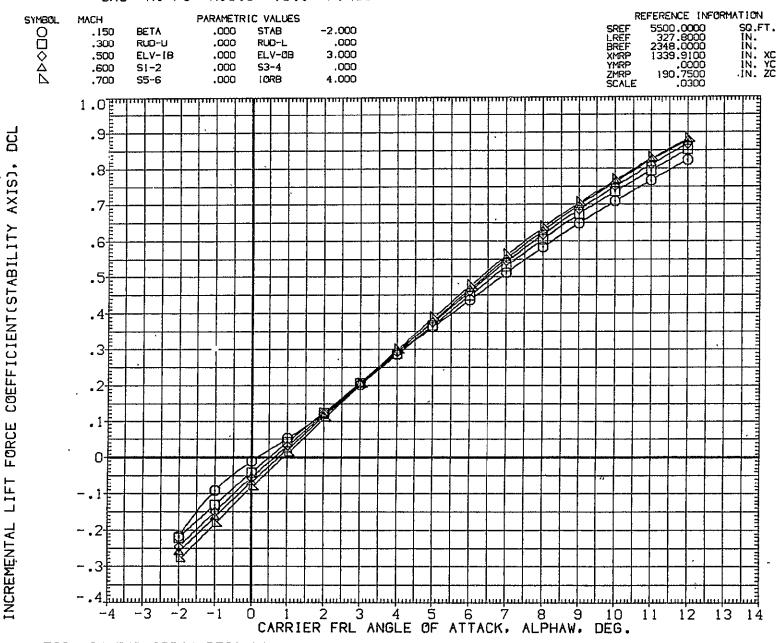


FIG. 84 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, BASELINE, (S=-2)

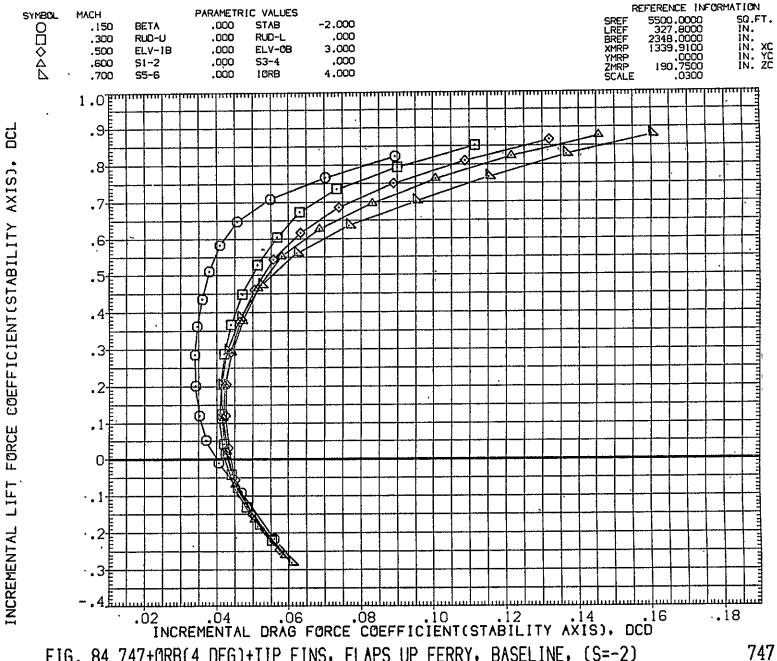
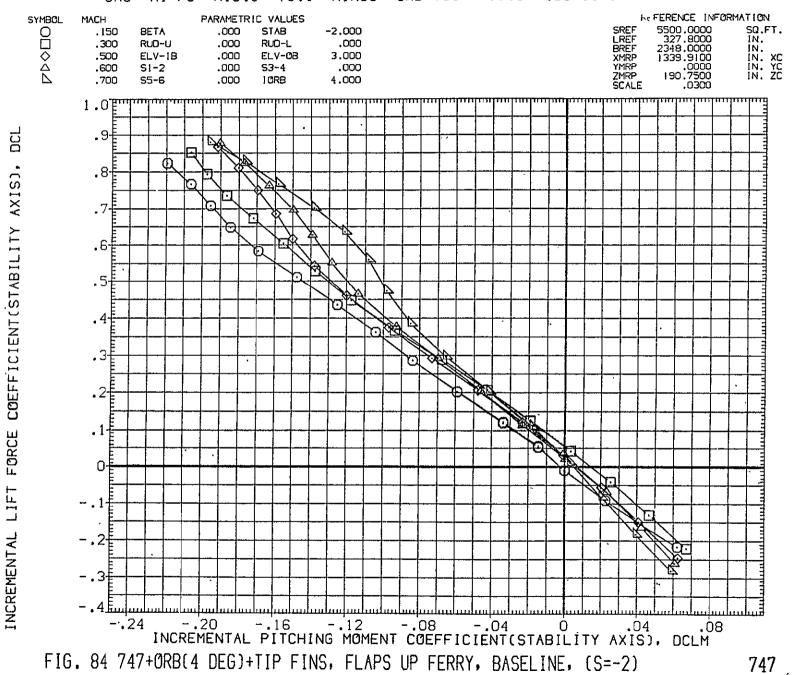


FIG. 84 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, BASELINE, (S=-2)

CA5 K1 FO H15.6 V9.1 MINUS ORB TC5 AT38.3 (BGM117)



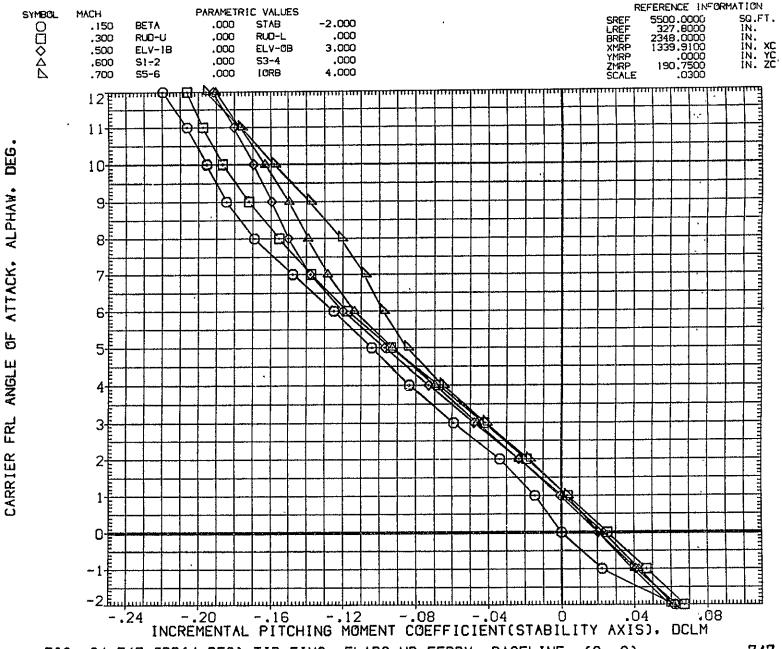


FIG. 84 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, BASELINE, (S=-2)

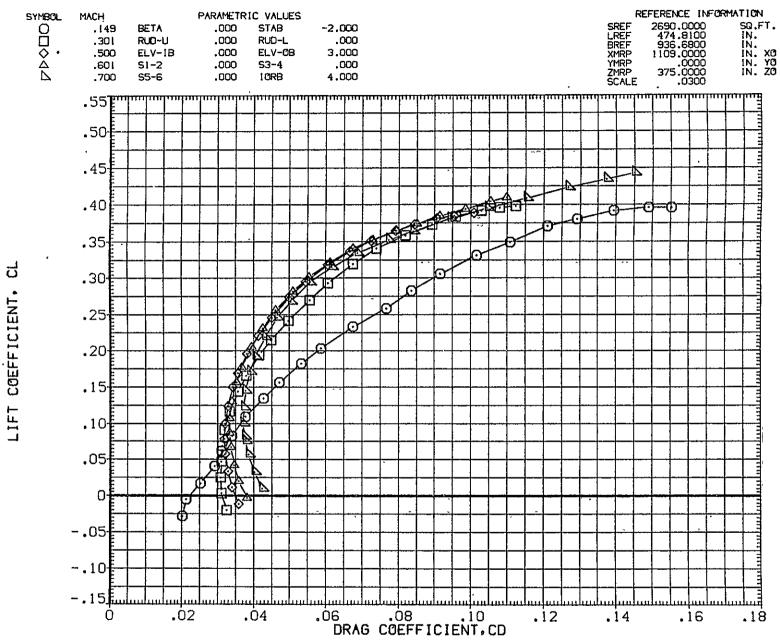
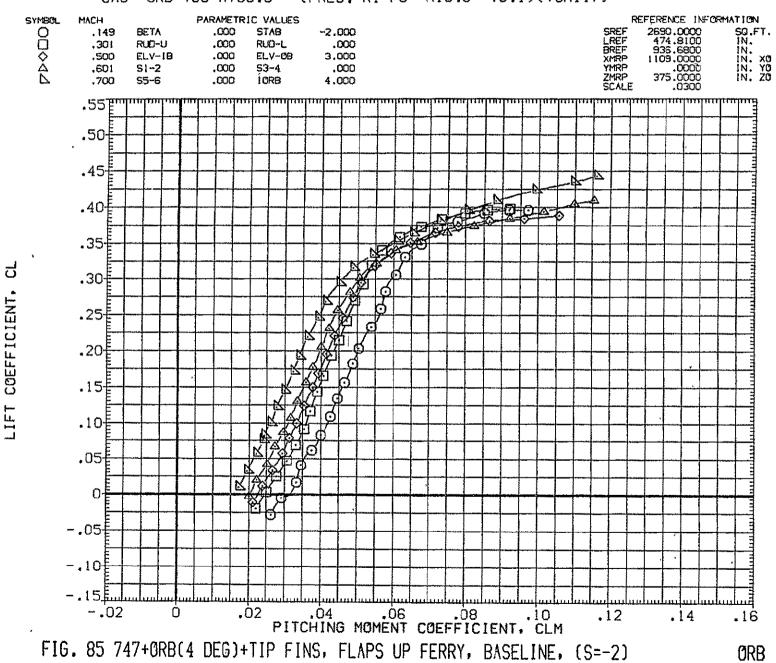


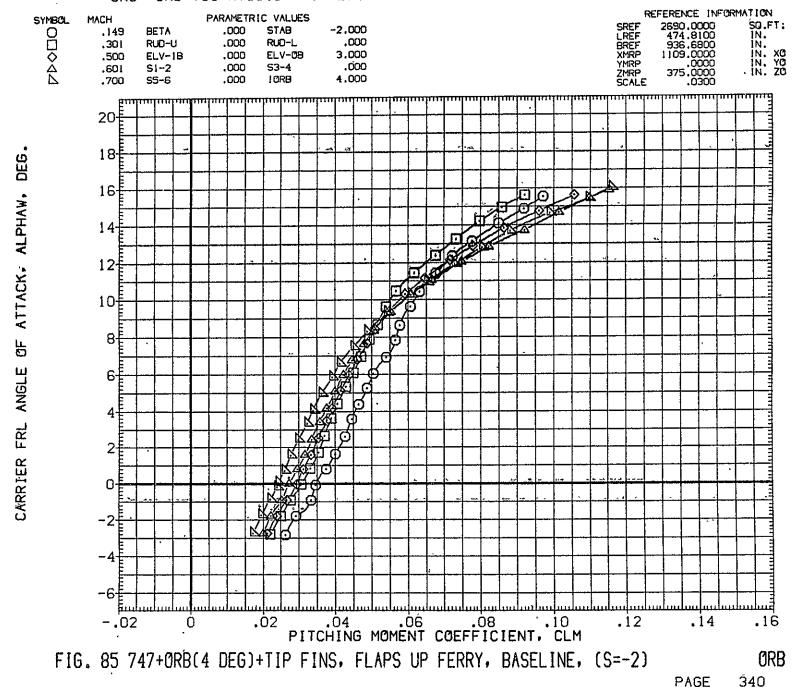
FIG. 85 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, BASELINE, (S=-2)

ØRB

CAS ORB TC5 AT38.3 (PRES. K1 FO H15.6 V9.1)(YGM117)



PAGE



CA5 K1 FO H15.6 V9.1 (PLUS. ØRB TC5 AT38.3)(RGM119) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH 5500.0000 327.8000 2348.0000 1339.9100 .0000 190.7500 .0300 SQ.FT. IN. IN. XC IN. YC IN. ZC SREF LREF BREF XMRP 000 BETA .000 STAB -2,000 .300 .000 .000 RUD-L .500 RUD-U ELV-0B -7,000 600 ELV-IB -10.000 YMRP ZMRP SCALE .000 53-4 .700 SI-2 .000 4.000 .000 LORB S5-6 1.0 . 8 [.7E .6[ರ LIFT COEFFICIENT, .5[·4[.3 0-1 2 3 4 5 6 7 8 9 10 CARRIER FRL ANGLE OF ATTACK, ALPHAW, DEG.

FIG. 86 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ELEV=-10 DEG

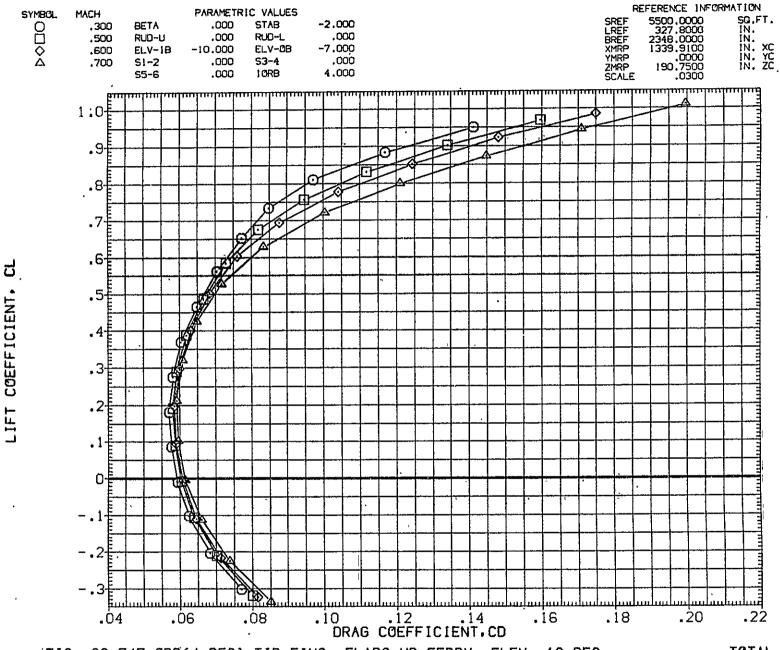


FIG. 86 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ELEV=-10 DEG

CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.3)(RGM119)

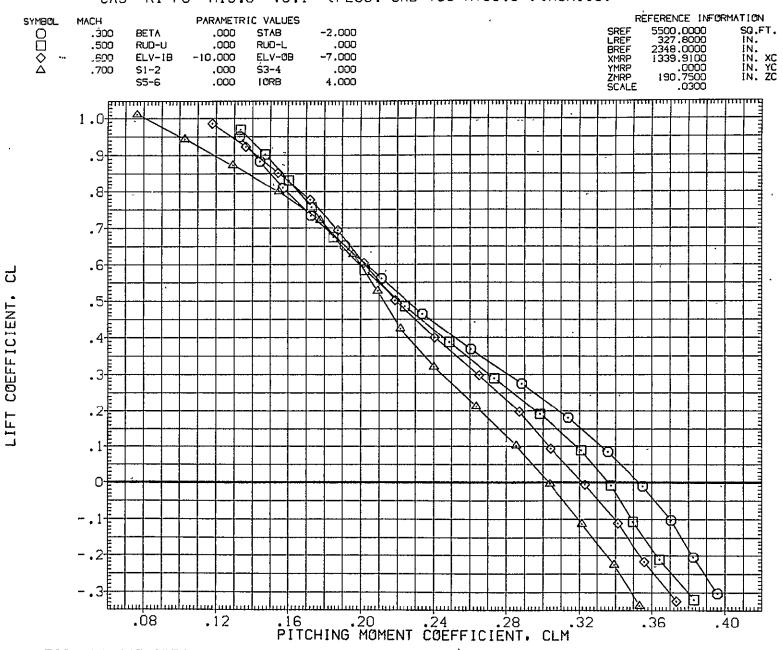


FIG. 86 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ELEV=-10 DEG

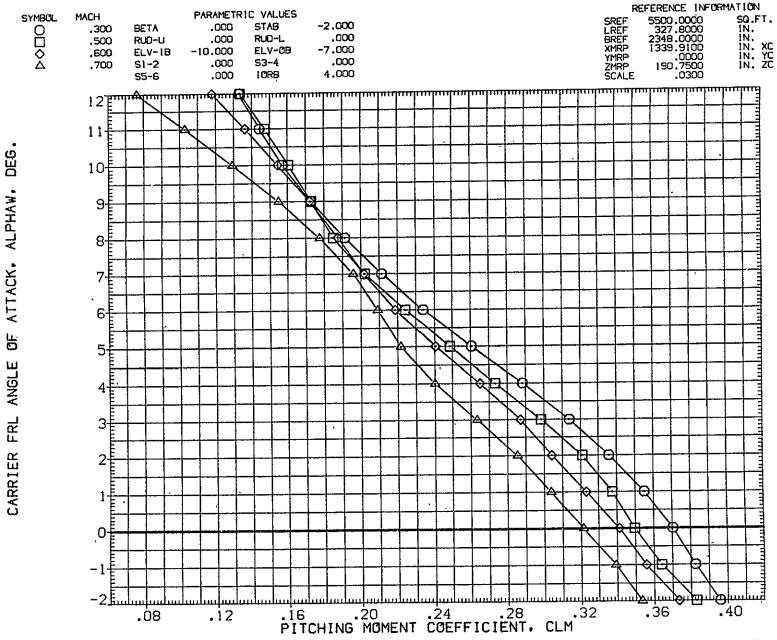


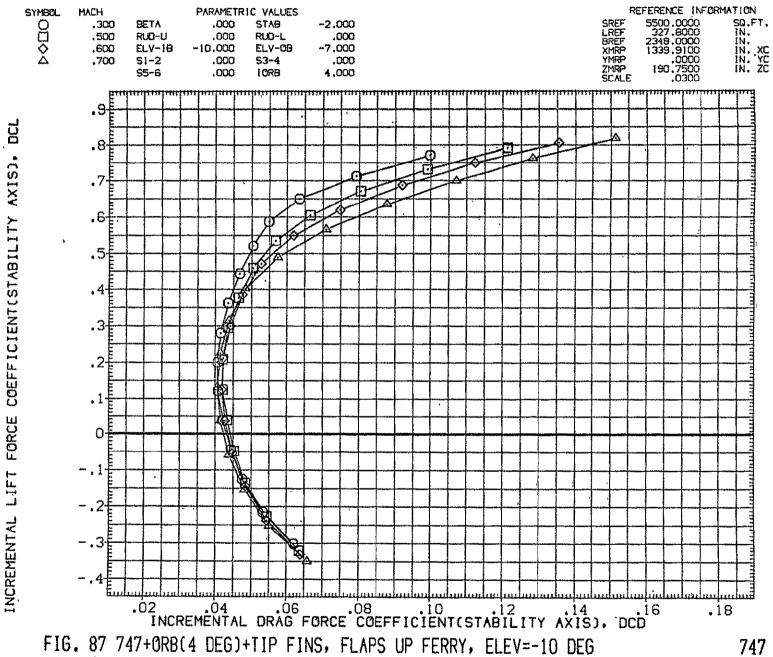
FIG. 86 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ELEV=-10 DEG

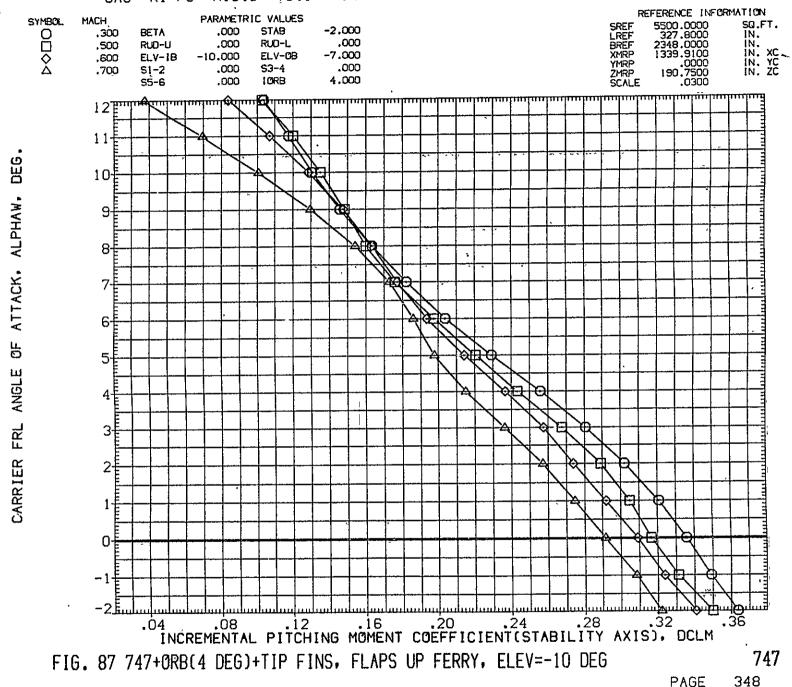


CA5 K1 FO H15.6 V9.1 MINUS ORB TC5 AT38.3 (BGM119) SYMBOL MACH PARAMETRIC VALUES REFERENCE INFORMATION SREF LREF BREF XMRP YMRP ZMRP SCALE .300 BETA .000 5500,0000 327,8000 2348,0000 1339,9100 .0000 190,7500 SQ.FT. IN. IN. XC IN. YC IN. ZC STAB -2,000 ,500 RUD-U .000 RUD-L .000 .600 ELV-18 -10.000 ELV-08 -7.000 .700 51-2 .000 53-4 .000 S5-6 .000 LORB 4.000 김 .8[COEFFICIENT(STABILITY, AXIS), .6‡ .5 ·4‡ .3 .2 ·1 INCREMENTAL LIFT FORCE 0 CARRIER FRL ANGLE OF ATTACK, ALPHAW. DEG. 12 13 FIG. 87 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ELEV=-10 DEG

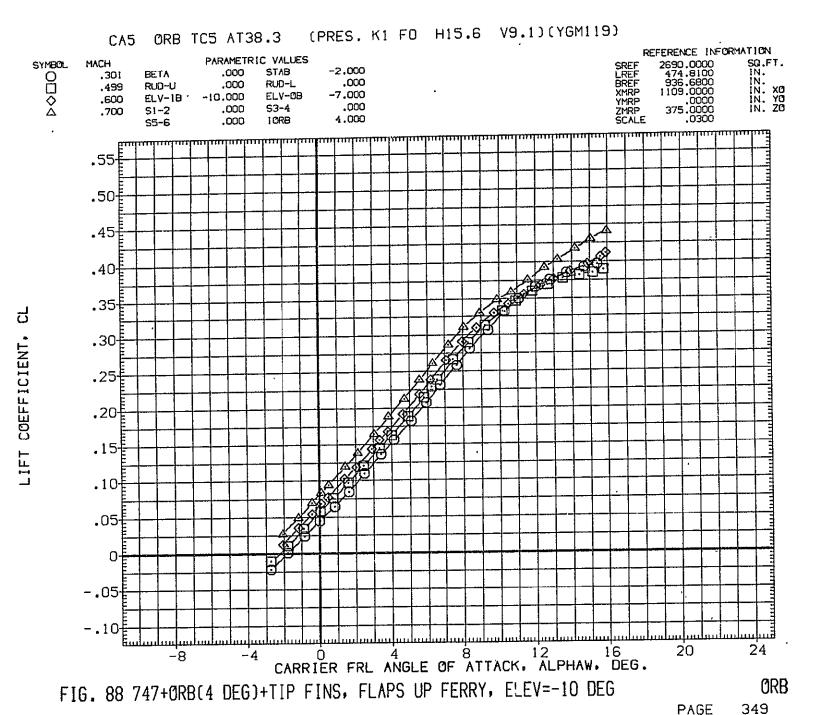
747

345









CA5 ORB TC5 AT38.3 (PRES. K1 FO H15.6 V9.1)(YGM119)

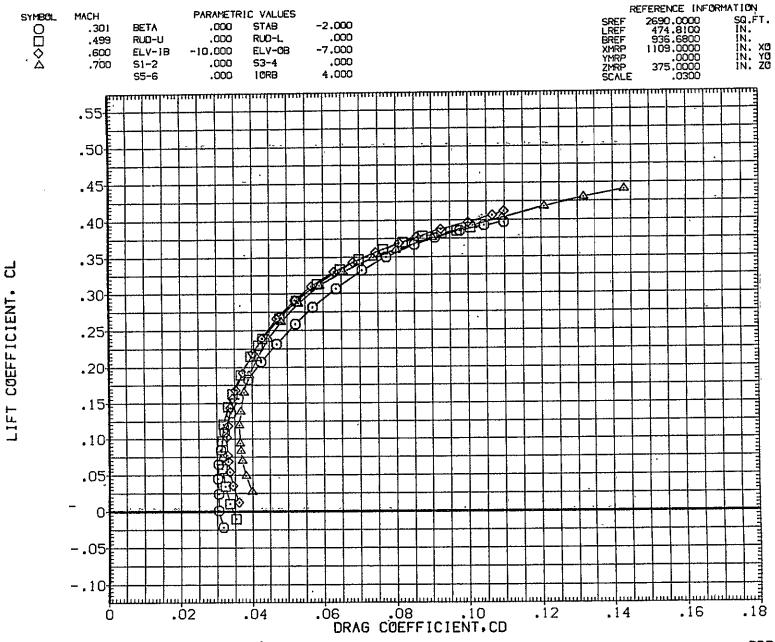


FIG. 88 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ELEV=-10 DEG

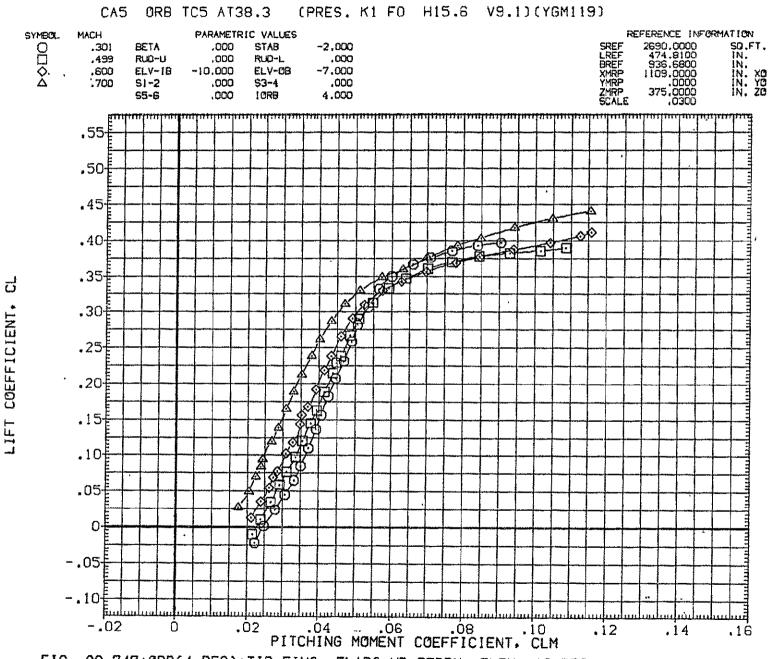
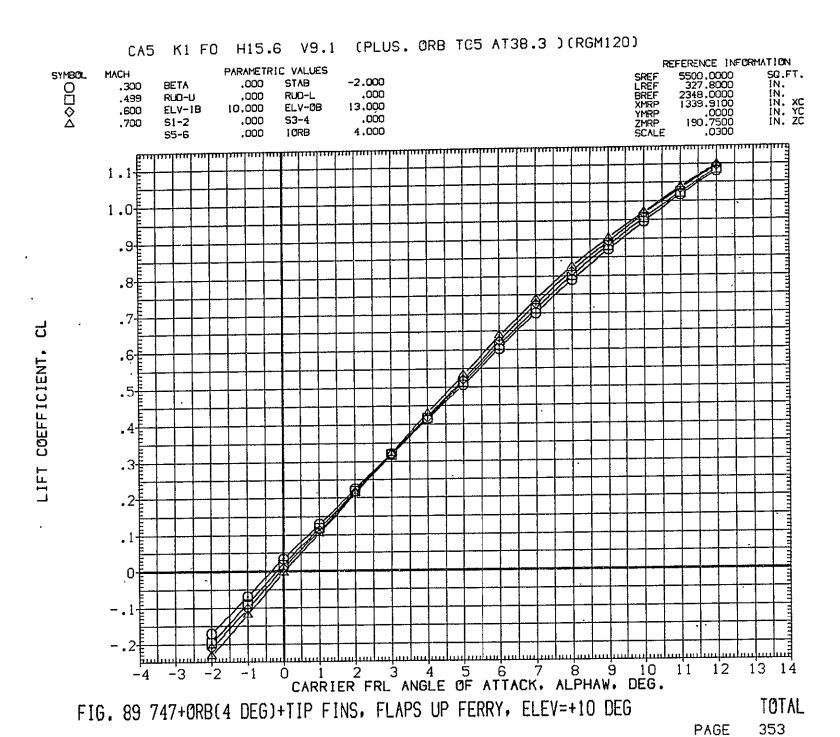


FIG. 88 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ELEV=-10 DEG

ORB

FIG. 88 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ELEV=-10 DEG



CA5 K1 FO H15.6 V9.1 (PLUS. ØRB TC5 A138.3) (RGM12U) REFERENCE INFORMATION PARAMETRIC VALUES .000 STAB SQ.FT. IN. IN. IN. XG IN. YC IN. ZC 5500.0000 327.8000 2348.0000 1339.9100 .0000 190.7500 SYMBOL MACH -2,000 000 BETA LREF BREF XMRP YMRP ZMRP SCALE .300 .000 ,000 RUD-L .499 RUD-U 13.000 ELV-08 .600 ELV-18 10,000 .000 .000 \$3-4 .700 S1~2 4.000 LORB S5-6 .000 1.1 O 1.0 .9-.8 占 .6-LIFT COEFFICIENT. .5[·4[

DRAG COEFFICIENT, CD TOTAL FIG. 89 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ELEV=+10 DEG PAGE 354

.06

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.10

.16

.i8

.20

.22

CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.3) (RGM120)

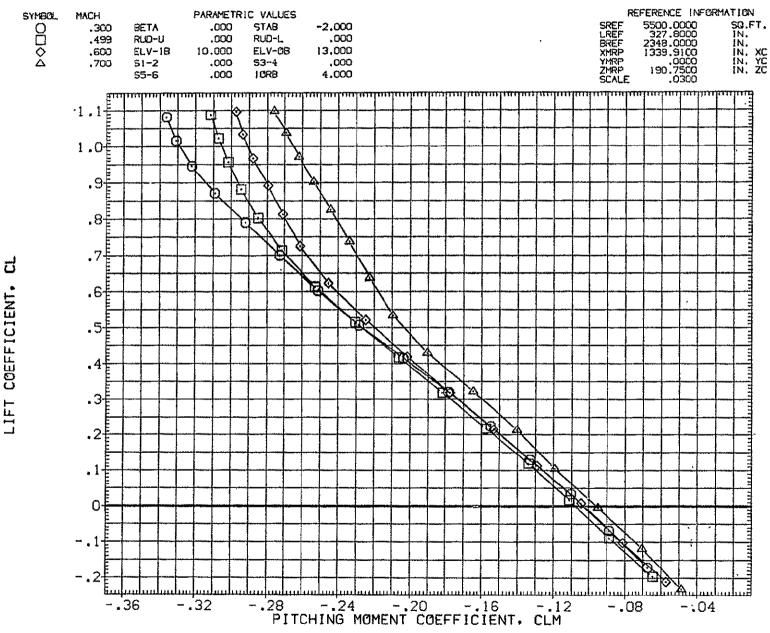


FIG: 89 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ELEV=+10 DEG

CA5 'K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.3)(RGM120) REFERENCE INFORMATION PARAMETRIC VALUES 5500.0000 327.8000 2348.0000 1339.9100 .0000 190.7500 .0300 SYMBOL MACH SQ.FT. IN. IN. IN. XC IN. YC IN. ZC SREF LREF BREF XMRP YMRP ZMRP SCALE -2.000 000 .000 STAB .300 BETA .000 .000 RUD-L RUD-U .499 13,000 ELV-0B .600 ELV~IB 10.000 S3-4 .000 .000 .700 51-2

DEG.

ATTACK, ALPHAW.

R

ANGLE

CARRIER FRL

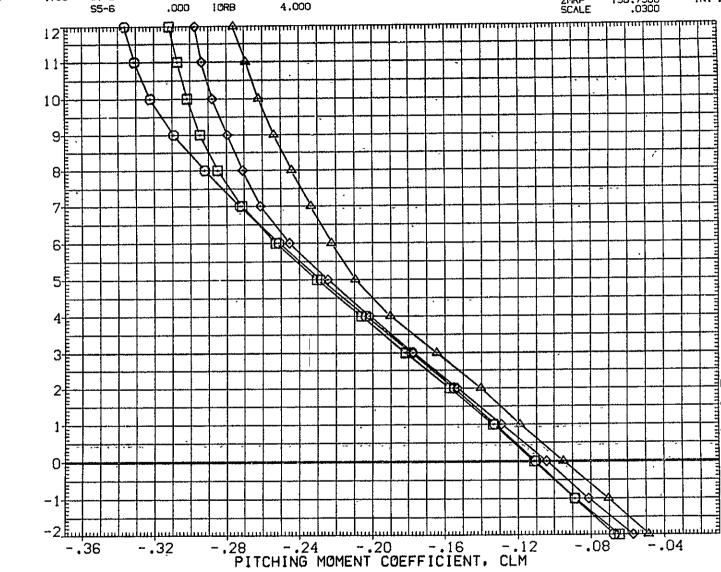
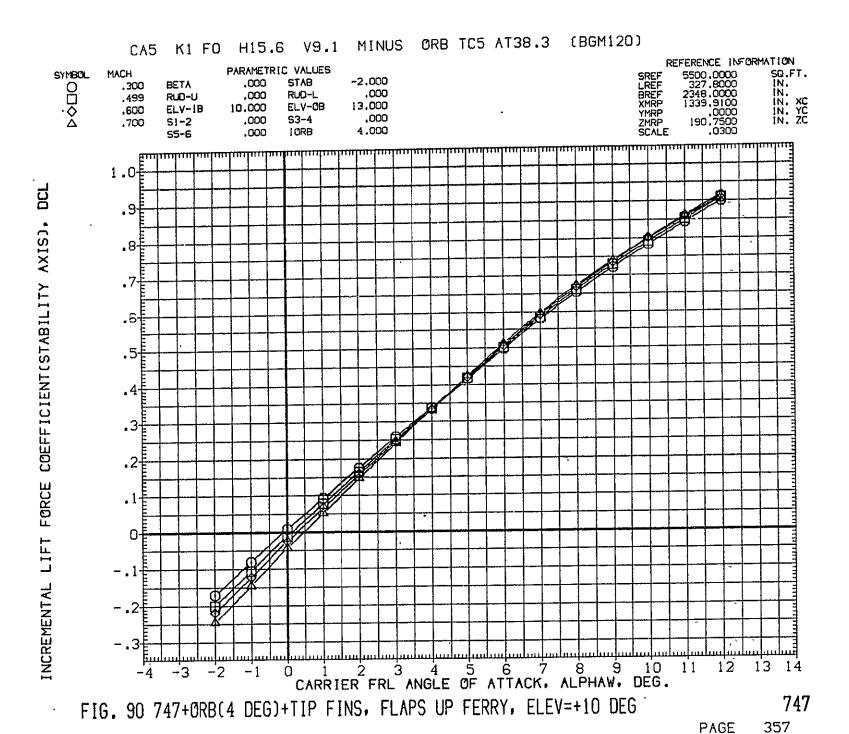


FIG. 89 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ELEV=+10 DEG

TOTAL



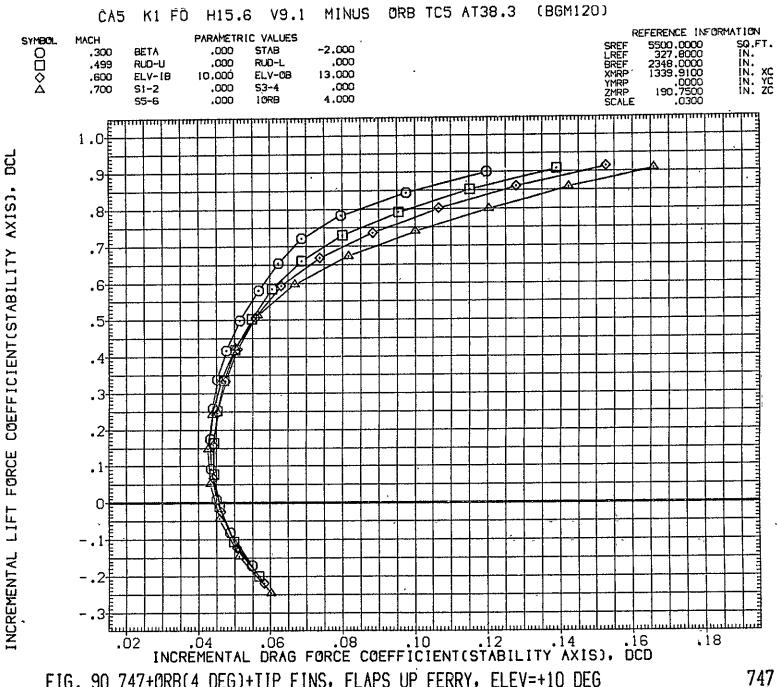
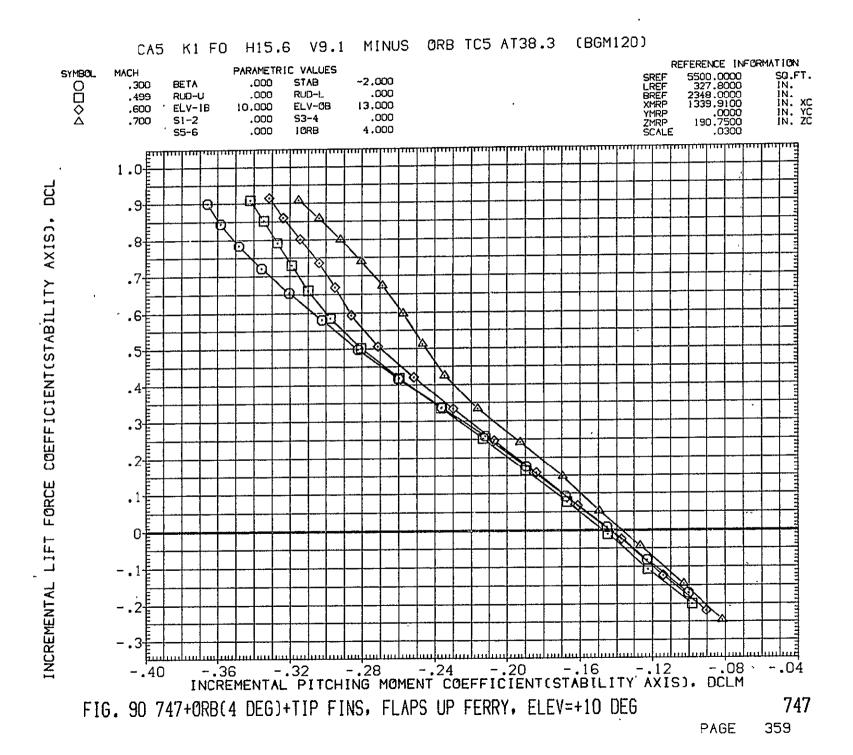


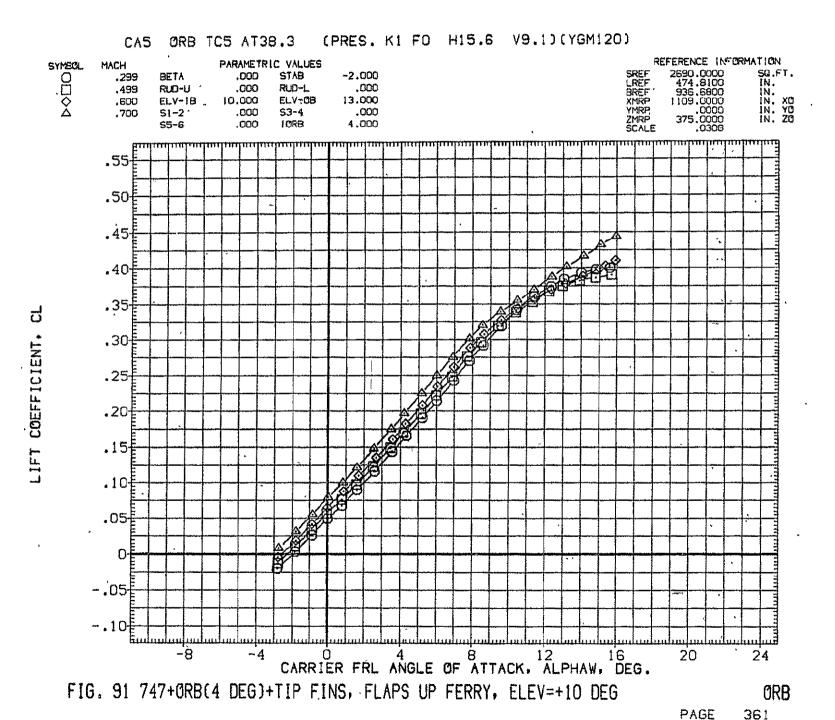
FIG. 90 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ELEV=+10 DEG



CA5 K1 FO H15.6 V9.1 MINUS ORB TC5 AT38.3 (BGM120) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH SG.FT. IN. IN. XC IN. YC IN. ZC 5500.0000 327.8000 2348.0000 SREF LREF BREF 0000 .000 STAB -2.000 .300 BETA .000 .000 RUO-L .499 RUD-U XMRP YMRP ZMRP SCALE 13.000 ELV-0B ELV-IB 10.000 .600 S3-4 .000 .700 S1-2 .000 4,000 I ORB .000 S5-6 12-<u>Բակավավակաֆափ</u>ակակակակակակարարա 11# DEG 10 ALPHAW. 9-8+ ATTACK, 6‡ PF. ANGLE 뭐 3€ CARRIER -1- -.36 -.32 -.28 -.24 -.20 -.16 -.12 -.08
INCREMENTAL PITCHING MOMENT COEFFICIENT(STABILITY AXIS). DCLM -.40 747

FIG. 90 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ELEV=+10 DEG





(PRES. K1 FO H15.6 V9.1)(YGM120) CA5 ORB TC5 AT38.3 REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH SQ.FT. IN. IN. XO IN. YO IN. ZO SREF LREF BREF XMRP YMRP ZMRP SCALE 2690.0000 474.8100 936.6800 1109.0000 .0000 375.0000 0□◊4 .000 STAB -2,000 .299 BETA .000 RUD-L. .000 RUD-U .499 13,000 ELV-08 .600 ELV-1B 10,000 .000 .000 \$3-4 .700 51-2 4.000 S5-6 .000 .55 .50 .45 .40 .35 تا LIFT COEFFICIENT. .30 .25 .20 .15 .10 .05 -.05 -.10

FIG. 91 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ELEV=+10 DEG

.02

.Ò4

.06 .08 .10 DRAG COEFFICIENT, CD .18

ORB

CAS ORB TC5 AT38.3 (PRES. K1 FO H15.6 V9.1)(YGM120) SYMBOL. MACH PARAMETRIC VALUES REFERENCE INFORMATION SQ.FT. .000 STAB -2.000 BETA .299 RUD-U .000 RUD-L 1000 .499 BREF XMRP YMRP ZMRP SCALE .600 ELV-18 10,000 ELV-OB 13.000 S1-2 53-4 .000 .000 S5-6 ,000 i CRB 4,000 .50 .45[.40E .35 겁 COEFFICIENT, .30[.25 .20[.15 .10 .05ŧ 0 -.05 -.10f -.02 Ò .02 .Ö4 .06 .08 .14 .12 .16 PITCHING MOMENT COEFFICIENT, CLM FIG. 91 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ELEV=+10 DEG ORB

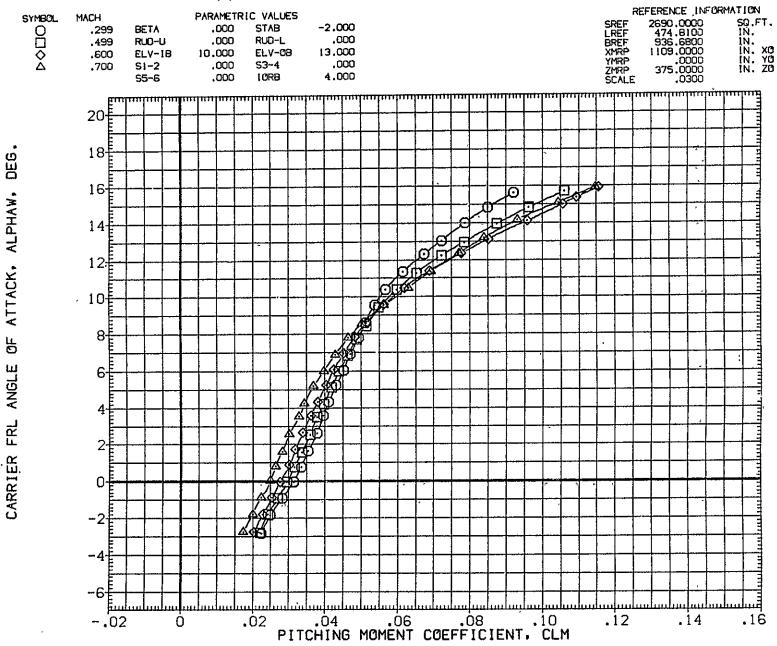
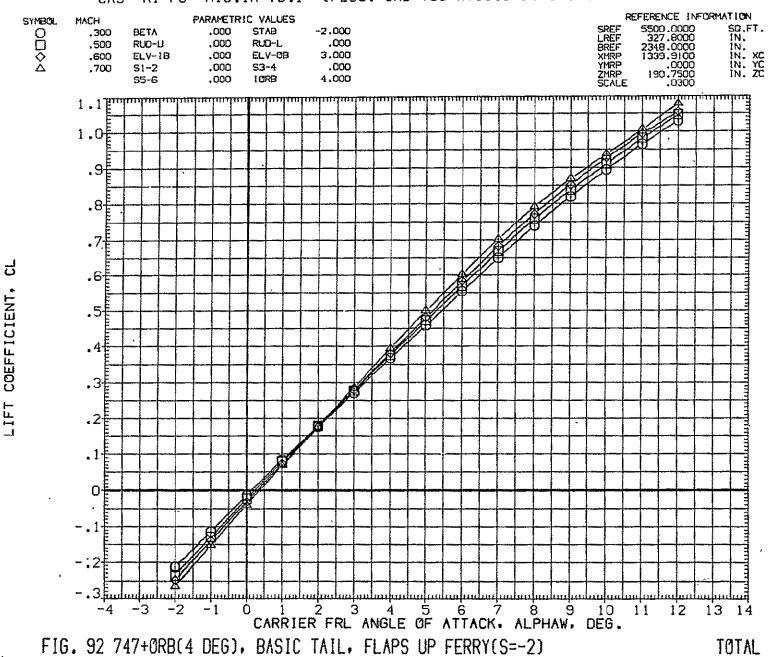


FIG. 91 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ELEV=+10 DEG

ORB

CA5 K1 FO H15.1A V9.1 (PLUS. ORB TC5 AT38.3)(RGM125)



CA5 K1 FO H15.1A V9.1 (PLUS. ORB TC5 AT38.3)(RGM125) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH SO.FT. IN. IN. XC IN. YC IN. ZC SREF LREF BREF XMRP 0000 STAB -2.000 .300 BETA .000 .500 RUD-U .000 RUD-L .000 3.000 ELV-0B .600 ELV-IB .000 YMRP ZMRP SCALE .000 S3-4 .000 .700 \$1-2 4.000 S5-6 .000 10RB 1.1 Էարախարարակատիուրա 1.0 .9ŧ .8[-.7년 .6 COEFFICIENT. .5

DRAG COEFFICIENT, CD FIG. 92 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY(S=-2) TOTAL PAGE 366

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.08

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.22

.20

.18

CA5 K1 FO H15.1A V9.1 (PLUS. ORB TC5 AT38.3)(RGM125)

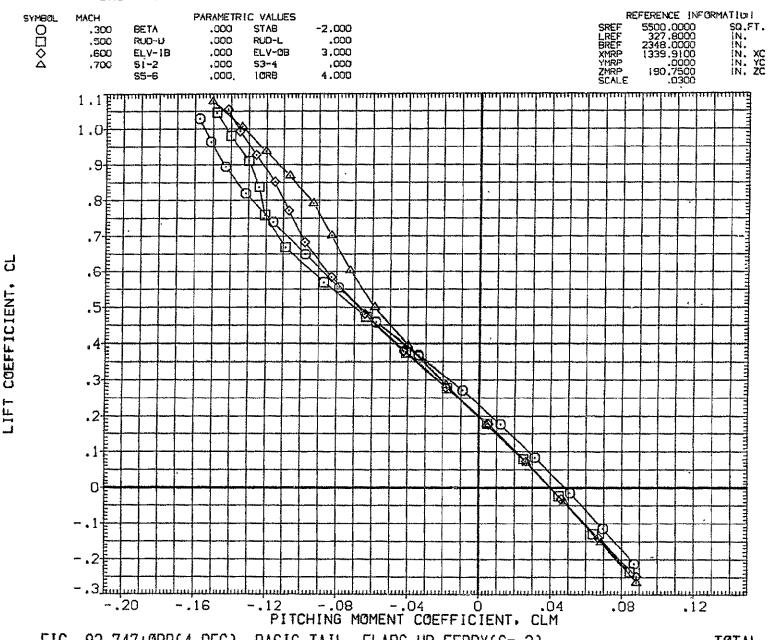


FIG. 92 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY(S=-2)

TOTAL

CA5 K1 FO H15.1A V9.1 (PLUS. ORB TC5 AT38.3)(RGM125) REFERENCE INFORMATION PARAMETRIC VALUES SQ.FT. IN. IN. IN. XC IN. YC IN. ZC 5500.0000 327.8000 2348.0000 1339.9100 .0000 190.7500 .0300 SYMBOL SREF LREF BREF XMRP YMRP ZMRP SCALE -2.000 0000 STAB BETA .000 .300 .000 RUD-L .500 RUD-U .000 3,000 ELV-08 .000 .600 ELV-18 .000 .000 53-4 .700 S1-2 4.000 LORB S5-6 .000 11+ DEG. 10[ATTACK. ALPHAW. 8 6 Ą ANGLE FR CARRIER 0 .08 .12 12 -.08 -.04 Ó .04 PITCHING MOMENT COEFFICIENT, CLM TOTAL

FIG. 92 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY(S=-2) PAGE

CA5 K1 FO H15.1A V9.1 MINUS ORB TC5 AT38.3 (BGM125) REFERENCE INFORMATION PARAMETRIC VALUES MACH SYMBOL SQ.FT. IN. IN. XC IN. YC IN. ZC 5500.0000 327.8000 2348.0000 SREF STAB 0000 .000 .-2.000 .300 BETA .000 RUD~L .000 .500 RUD-U 3.000 XMRP YMRP ZMRP SCALE 1339.9100 .0000 190.7500 .000 ELV-08 .600 ELV-IB .700 \$1-2 .000 53-4 4.000 .000 LORB S5-6 1.0 Emmini .8 .6 0-

1 2 3 4 5 6 7 8 9 10 CARRIER FRL ANGLE OF ATTACK, ALPHAW, DEG.

FIG. 93 747+ORB(4 DEG), BASIC TAIL, FLAPS UP FERRY(S=-2)

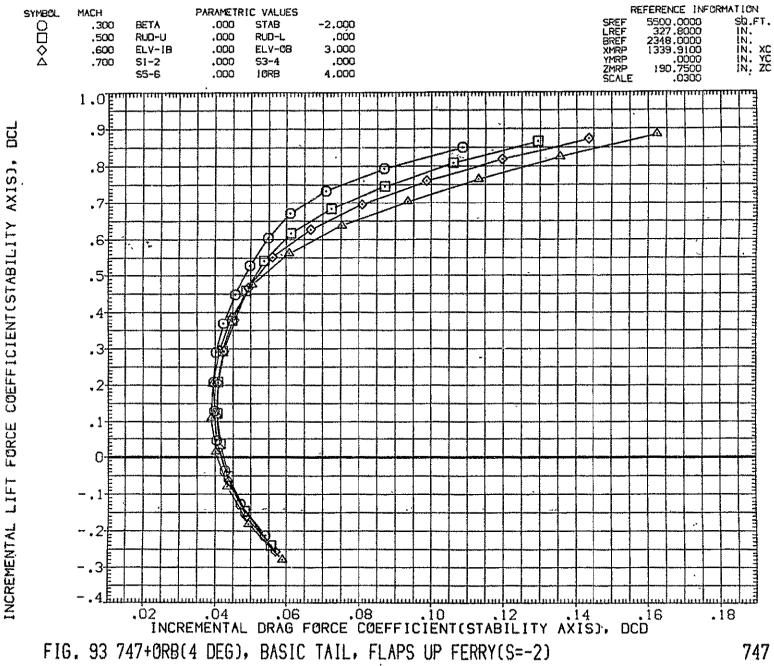
DCL

AXIS).

COEFFICIENT(STABILITY

FORCE

INCREMENTAL

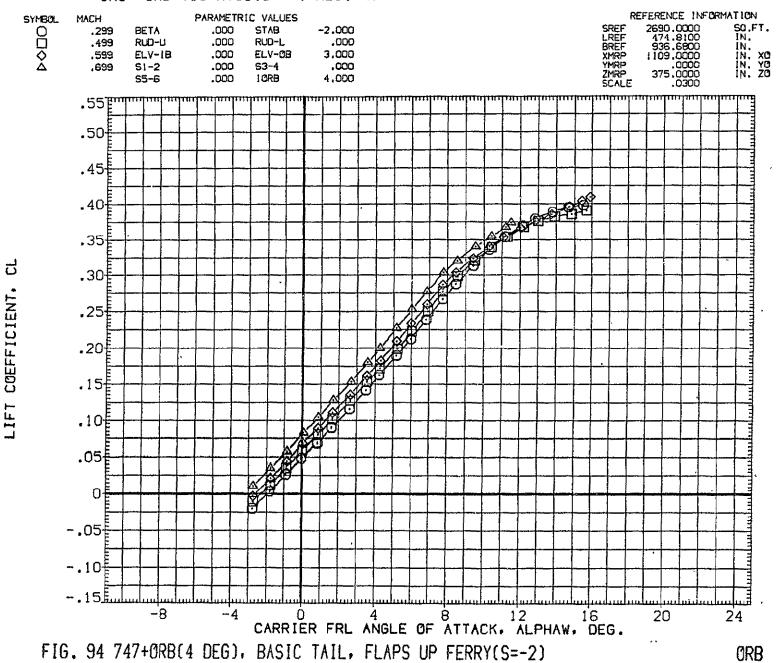


ORB TC5 AT38.3 (BGM125) CA5 K1 FO H15.1A V9.1 MINUS REFERENCE INFORMATION O O O SYMBOL PARAMETRIC VALUES MACH 5500.0000 327.8000 2348.0000 1339.9100 .0000 190.7500 SREF LREF BREF XMRP YMRP ZMRP SCALE SQ.FT. IN. IN. IN. XC IN. YC IN. ZC -2,000 STAB .300 BETA .000 .500 RUD-U .000 RUD-L ELV-08 3.000 .000 .600 ELV-IB .000 S3-4 .700 \$1-2 .000 55-6 .000 IORB 4.000 1.0E" 덤 COEFFICIENT(STABILITY AXIS), .8 .6ŧ •3[FORCE INCREMENTAL LIFT -.20 -.16 -.12 -.08 -.04 0 .04 .08 INCREMENTAL PITCHING MOMENT COEFFICIENT(STABILITY AXIS), DCLM .12 FIG. 93 747+ORB(4 DEG), BASIC TAIL, FLAPS UP FERRY(S=-2) 747

CA5 K1 FO H15.1A V9.1 MINUS ORB TC5 AT38.3 (BGM125) REFERENCE INFORMATION SYMBOL MACH PARAMETRIC VALUES SREF LREF BREF XMRP YMRP ZMRP SCALE 5500.0000 327.8000 2348.0000 1339.9100 .0000 190.7500 SQ.FT. IN. IN. XC IN. YC IN. ZC 000 STAB -2,000 .000 .300 BETA RUD-U .000 RUD-L .000 .500 ELV-1B .000 ELV-0B 3.000 .600 .000 .700 S1-2 .000 53-4 S5-6 .000 IORB 4.000 11+ OEC 10卡 ALPHAW. 9€ 8 ATTACK. 6[∵ R ANGLE 뛾 3-CARRIER 0--.20 -.16 -.12 -.08 -.04 0 .04 .08
INCREMENTAL PITCHING MOMENT COEFFICIENT(STABILITY AXIS). DCLM .12 747

FIG. 93 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY(S=-2)

CA5 ORB TC5 AT38.3 (PRES. K1 FO H15.1A V9.1)(YGM125)



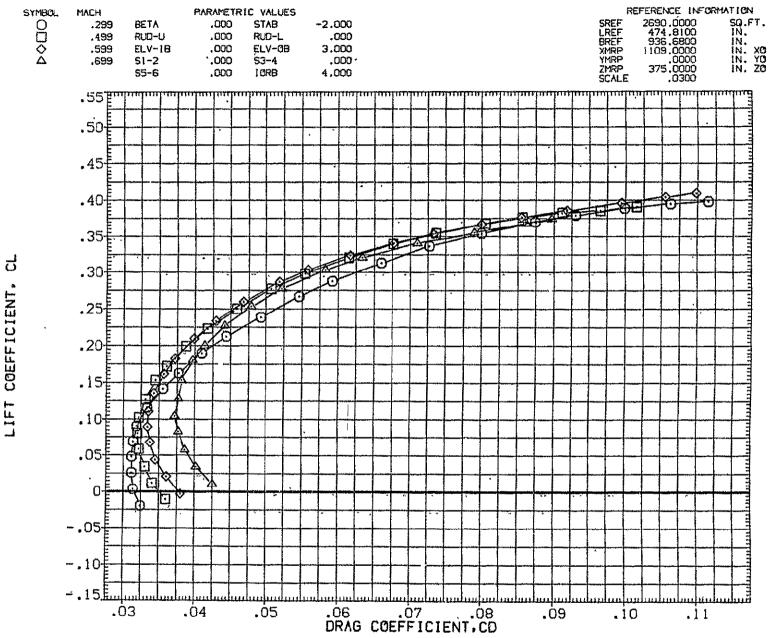
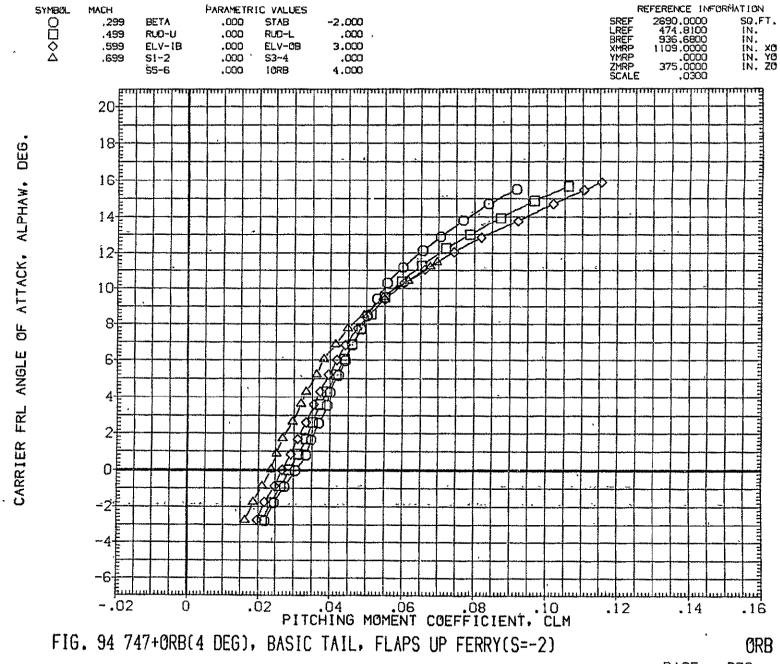


FIG. 94 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY(S=-2)

ORB



(PRES. K1 FO H15.1A V9.1)(YGM125) CA5 ORB TC5 AT38.3 REFERENCE INFORMATION MACH PARAMETRIC VALUES SYMBOL SREF LREF BREF XMRP YMRP ZMRP SCALE 2690.0000 474.8100 936.5600 1109.0000 SQ.FT. ₽<0□<4 .299 BETA .000 STAB -2.000 .000 .000 .499 RUD-U RUD-L IN. XO IN. YO IN. ZO .599 ELV-18 .000 ELV-08 3.000 51-2 .000 53-4 .000 S5-6 .000 IORB 4,000 .55E'''''''' **.**50₽ .45 .40 .35£ 占 .30£ COEFFICIENT, .25 .20 .15[.10E .05 -.05E -.10[-.15 £u.l. - .02 .04 .06 .08 .10 PITCHING MOMENT COEFFICIENT. CLM .12 .16 ·FIG. 94 747+ORB(4 DEG). BASIC TAIL. FLAPS UP FERRY(S=-2) ORB

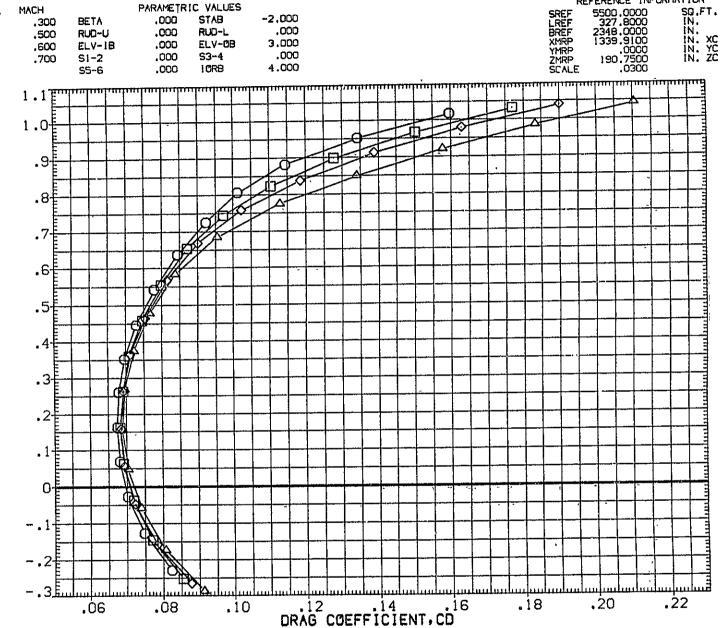


CA5 K1 FO H15.6 V9.1 (PLUS. ORB NOZ AT38.3) (RGM126) REFERENCE INFORMATION PARAMETRIC VALUES 5500.0000 327.8000 2348.0000 1339.9100 .0000 190.7500 SO.FT. IN. IN. IN. XC IN. YC IN. ZC SYMBOL. MACH SREF LREF BREF XMRP YMRP ZMRP SCALE -2.000 STAB 000 .000 ,300 BETA .000 RUD-U .000 RUD-L .500 3.000 .000 ELV-08 .600 ELV-IB S3-4 .000 .700 S1-2 .000 4,000 .000 LORB S5-6 1.1 բարարարարարարարարարար 1.0 .7 ᄓ .6[COEFFICIENT. 4 7 9 10 3 Ġ CARRIER FRL ANGLE OF ATTACK, ALPHAW, DEG.

FIG. 95 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, TAIL CONE OFF

TOTAL

377



REFERENCE INFORMATION

FIG. 95 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, TAIL CONE OFF
PAGE 378

CA5 K1 FO H15.6 V9.1 (PLUS. ORB NOZ AT38.3)(RGM126) MACH PARAMETRIC VALUES REFERENCE INFORMATION SYMBOL. 0000 SREF LREF BREF XMRP YMRP ZMRP SCALE SQ.FT. 5500.0000 .000 -2.000 .300 BETA STAB .500 RUD-U .000 RUD-L .000 ELV-08 3,000 IN. XC IN. YC IN. ZC .600 ELV-1B .000 0000 .700 S!-2 ,000 53-4 .000 190.7500 S5-6 .000 IORB 4,000 O 2 .6- COEFFICIENT. LIFT

- .04

PITCHING MOMENT COEFFICIENT, CLM

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FIG. 95 747+ORB(4 DEG)+TIP FINS, FLAPS UP FERRY, TAIL CONE OFF

- .08

-.16

TOTAL

.16

.12

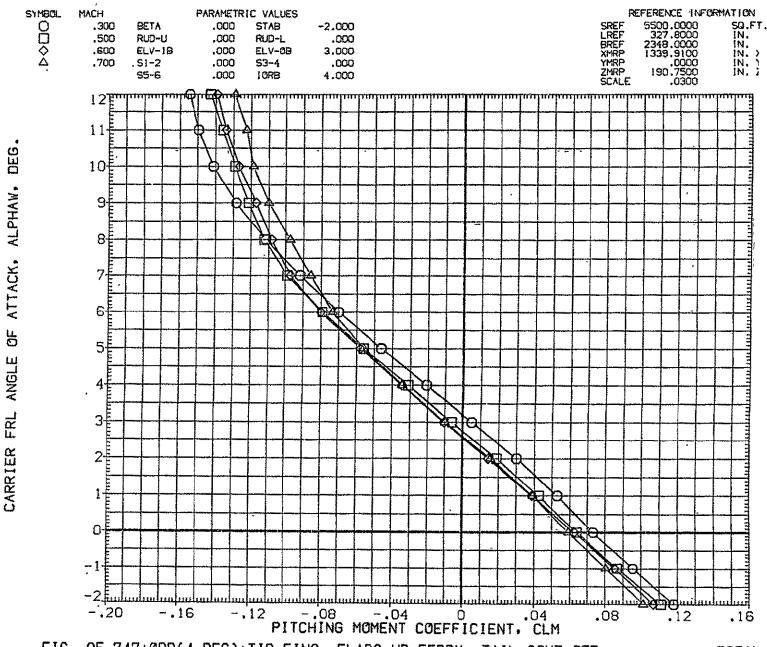


FIG. 95 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, TAIL CONE OFF

TOTAL

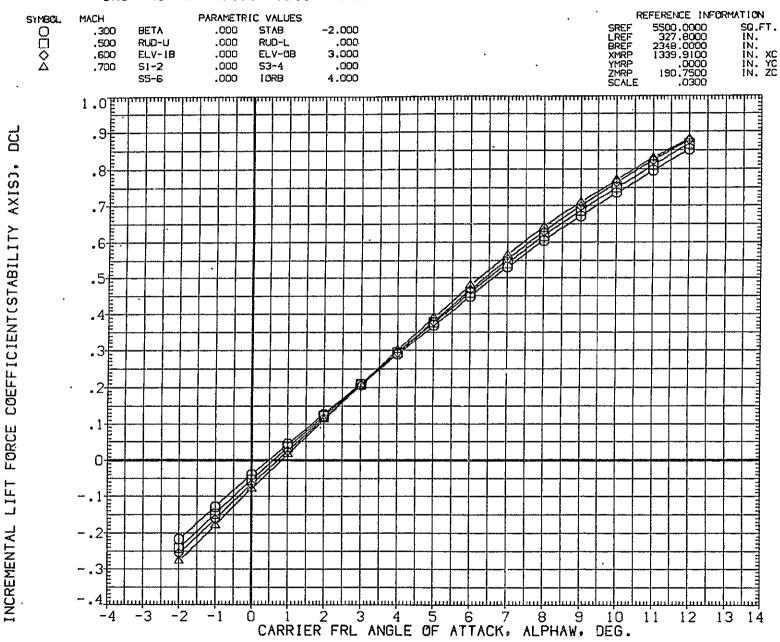
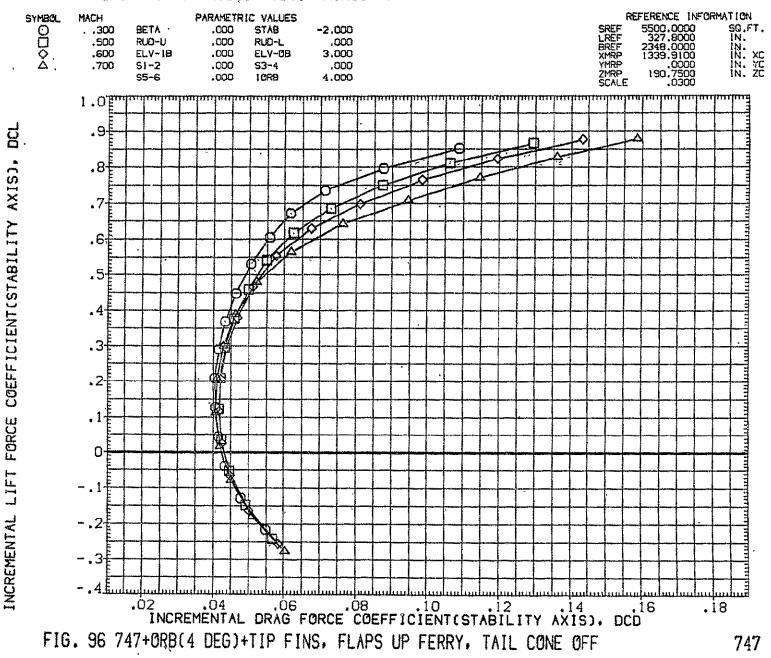
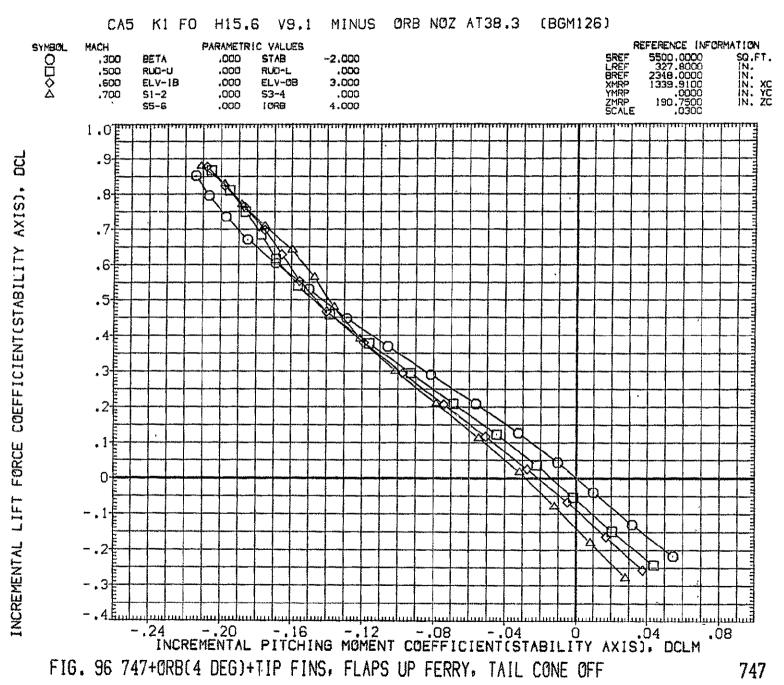


FIG. 96 747+ORB(4 DEG)+TIP FINS, FLAPS UP FERRY, TAIL CONE OFF





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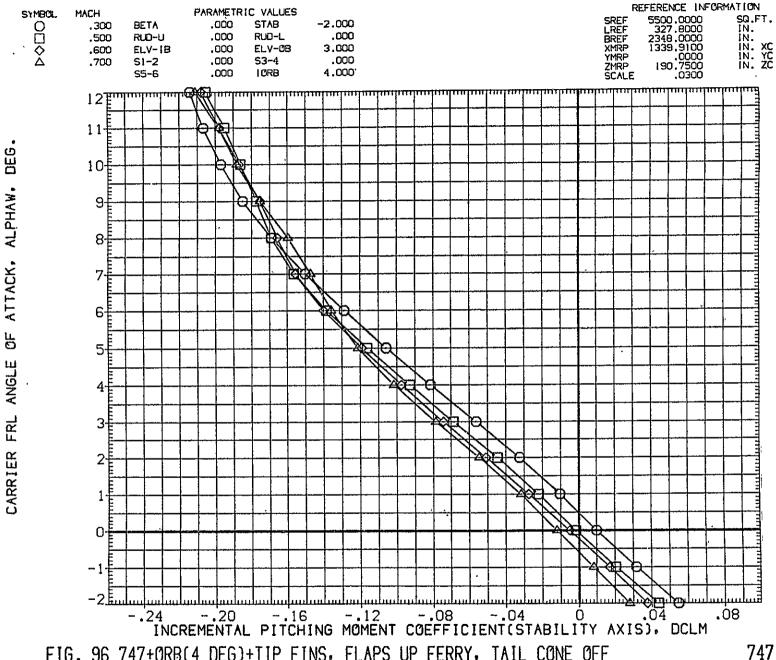


FIG. 96 747+ORB(4 DEG)+TIP FINS, FLAPS UP FERRY, TAIL CONE OFF

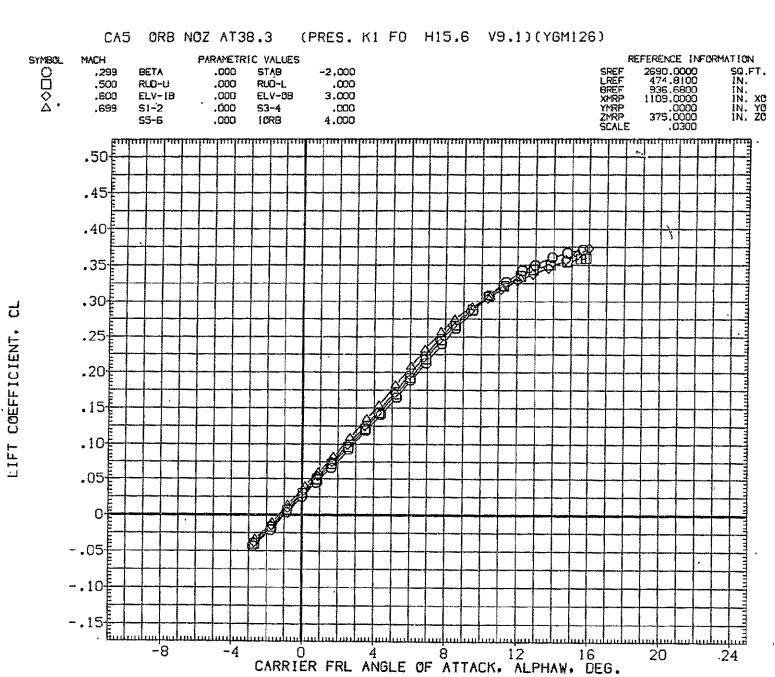


FIG. 97 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, TAIL CONE OFF

ORB

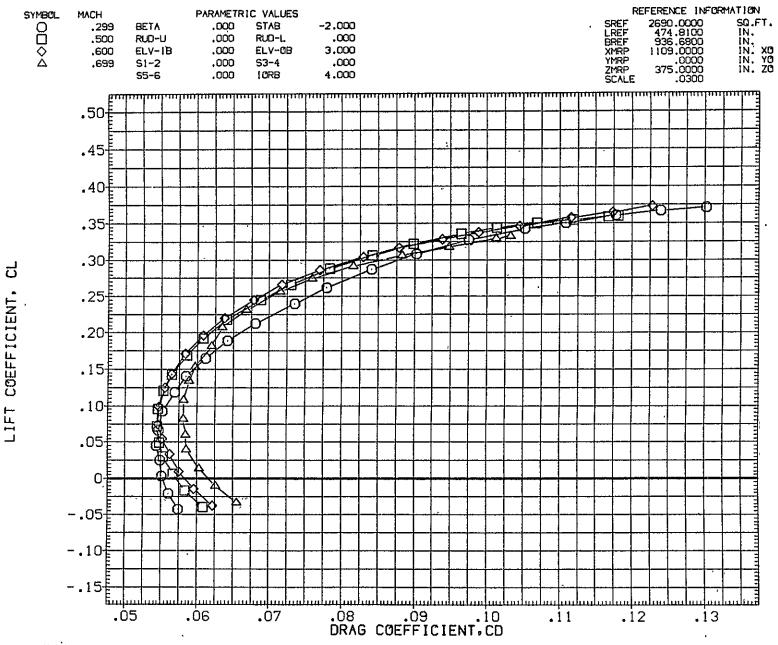


FIG. 97 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, TAIL CONE OFF

ORB

FIG. 97 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, TAIL CONE OFF

ORB

387

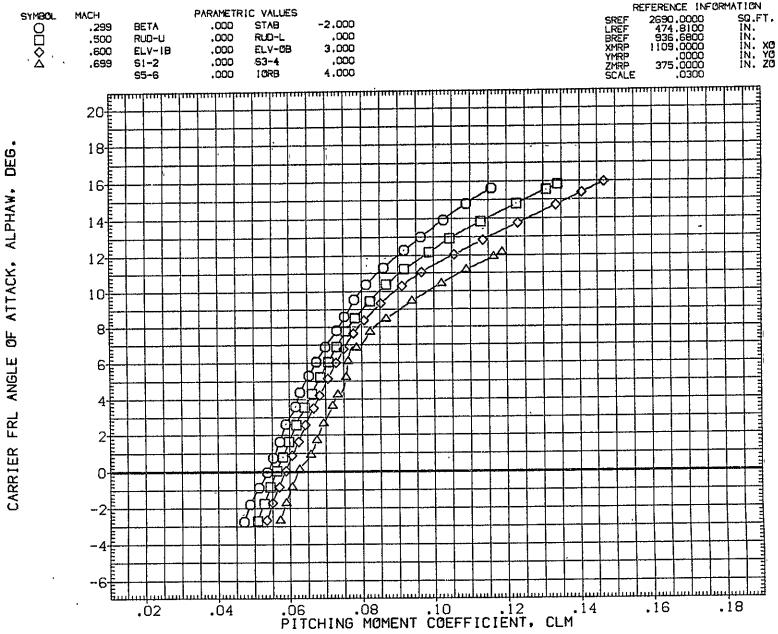


FIG. 97 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, TAIL CONE OFF

ORB

FIG. 98 ORBITER ALONE, TAIL CONE OFF

FIG. 98 ORBITER ALONE, TAIL CONE OFF

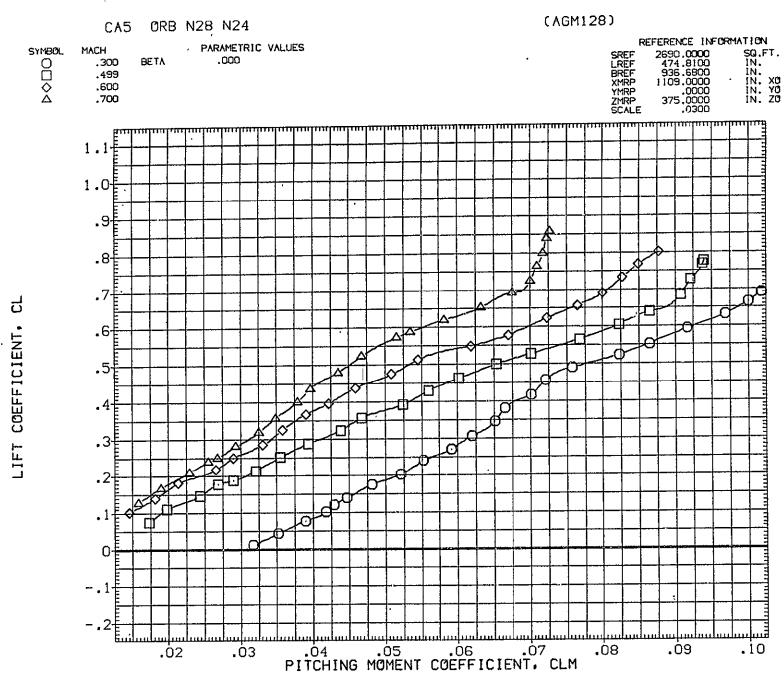


FIG. 98 ORBITER ALONE, TAIL CONE OFF

.04 .05 .06 .07 PITCHING MOMENT COEFFICIENT, CLM

FIG. 98 ORBITER ALONE, TAIL CONE OFF

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eó.

.08

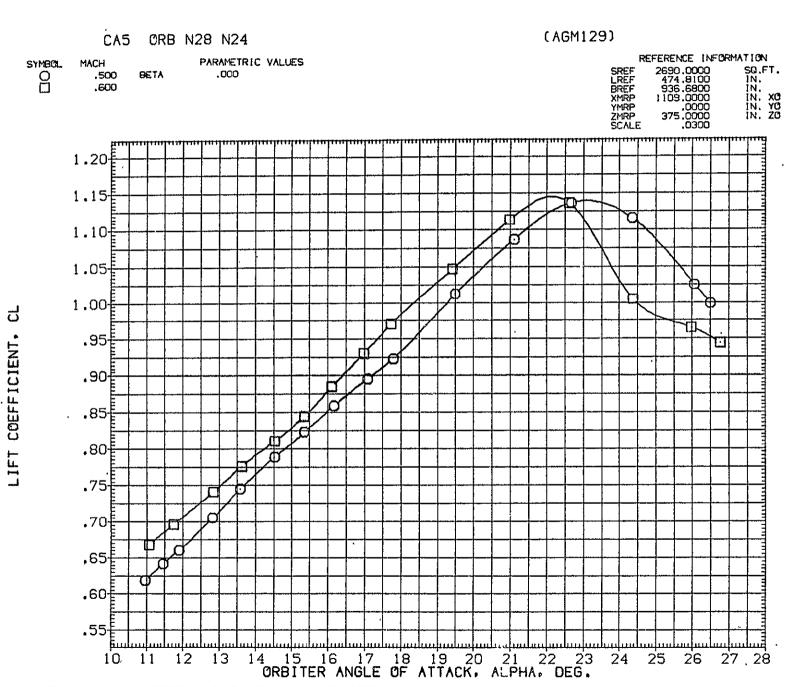


FIG. 98 ORBITER ALONE, TAIL CONE OFF

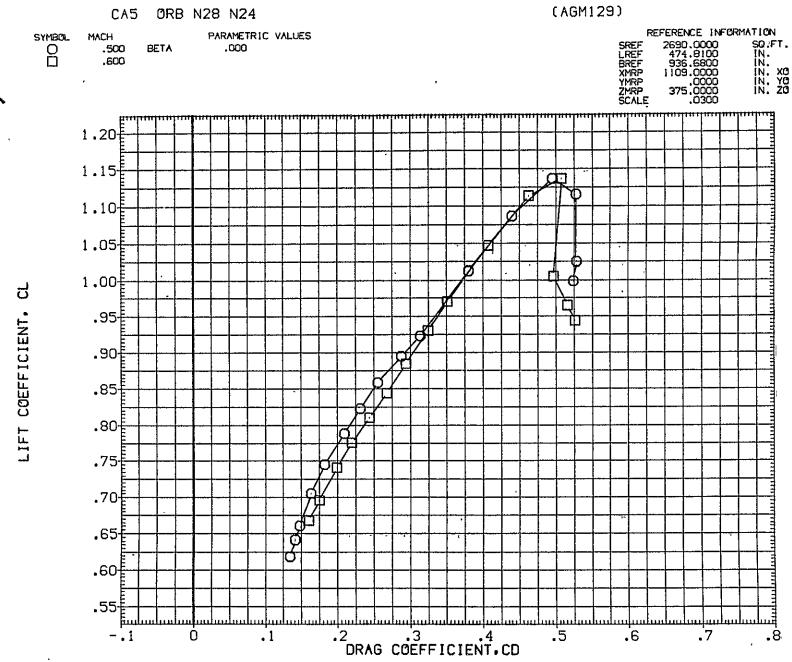


FIG. 98 ORBITER ALONE, TAIL CONE OFF

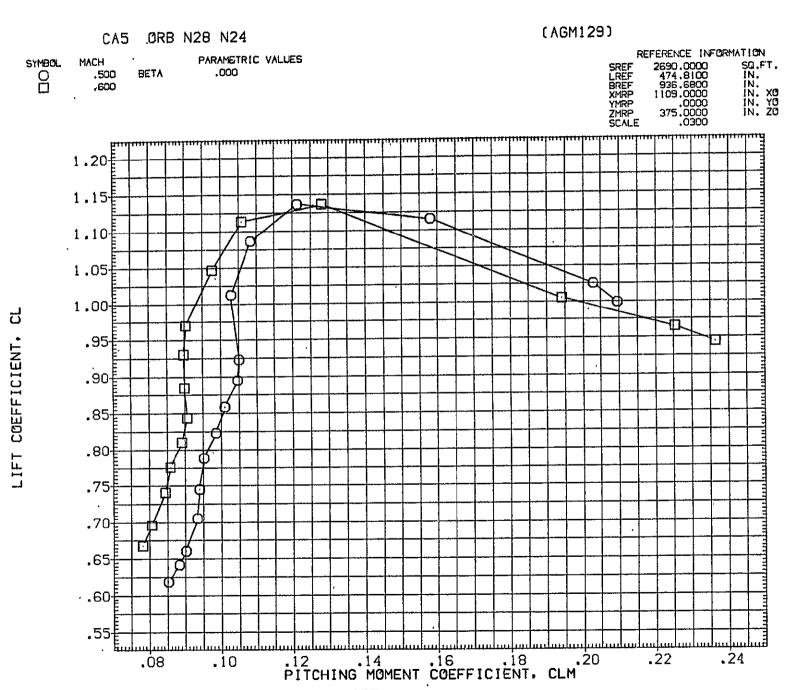


FIG. 98 ORBITER ALONE, TAIL CONE OFF

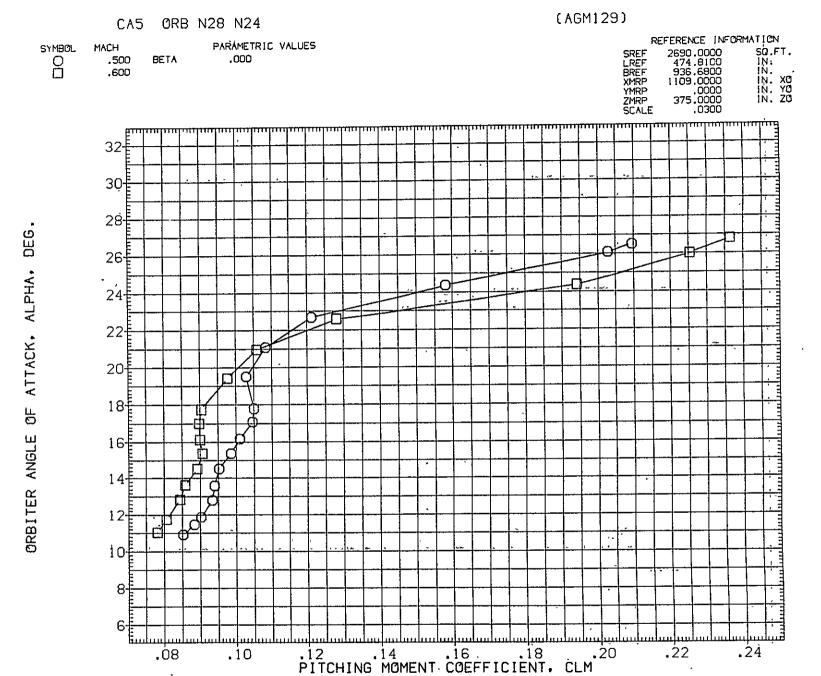


FIG. 98 ORBITER ALONE, TAIL CONE OFF

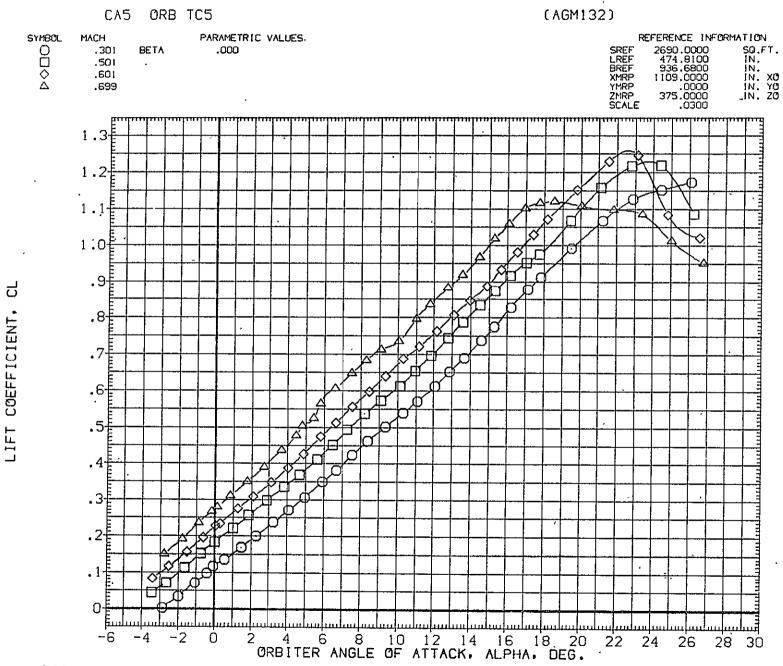


FIG. 99 ORBITER ALONE, TAIL CONE ON

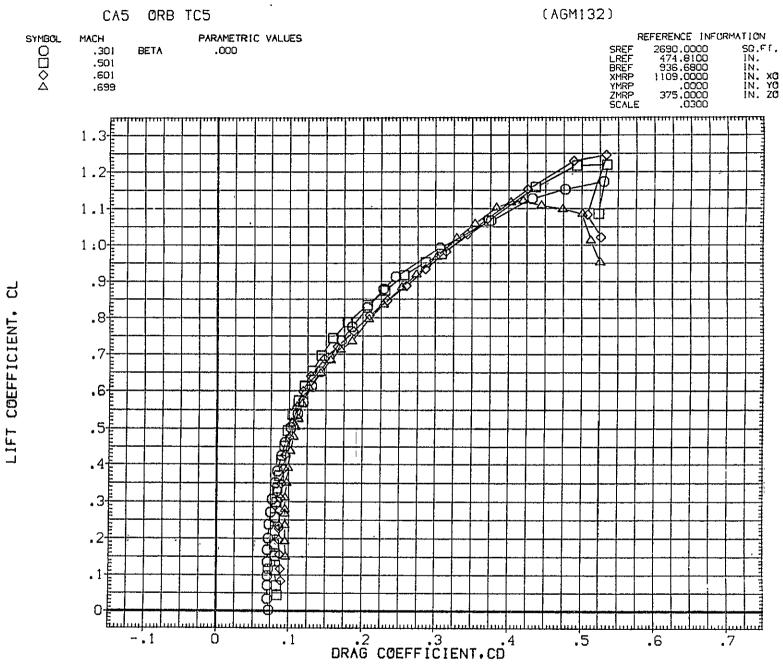


FIG. 99 ORBITER ALONE, TAIL CONE ON

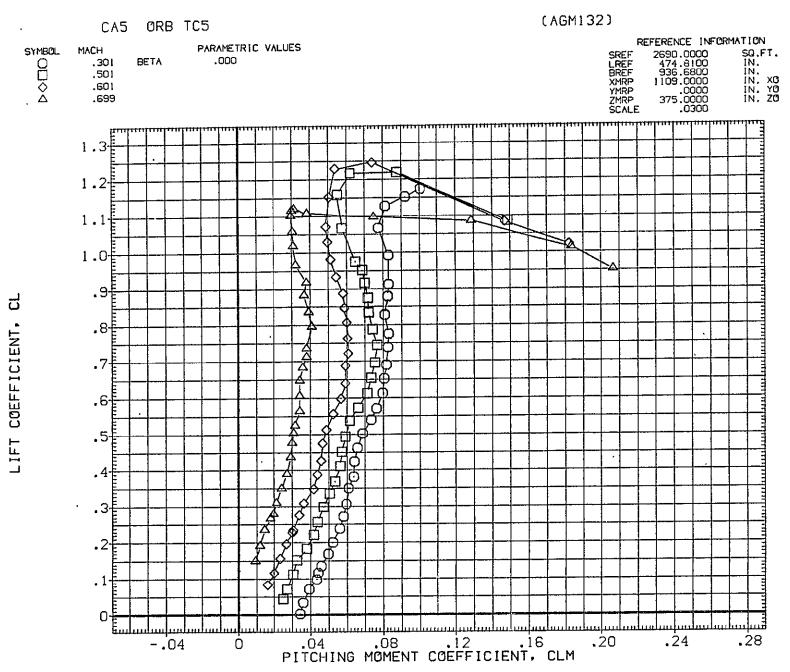


FIG. 99 ORBITER ALONE, TAIL CONE ON

FIG. 99 ORBITER ALONE, TAIL CONE ON

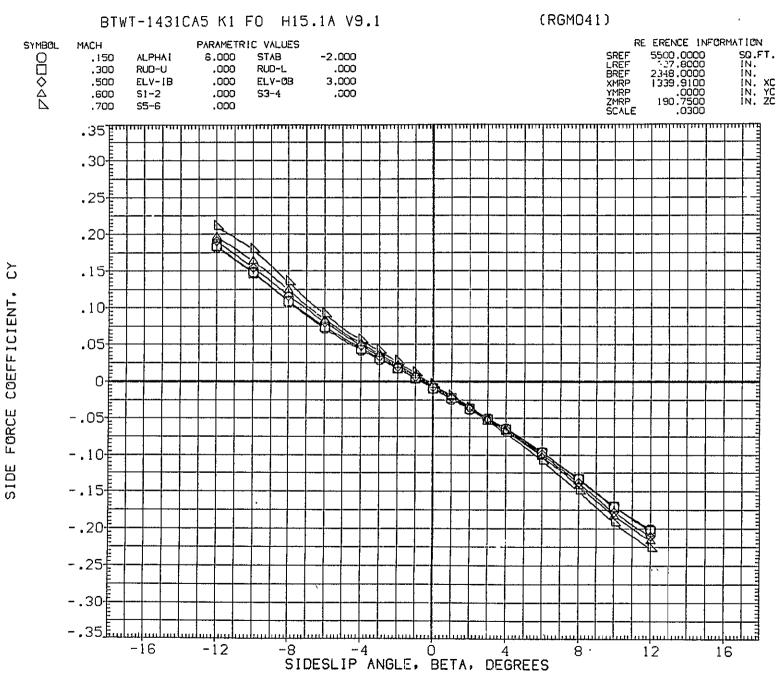


FIG.100 747 ALONE, BASIC AIRPLANE, FLAPS UP, TAIL ON, ALPHAI=6 DEG

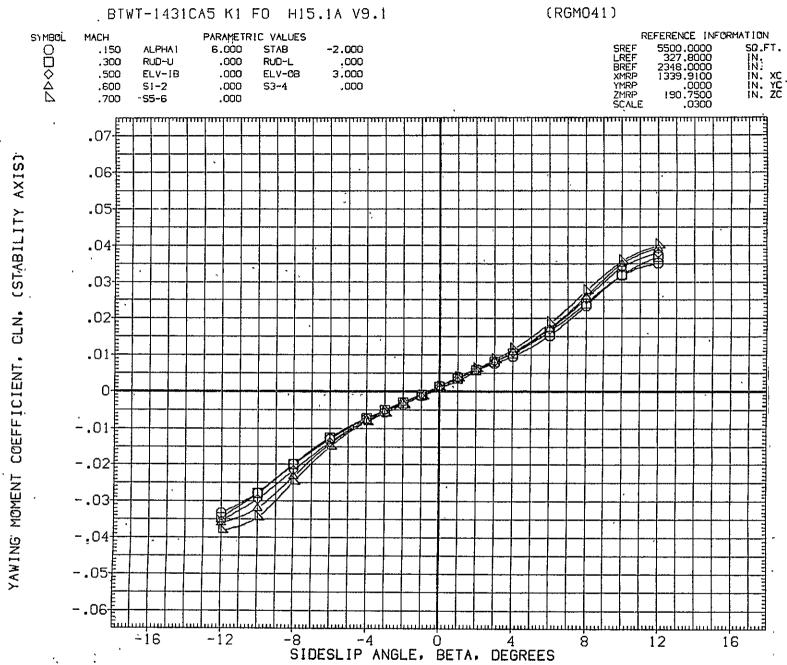
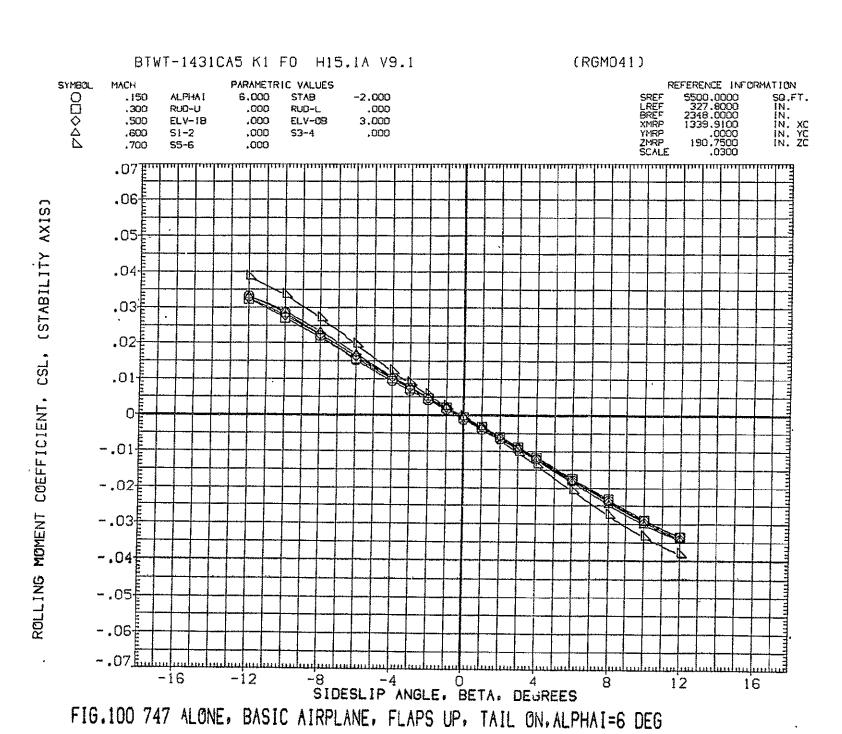


FIG.100 747 ALONE, BASIC AIRPLANE, FLAPS UP, TAIL ON, ALPHAI=6 DEG



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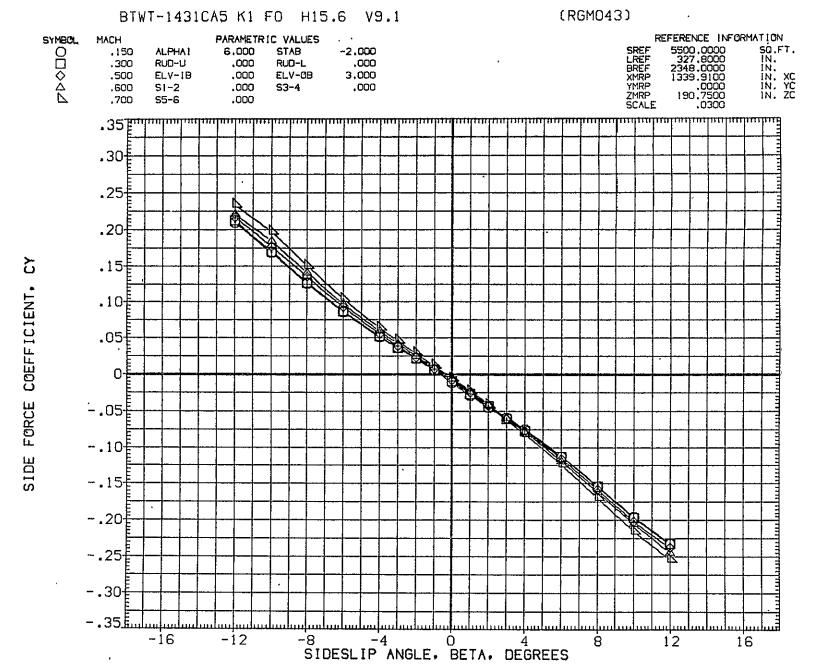


FIG.101 747 ALONE+ TIP FINS, FLAPS UP, TAIL ON, ALPHAI= 6 DEG .

FIG.101 747 ALONE+ TIP FINS, FLAPS UP, TAIL ON, ALPHAI= 6 DEG

FIG.101 747 ALONE+ TIP FINS, FLAPS UP, TAIL ON, ALPHAI = 6 DEG

-4

SIDESLIP ANGLE, BETA, DEGREES

-16

-12

16

12



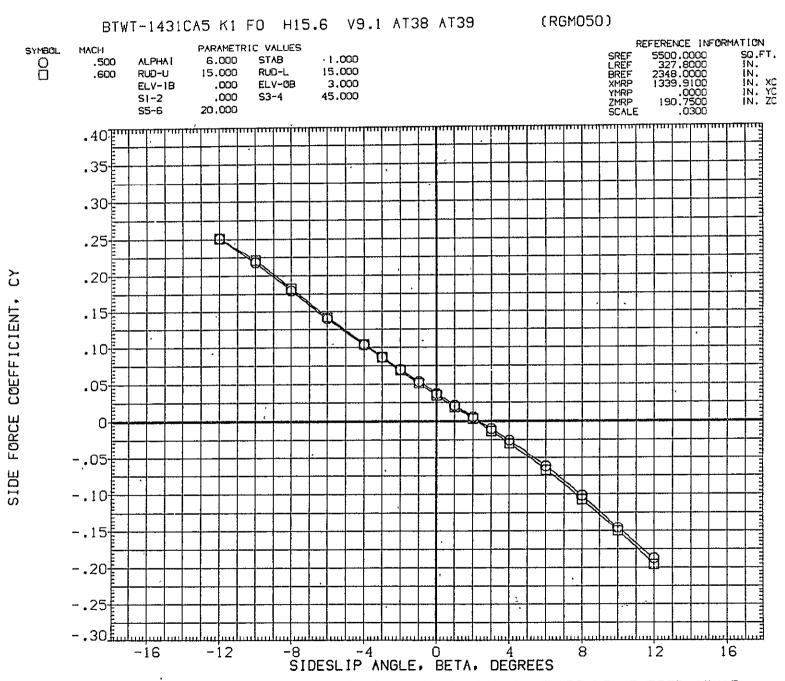


FIG.102 747 ALONE+ TIP FINS+STD SPD BRK+ORB SUPPORTS, FLAPS UP, RUDDER=15/15

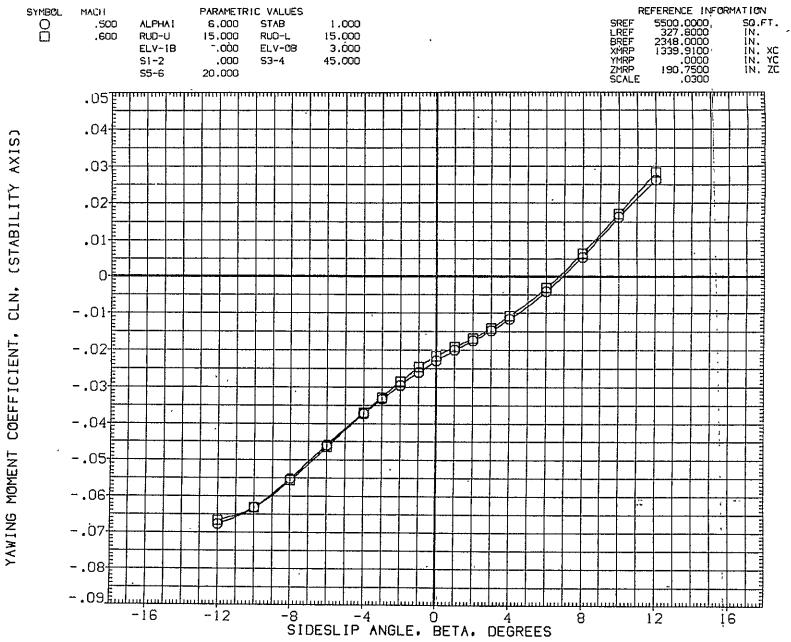


FIG.102 747 ALONE+ TIP FINS+STD SPD BRK+ORB SUPPORTS, FLAPS UP, RUDDER=15/15

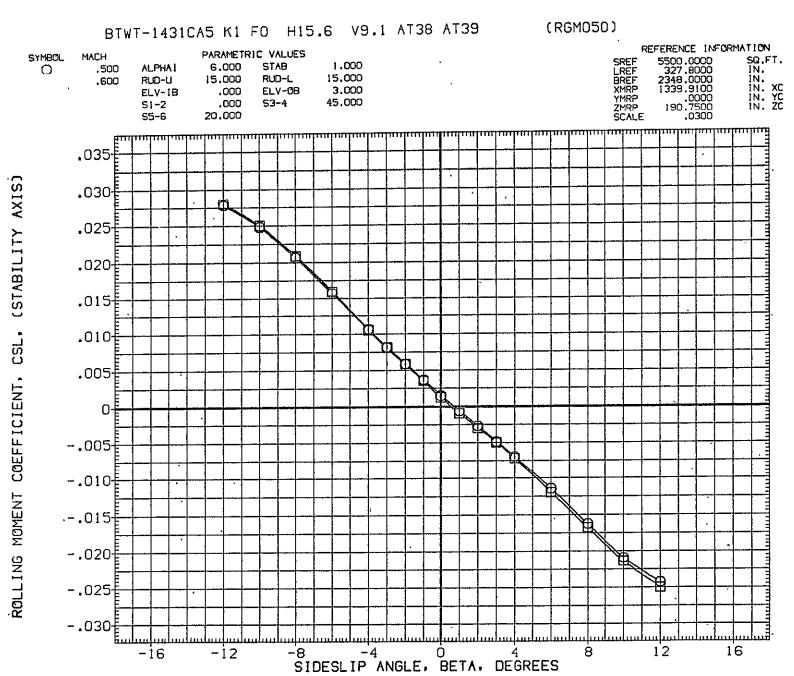


FIG.102 747 ALONE+ TIP FINS+STD SPD BRK+ORB SUPPORTS, FLAPS UP, RUDDER=15/15

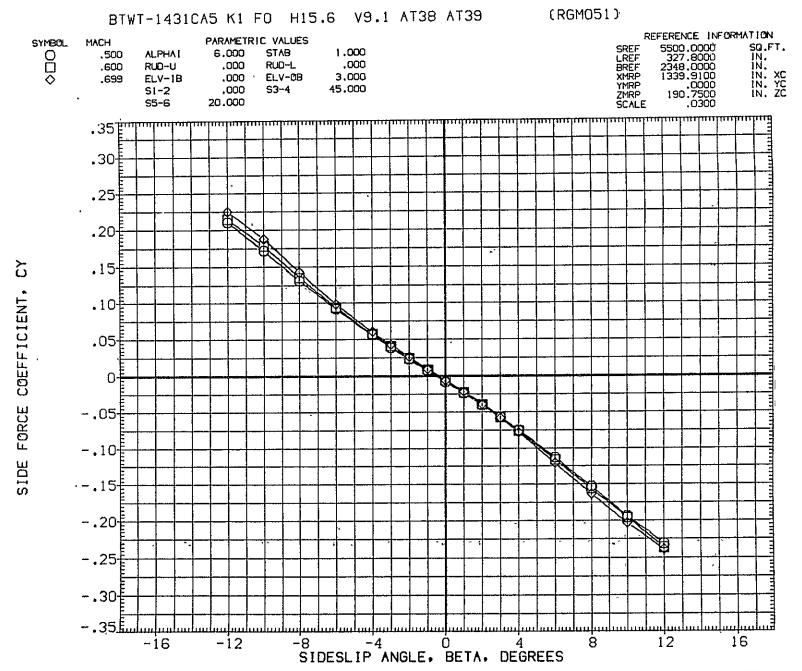


FIG.103 747 ALONE+ TIP FINS+STD SPD BRK+ORB SUPPORTS, FLAPS UP, ALPHAI= 6 DEG



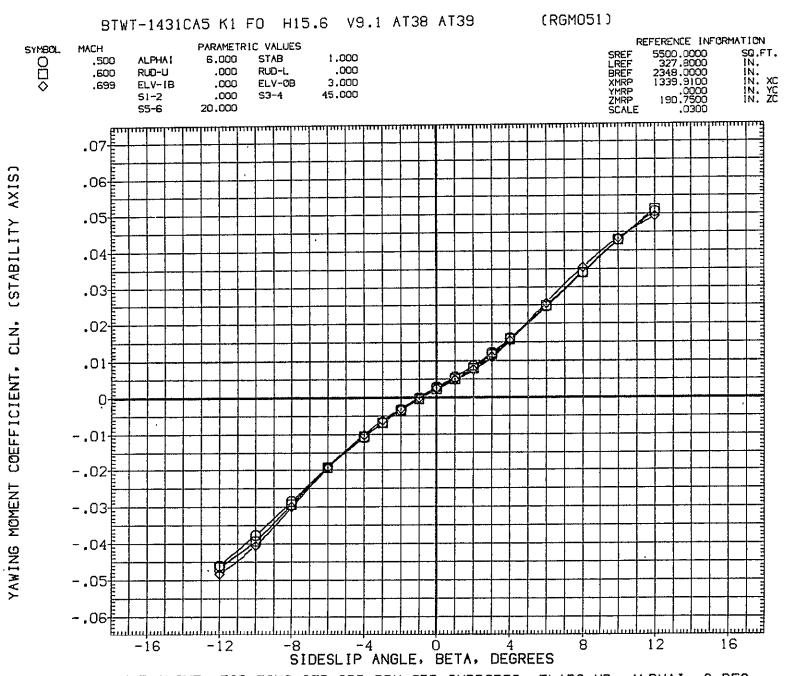


FIG.103 747 ALONE+ TIP FINS+STD SPD BRK+ORB SUPPORTS, FLAPS UP, ALPHAI= 6 DEG

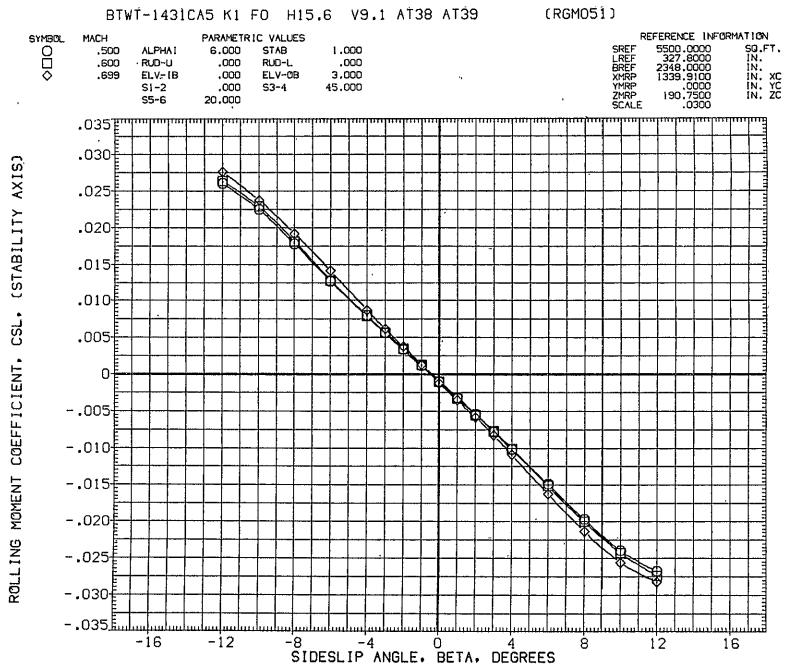


FIG.103 747 ALONE+ TIP FINS+STD SPD BRK+ORB SUPPORTS, FLAPS UP, ALPHAI= 6 DEG

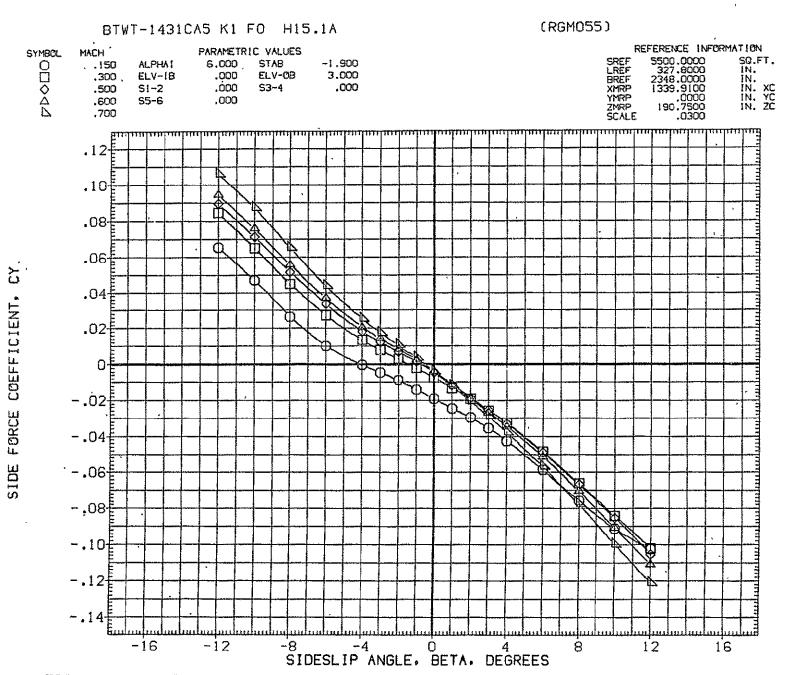


FIG.104 747 ALONE, VERTICAL TAIL OFF, FLAPS 20, ALPHAI = 6 DEG (S=-1.9)

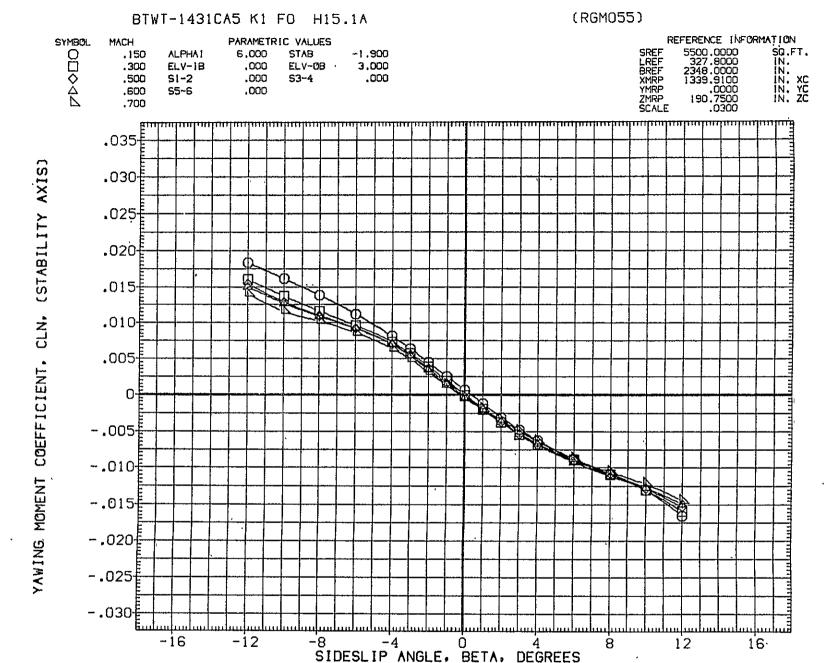


FIG.104 747 ALONE, VERTICAL TAIL OFF, FLAPS 20, ALPHAI = 6 DEG (S=-1.9)



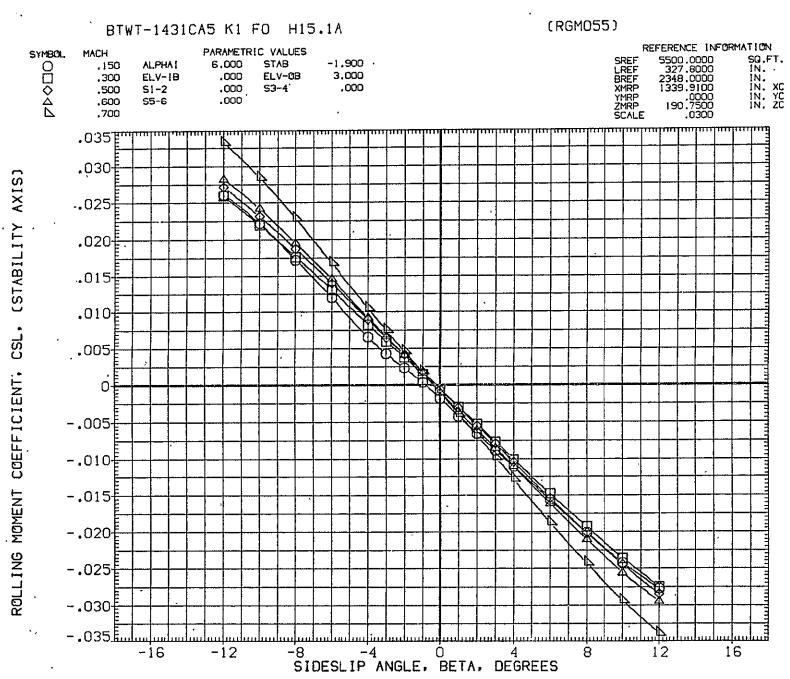
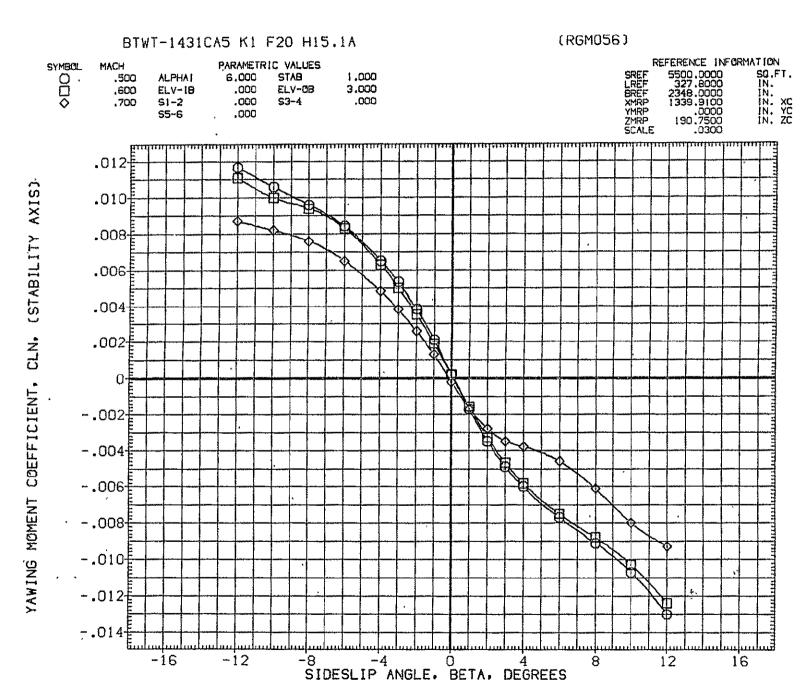


FIG. 104 747 ALONE, VERTICAL TAIL OFF, FLAPS 20, ALPHAI = 6 DEG (S=-1.9)

FIG.105 747 ALONE, VERTICAL TAIL OFF, FLAPS 20, ALPHAI = 6 DEG (S=+1)

-12

-8 -4 0 4
SIDESLIP ANGLE, BETA, DEGREES



-FIG.105 747 ALONE, VERTICAL TAIL OFF, FLAPS 20, ALPHAI= 6 DEG (S=+1)

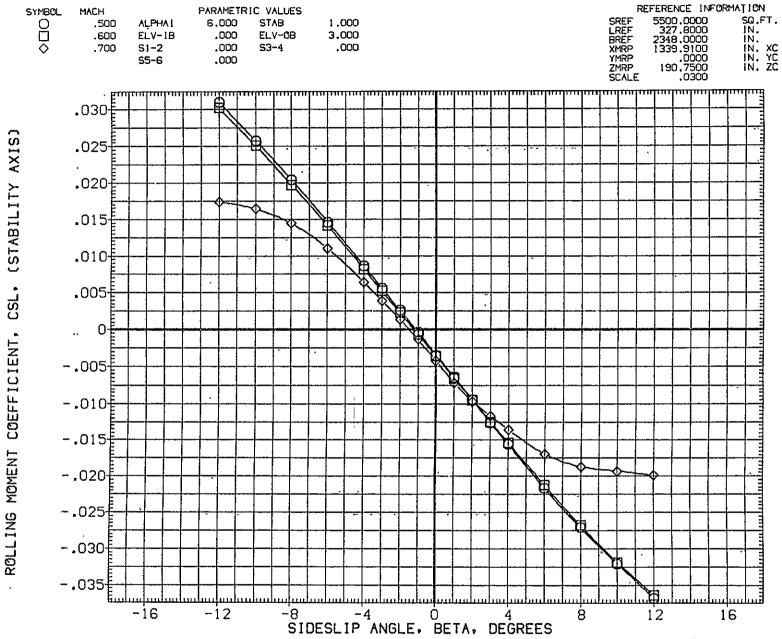


FIG.105 747 ALONE, VERTICAL TAIL OFF, FLAPS 20, ALPHAI= 6 DEG (S=+1)

FIG.106 747 ALONE, BASIC AIRPLANE, FLAPS 20, ALPHAI= 6 DEG (S=+1)

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SIDESLIP ANGLE, BETA, DEGREES

-8

SYMBOL.

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FORCE COEFFICIENT,

SIDE

-.20

-.25E

-:30

-.35 E...

-16

-12

16

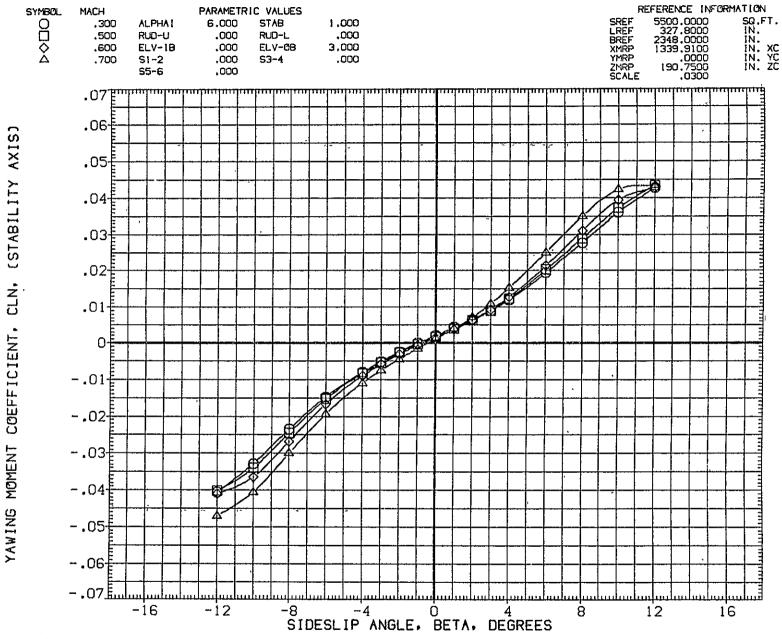


FIG.106 747 ALONE, BASIC AIRPLANE, FLAPS 20, ALPHAI= 6 DEG (S=+1)



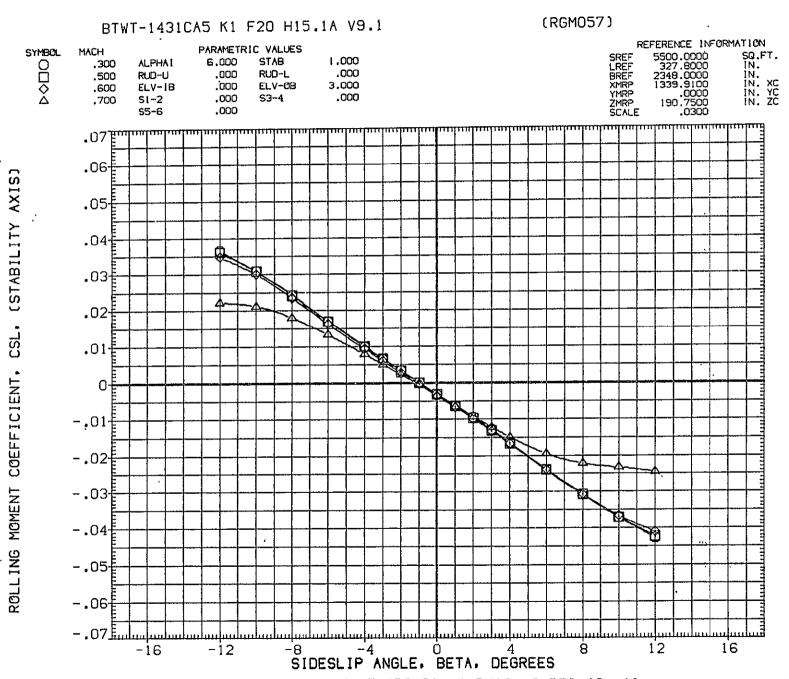


FIG. 106 747 ALONE, BASIC AIRPLANE, FLAPS 20, ALPHAI = 6 DEG (S=+1)

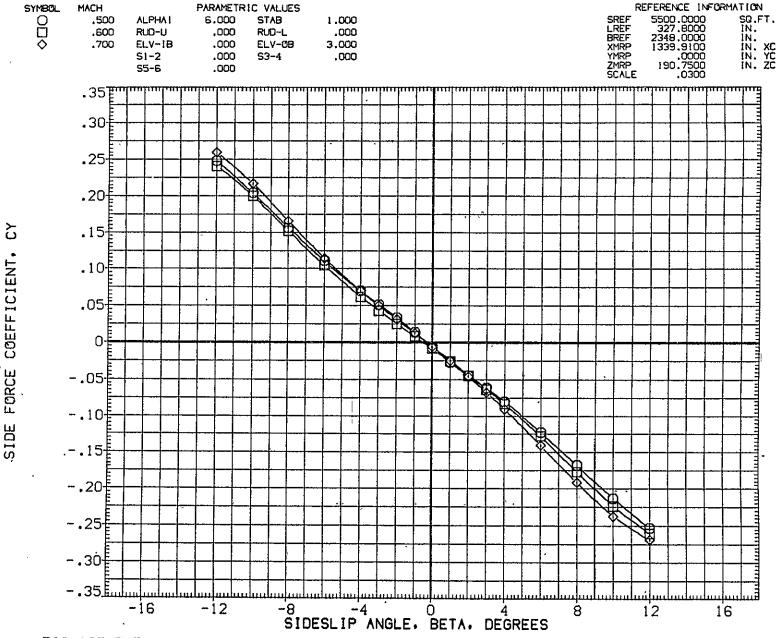
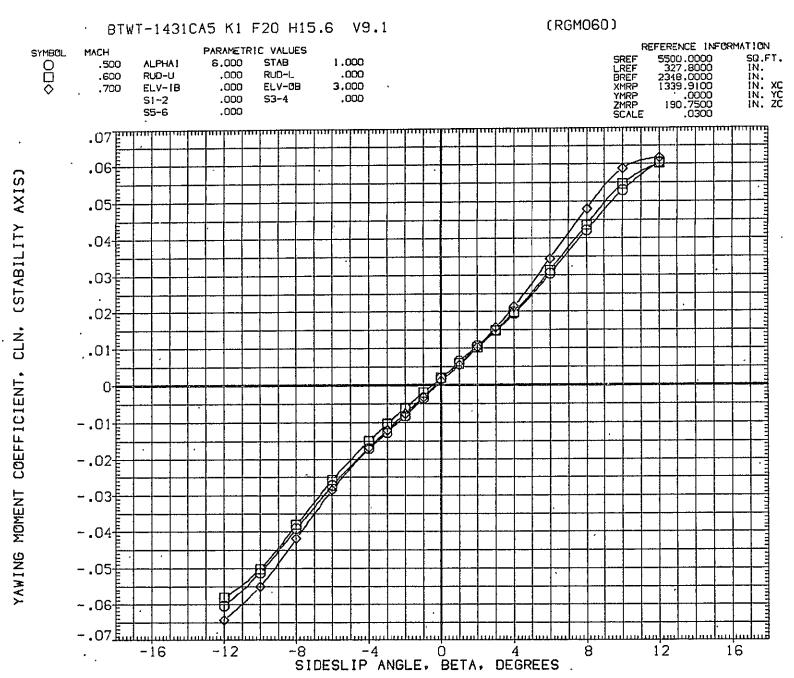


FIG.107 747 ALGNE+ TIP FINS, FLAPS 20, ALPHAI= 6 DEG (S=+1)



·FIG.107 747 ALONE+ TIP FINS, FLAPS 20, ALPHAI= 6 DEG (S=+1)

FIG.107 747 ALØNE+ TIP FINS, FLAPS 20, ALPHAI= 6 DEG (S=+1)

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SIDESLIP ANGLE, BETA, DEGREES

-12

-.07<u>E</u>…

16

-:35<u>E...l...</u>

-16

-12

CA5 K1 F20 H15.6 V9.1 (PLUS. ORB TC5 AT38.1)(RGMO67) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH SREF LREF BREF XMRP YMRP ZMRP SCALE SQ.FT. 5500,0000 ALPHA I 6.000 STAB 1.000 .500 IN. IN. IN. XC IN. YC IN. ZC 327.8000 2348.0000 .000 RUD-L .000 .600 RUD-U ELV-18 .000 ELV-0B 3.000 .700 S!-2 45,000 53-4 45.000 9,700 **S5-6** 45.000 LORG .35Tm .30€ .25 .20 .15[COEFFICIENT, .10 .05[0 FORCE -.05[-.10[SIDE -.15[-.20[-.25[-.30f

-8 -4 0 4
SIDESLIP ANGLE, BETA, DEGREES FIG.108 747+0RB(10 DEG)+TIP FINS+SP45.FLAPS 20 LAUNCH.BASELINE.ALPHAI= 6 TOTAL PAGE 425

12

16

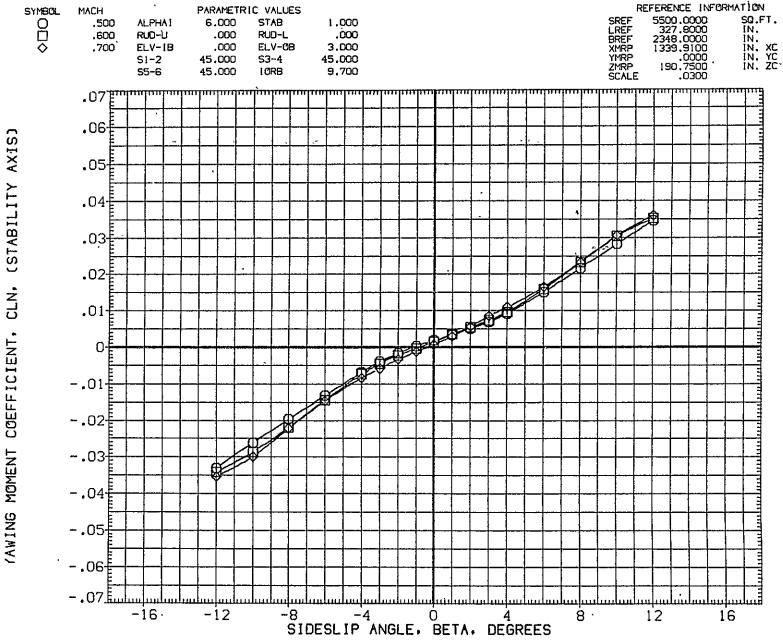


FIG.108 747+0RB(10 DEG)+TIP FINS+SP45,FLAPS 20 LAUNCH,BASELINE,ALPHAI= 6 TOTAL
PAGE 426

CA5 K1 F20 H15.6 V9.1 (PLUS. ORB TC5 AT38.1)(RGMO67)

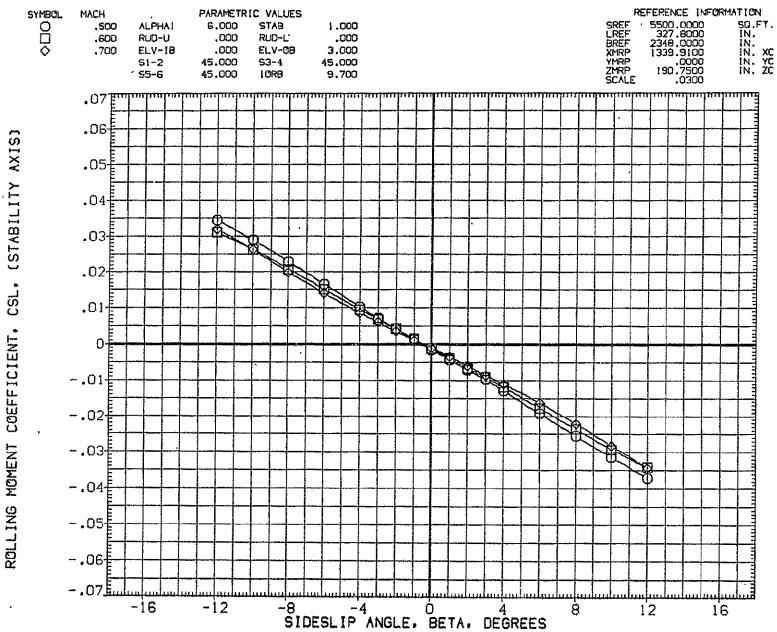


FIG.108 747+0RB(10 DEG)+TIP FINS+SP45, FLAPS 20 LAUNCH, BASELINE, ALPHAI= 6 TOTAL
PAGE 427

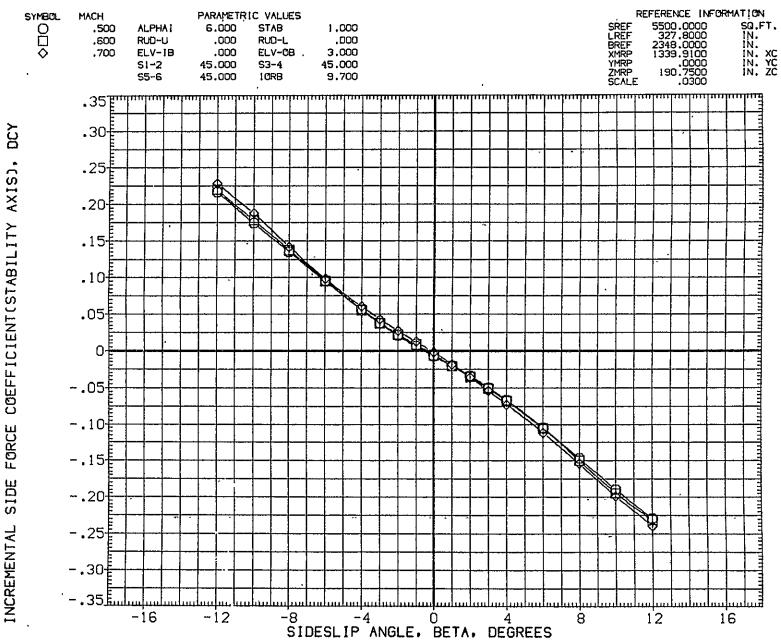


FIG.109 747+0RB(10 DEG)+TIP FINS+SP45,FLAPS 20 LAUNCH, BASELINE, ALPHAI= 6 747

PAGE 428

CA5 K1 F20 H15.6 V9.1 MINUS ORB TC5 AT38.1 (BGM067)

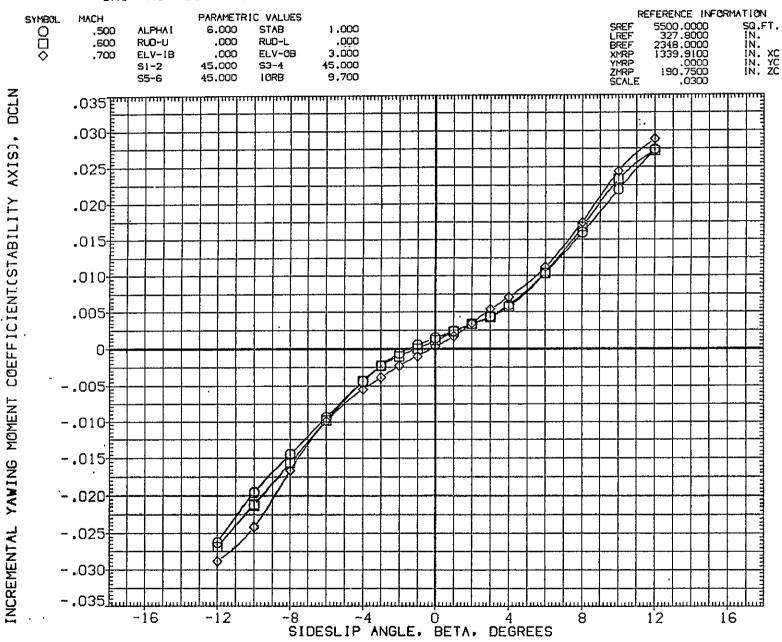


FIG.109 747+0RB(10 DEG)+TIP FINS+SP45,FLAPS 20 LAUNCH,BASELINE,ALPHAI= 6
PAGE 429

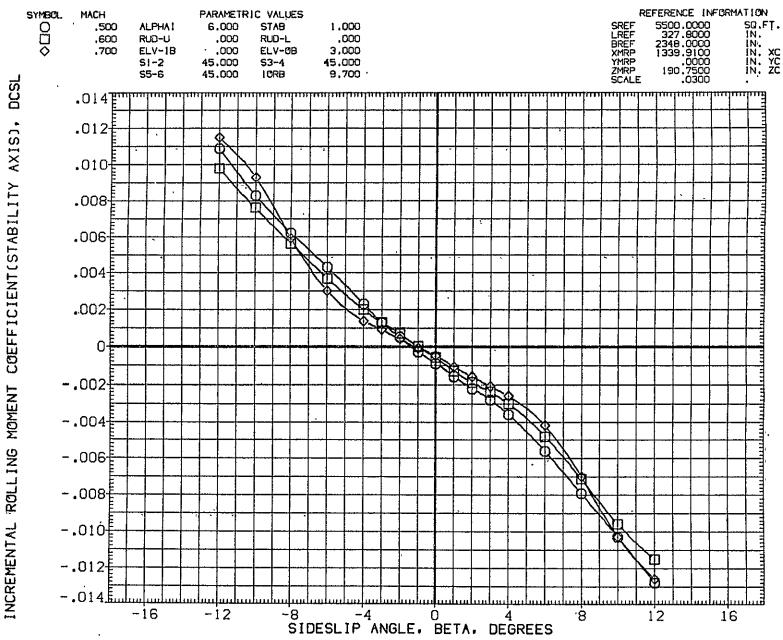


FIG.109 747+ORB(10 DEG)+TIP FINS+SP45,FLAPS 20 LAUNCH,BASELINE,ALPHAI= 6 PAGE 430

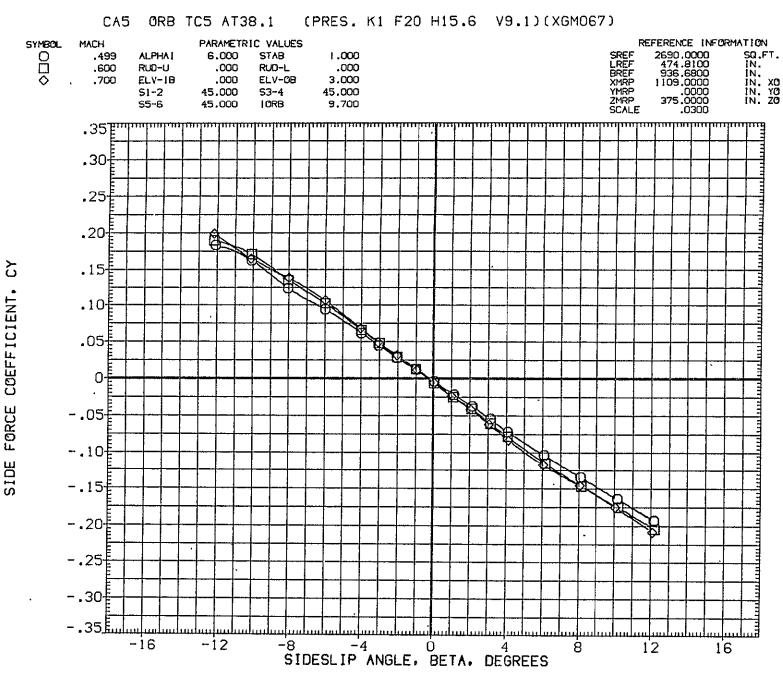


FIG.110 747+0RB(10 DEG)+TIP FINS+SP45,FLAPS 20 LAUNCH,BASELINE,ALPHAI= 6 ORB

CA5 ORB TC5 AT38.1 (PRES. K1 F20 H15.6 V9.1)(XGMO67) REFERENCE INFORMATION SYMBOL PARAMETRIC VALUES SREF LREF BREF 2690.0000 000 6.000 1.000 .499 **ALPHAI** STAB 474.8100 936.6800 1109.0000 IN. .000 RUD-L .000 .600 RUÐ-U IN. XO IN. YO IN. ZO ELV-1B .000 ELV-0B 3.000 XMRP YMRP ZMRP SCALE S1-2 45.000 S3-4 45.000 LORB 9.700 · S5-6 45.000 .030[.025 **(STABILITY** .020 .015 .010[CLN. .005 YAWING MOMENT COEFFICIENT. 0-..005₽ ·.010 .015 ..020 -.025 -.030 -.035<u>E</u>

FIG.110 747+0RB(10 DEG)+TIP FINS+SP45,FLAPS 20 LAUNCH,BASELINE,ALPHAI= 6
PAGE 432

SIDESLIP ANGLE, BETA, DEGREES

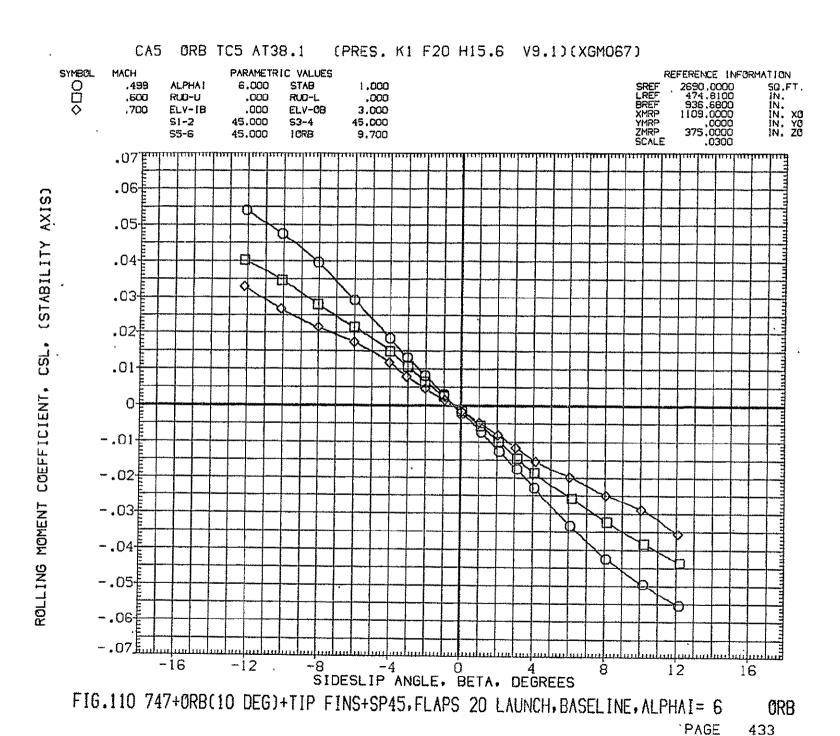
12

16

-16

-12

-.8



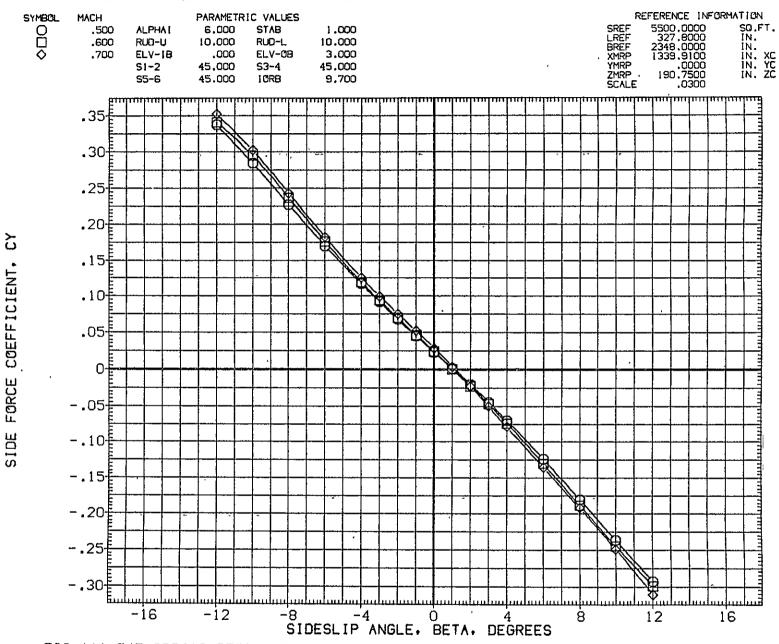


FIG.111 747+0RB(10 DEG)+TIP FINS+SP45,FLAPS 20 LNCH,ALPHAI=6,RUDDER=10/10 TOTAL 3E 434



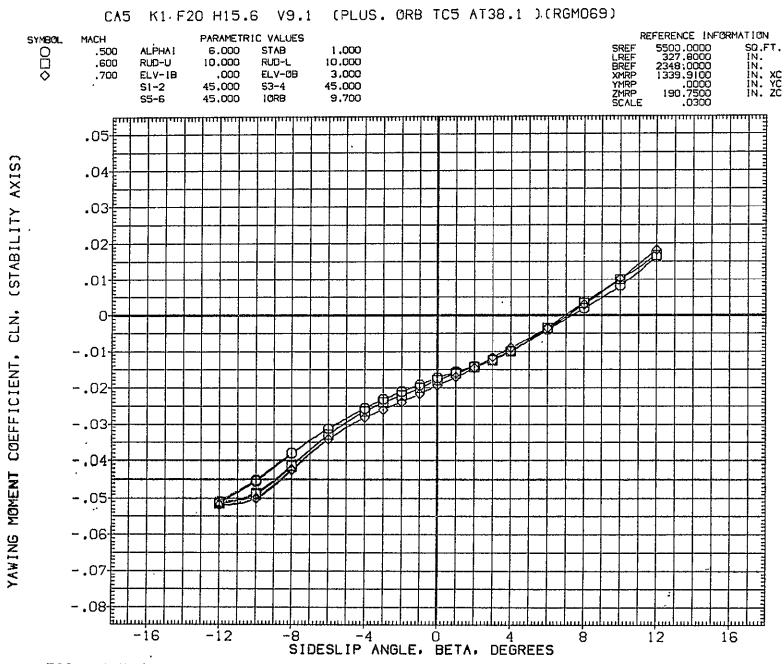


FIG.111 747+0RB(10 DEG)+TIP FINS+SP45,FLAPS 20 LNCH,ALPHAI=6,RUDDER=10/10 TOTAL PAGE 435

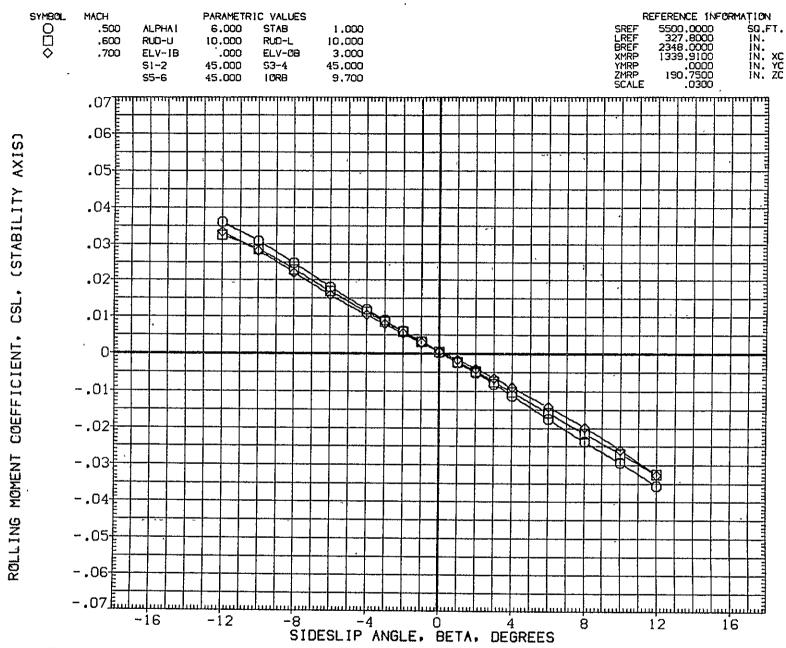


FIG.111 747+0RB(10 DEG)+TIP FINS+SP45.FLAPS 20 LNCH.ALPHAI=6.RUDDER=10/10 TOTAL PAGE 436

ORB TC5 AT38.1 (BGM069) CA5 K1 F20 H15.6 V9.1 MINUS PARAMETRIC VALUES REFERENCE INFORMATION SYMBOL MACH SREF LREF BREF XMRP YMRP ZMRP SCALE 5500,0000 327,8000 2348,0000 1339,9100 SQ.FT. 000 .500 ALPHA! 6,000 STAB 1,000 IN. IN. IN. XC IN. YC IN. ZC .600 RUD-U 10,000 RUD-L 10.000 .700 ELV-1B .000 ELV-08 3.000 .0000 190.7500 0300 S3-4 45.000 S1-2 45,000 45,000 10RB 9.700 S5~6 .35 DCY .30 COEFFICIENT(STABILITY AXIS), .25 .20 .15 .10 .05[0 SIDE FORCE -.10[INCREMENTAL -.30E -16 -12 12 16

SIDESLIP ANGLE, BETA, DEGREES
FIG.112 747+ORB(10 DEG)+TIP FINS+SP45, FLAPS 20 LNCH, ALPHAI=6, RUDDER=10/10 7

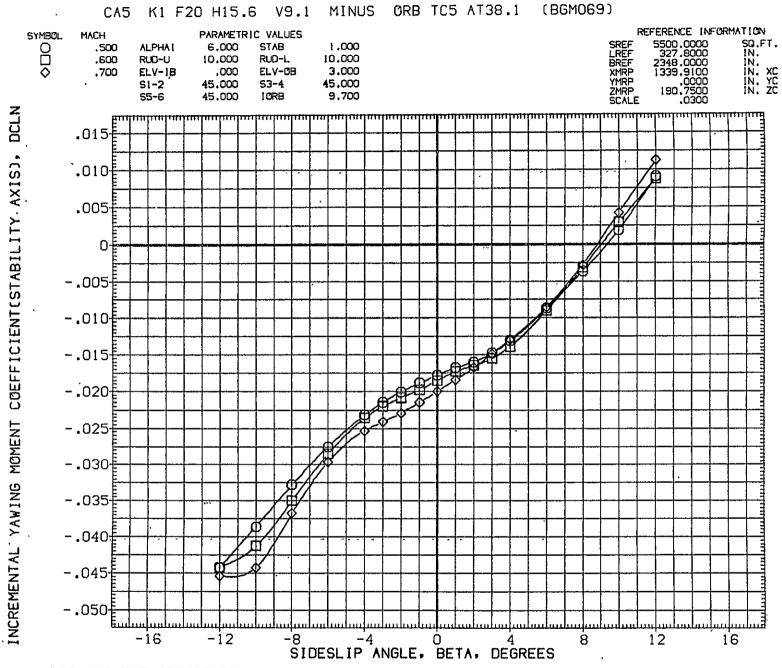


FIG.112 747+0RB(10 DEG)+TIP FINS+SP45,FLAPS 20 LNCH,ALPHAI=6,RUDDER=10/10 747

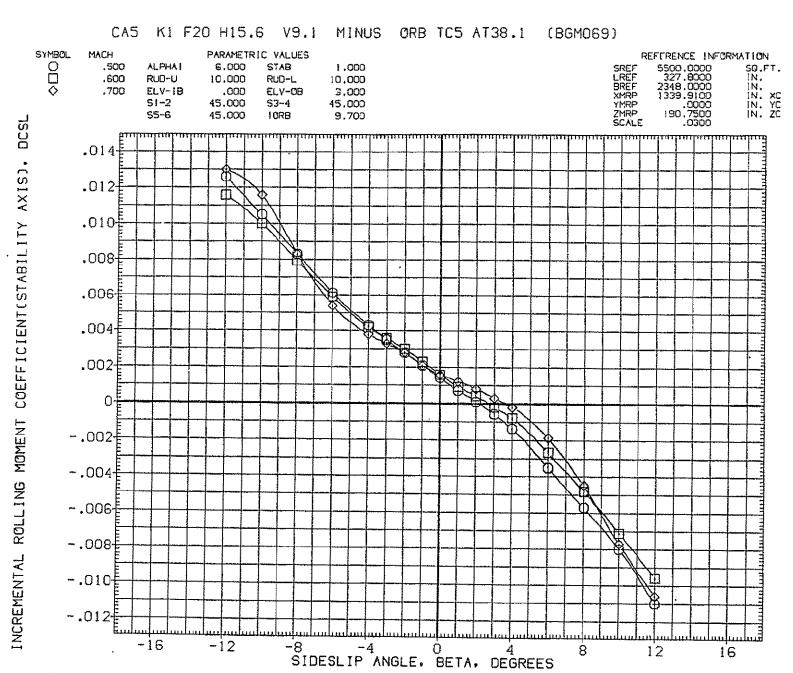


FIG.112 747+0RB(10 DEG)+TIP FINS+SP45,FLAPS 20 LNCH, ALPHAI=6, RUDDER=10/10 747

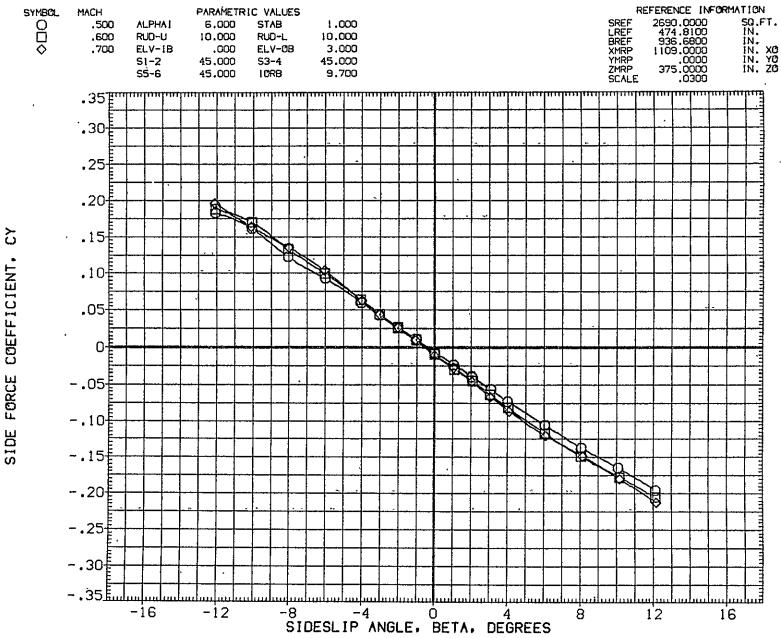


FIG.113 747+0RB(10 DEG)+TIP FINS+SP45.FLAPS 20 LNCH.ALPHAI=6.RUDDER=10/10 ORB

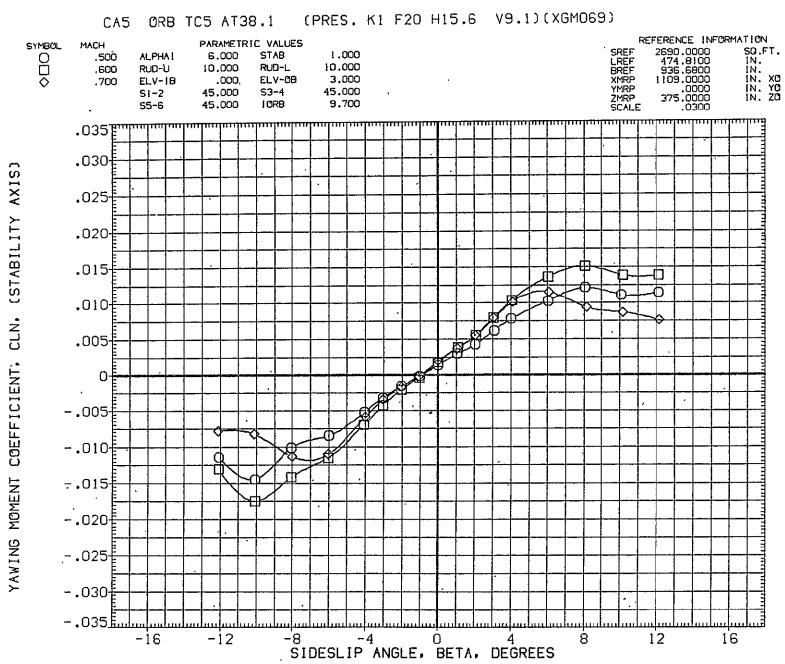


FIG.113 747+0RB(10 DEG)+TIP FINS+SP45,FLAPS 20 LNCH, ALPHAI=6, RUDDER=10/10 ORB

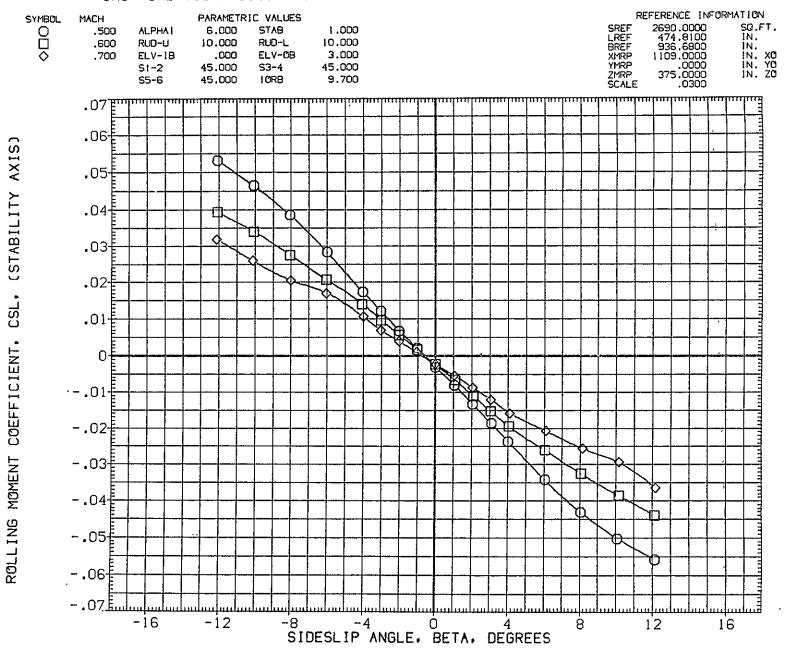


FIG.113 747+0RB(10 DEG)+TIP FINS+SP45,FLAPS 20 LNCH,ALPHAI=6,RUDDER=10/10 ORB

CA5 K1 F20 H15.6 V9.1 (PLUS. ORB TC5 AT38.1)(RGMO72)

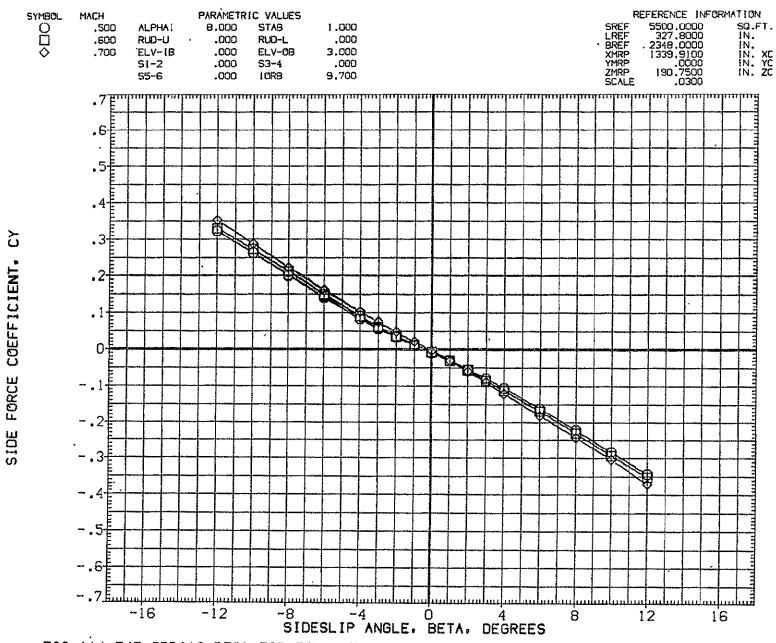


FIG.114 747+0RB(10 DEG)+TIP FINS, FLAPS 20 LAUNCH, ALPHAI= 8 DEG

TOTAL

CA5 K1 F20 H15.6 V9.1 (PLUS. ORB TC5 AT38.1)(RGM072)

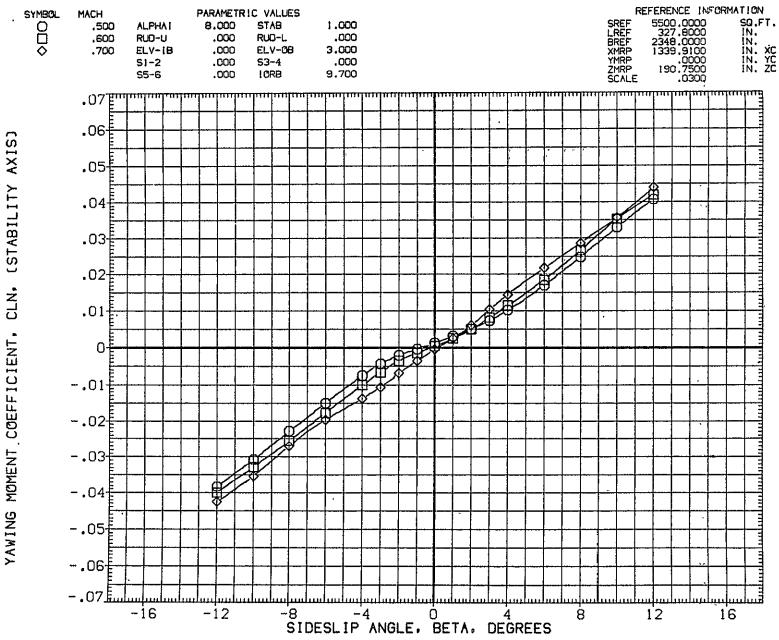


FIG.114 747+0RB(10 DEG)+TIP FINS, FLAPS 20 LAUNCH, ALPHAI= 8 DEG

TOTAL

CA5 K1 F20 H15.6 V9.1 (PLUS. ORB TC5 AT38.1)(RGM072) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH SREF LREF BREF XMRP YMRP ZMRP SCALE SQ.FT, 5500,0000 327.8000 000 ALPHA I 8,000 STAB 1.000 .500 .000 RUD-L .000 2348.0000 1339.9100 .0000 190.7500 .0300 IN. XC IN. XC IN. YC IN. ZC .600 RUD-U 3.000 .700 ELV-18 .000 ELV-08 .000 S1-2 .000 53-4 10RB 9,700 55-6 .000 .07Fm .06‡ (STABILITY AXIS) 盘 .05# . .04 .03 .02 CSL, .01 ROLLING MOMENT COEFFICIENT, 0. -.01[-.02[-.03[-.04 - .05 -.06 - 07<u>L...</u> -12 12 -16 -8 16 -4 Ò 8 SIDESLIP ANGLE, BETA, DEGREES FIG.114 747+0RB(10 DEG)+TIP FINS, FLAPS 20 LAUNCH, ALPHAI= 8 DEG

TOTAL

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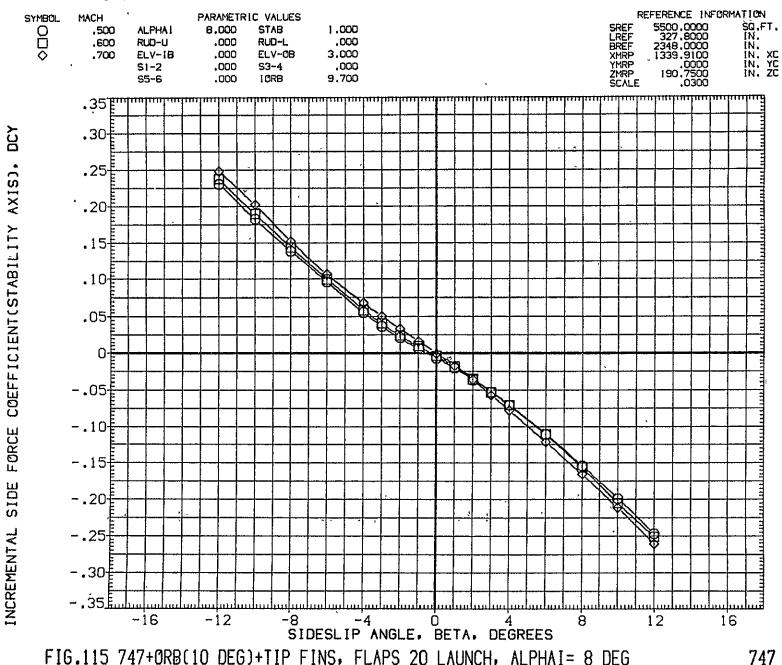


FIG.115 747+0RB(10 DEG)+TIP FINS, FLAPS 20 LAUNCH, ALPHAI= 8 DEG

-12 -16 12 -8 16 8 SIDESLIP ANGLE, BETA, DEGREES FIG.115 747+0RB(10 DEG)+TIP FINS, FLAPS 20 LAUNCH, ALPHAI= 8 DEG 747 PAGE 447

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DCLN

YAWING MOMENT COEFFICIENT(STABILITY AXIS),

INCREMENTAL

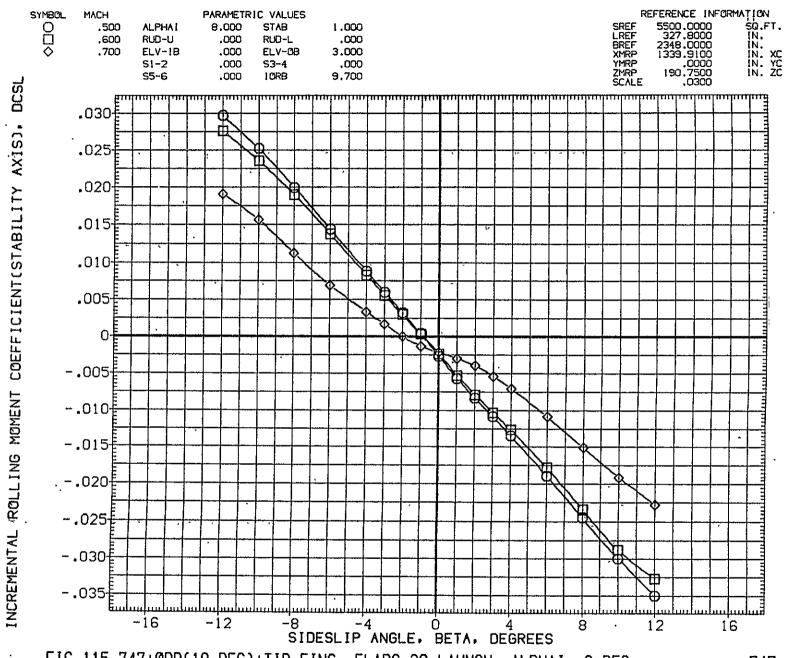
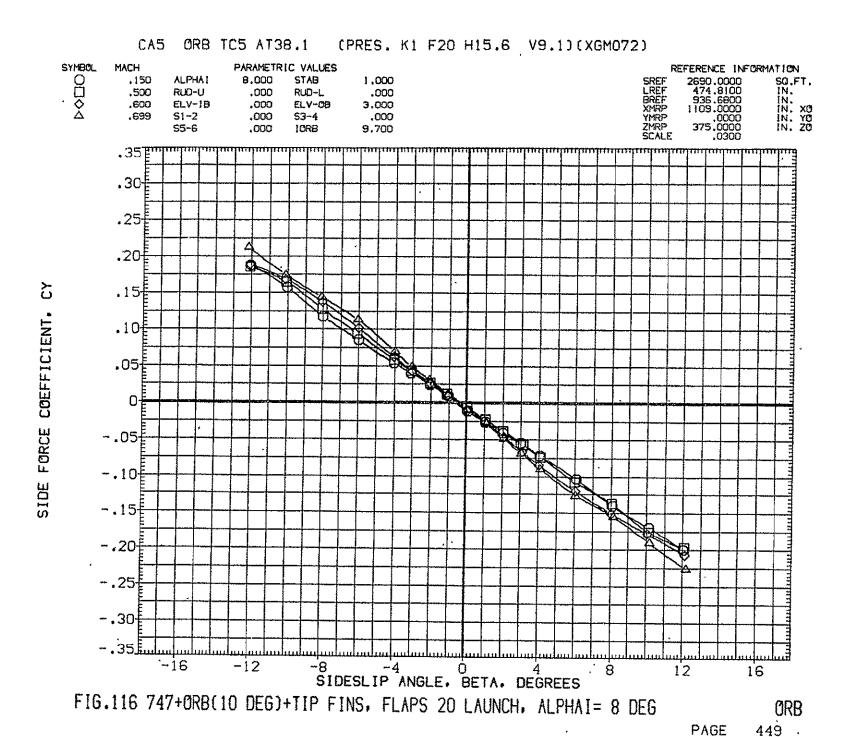


FIG.115 747+0RB(10 DEG)+TIP FINS, FLAPS 20 LAUNCH, ALPHAI= 8 DEG



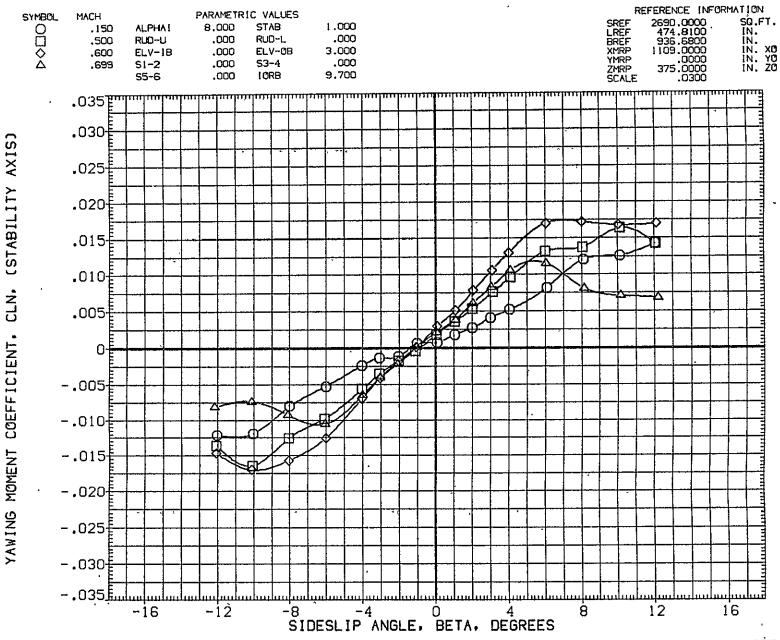


FIG.116 747+ORB(10 DEG)+TIP FINS, FLAPS 20 LAUNCH, ALPHAI= 8 DEG



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CA5 K1 F20 H15.6 V9.1 (PLUS. ØRB TC5 AT38.1)(RGM073)

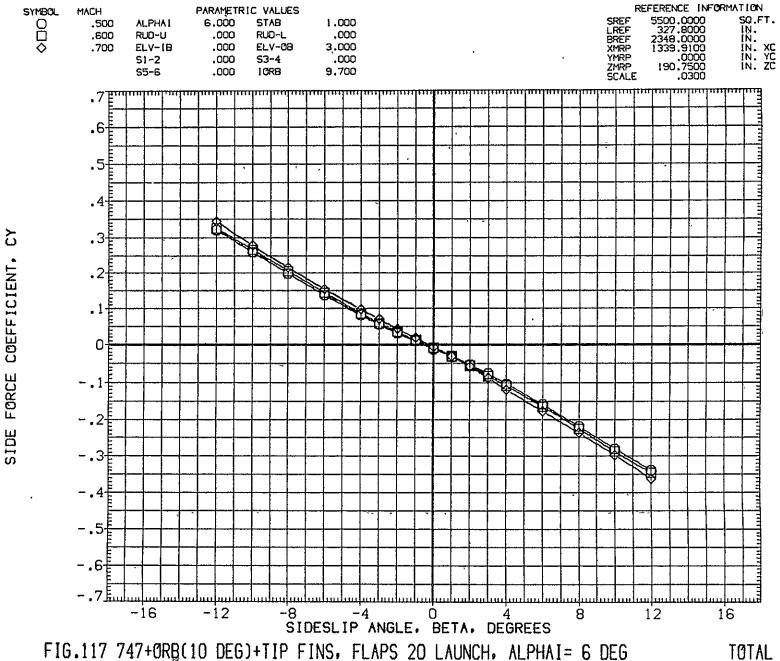


FIG.117 747+0RB(10 DEG)+TIP FINS, FLAPS 20 LAUNCH, ALPHAI= 6 DEG

CA5 K1 F20 H15.6 V9.1 (PLUS. ORB TC5 AT38.1)(RGMO73)

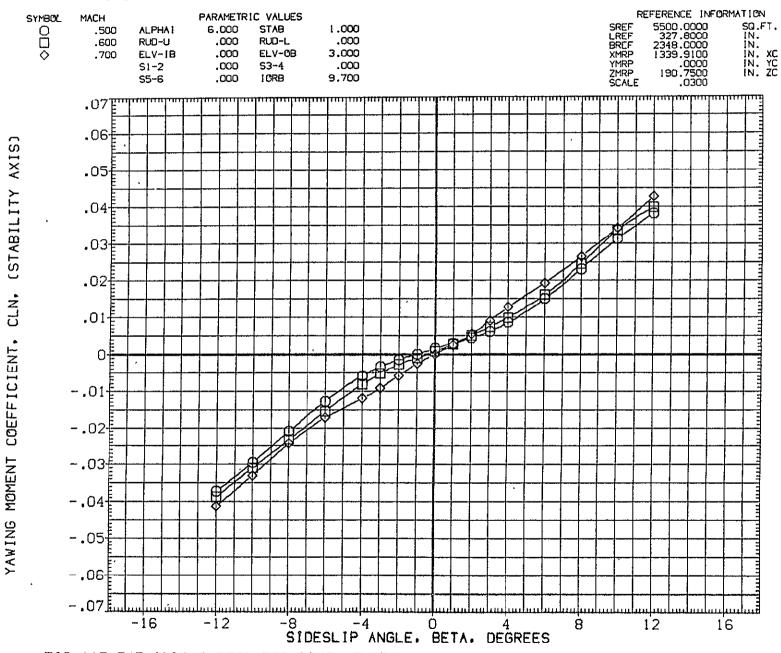
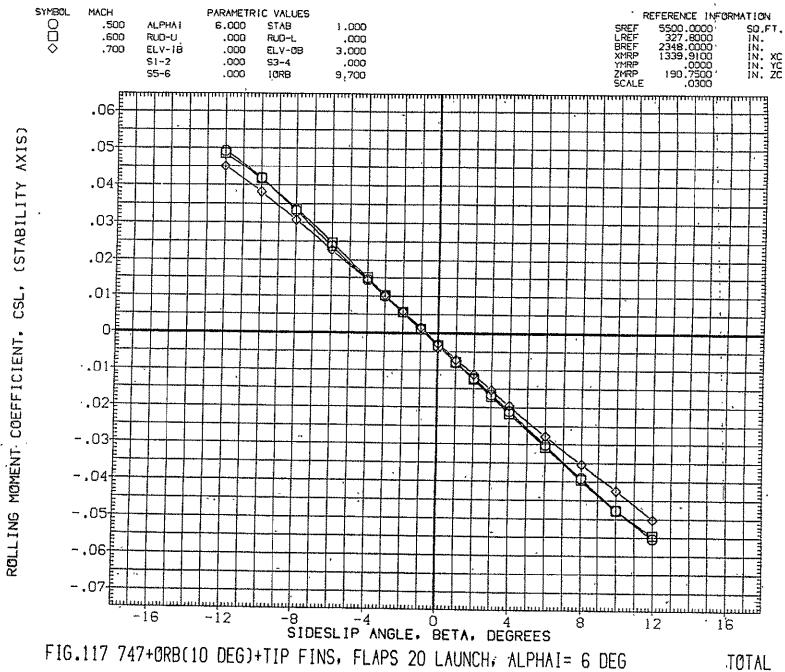
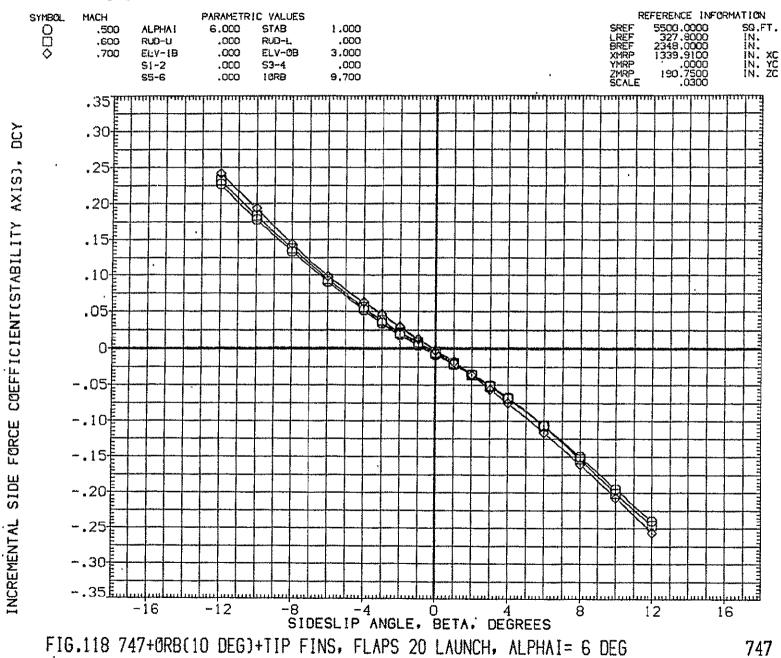


FIG.117 747+0RB(10 DEG)+TIP FINS, FLAPS 20 LAUNCH, ALPHAI= 6 DEG



CA5 K1 F20 H15.6 V9.1 MINUS ORB TC5 AT38.1 (BGM073)



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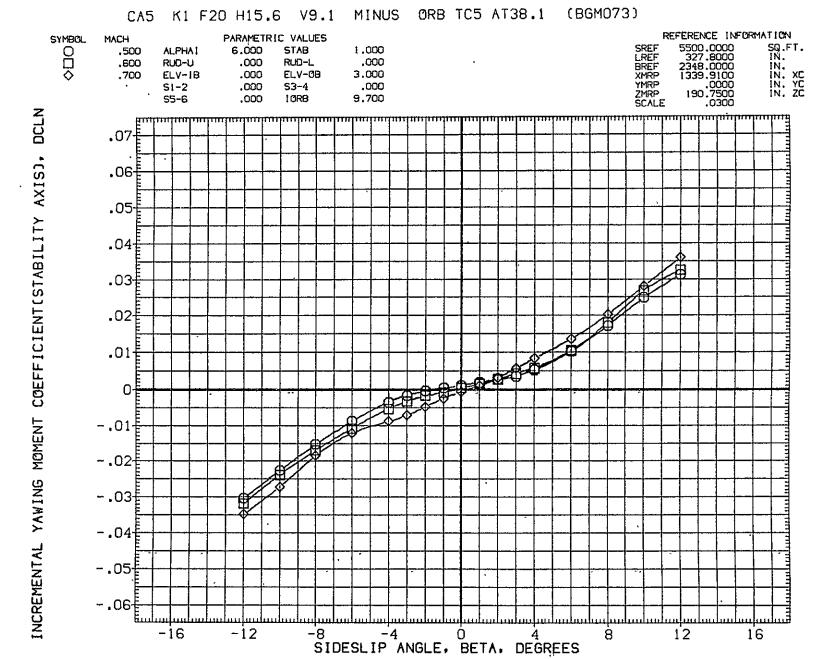
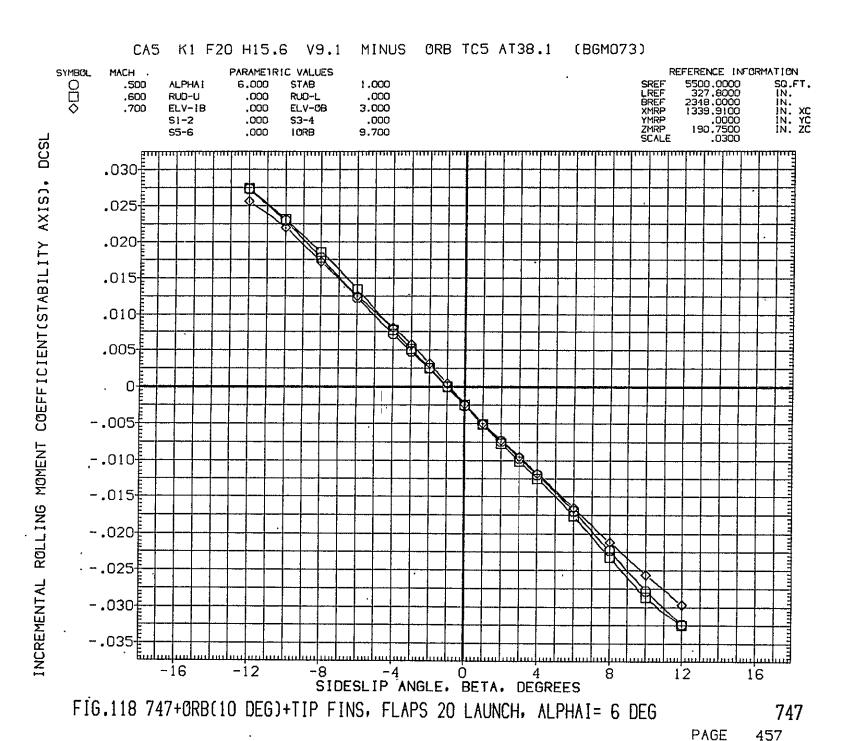


FIG.118 747+0RB(10 DEG)+TIP FINS, FLAPS 20 LAUNCH, ALPHAI= 6 DEG

747



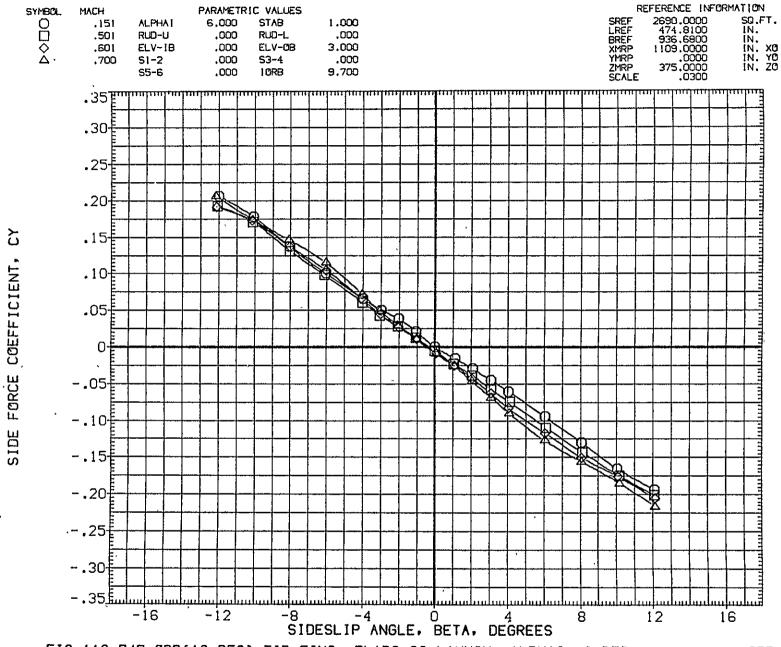
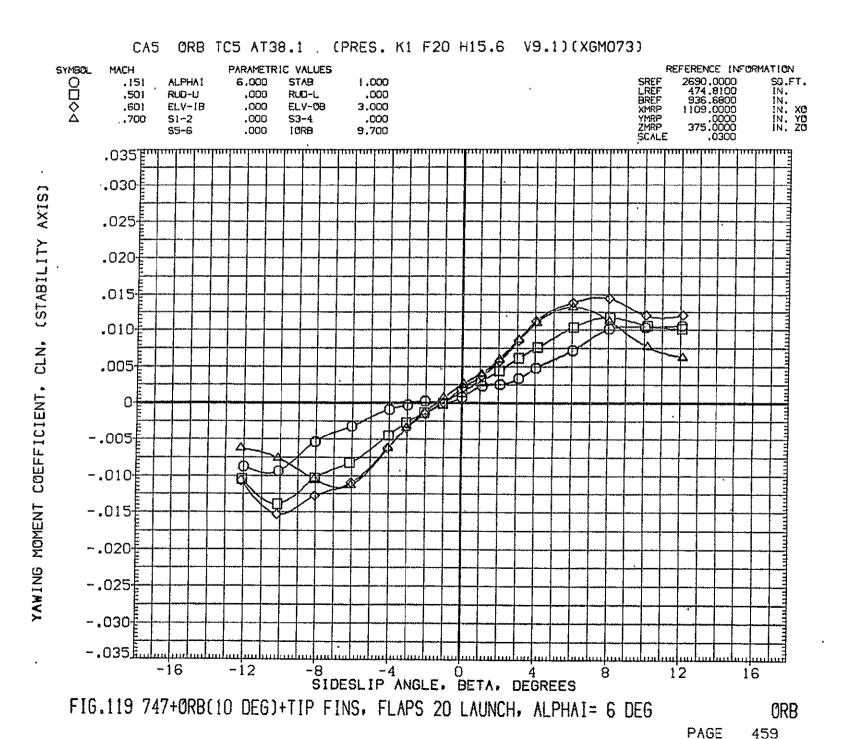


FIG.119 747+0RB(10 DEG)+TIP FINS, FLAPS 20 LAUNCH, ALPHAI= 6 DEG



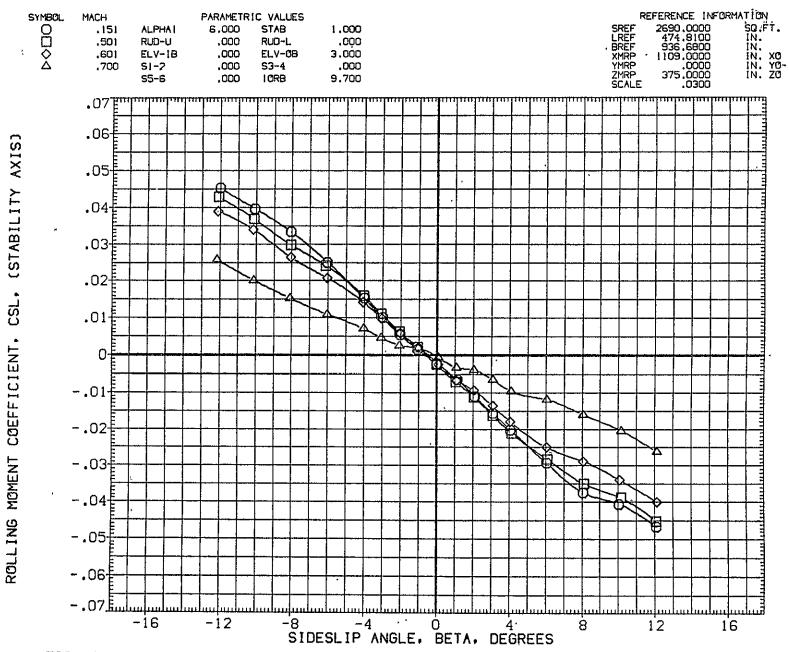
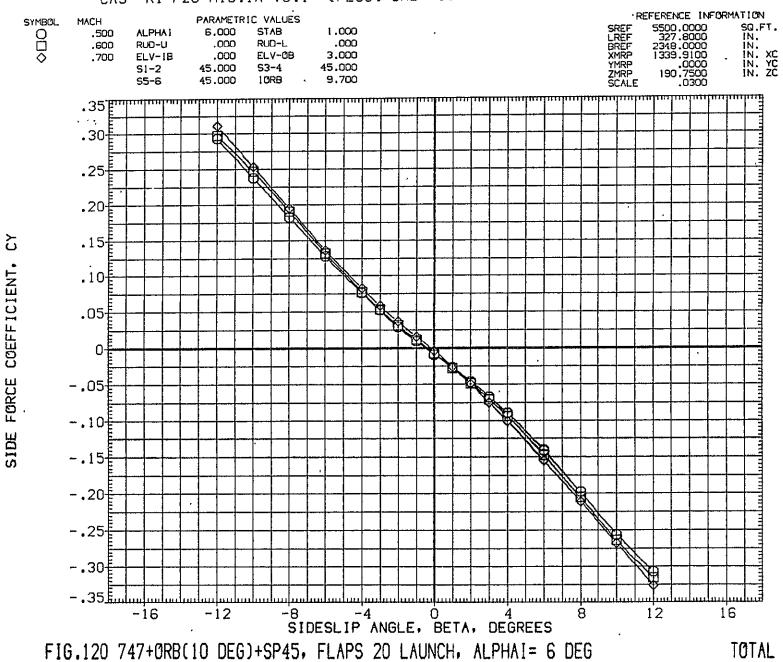


FIG.119 747+0RB(10 DEG)+TIP FINS, FLAPS 20 LAUNCH, ALPHAI= 6 DEG

CA5 K1 F20 H15.1A V9.1 (PLUS. ØRB TC5 AT38.1)(RGM076)



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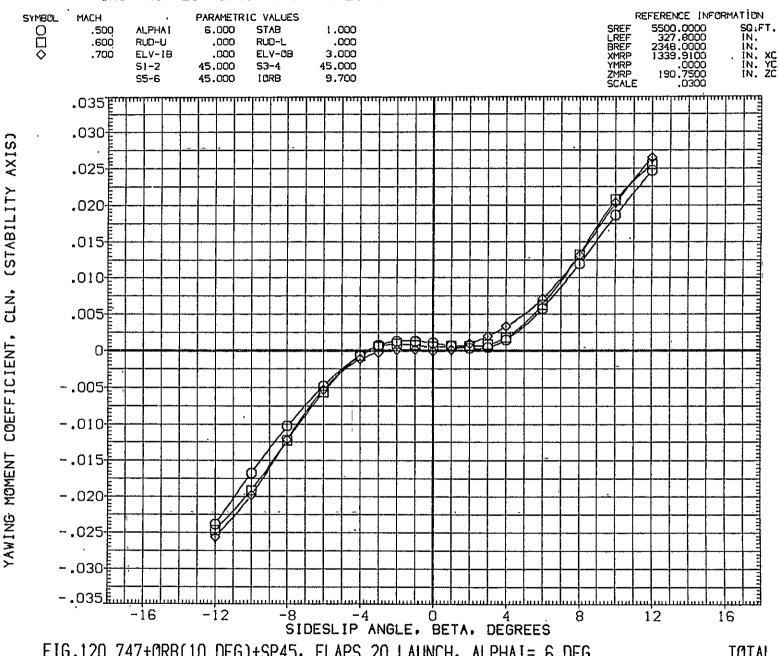


FIG.120 747+0RB(10 DEG)+SP45, FLAPS 20 LAUNCH, ALPHAI= 6 DEG

TOTAL

CA5 K1 F20 H15.1A V9.1 (PLUS. ØRB TC5 AT38.1)(RGM076)

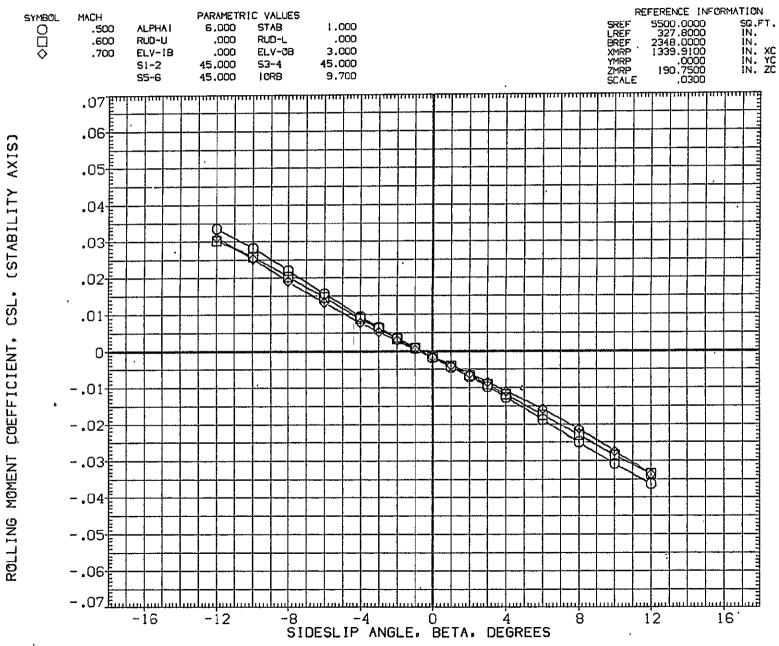
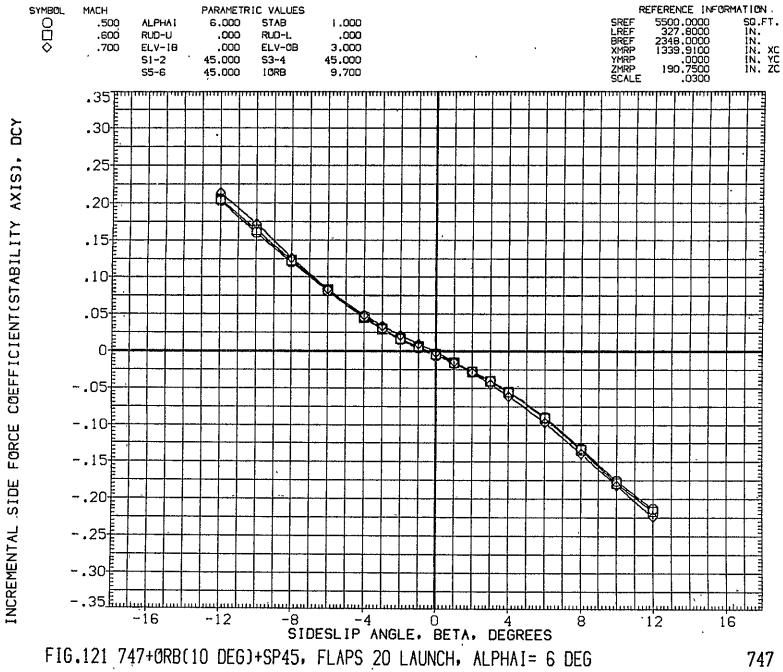


FIG.120 747+0RB(10 DEG)+SP45, FLAPS 20 LAUNCH, ALPHAI= 6 DEG

TOTAL

CA5 K1 F20 H15.1A V9.1 MINUS ORB TC5 AT38.1 (BGM076)



CA5 K1 F20 H15.1A V9.1 MINUS ORB TC5 AT38.1 (BGM076)

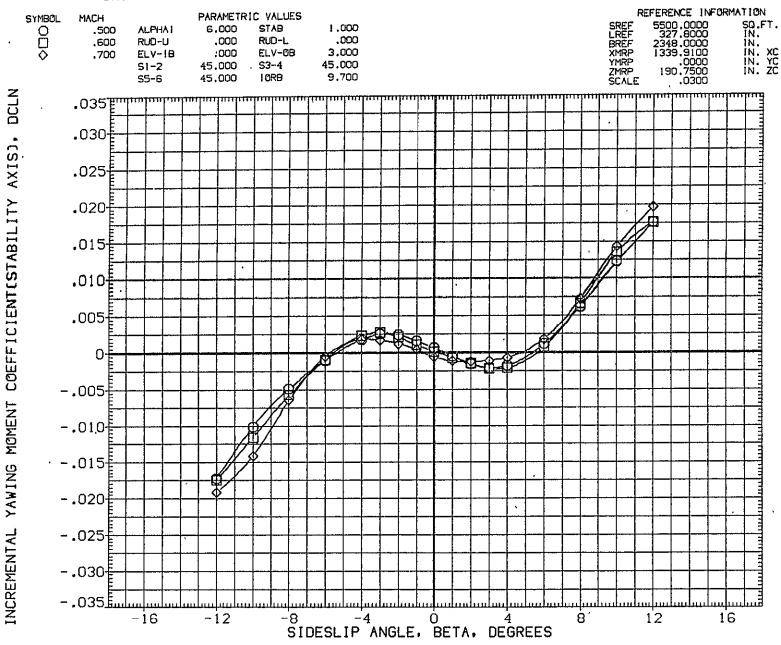


FIG.121 747+0RB(10 DEG)+SP45, FLAPS 20 LAUNCH, ALPHAI= 6 DEG

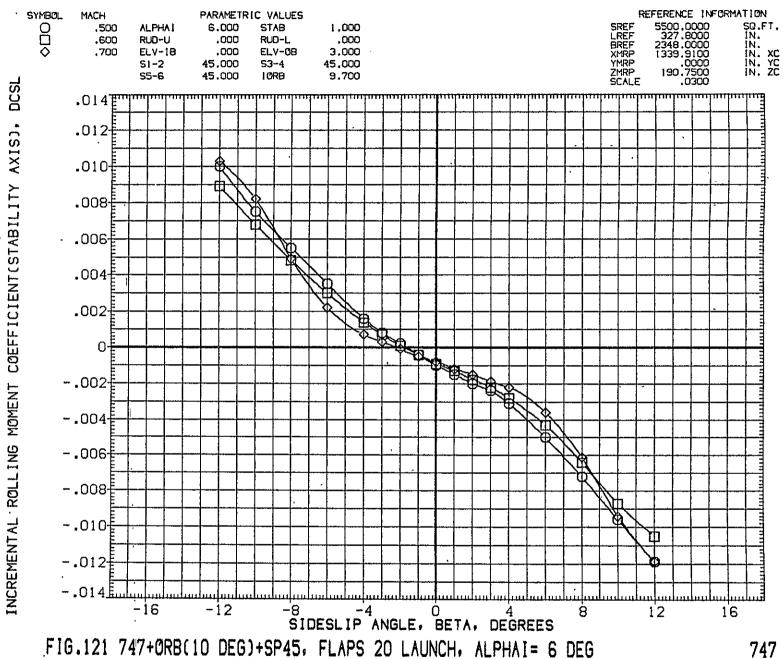
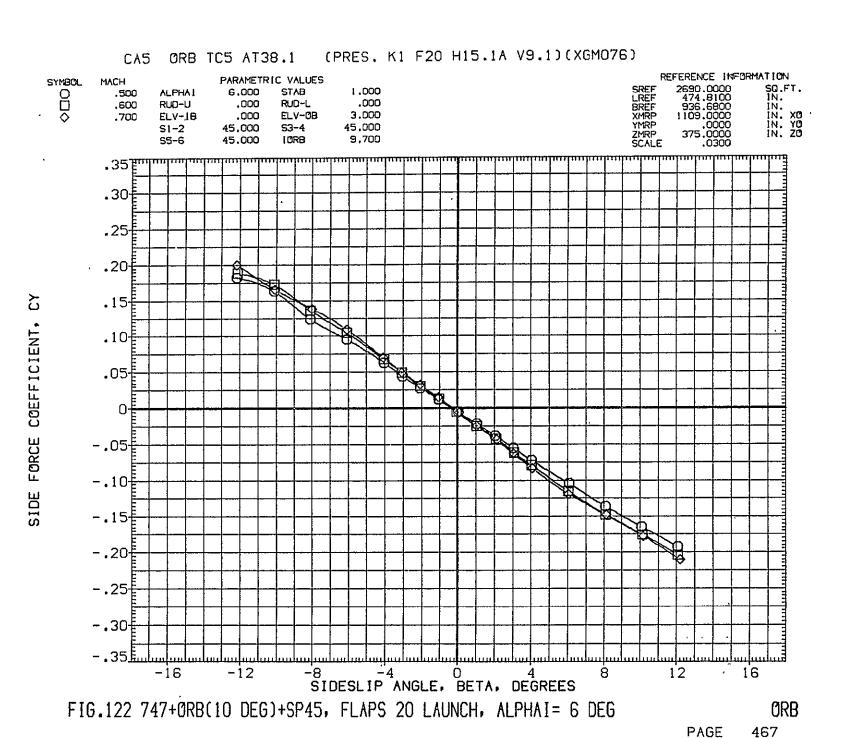


FIG.121 747+0RB(10 DEG)+SP45, FLAPS 20 LAUNCH, ALPHAI = 6 DEG





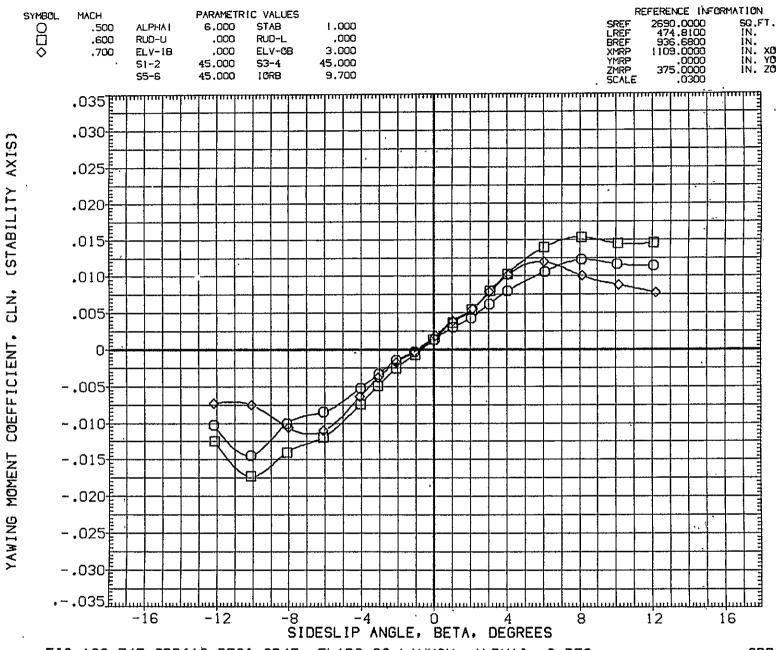


FIG.122 747+0RB(10 DEG)+SP45, FLAPS 20 LAUNCH, ALPHAI= 6 DEG

CA5 ORB TC5 AT38.1 (PRES. K1 F20 H15.1A V9.1)(XGM076) REFERENCE INFORMATION SYMBOL. MACH PARAMETRIC VALUES 2690.0000 474.8100 936.6800 1109.0000 .0000 375.0000 .0300 SQ.FT. IN. IN. XO IN. YO IN. ZO SREF LREF BREF XMRP YMRP ZMRP SCALE 0□\$ 1,000 ALPHA! 6.000 STAB .500 RUD-L .000 .000 ,600 RUD-U ELV-0B 3.000 .700 ELV-IB .000 45,000 53-4 45,000 S1-2 9.700 S5-6 45.000 IORB .07 .06ŧ .05ŧ **CSTABILITY** .04 .03[.02[CSL .01[COEFFICIENT, 0--.01[-.02 ROLLING MOMENT -.03[-.04- -.05[-.06[-.07<u>.</u>E...l.... -12 12 16 -16 SIDESLIP ANGLE, BETA, DEGREES

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FIG.122 747+0RB(10 DEG)+SP45, FLAPS 20 LAUNCH, ALPHAI= 6 DEG

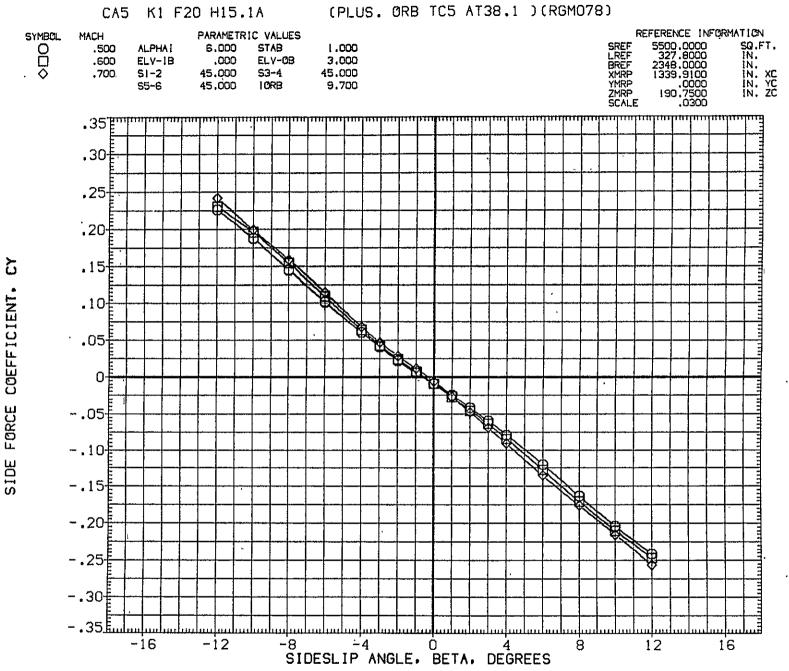
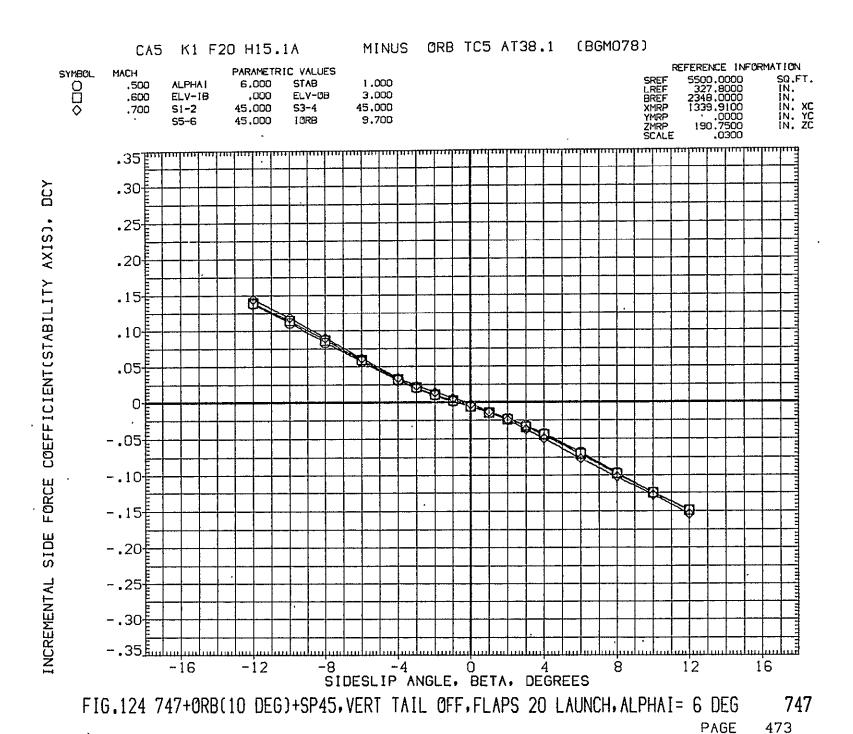


FIG.123 747+0RB(10 DEG)+SP45, VERT TAIL OFF, FLAPS 20 LAUNCH, ALPHAI = 6 DEG TOTAL PAGE 470

FIG.123 747+0RB(10 DEG)+SP45, VERT TAIL OFF, FLAPS 20 LAUNCH, ALPHAI= 6 DEG TOTAL

FIG.123 747+0RB(10 DEG)+SP45, VERT TAIL OFF, FLAPS 20 LAUNCH, ALPHAI = 6 DEG TOTAL PAGE 472





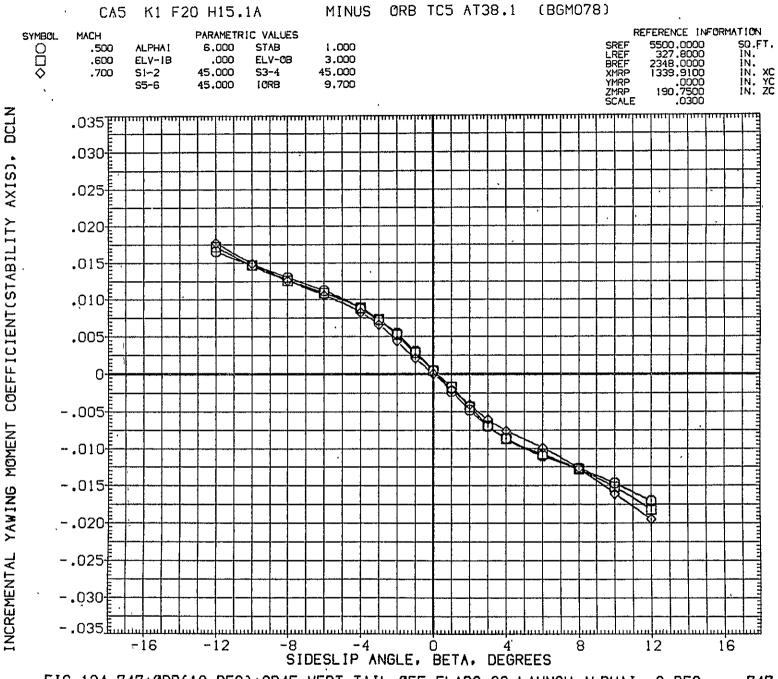
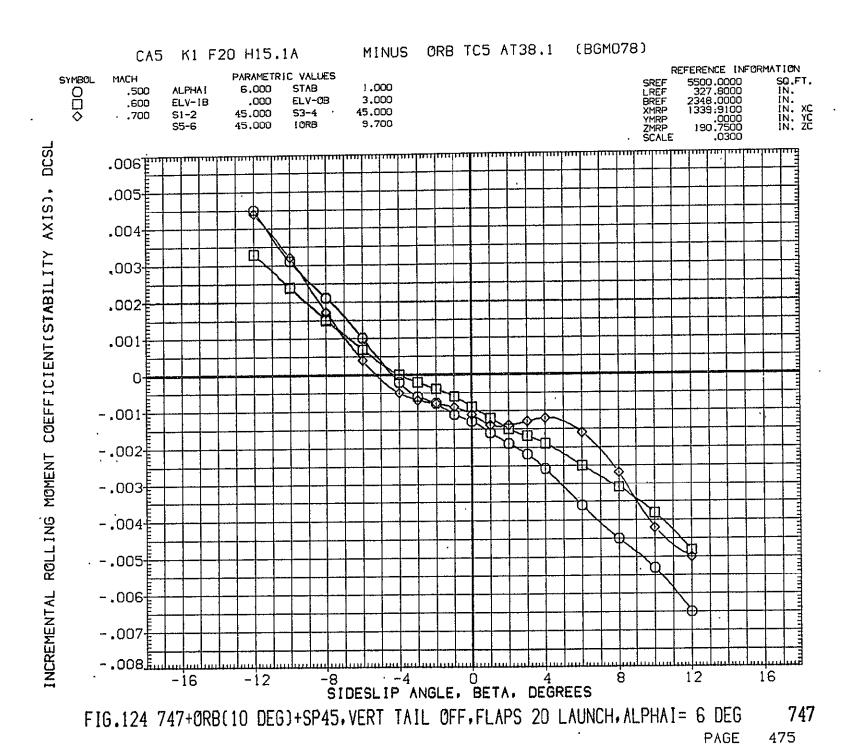


FIG.124 747+0RB(10 DEG)+SP45, VERT TAIL OFF, FLAPS 20 LAUNCH, ALPHAI = 6 DEG
PAGE 474



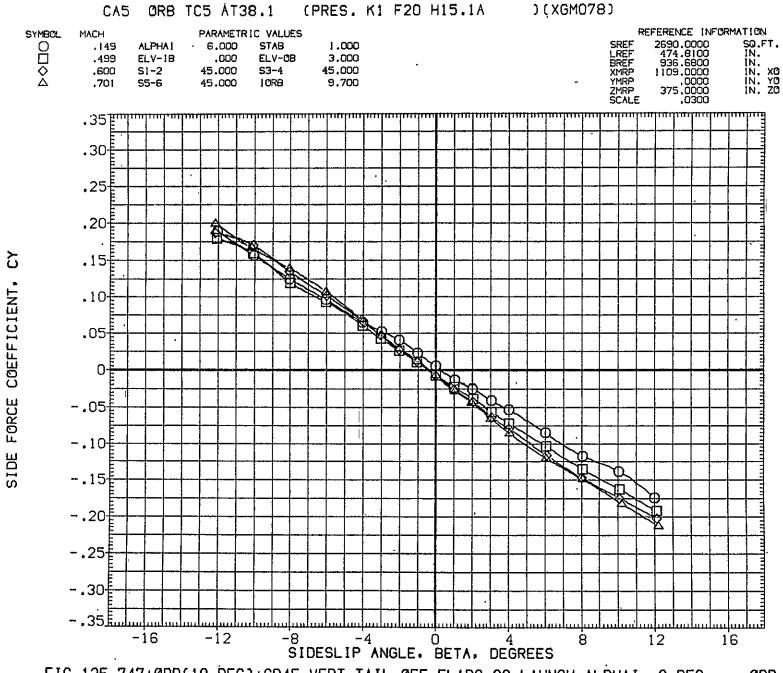
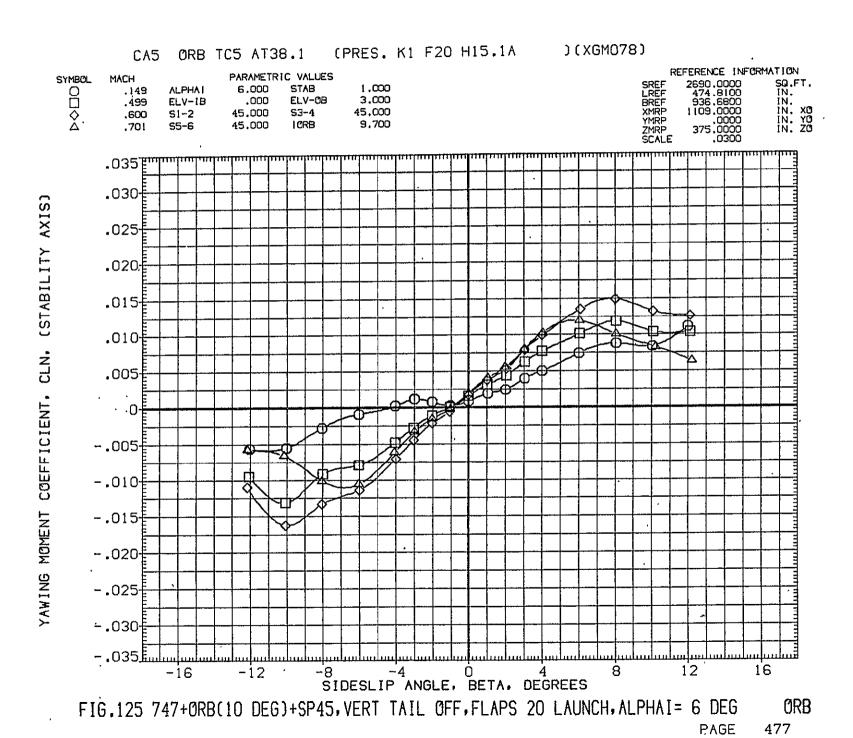


FIG.125 747+0RB(10 DEG)+SP45, VERT TAIL OFF, FLAPS 20 LAUNCH, ALPHAI = 6 DEG ORB

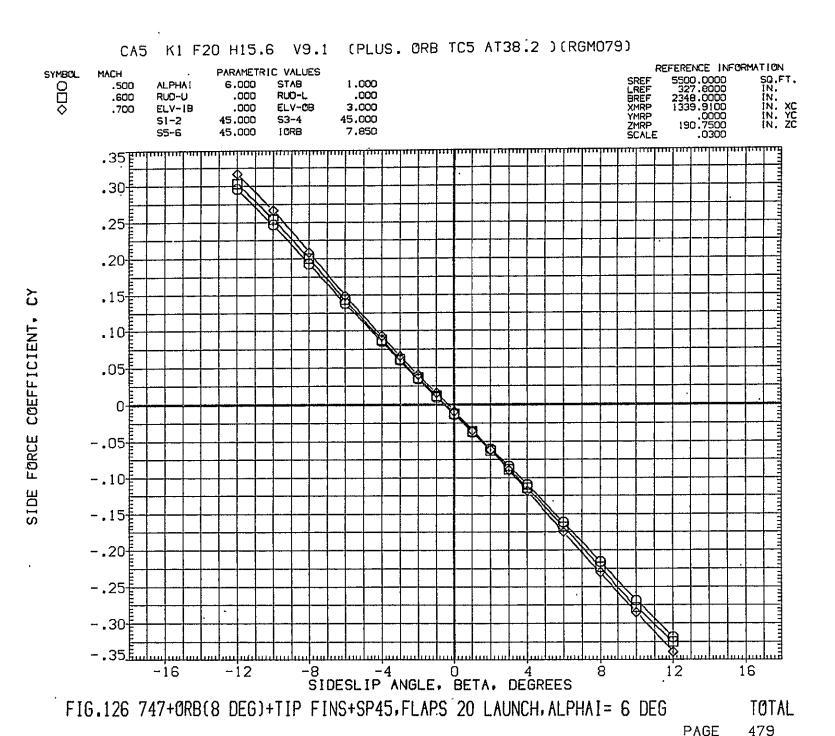


CA5 ORB TC5 AT38.1 (PRES. K1 F20 H15.1A)(XGM078) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH 2690.0000 474.8100 936.6800 1109.0000 SO.FT. SREF LREF BREF XMRP 000 6.000 STAB 1.000 ALPHA I .149 3.000 ELV-IB .000 ELV-0B .499 IN. XO IN. YO IN. ZO .600 51-2 45.000 S3-4 45.000 YMRP ZMRP SCALE .0000 375.0000 .0300 9.700 .701 45.000 10RB S5~6 .06- (STABILITY AXIS) .05 .04 .03 .02 CSL. .01-COEFFICIENT, 0 -.01 -.02 ROLLING MOMENT -.03 -.04 -.05 -.06-- .07<u>L</u>... ավավավավումուկու -16 12 16 -12-4 -8 8 SIDESLIP ANGLE, BETA, DEGREES ORB . FIG.125 747+0RB(10 DEG)+SP45.VERT TAIL OFF.FLAPS 20 LAUNCH.ALPHAI= 6 DEG

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and the same of th



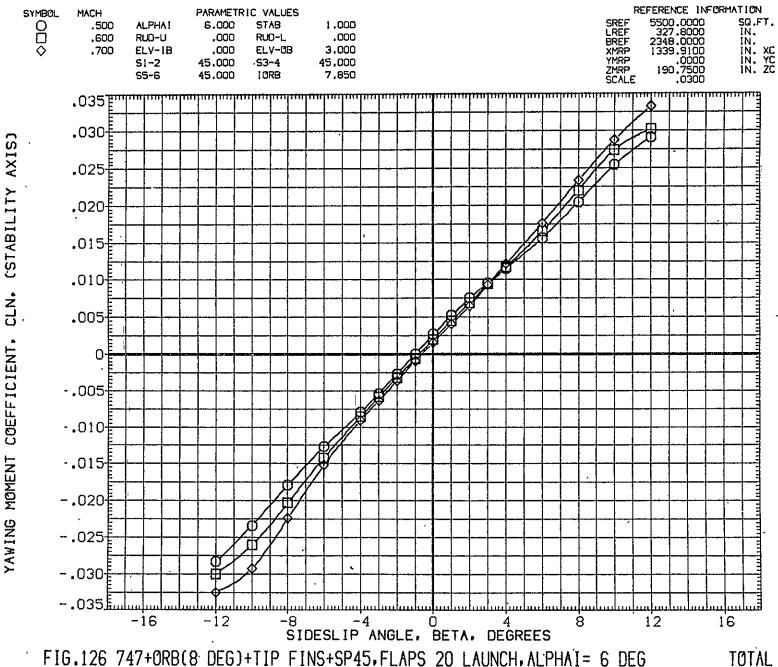


FIG.126 747+0RB(8 DEG)+TIP FINS+SP45,FLAPS 20 LAUNCH,ALPHAI= 6 DEG

CA5 K1 F20 H15.6 V9.1 (PLUS. ORB TC5 AT38.2)(RGM079)

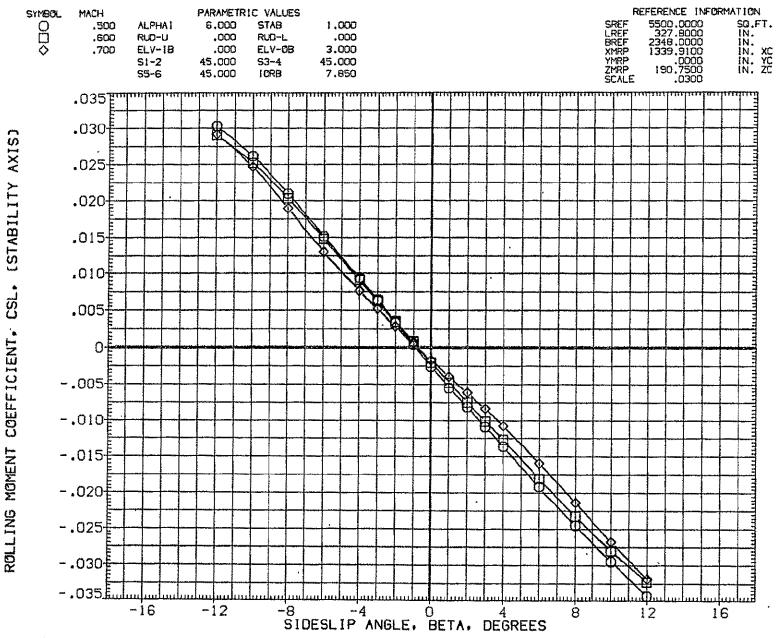


FIG.126 747+0RB(8 DEG)+TIP FINS+SP45, FLAPS 20 LAUNCH, ALPHAI= 6 DEG

TOTAL

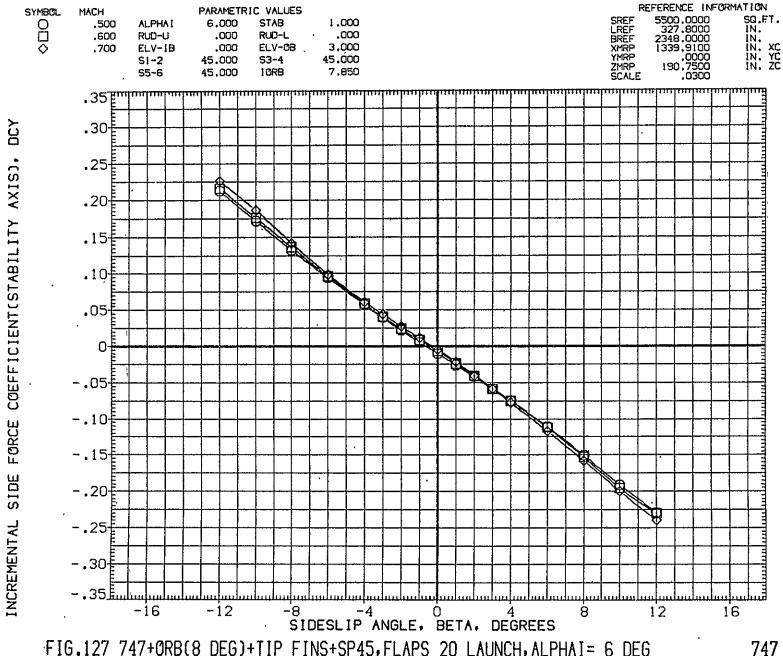


FIG.127 747+0RB(8 DEG)+TIP FINS+SP45,FLAPS 20 LAUNCH,ALPHAI= 6 DEG

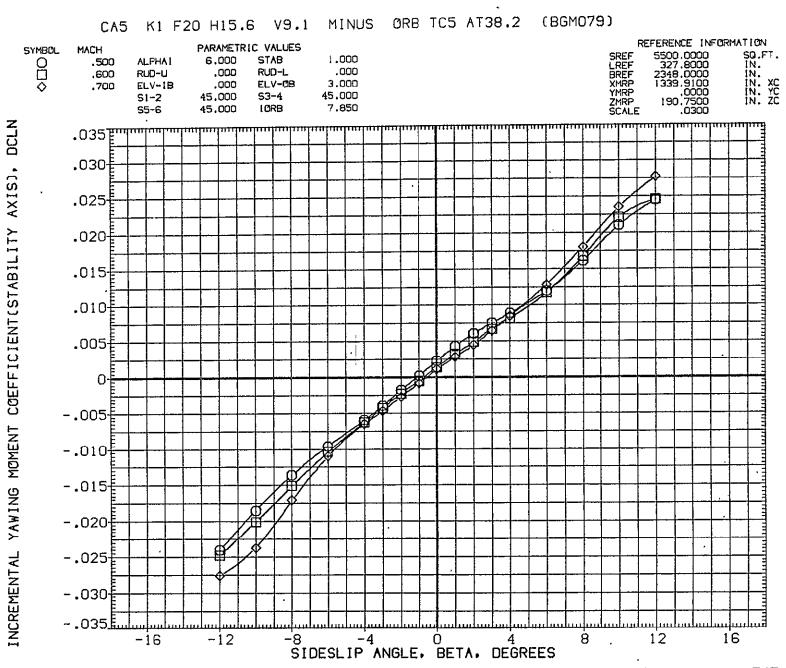


FIG.127 747+0RB(8 DEG)+TIP FINS+SP45,FLAPS 20 LAUNCH,ALPHAI= 6 DEG

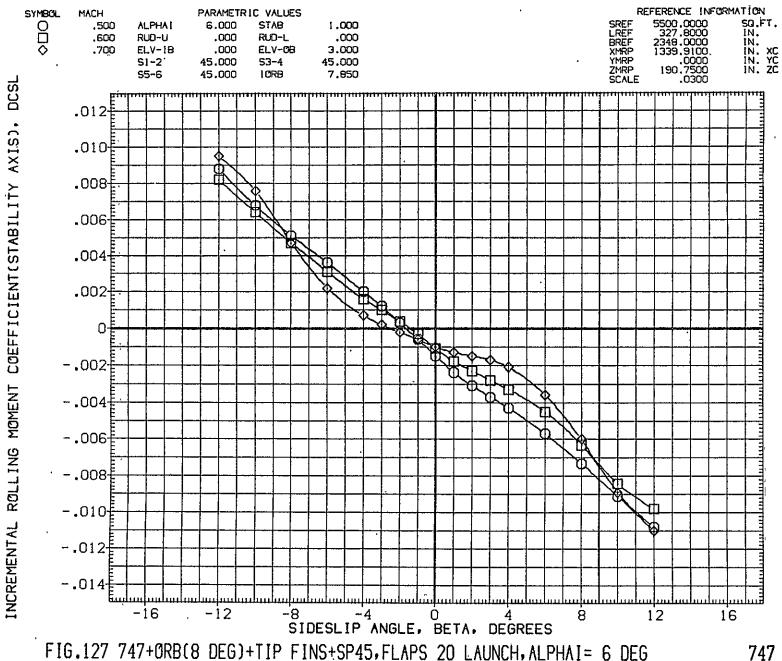
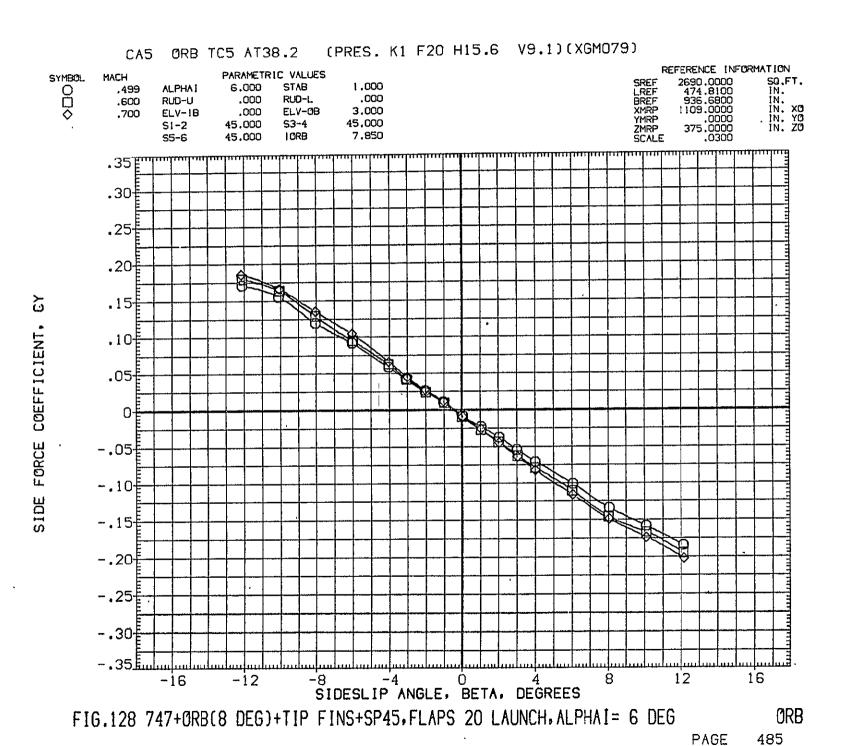


FIG.127 747+0RB(8 DEG)+TIP FINS+SP45,FLAPS 20 LAUNCH, ALPHAI= 6 DEG



CA5 ORB JC5 AT38.2 (PRES. K1 F20 H15.6 V9.1)(XGM079)

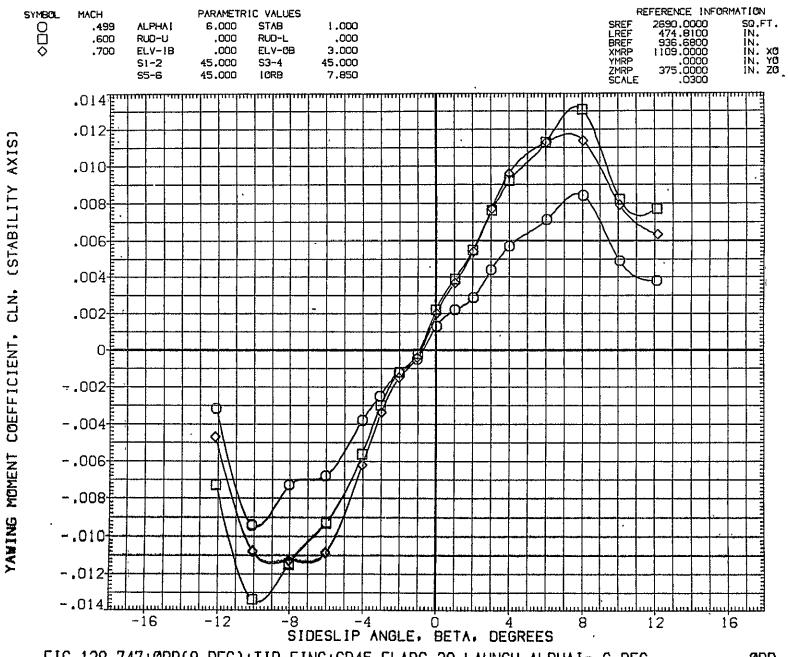


FIG.128 747+0RB(8 DEG)+TIP FINS+SP45.FLAPS 20 LAUNCH.ALPHAI= 6 DEG

ORB

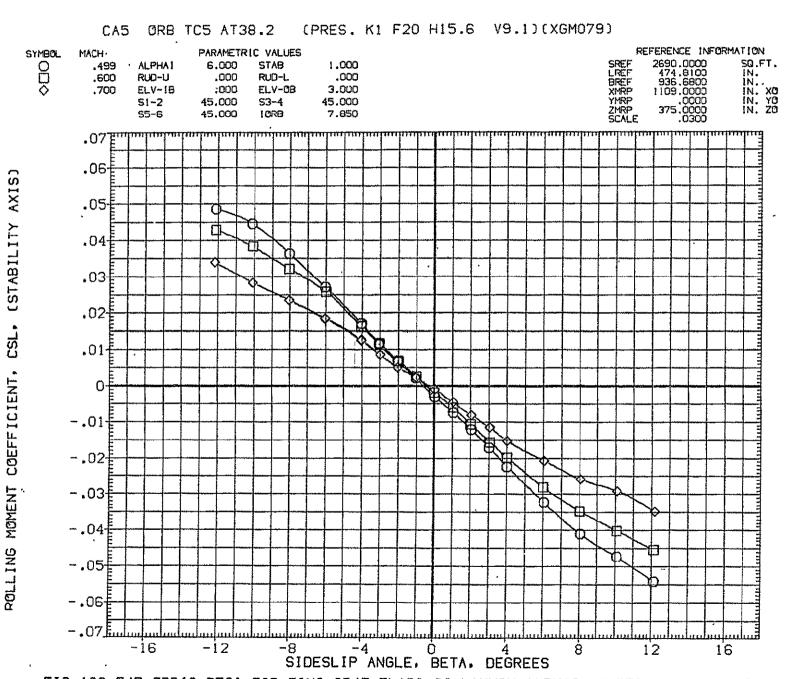


FIG.128 747+0RB(8 DEG)+TIP FINS+SP45,FLAPS 20 LAUNCH, ALPHAI= 6 DEG

ORB

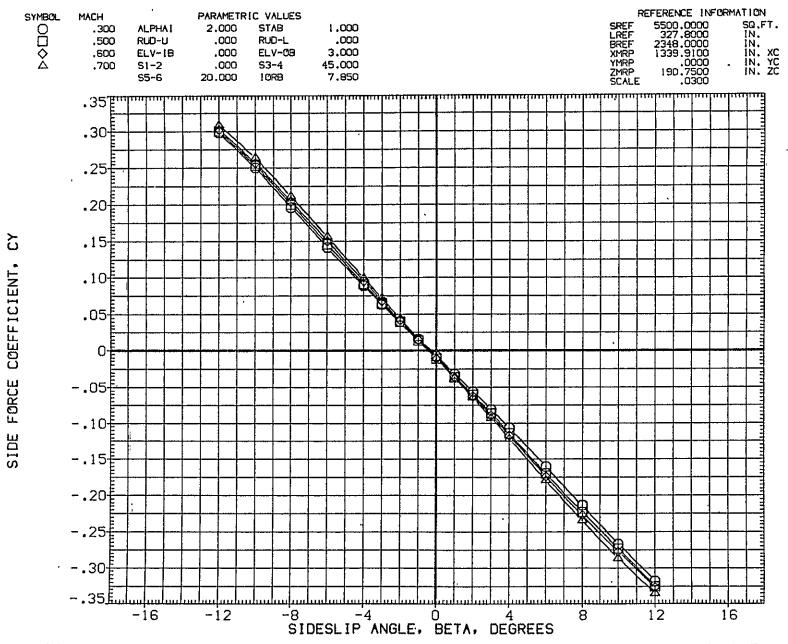


FIG.129 747+ORB(8 DEG)+TIP FINS+STD SPD BRK,FLAPS UP LAUNCH,ALPHAI= 2 DEG TOTAL
PAGE 488

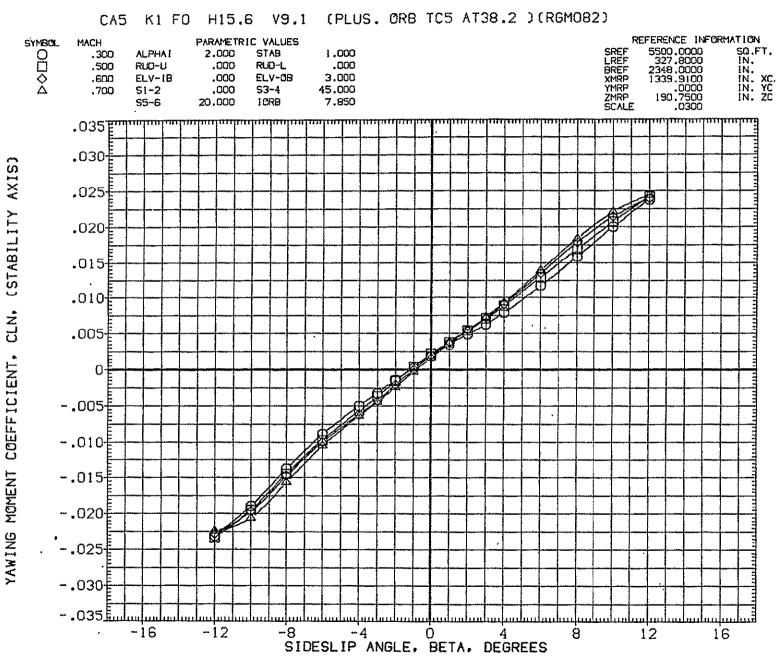


FIG.129 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPHAI= 2 DEG TOTAL PAGE 489

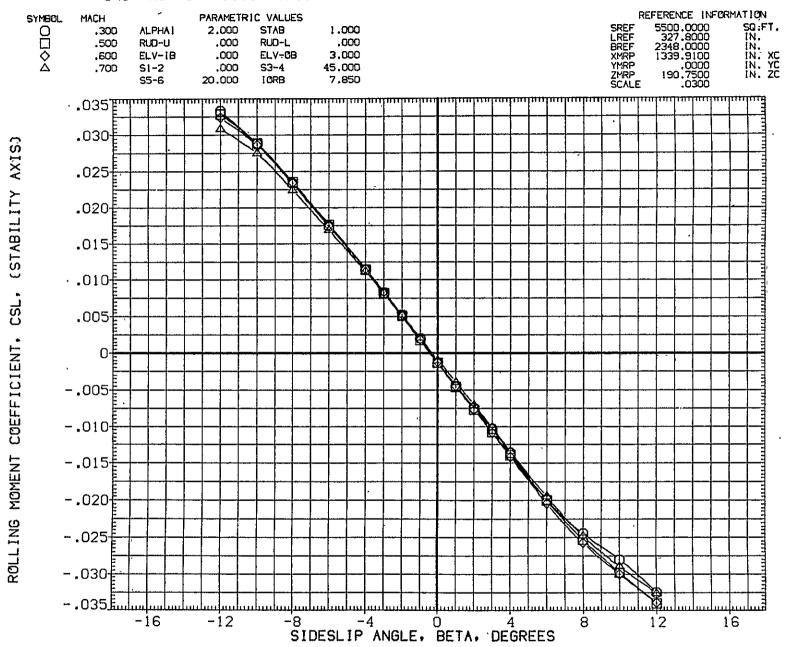
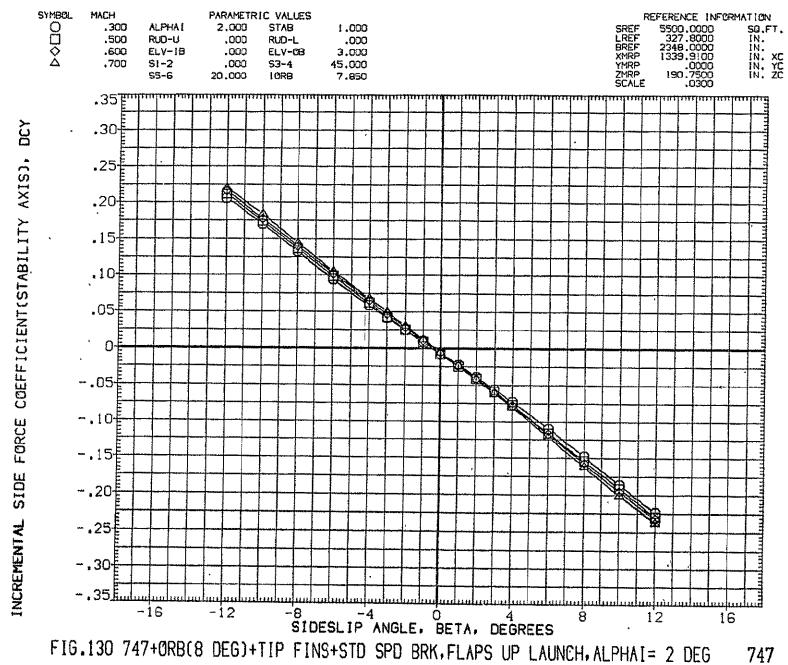


FIG.129 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPHAI = 2 DEG TOTAL

(,)

CA5 K1 FO H15.6 'V9.1 MINUS ORB TC5 AT38.2 (BGMO82)



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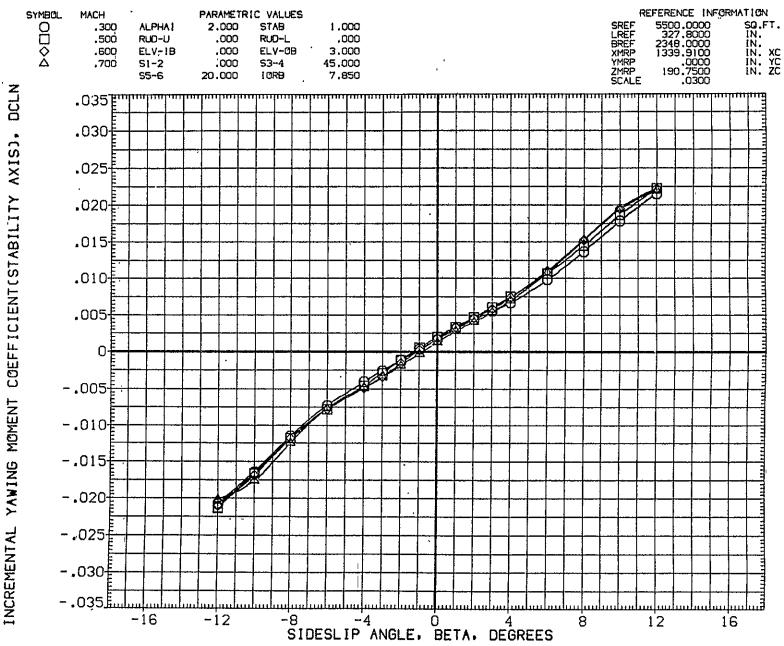


FIG.130 747+0RB(8 DEG)+TIP FINS+STD SPD BRK,FLAPS UP LAUNCH, ALPHAI= 2 DEG 747
PAGE 492





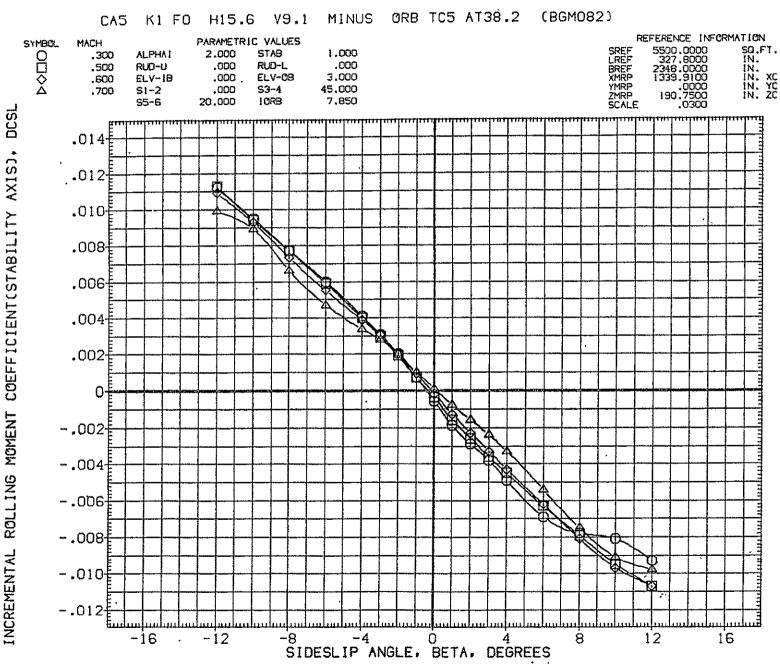


FIG.130 747+0RB(8 DEG)+TIP FINS+STD SPD BRK,FLAPS UP LAUNCH,ALPHAI= 2 DEG PAGE 493

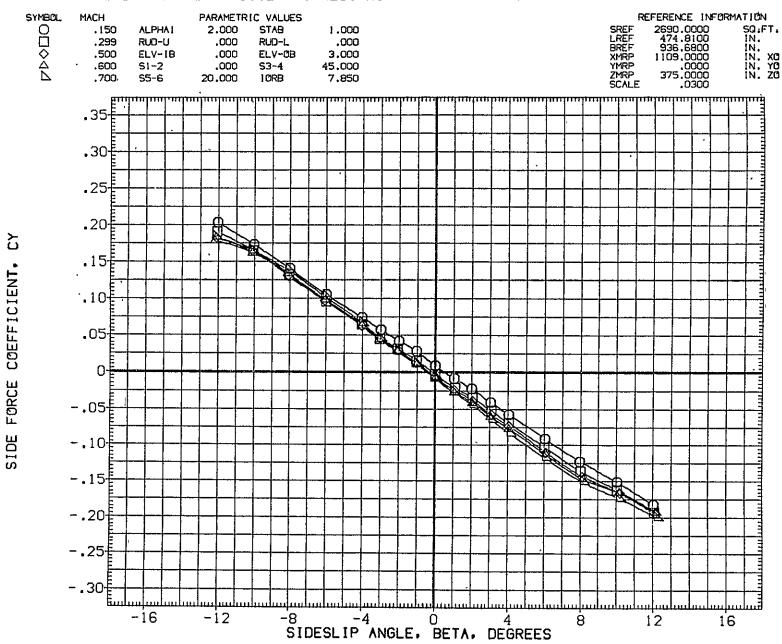


FIG.131 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPHAI= 2 DEG ORB

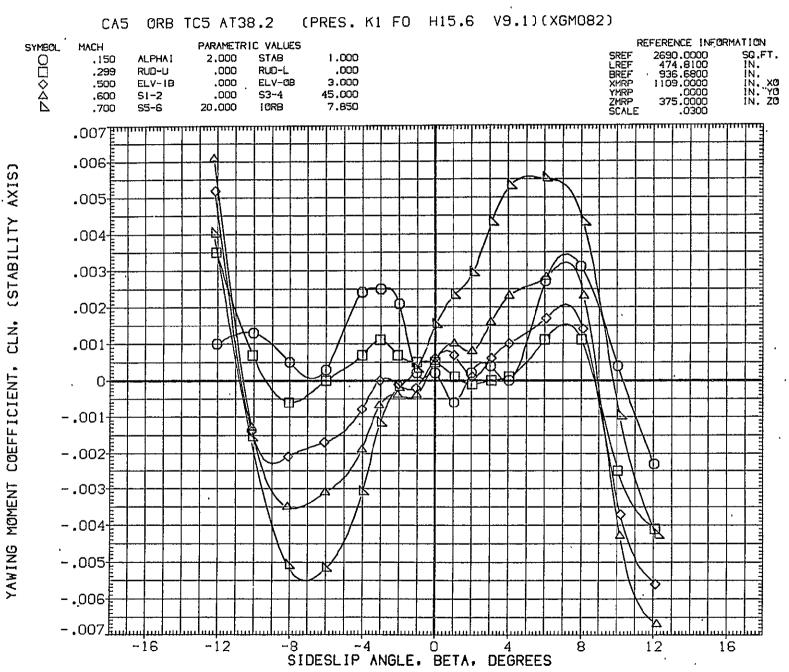


FIG.131 747+0RB(8 DEG)+TIP FINS+STD SPD BRK,FLAPS UP LAUNCH, ALPHAI= 2 DEG ORB

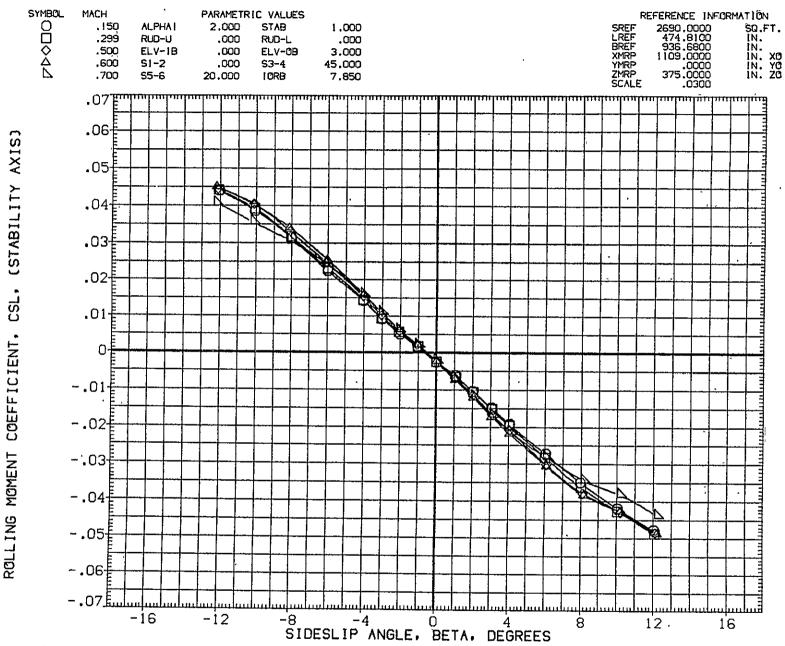


FIG.131 747+0RB(8 DEG)+TIP FINS+STD SPD BRK,FLAPS UP LAUNCH, ALPHAI= 2 DEG ORB

CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.2)(RGMO83)

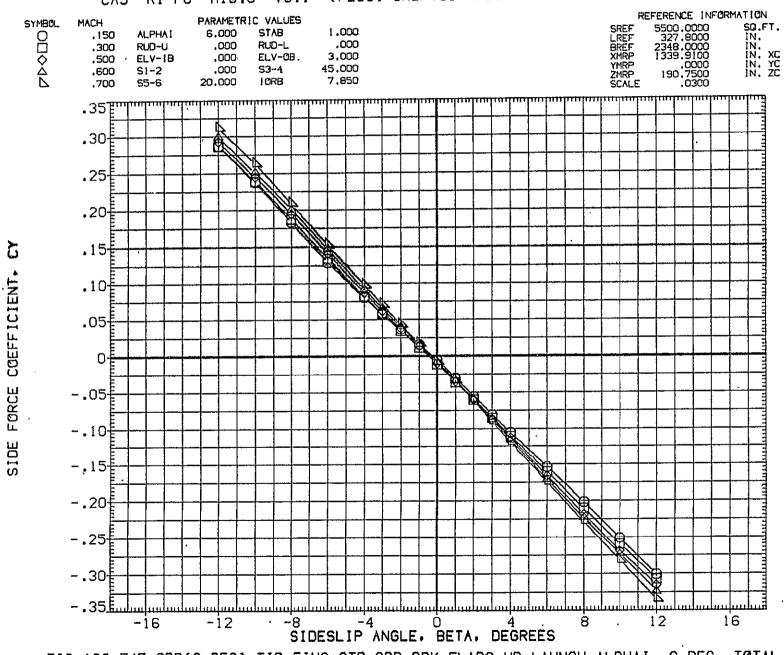


FIG.132 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPHAI= 6 DEG TOTAL
PAGE 497

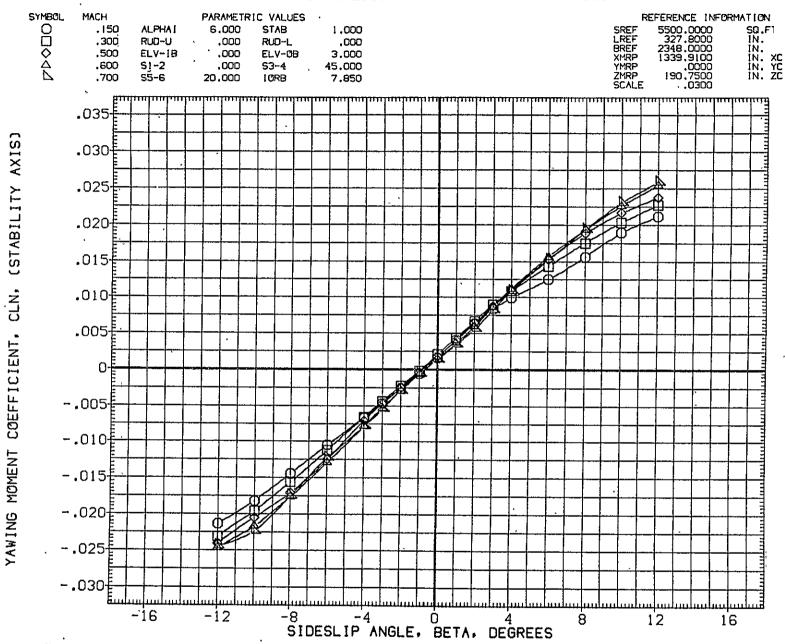


FIG.132 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPHAI= 6 DEG TOTAL PAGE 498

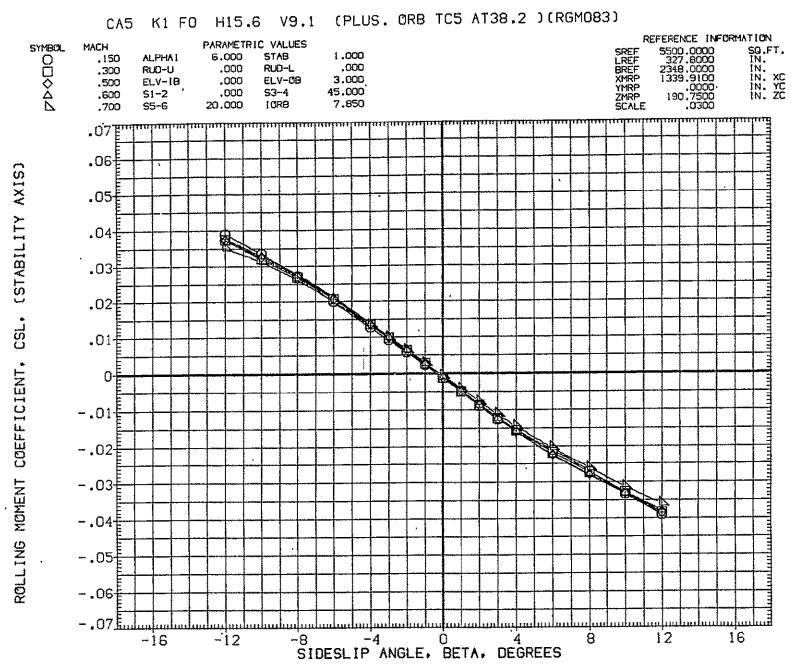


FIG.132 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPHAI= 6 DEG TOTAL

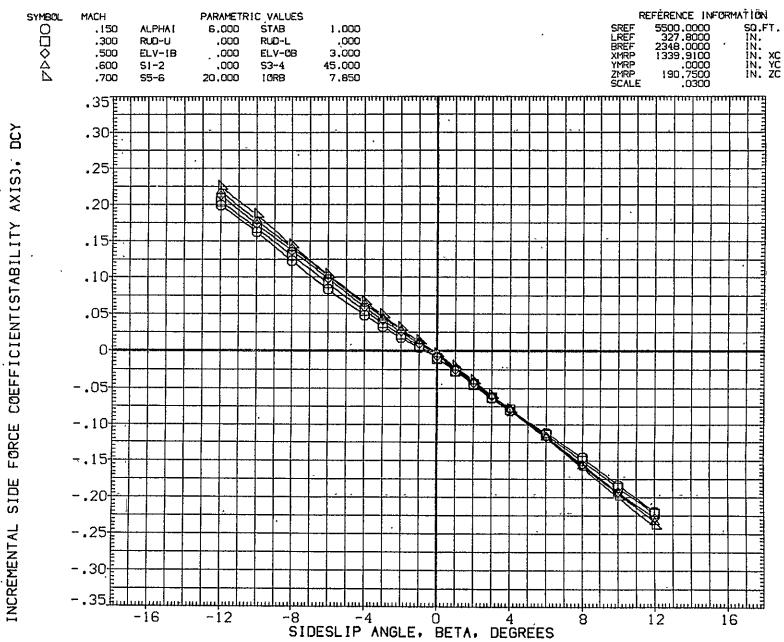
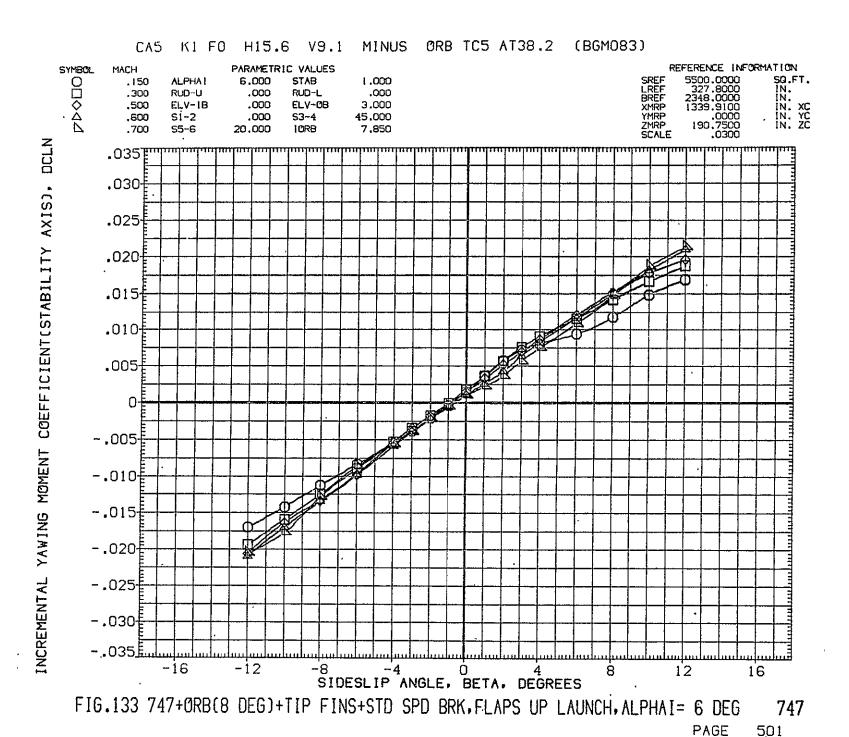


FIG.133 747+0RB(8 DEG)+TIP FINS+STD SPD BRK,FLAPS UP LAUNCH, ALPHAI= 6 DEG PAGE 500



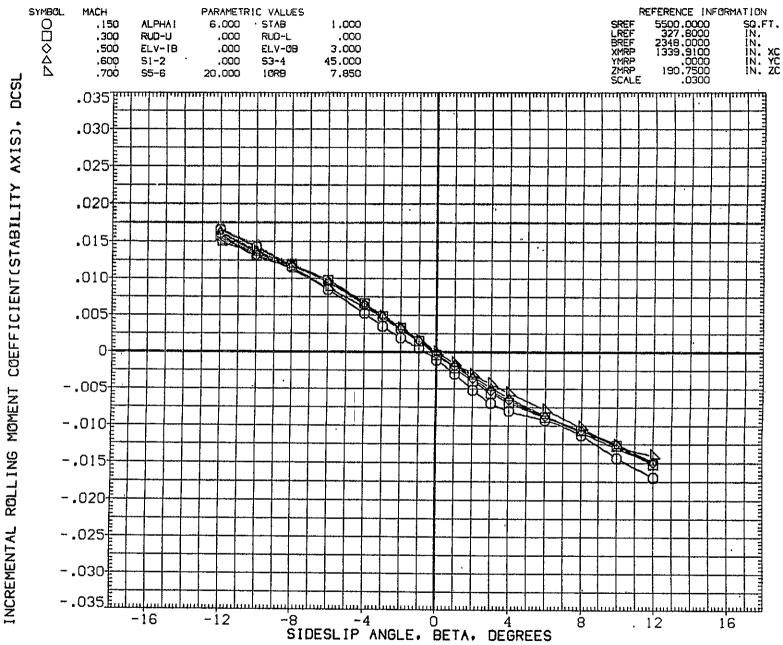


FIG.133 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPHAI= 6 DEG 747



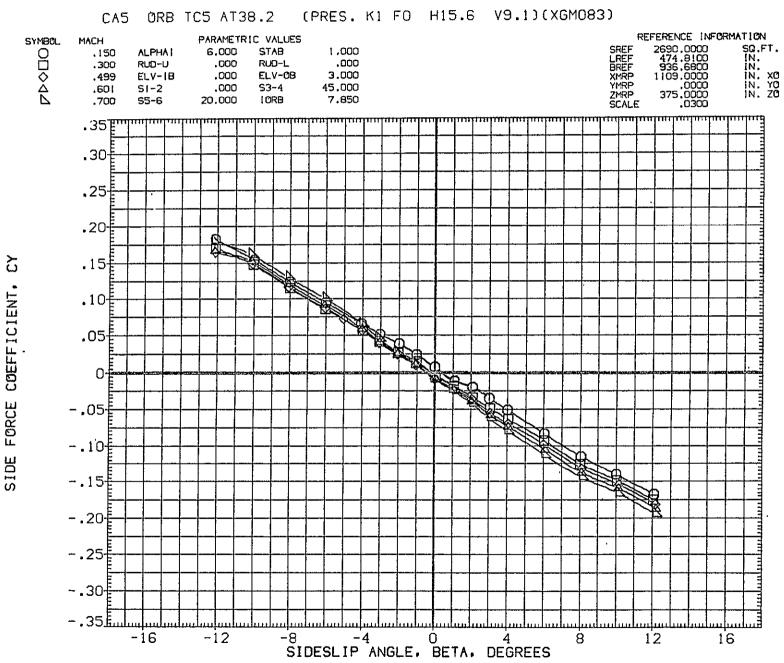


FIG.134 747+0RB(8 DEG)+TIP FINS+STD SPD BRK,FLAPS UP LAUNCH,ALPHAI= 6 DEG ORB

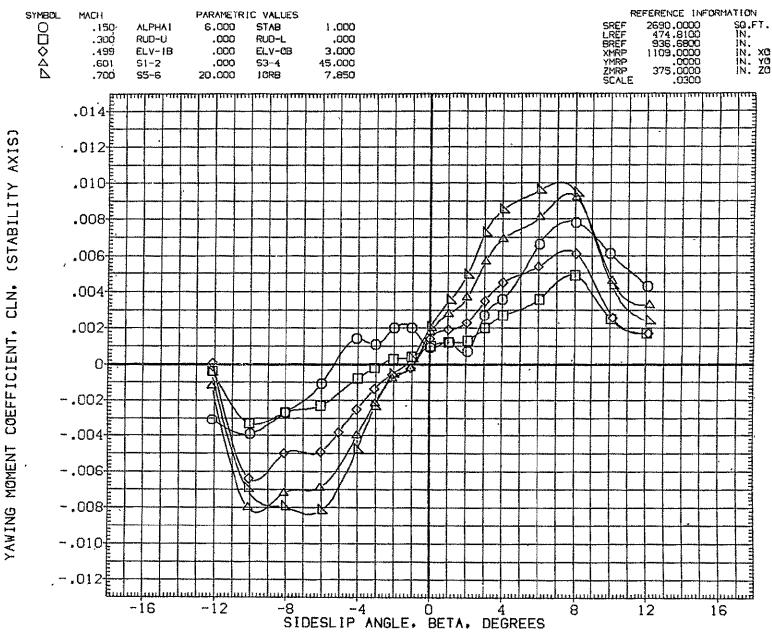


FIG.134 747+0RB(8 DEG)+TIP FINS+STD SPD BRK,FLAPS UP LAUNCH, ALPHAI= 6 DEG ORB

CA5 ORB TC5 AT38.2 (PRES. K1 FO H15.6 V9.1)(XGMO83)

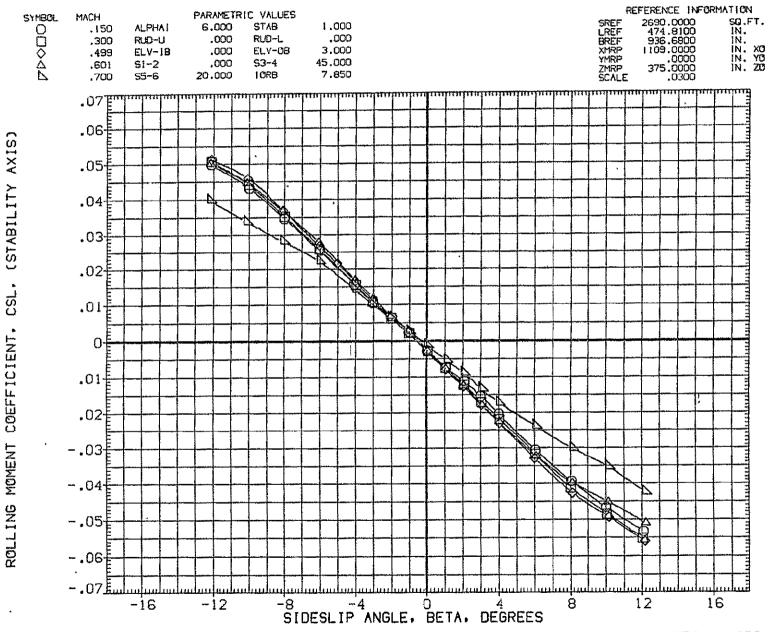


FIG.134 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPHAI= 6 DEG ORB

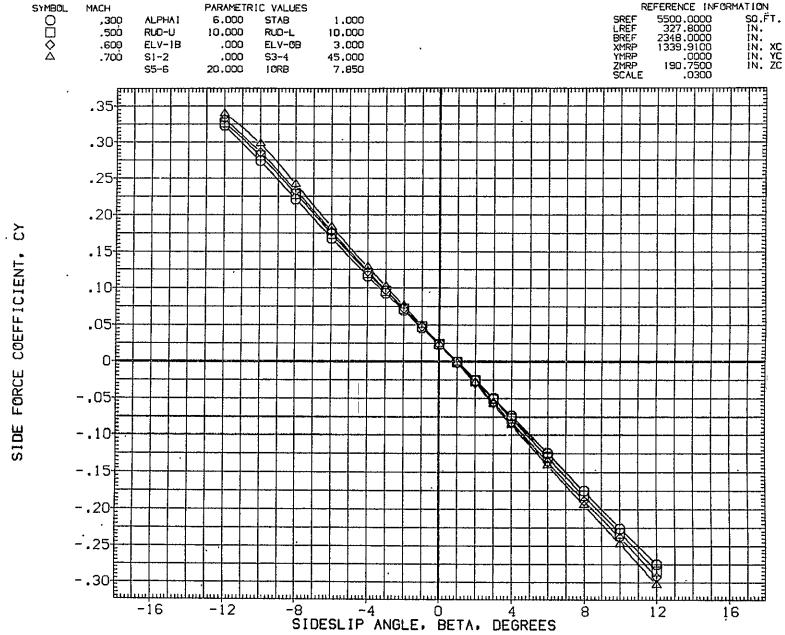


FIG.135 747+0RB(8 DEG)+TIP FINS+STD SPD BRK,FLAPS UP LAUNCH, ALPI=6.RUD=10/10 TOT
PAGE 506

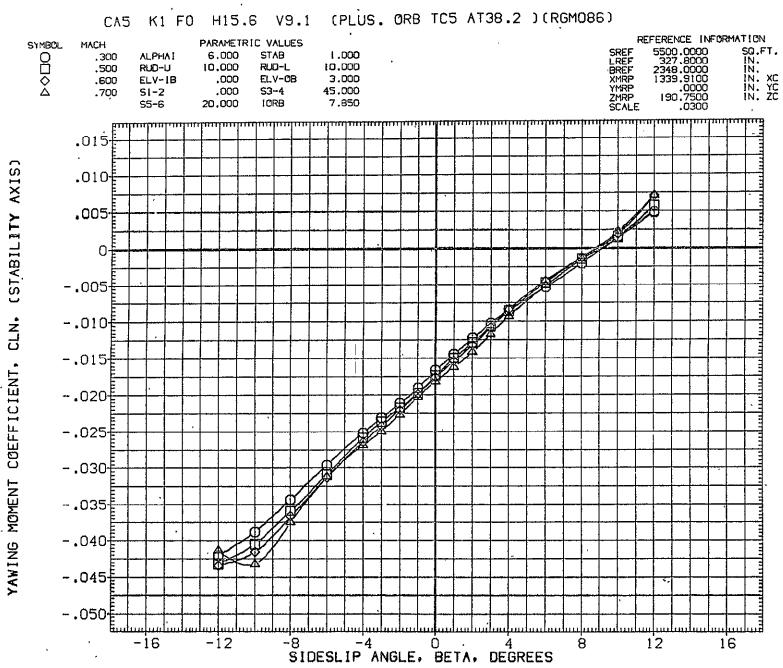


FIG.135 747+0RB(8 DEG)+TIP FINS+STD SPD BRK,FLAPS UP LAUNCH, ALPI=6.RUD=10/10 TOT
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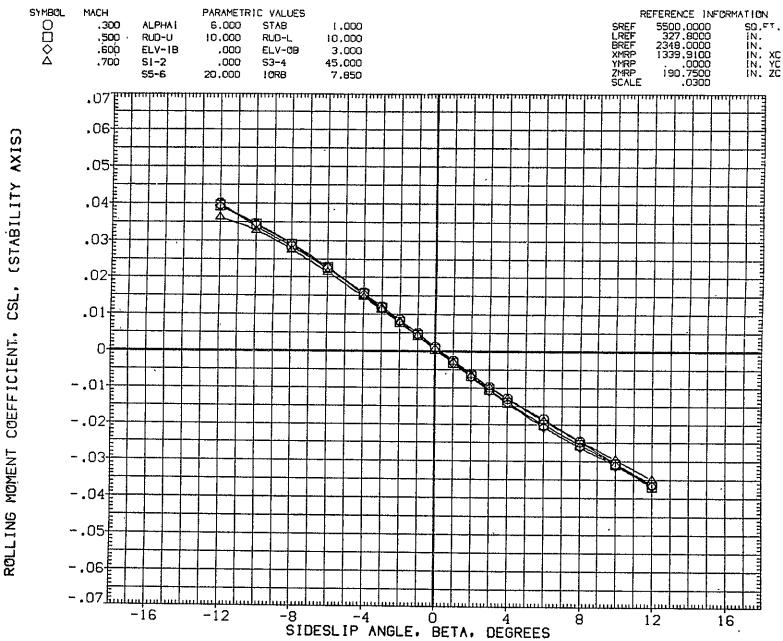


FIG.135 747+0RB(8 DEG)+TIP FINS+STD SPD BRK,FLAPS UP LAUNCH, ALPI=6, RUD=10/10 TOT
PAGE 508



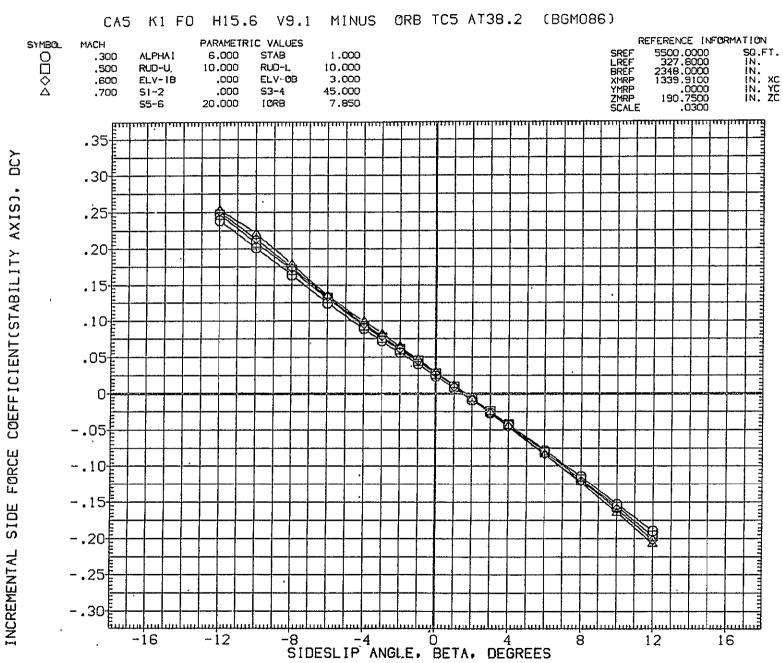


FIG.136 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPI=6, RUD=10/10 747

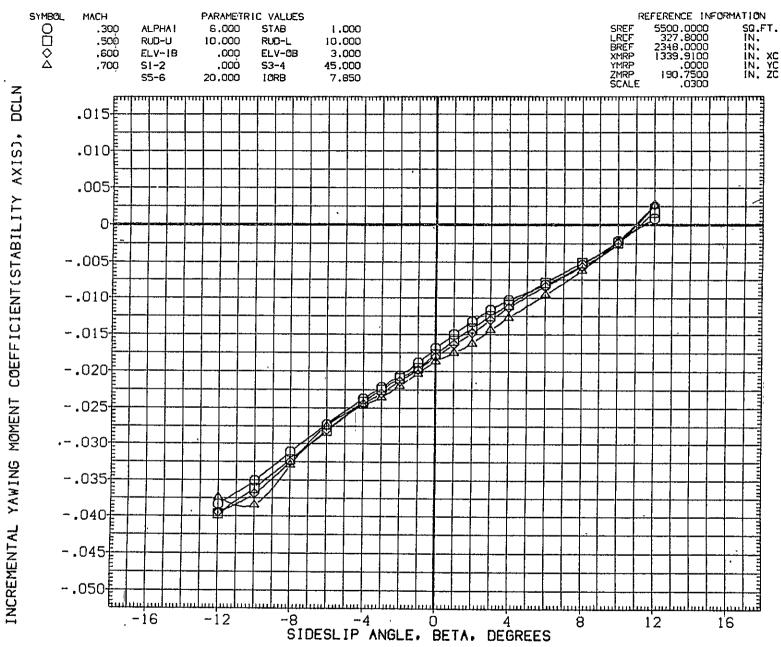


FIG.136 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPI=6, RUD=10/10 747

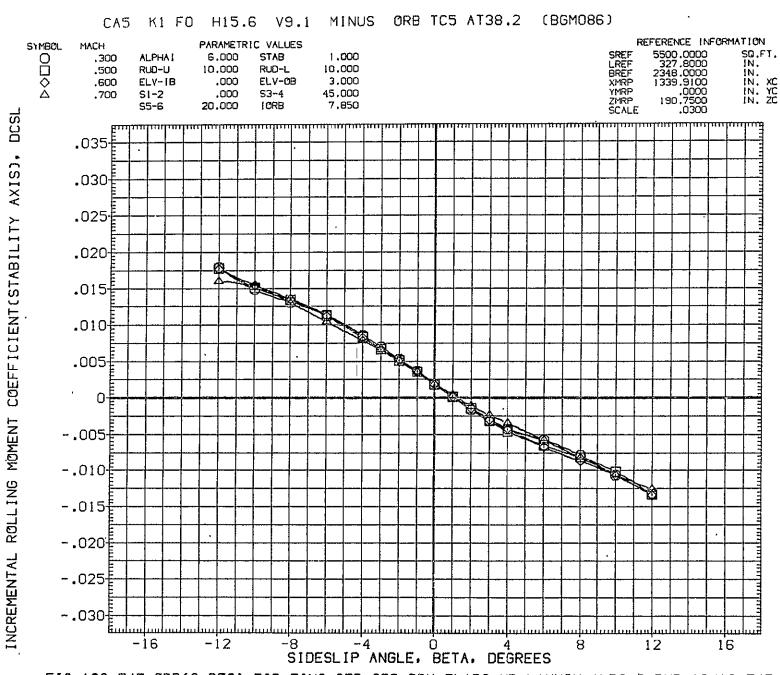


FIG.136 747+0RB(8 DEG)+TIP FINS+STD SPD BRK,FLAPS UP LAUNCH,ALPI=6,RUD=10/10 747

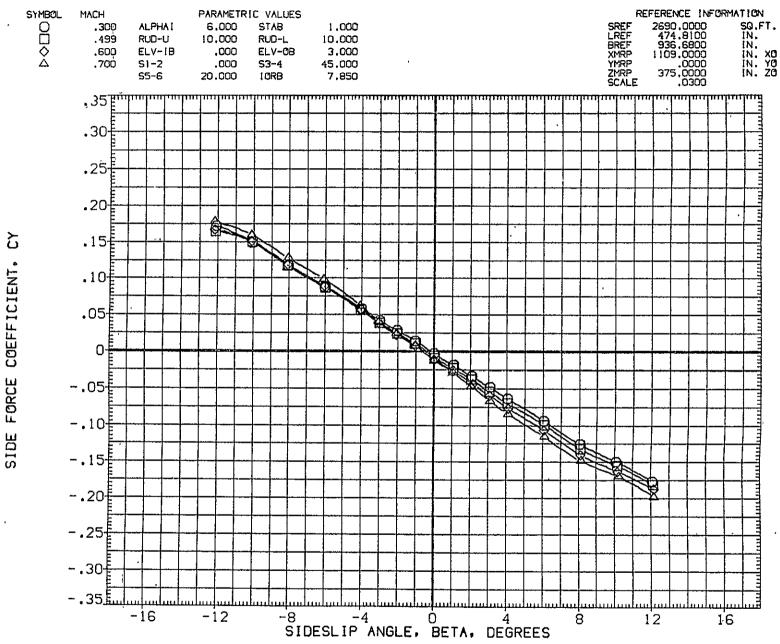


FIG.137 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPI=6, RUD=10/10 ORB

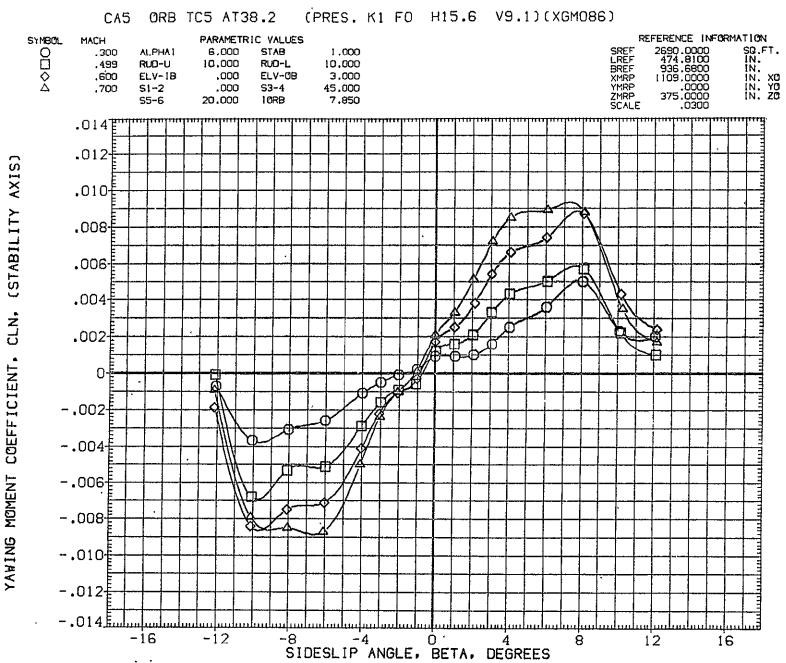


FIG.137 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPI=6, RUD=10, 10 ORB

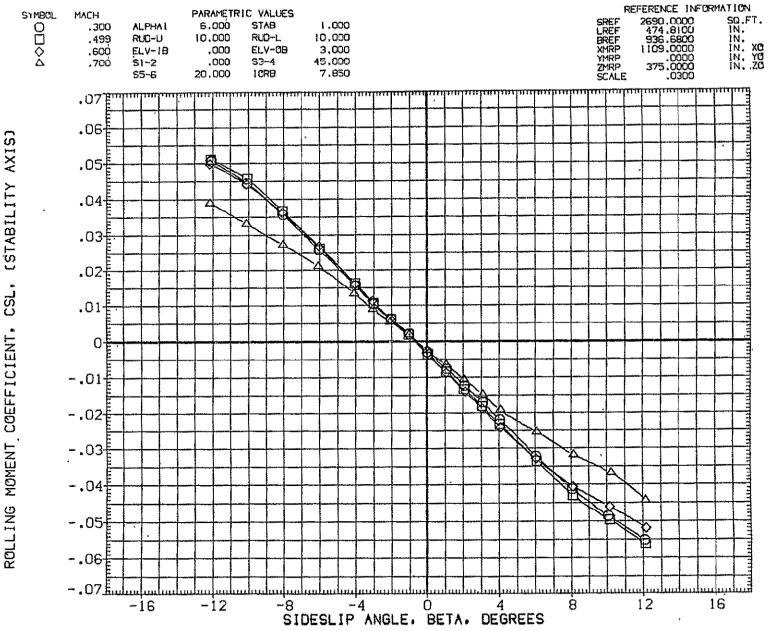
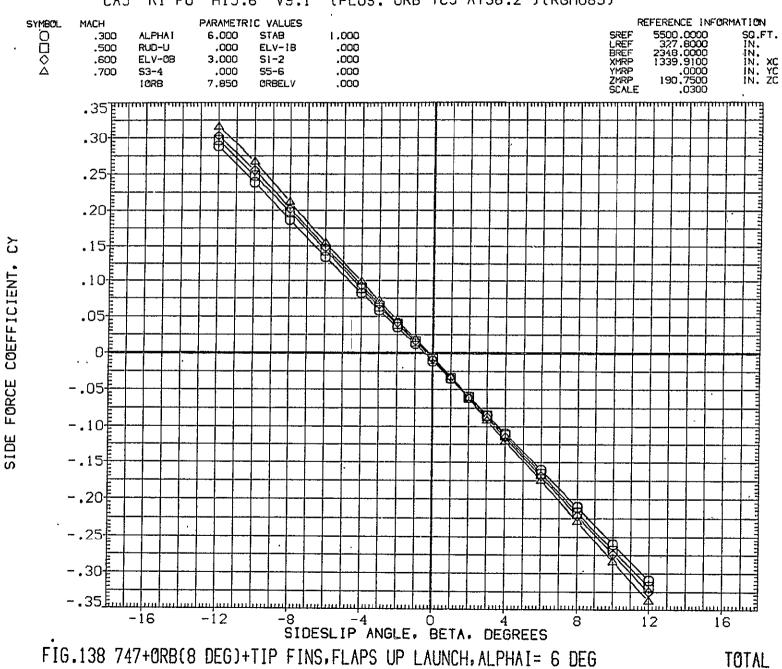


FIG.137 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPI=6, RUD=10/10 ORB



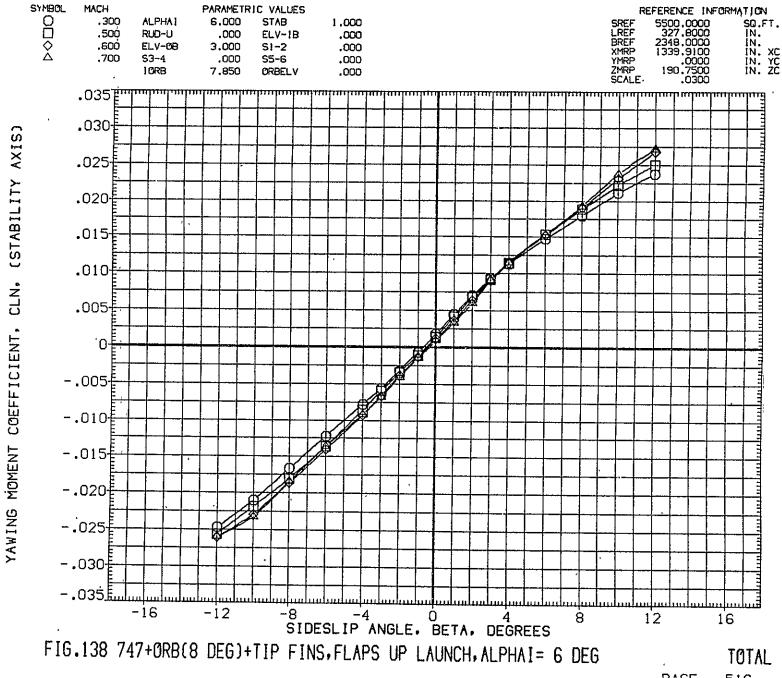
CA5 K1 FO H15.6 V9.1 (PLUS. ØRB TC5 AT38.2)(RGMO89)



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CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.2)(RGMO89)



CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.2)(RGMO89) SYMBOL MACH PARAMETRIC VALUES REFERENCE INFORMATION 0000 SG.FT. IN. IN. IN. XC IN. YC IN. ZC 5500.0000 327.8000 2348.0000 6.000 SREF LREF BREF .300 ALPHA! STAB 1,000 .500 RUD-U .000 ELV-1B .000 .600 ELV-0B 3.000 SI-2 .000 1339.9100 .0000 190.7500 XMRP YMRP ZMRP SCALE ,700 53-4 .000 S5-6 .000 10RB 7.850 ORBELV .000 .07FM .06[.05ŧ .04[.03[.02 .01# 0 -.01[-.02 -.03 -.04 -.05 -.06 - **.**07<u>.</u>L.... -8 -4 0 4 SIDESLIP ANGLE, BETA, DEGREES -16 -i2 12 16

FIG.138 747+ORB(8 DEG)+TIP FINS, FLAPS UP LAUNCH, ALPHAI= 6 DEG

(STABILITY AXIS)

CSL,

COEFFICIENT,

ROLLING MOMENT

TOTAL PAGE 517

CA5 K1 FO H15.6 V9.1 MINUS ORB TC5 AT38.2 (BGM089) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH 5500.0000 327.8000 2348.0000 1339.9100 .0000 190.7500 SQ.FT. IN. IN. XC IN. YC IN. ZC SREF LREF BREF XMRP YMRP ZMRP SCALE 0000 6,000 STAB 1,000 ALPHA1 .300 .000 .000 ELV-IB RUD-U .50Ò .000 3,000 S1-2 .600 ELV-08 55-6 .000 .000 .700 53-4 7.850 ORBELV .006 10RB .35 Em DC≺ •30[AXIS), .25‡ .20 .15[.10[.05

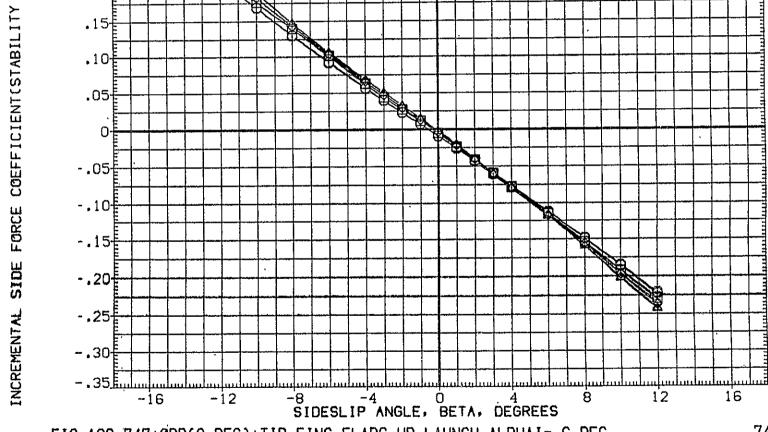
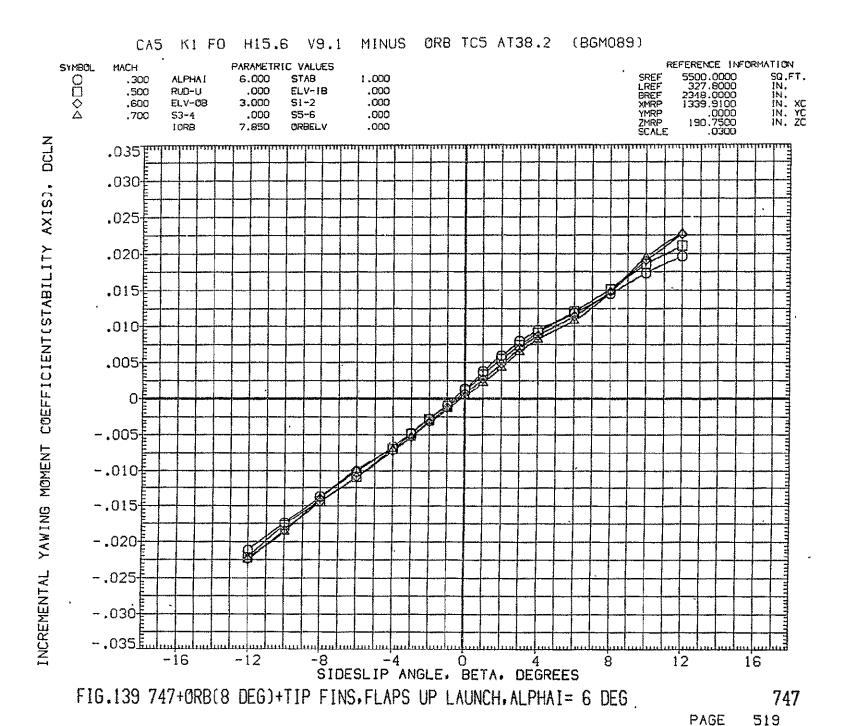
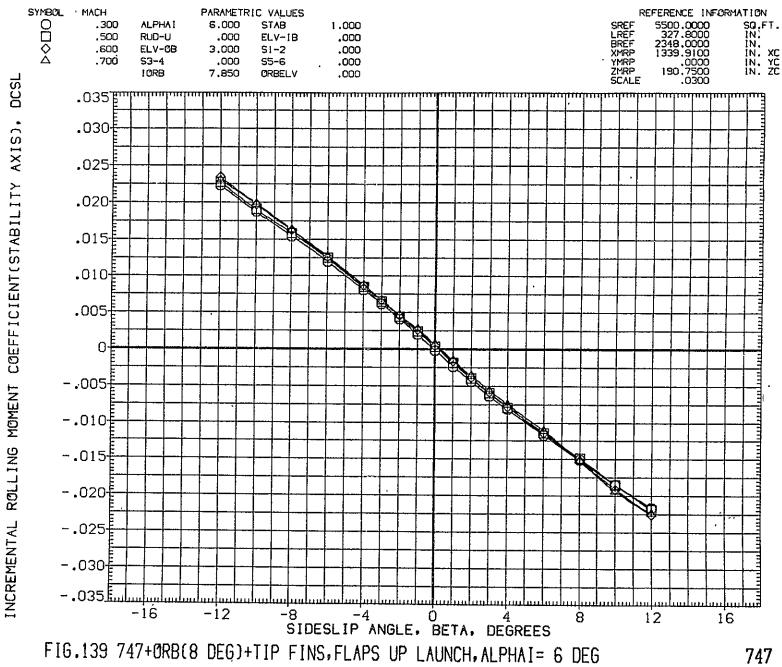


FIG.139 747+0RB(8 DEG)+TIP FINS, FLAPS UP LAUNCH, ALPHAI= 6 DEG

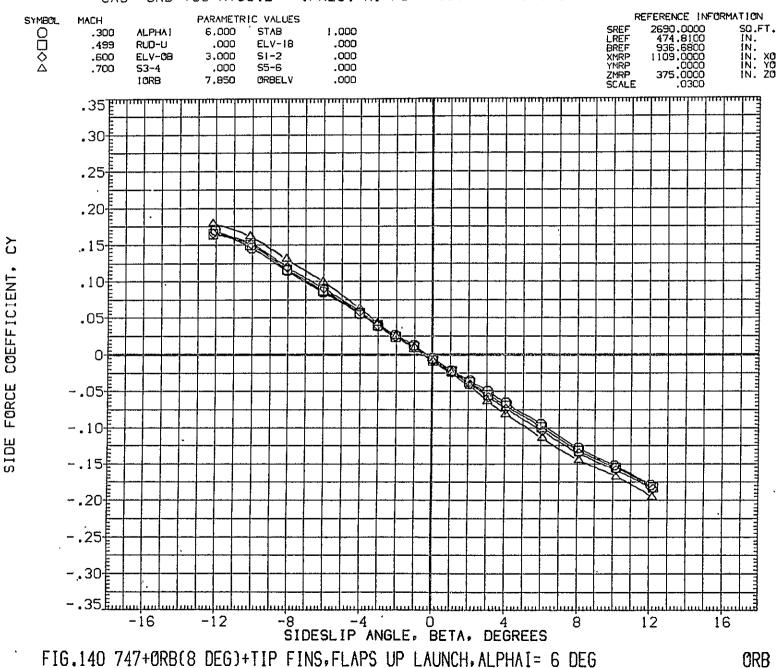
747





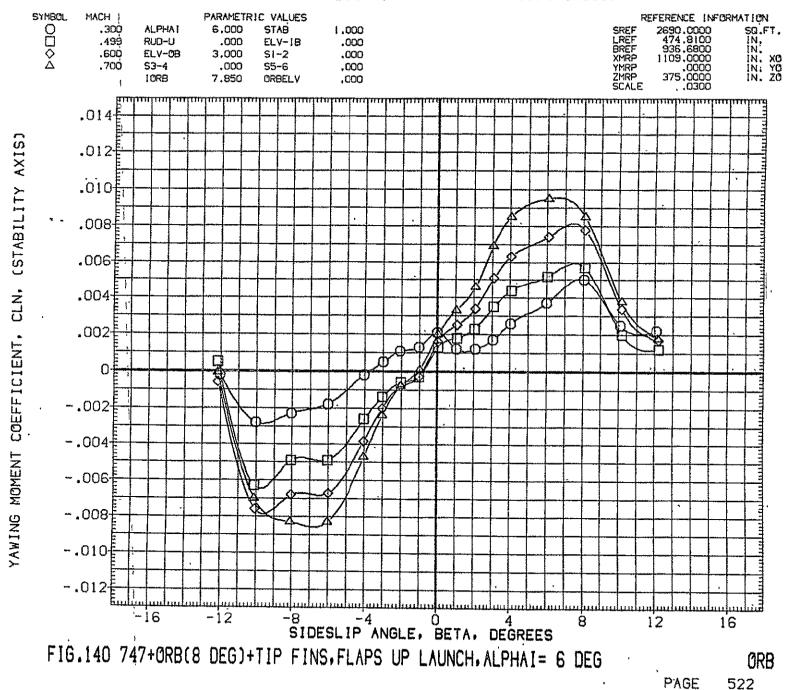


CA5 ORB TC5 AT38.2 (PRES. K1 FO H15.6 V9.1)(XGMO89)



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ORB TC5 AT38.2 (PRES. K1 FO H15.6 V9.1)(XGM089) CA5 SYMBOL MACH PARAMETRIC VALUES REFERENCE INFORMATION SO.FT. IN. IN. XO IN. YO IN. ZO SREF LREF BREF XMRP YMRP ZMRP SCALE 2690.0000 474.9100 936.6800 1109.0000 .300 ALPHA1 6.000 STAB 1,000 ,000 .499 RUD-U .000 ELV-18 .600 ELV-09 3,000 S1-2 .000 ,0000 375,0000 ,0300 .700 53-4 .000 S5-6 .000 10RB 7.850 ORBELV ,000 .07 Emliminimini .06[(STABILITY AXIS) .05₽ .04[.03[.02 CSL. .01- ROLLING MOMENT COEFFICIENT, . 0--.01 -.02 -.03 - 04 .-.05 -.06 ساستا 07. -12 -8 -4 0 4 SIDESLIP ANGLE, BETA, DEGREES 12 16 FIG.140 747+0RB(8 DEG)+TIP FINS, FLAPS UP LAUNCH, ALPHAI= 6 DEG

ORB

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PAGE

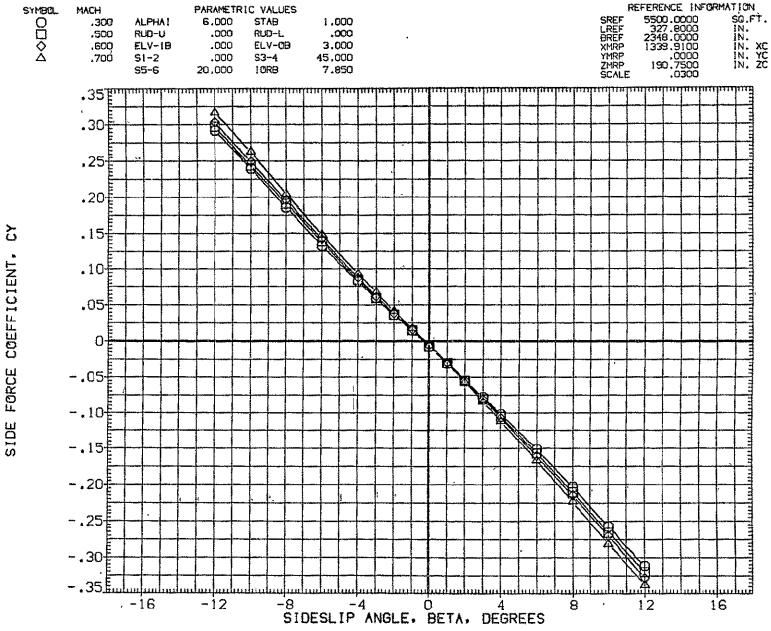


FIG.141 747+ORB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPHAI=6, TC OFF TOT
PAGE 524

CA5 K1 FO H15.6 V9.1 (PLUS. ORB NOZ AT38.2)(RGM093) SYMBOL PARAMETRIC VALUES MACH REFERENCE INFORMATION 0000 5500.0000 327.8000 2348.0000 1339.9100 .0000 190.7500 .0300 SQ.FT. IN. IN. XC IN. YC IN. ZC SREF LREF BREF XMRP YMRP ZMRP SCALE .300 ALPHA I 6.000 STAB 1.000 .500 RUD-U .000 RUD-L .000 .600 ELV-IB .000 ELV-0B 3.000 .700 S1-2 .000 S3-4 45,000 S5-6 20,000 10RB 7.850 .035 (STABILITY AXIS) .030[.025 .020 .015 CLN. .010 .005 YAWING MOMENT COEFFICIENT, 0--.005 -.010 -.015[-.020 -.025 -.030 -8 -4 0 4 SIDESLIP ANGLE, BETA, DEGREES -16 -i2 8 12 16

FIG.141 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPHAI=6, TC OFF TOT

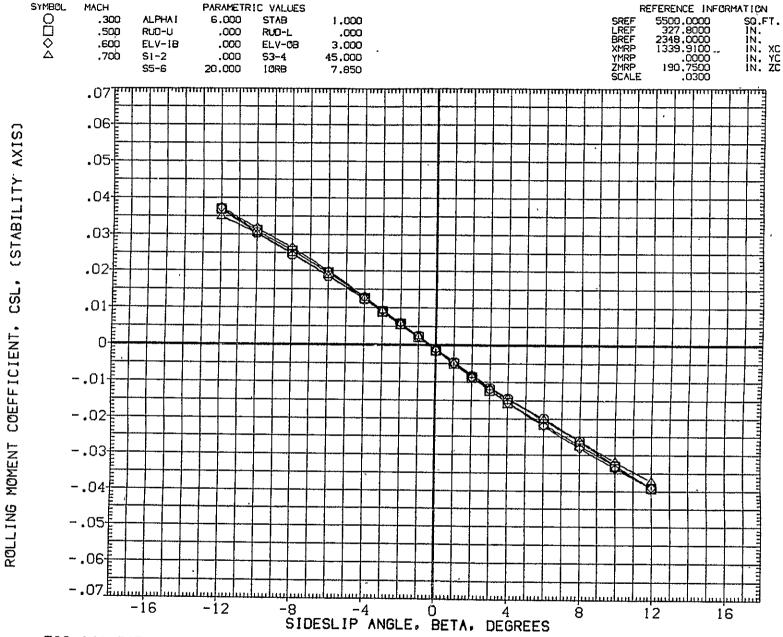


FIG.141 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPHAI=6, TC OFF TOT

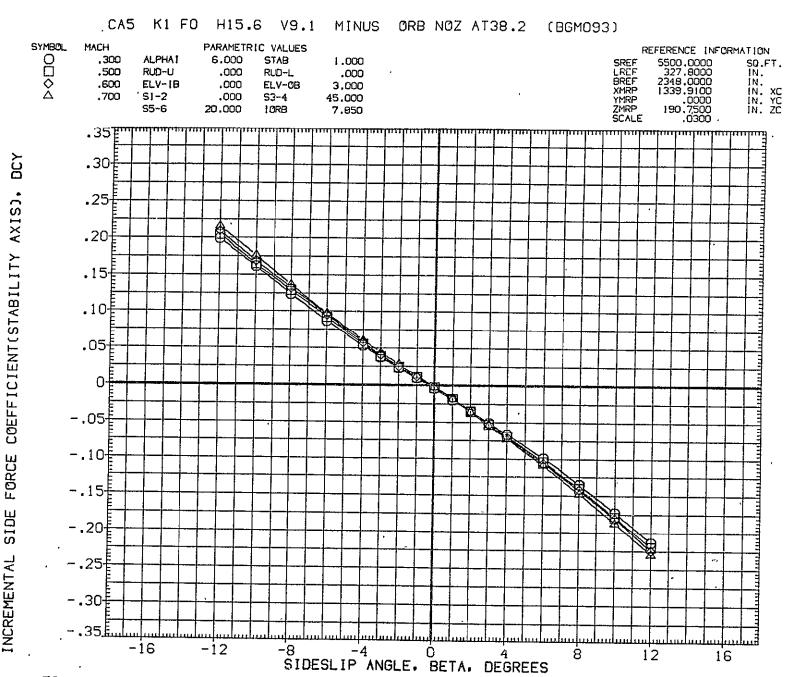


FIG.142 747+0RB(8 DEG)+TIP FINS+STD SPD BRK,FLAPS UP LAUNCH, ALPHAI=6,TC 0FF 747

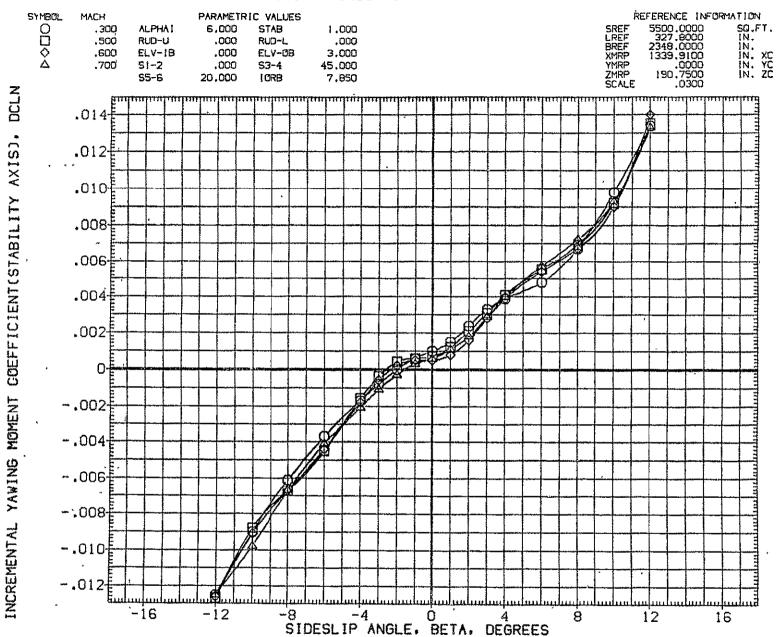


FIG.142 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPHAI=6, TC 0FF 747

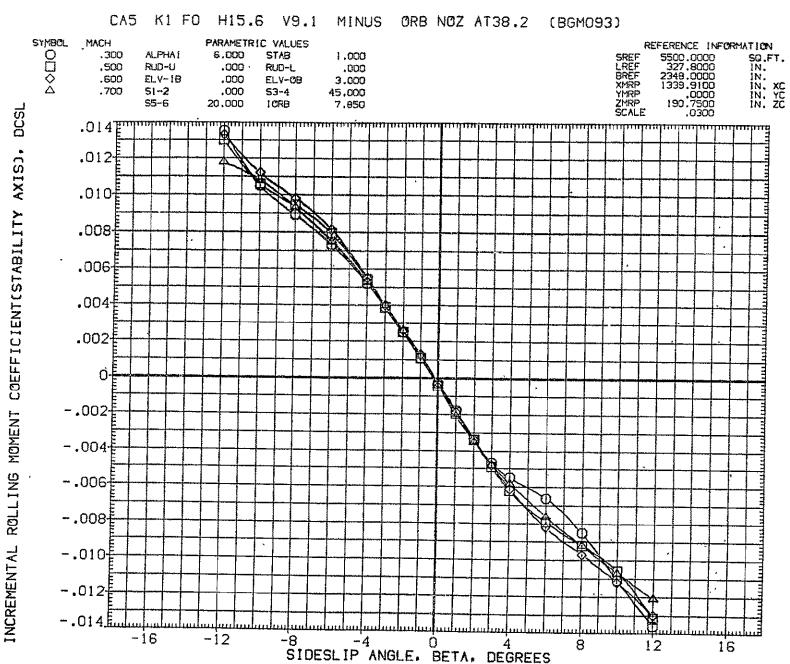


FIG.142 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPHAI=6.TC OFF 747

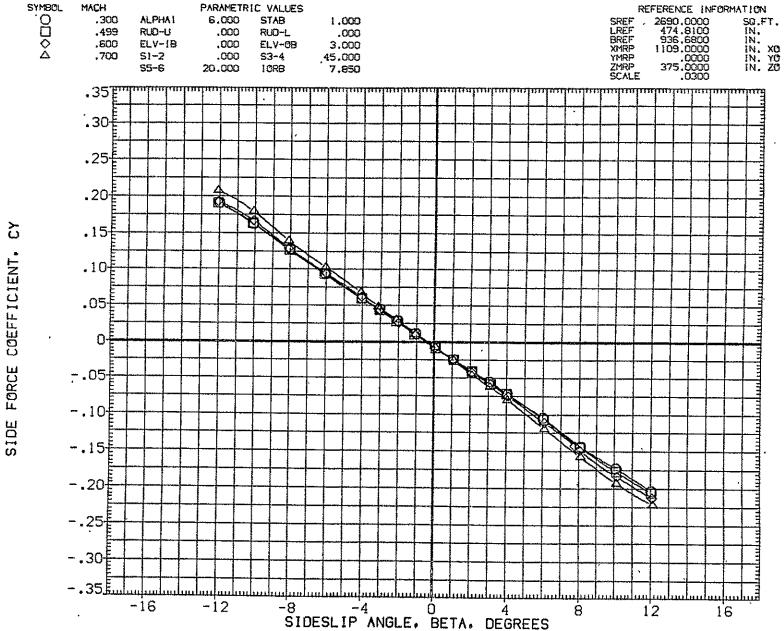


FIG.143 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPHAI=6.TC OFF ORB

ØRB NOZ AT38.2 (PRES. K1 FO H15.6 V9.1)(XGMO93) CA5 REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH SQ.FT. IN. IN. XO IN. YO IN. ZO SREF LREF BREF XMRP YMRP ZMRP SCALE 2690.0000 474.8100 936.6800 1109.0000 .0000 375.0000 .300 ALPHA I 5,000 STAB 1.000 .000 .499 RUD-U .000 RUD-L .600 ELV-IB .000 ELV-0B 3.000 .000 .700 51-2 S3-4 45.000 10RB 55-6 20,000 7.850 .035 (STABILITY AXIS) .030‡ .025[.020[.015[CLN .010[.005‡ YAWING MOMENT COEFFICIENT. 0-6 -.005[-.010 -.015[-.020[-.025 -.030[] SIDESLIP ANGLE. BETA. DEGREES -16 -i2 12 8 16

FIG.143 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPHAI=6, TC OFF ORB

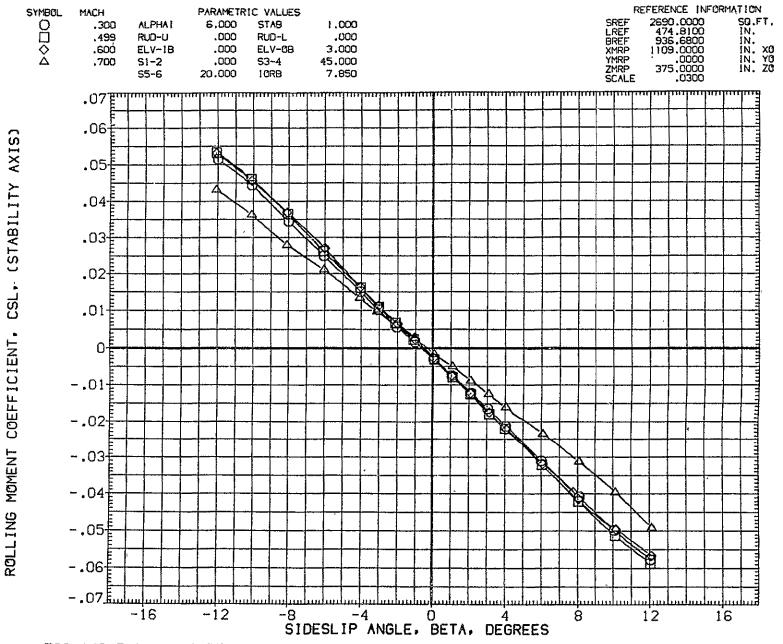


FIG.143 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPHAI=6, TC OFF ORB

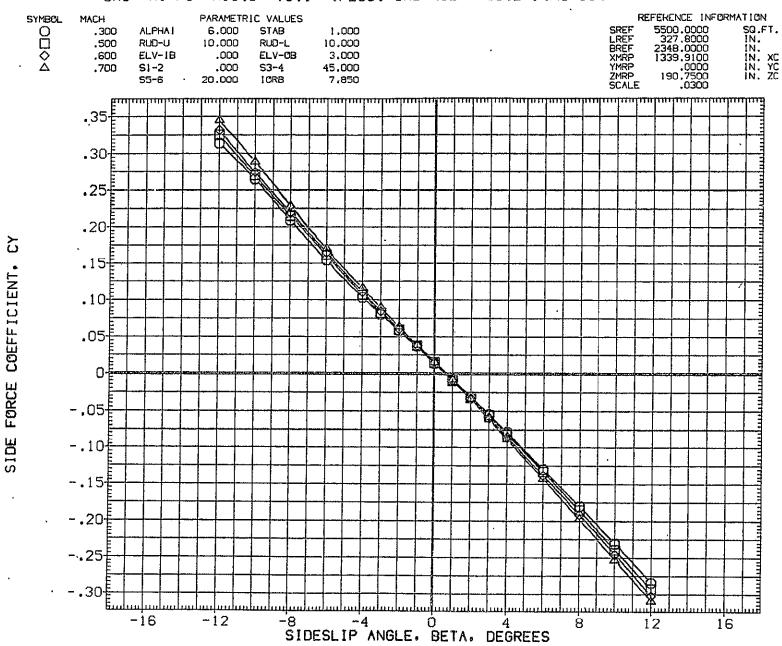


FIG.144 747+0RB(8 DEG)+TIP FINS+STD SPD BRK,FLAPUP LNCH,ALP=6,TC OFF,R=10/10 TOT

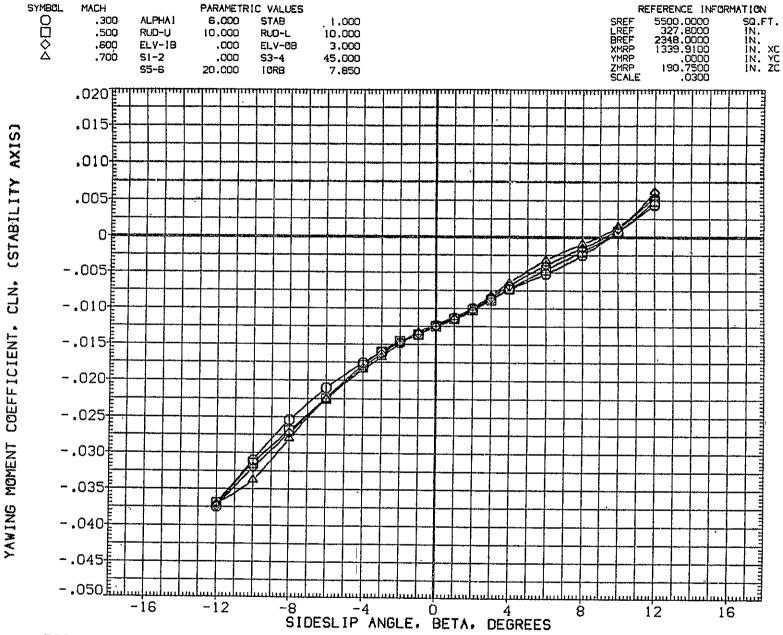


FIG.144 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPUP LNCH, ALP=6, TC OFF, R=10/10 TOT PAGE 534

K1 FO H15.6 V9.1 (PLUS. ORB NOZ AT38.2)(RGM094) REFERENCE INFORMATION SYMBOL. PARAMETRIC VALUES MACH 5500.0000 327.8000 2348.0000 1339.3100 .0000 190.7500 SQ.FT. SREF LREF BREF XMRP 6.000 STAB 1,000 ,300 ALPHA1 10.000 .500 ,600 RUD-U 10.000 RUD-L N. XC IN. YC IN. ZC 3,000 ELV-18 .000 ELY-08 45.000 7.850 YMRP ZMRP SCALE .000 S3-4 .700 S1-2 55-6 20.000 IORB .07 քարարա .06 AXIS) .05# (STABILITY .04[.03[.02 CSL. .01 COEFFICIENT, 0# -.01 -.02 ROLLING MOMENT -.03 - ..04 -.05 -.06

FIG.144 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPUP LNCH, ALP=6, TC 0FF, R=10/10 T0T
PAGE 535

SIDESLIP ANGLE, BETA, DEGREES

12

16

-16

-12

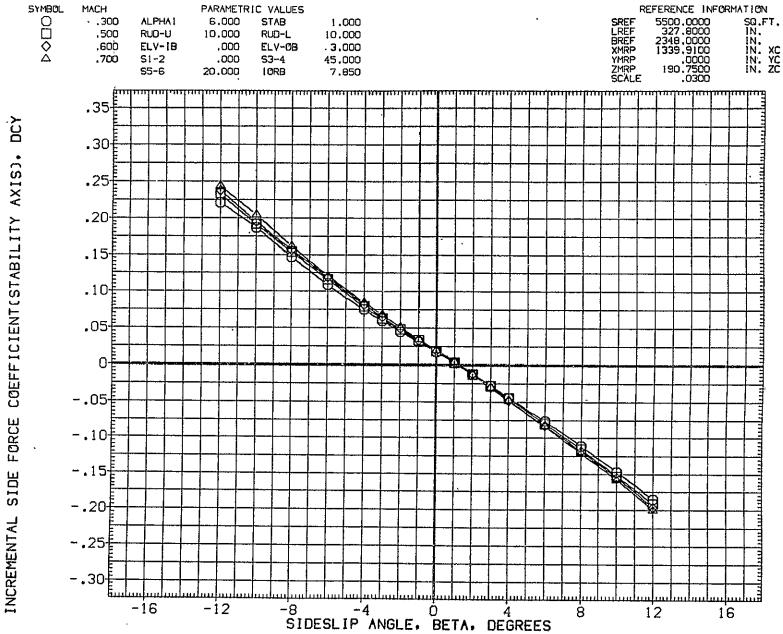


FIG.145 747+ORB(8 DEG)+TIP FINS+STD SPD BRK, FLAPUP LNCH, ALP=6, TC OFF, R=10/10 747



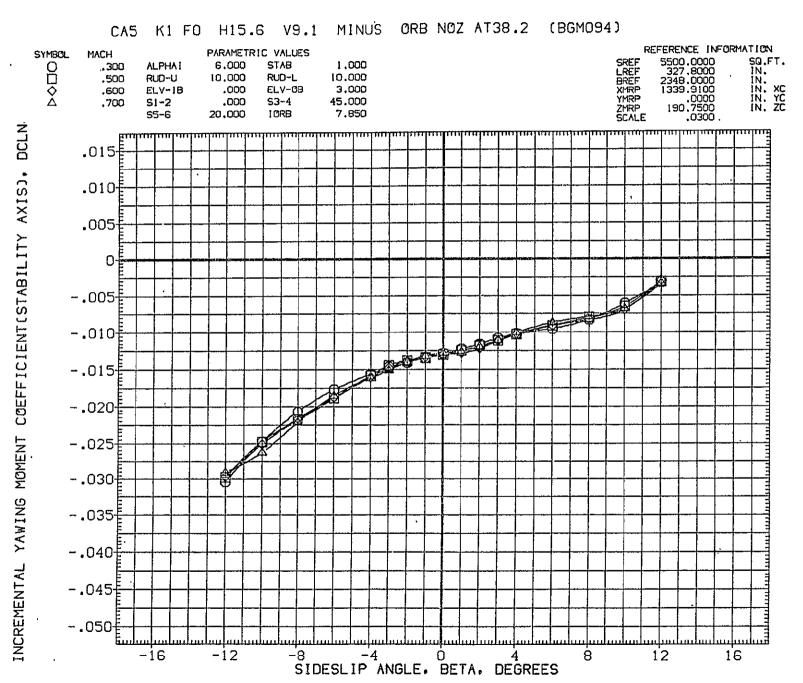


FIG.145 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPUP LNCH, ALP=6,TC OFF, R=10/10 747

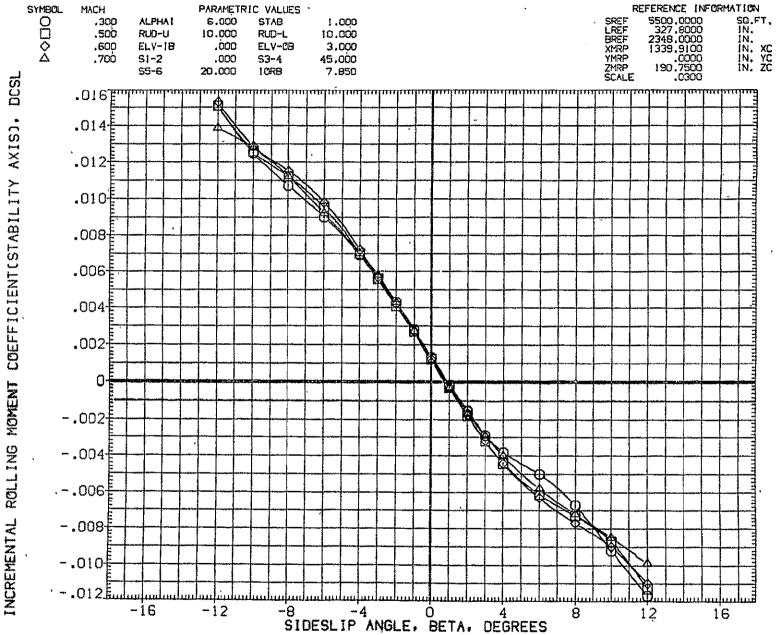


FIG.145 747+0RB(8 DEG)+TIP FINS+STD SPD BRK,FLAPUP LNCH,ALP=6,TC 0FF,R=10/10 747

CA5 ORB NOZ AT38.2 (PRES. K1 FO H15.6 V9.1)(XGMO94) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH SQ.FT. 2690.0000 0000 6.000 STAB 1.000 ,299 ALPHA1 LREF BREF 474.8100 936.6800 10.000 RUD-L 10.000 RUD~U IN. XO IN. YO IN. ZO .500 1 109 .0000 .0000 375 .0000 .000 ELV-OB 3,000 ELV-IB .601 XMRP YMRP ZMRP SCALE .700 S1-2 .000 53-4 45.000 S5-6 20.000 TORB 7,850 .35 Em .30= .20 .15£ COEFFICIENT. .10[.05ŧ 0 FORCE -.05 -.10E SIDE -.15[-.20E -.25 -.30₽ -12 -16

FIG.146 747+ORB(8 DEG)+TIP FINS+STD SPD BRK, FLAPUP LNCH, ALP=6.TC OFF, R=10/10 ORB

0 SIDESLIP ANGLE, BETA, DEGREES 16

12

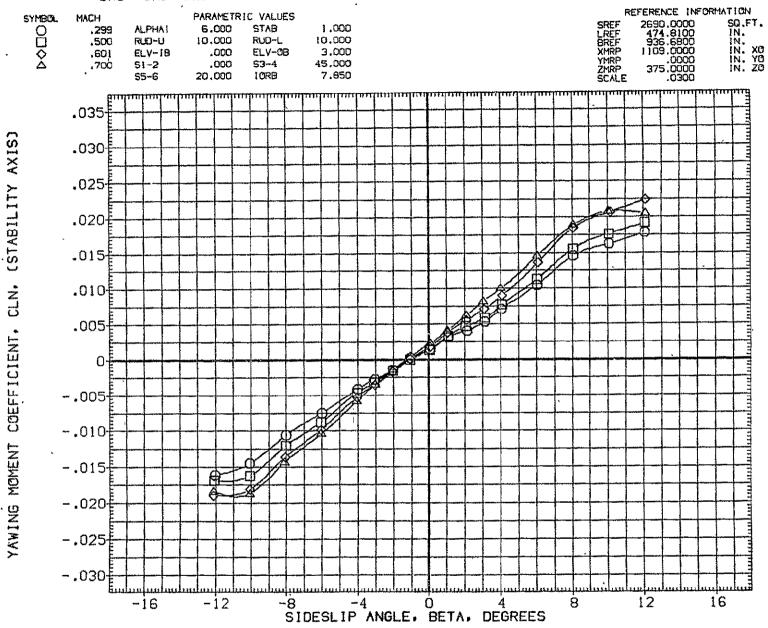


FIG.146 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPUP LNCH, ALP=6, TC OFF, R=10/10 ORB



CAS ORB NOZ AT38.2 (PRES. K1 FO H15.6 V9.1)(XGM094)

H PARAMETRIC VALUES
299 ALPHAI 6.000 STAB 1.000 S

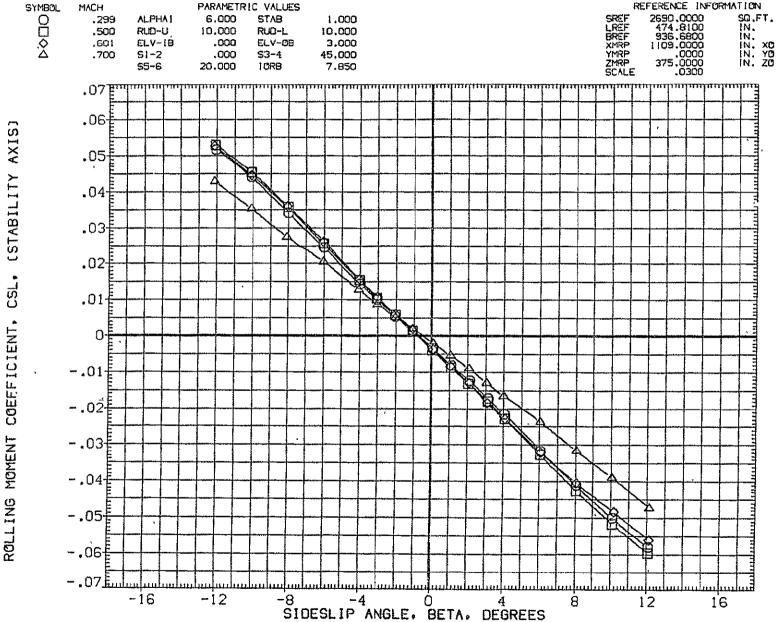


FIG.146 747+0RB(8 DEG)+TIP FINS+STD SPD BRK, FLAPUP LNCH, ALP=6, TC OFF, R=10/10 ORB

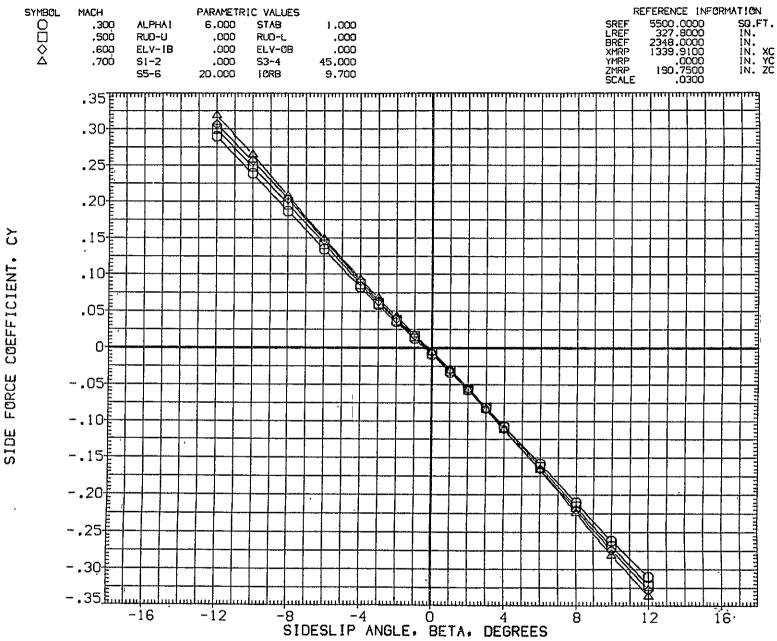


FIG.147 747+0RB(10 DEG)+TIP FINS+STD SPD BRK,FLAPS UP LAUNCH,ALPHAI= 6 DEG TOTAL

CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.2)(RGMO97)

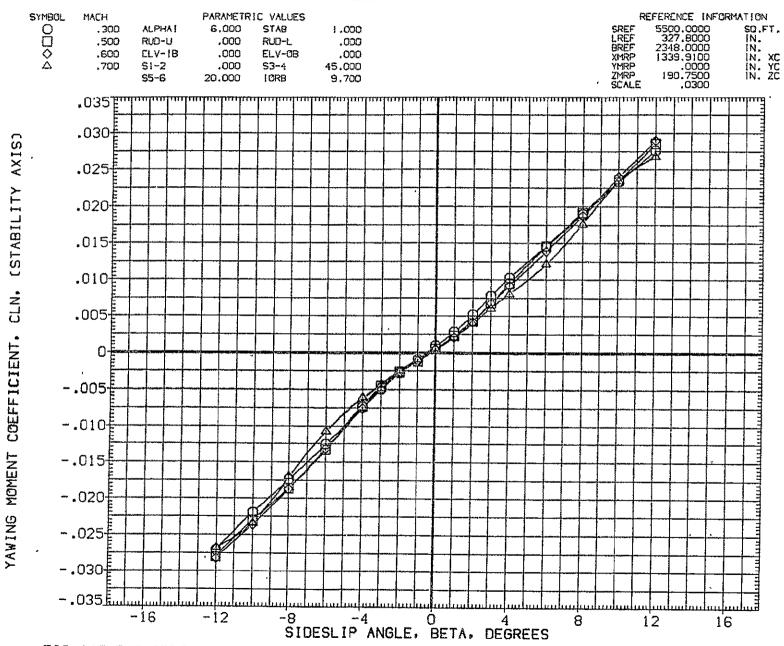


FIG.147 747+0RB(10 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPHAI= 6 DEG TOTAL

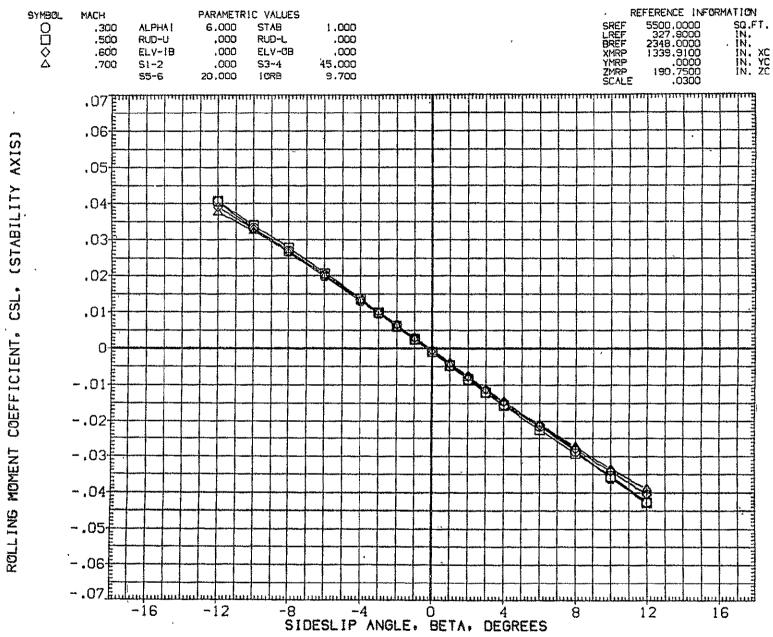
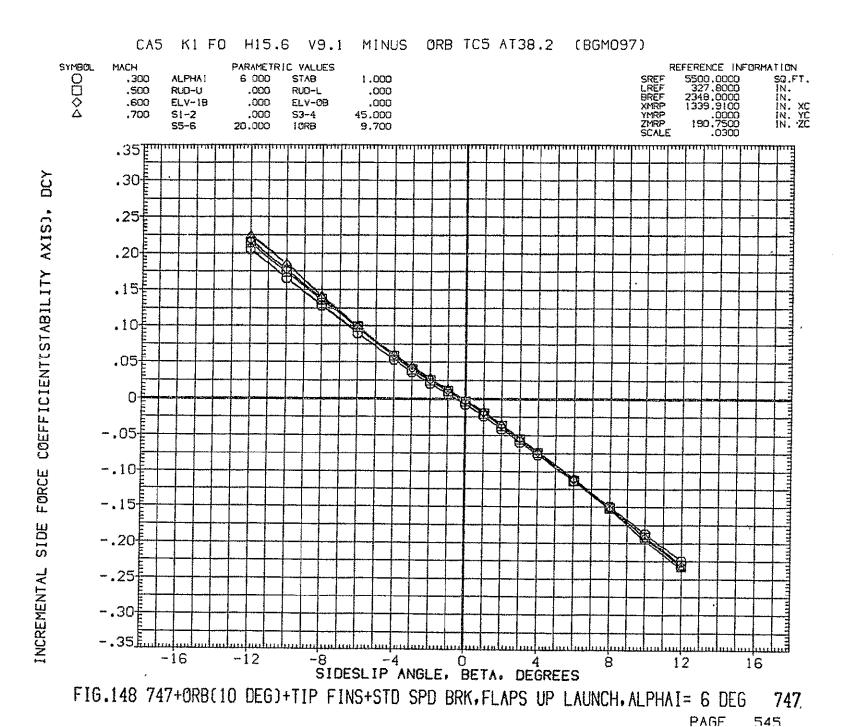


FIG.147 747+ORB(10 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPHAI = 6 DEG TOTAL
PAGE 544





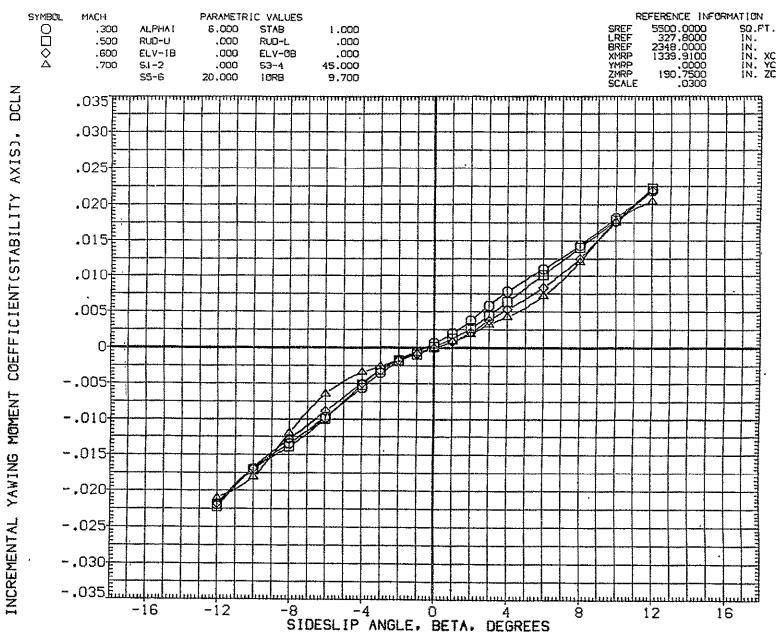


FIG.148 747+0RB(10 DEG)+TIP FINS+STD SPD BRK, FLAPS UP LAUNCH, ALPHAI = 6 DEG 743



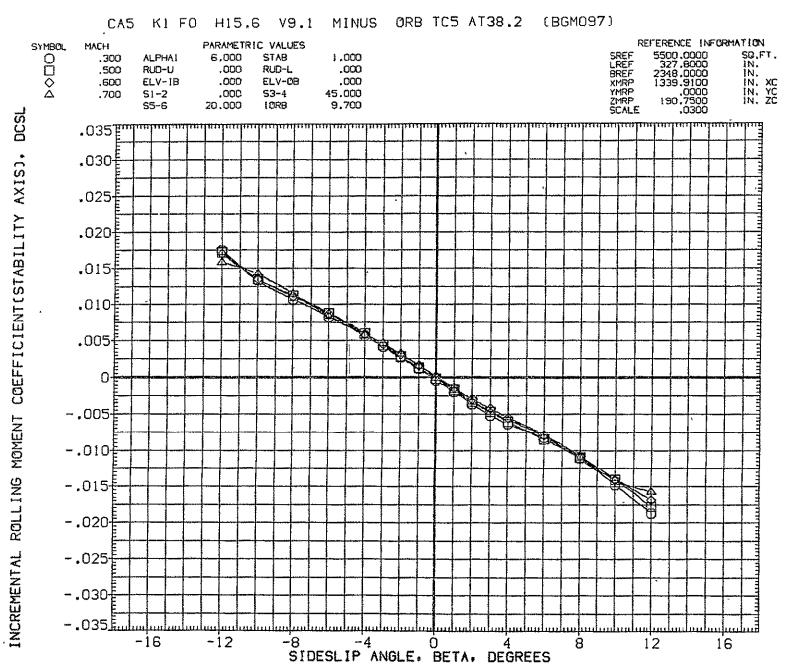


FIG.148 747+0RB(10 DEG)+TIP FINS+STD SPD BRK,FLAPS UP LAUNCH,ALPHAI= 6 DEG 747

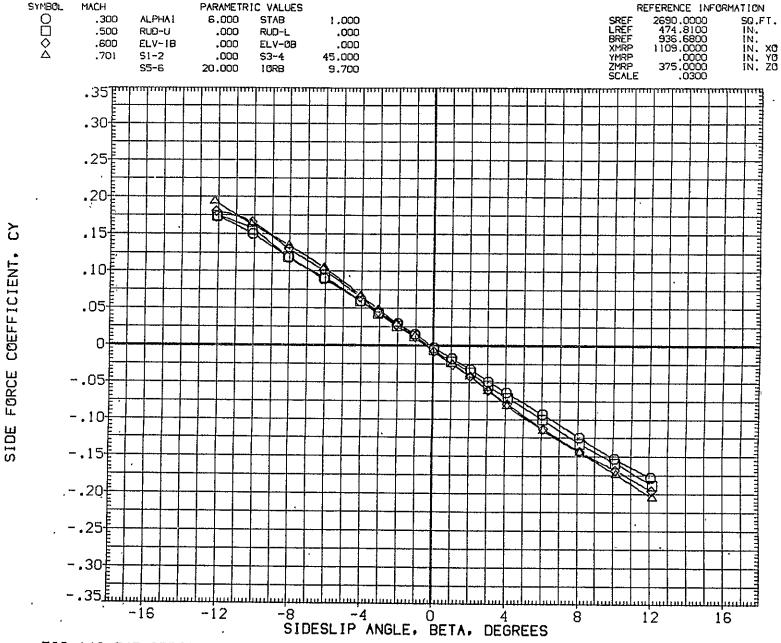


FIG.149 747+0RB(10 DEG)+TIP FINS+STD SPD BRK,FLAPS UP LAUNCH, ALPHAI= 6 DEG ORB

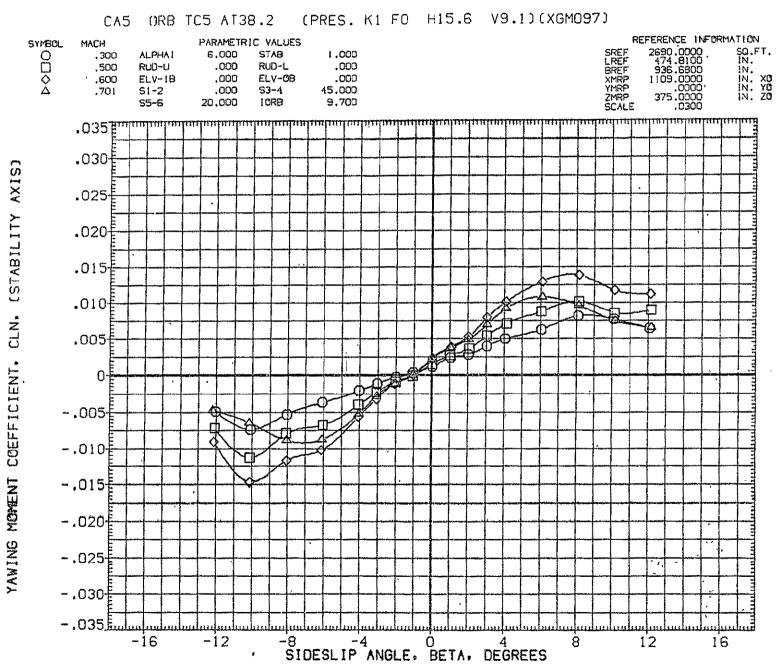


FIG.149 747+0RB(10 DEG)+TIP FINS+STD SPD BRK.FLAPS UP LAUNCH.ALPHAI= 6 DEG OR

CA5 ORB TC5 AT38.2 (PRES. K1 FO H15.6 V9.1)(XGMO97)

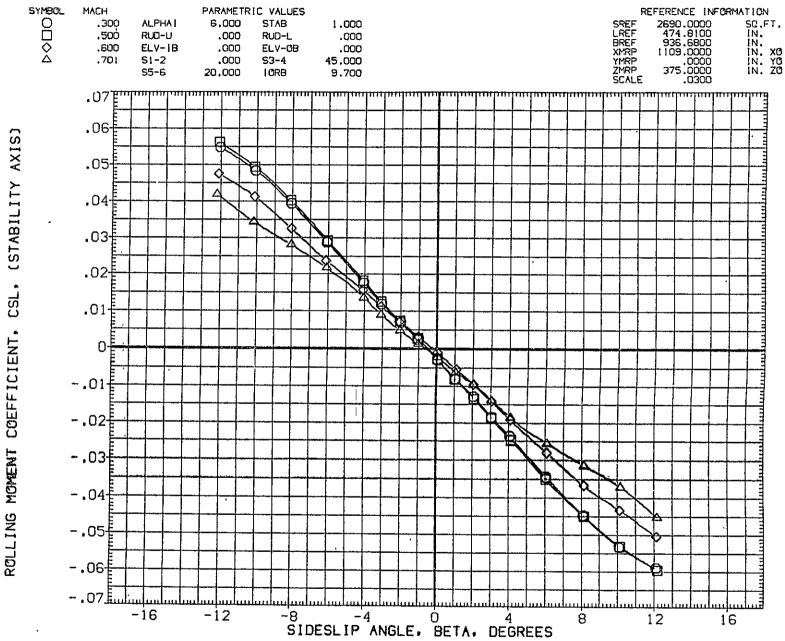


FIG.149 747+0RB(10 DEG)+TIP FINS+STD SPD BRK,FLAPS UP LAUNCH, ALPHAI = 6 DEG ORI

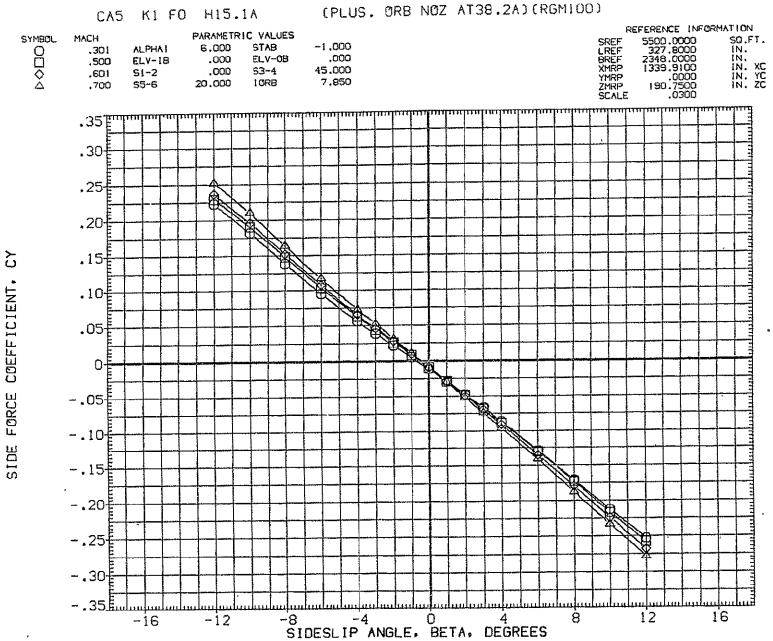


FIG.150 747+0RB(8 DEG)+STD SP, VERT TAIL OFF, FLAP UP LNCH, ALPHAI = 6.TC OFF TOTAL PAGE 551

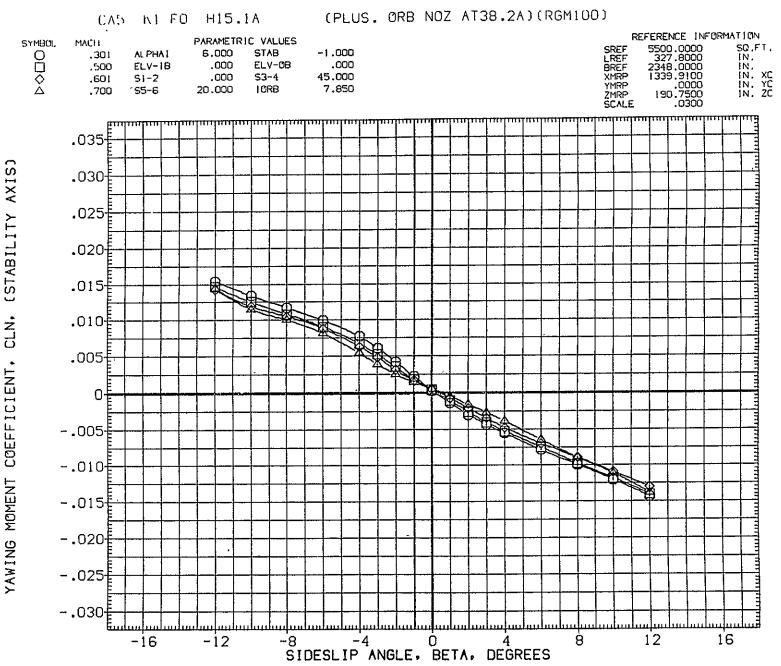


FIG.150 747+ORB(8 DEG)+STD SP, VERT TAIL OFF, FLAP UP LNCH, ALPHAI= 6, TC OFF TOTAL
PAGE 552

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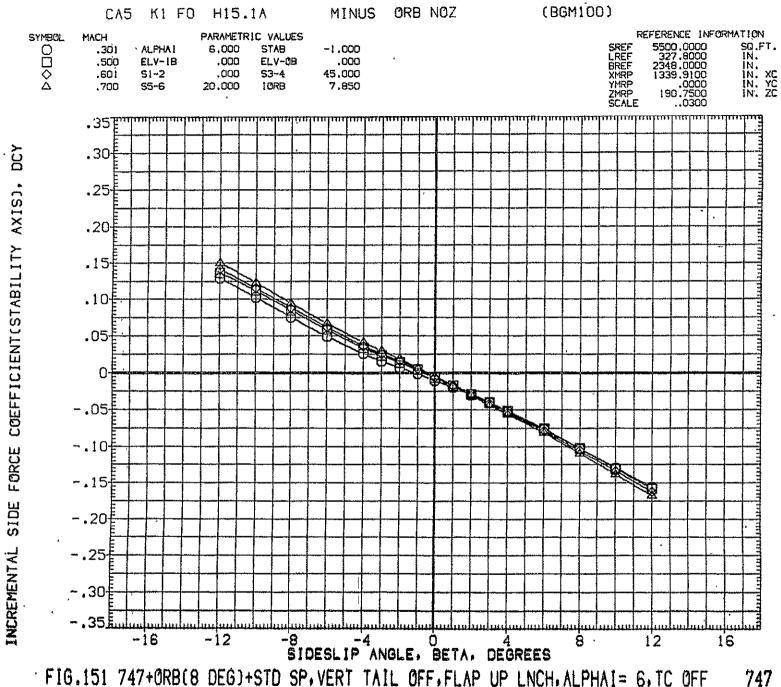


FIG.151 747+0RB(8 DEG)+STD SP, VERT TAIL OFF, FLAP UP LNCH, ALPHAI= 6, TC OFF 554 PAGE



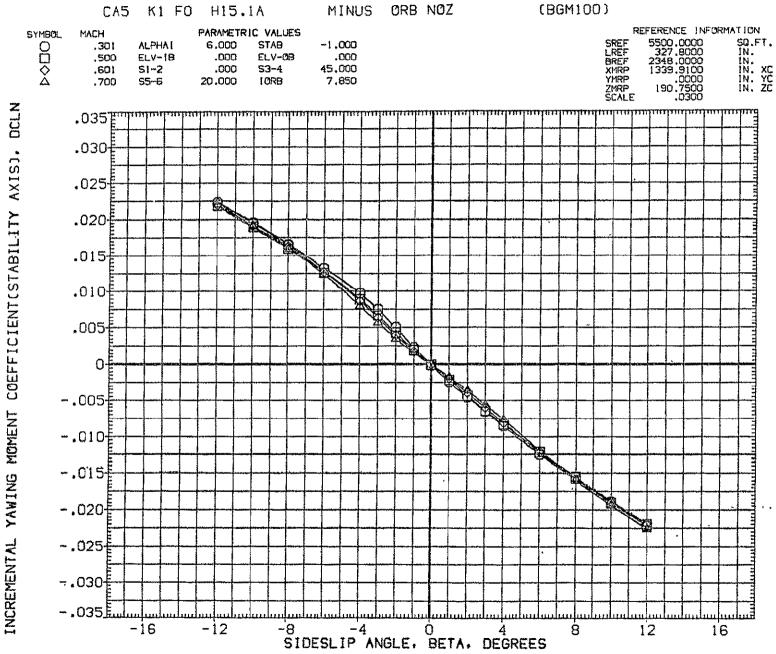


FIG.151 747+0RB(8 DEG)+STD SP. VERT TAIL OFF, FLAP UP LNCH, ALPHAI= 6,TC OFF
PAGE 555

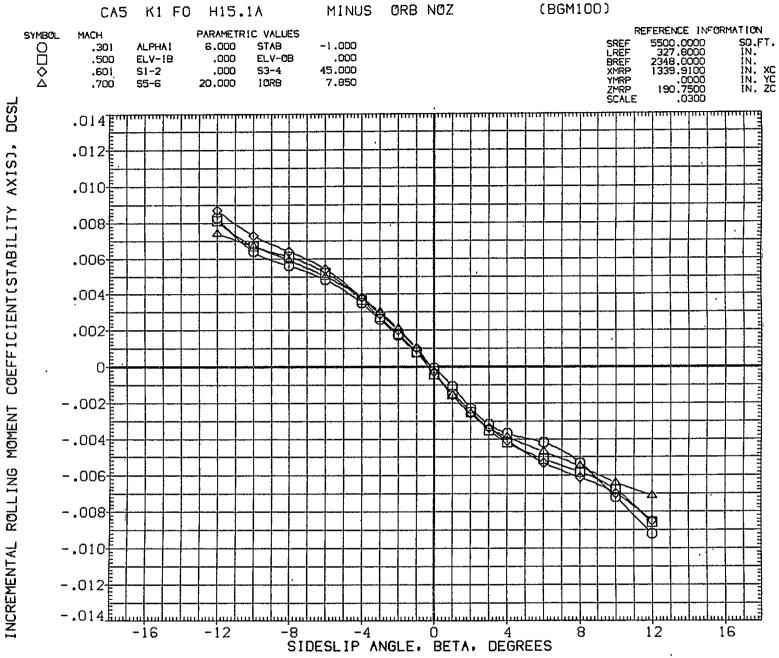


FIG.151 747+0RB(8 DEG)+STD SP, VERT TAIL OFF, FLAP UP LNCH, ALPHAI= 6, TC OFF
PAGE 556



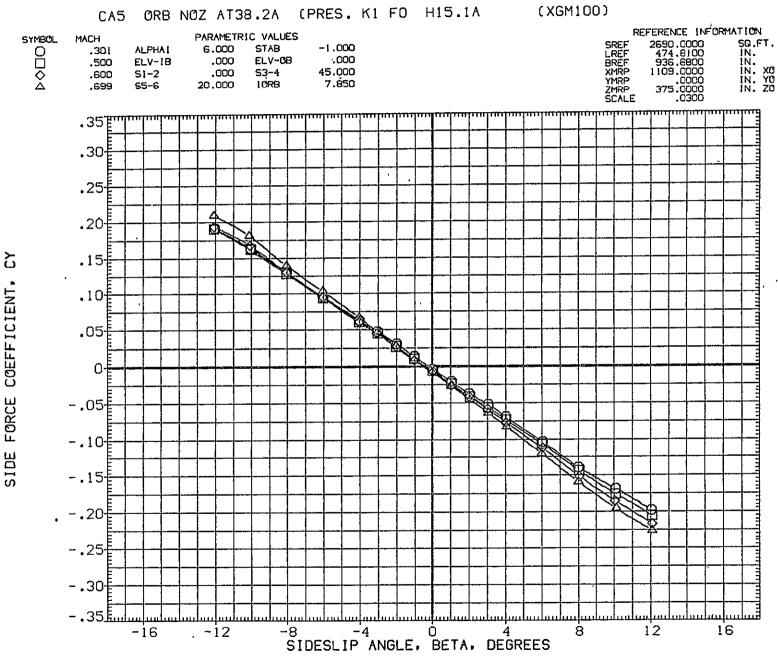


FIG.152 747+0RB(8 DEG)+STD SP, VERT TAIL OFF, FLAP UP LNCH, ALPHAI= 6, TC OFF ORB

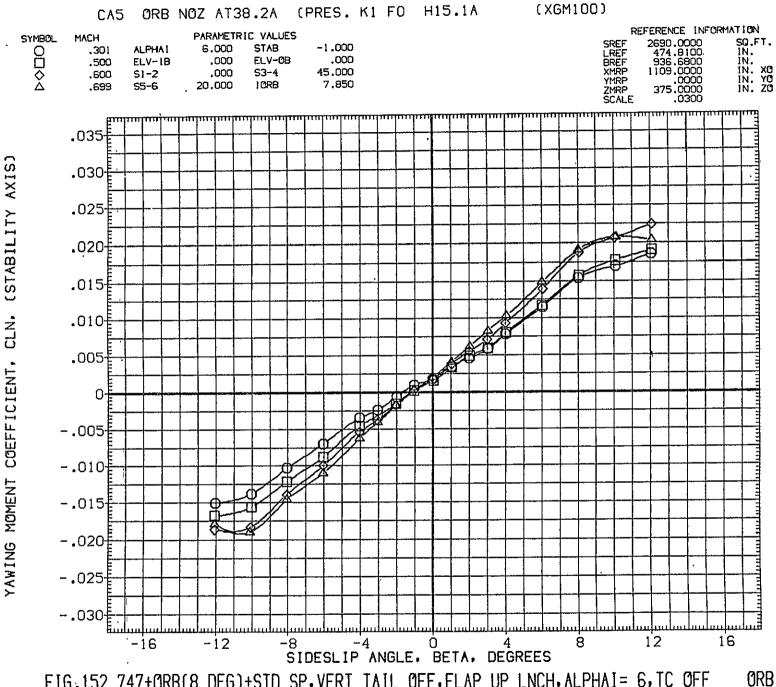


FIG.152 747+0RB(8 DEG)+STD SP, VERT TAIL OFF, FLAP UP LNCH, ALPHAI= 6, TC OFF
PAGE 558



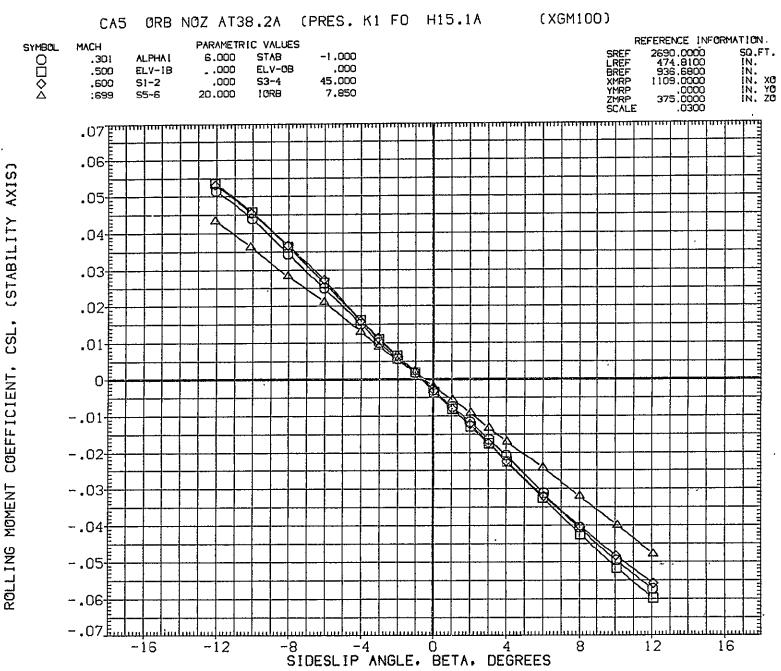
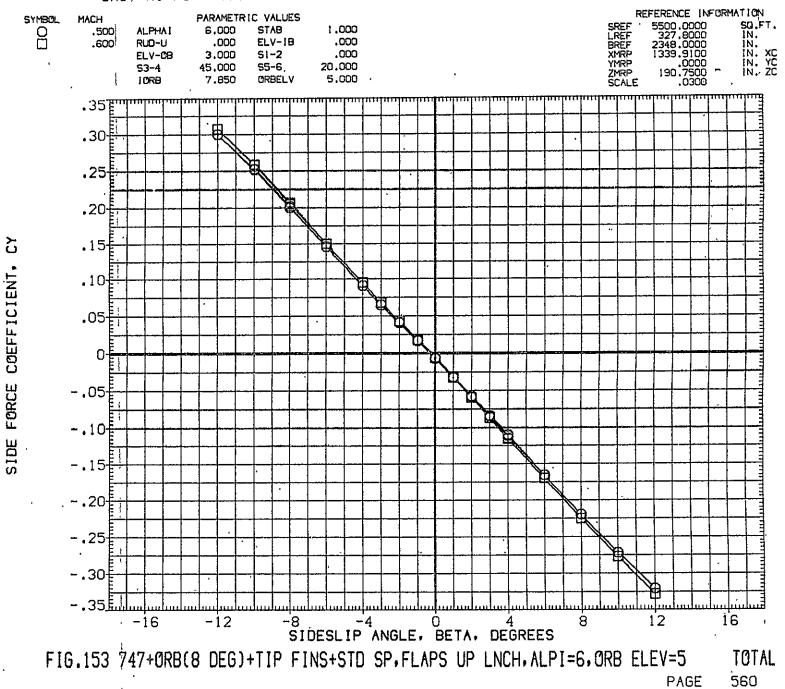
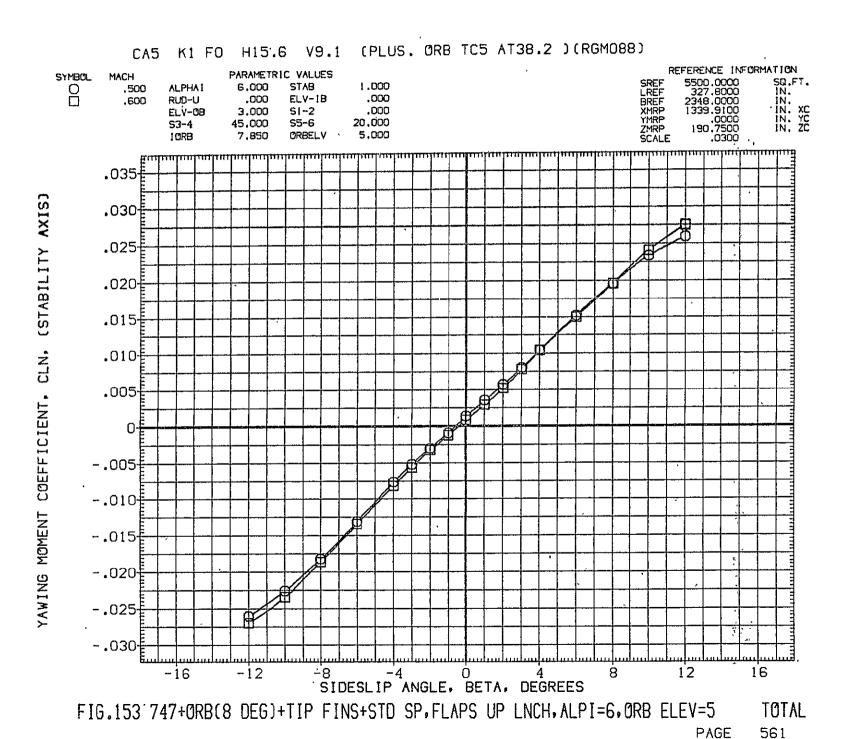


FIG.152 747+0RB(8 DEG)+STD SP, VERT TAIL OFF, FLAP UP LNCH, ALPHAI = 6, TC OFF ORB

CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.2)(RGMO88)







CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.2)(RGMO88)

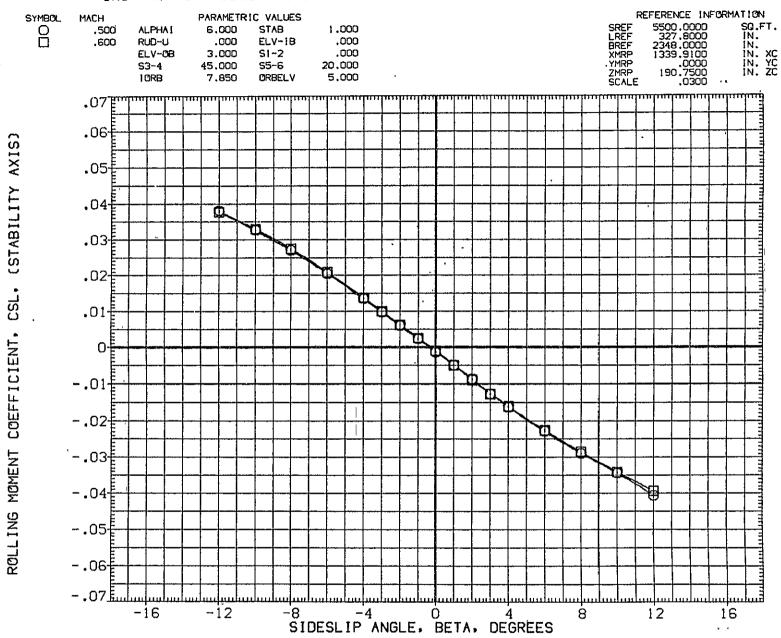
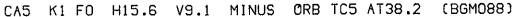


FIG.153 747+0RB(8 DEG)+TIP FINS+STD SP.FLAPS UP LNCH.ALPI=6.0RB ELEV=5

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FIG.154 747+0RB(8 DEG)+TIP FINS+STD SP.FLAPS UP LNCH.ALPI=6.0RB ELEV=5



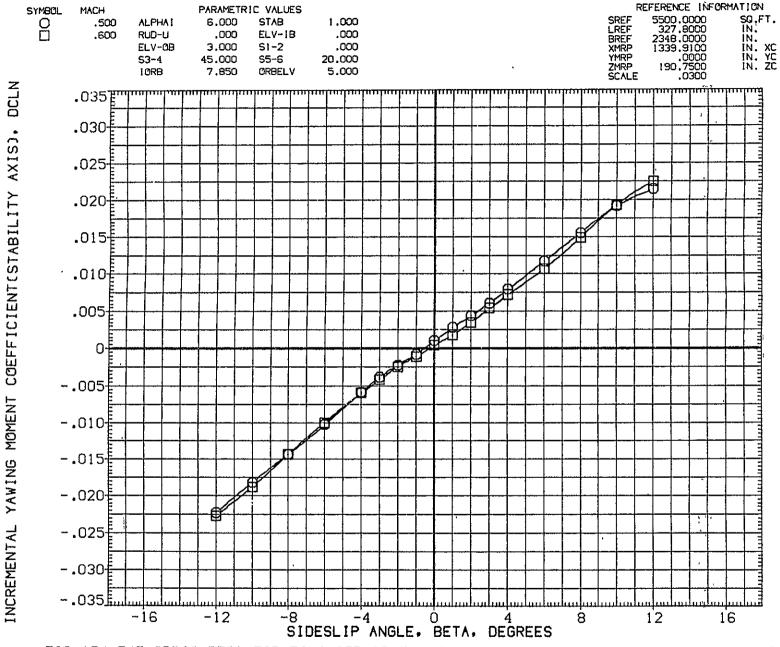
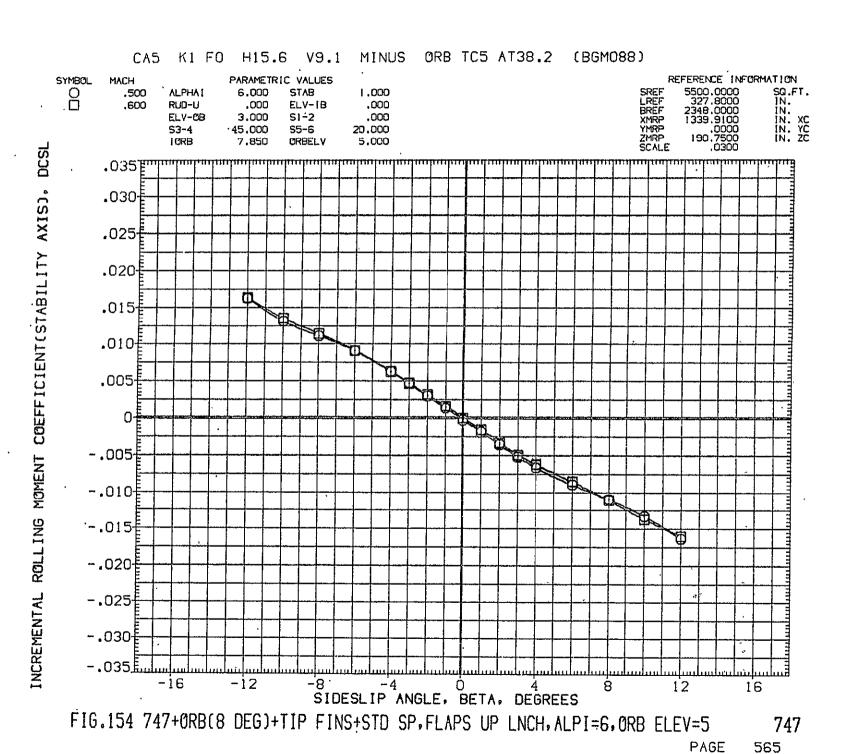


FIG.154 747+0RB(8 DEG)+TIP FINS+STD SP.FLAPS UP LNCH, ALPI=6, ORB ELEV=5
PAGE 564



CAS ORB TC5 AT38.2 (PRES. K1 FO H15.6 V9.1)(XGM088) REFERENCE INFORMATION PARAMETRIC VALUES MACH : 2690.0000 474.8100 936.6800 1109.0000 .0000 375.0000 SYMBOL. 1.000 IN. IN. IN. XO IN. YO IN. ZO 0 ALPHA I 6,000 STAB .000 .000 ELV-(B ,599 RUD-U BREF .000 3.000 51-2 ELV-08 YMRP ZMRP SCALE 20.000 -S5-6 S3-4 45,000 5,000 1 ORB 7,850 ORBELV .35TT .30- .25卡 .20[Շ .15- COEFFICIENT .10[.05 0 FORCE -.05 -.10[SIDE -.15[-.20[-.25[-.30

FIG.155 747+0RB(8 DEG)+TIP FINS+STD SP,FLAPS UP LNCH,ALPI=6,0RB ELEV=5 ORB PAGE 566 ·

Ò SIDESLIP ANGLE. BETA. DEGREES

-4

-i2

12

16

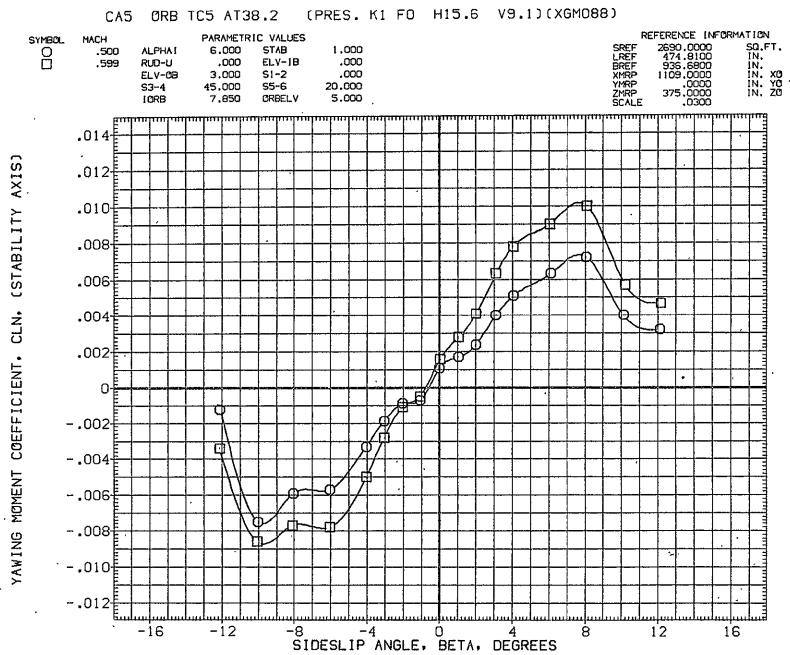


FIG.155 747+0RB(8 DEG)+TIP FINS+STD SP,FLAPS UP LNCH, ALPI=6,0RB ELEV=5

ORB

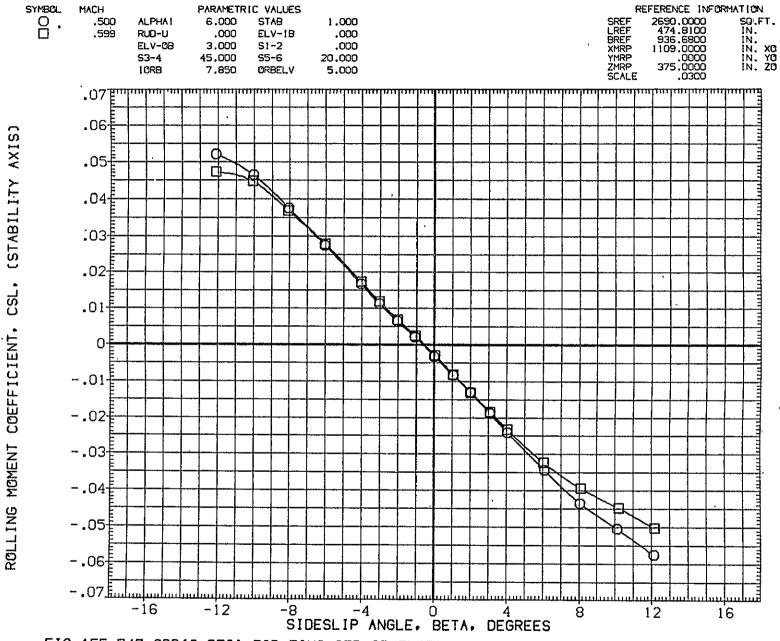


FIG.155 747+0RB(8 DEG)+TIP FINS+STD SP.FLAPS UP LNCH, ALPI=6.0RB ELEV=5 ORB

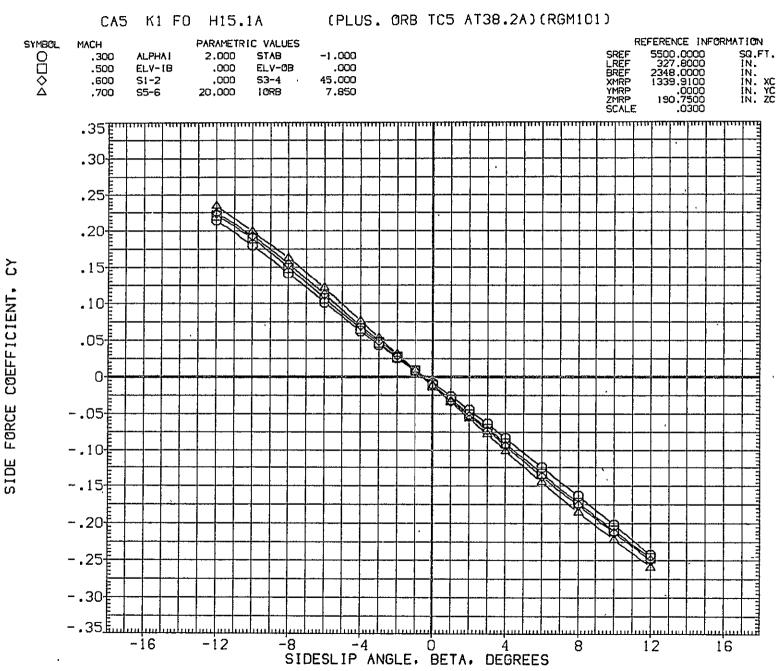


FIG.156 747+0RB(8 DEG)+STD SP. VERT TAIL OFF, FLAPS UP, ALPI=2, DE=0/0 (S=-1) TOTAL
PAGE 569

FIG.156 747+0RB(8 DEG)+STD SP, VERT TAIL OFF, FLAPS UP, ALPI=2, DE=0/0 (S=-1) TOT/

CA5 K1 FO H15.1A (PLUS. ORB TC5 AT38.2A)(RGM101) SYMBOL O O · \diamondsuit PARAMETRIC VALUES REFERENCE INFORMATION SREF LREF BREF XMRP YMRP ZMRP SCALE 5500.0000 327.8000 2348.0000 SQ.FT. STAB .300 ALPHA! 2.000 ~1.000 .000 .500 ELV-IB .000 ELV-08 IN. XC IN. YC IN. ZC .000 **\$3-4** 45.000 ..600 \$1-2 1339.9100 .0000 190.7500 7.850 IORB .700 S5~6 20,000 .035TT .030£ (STABILITY AXIS) .025 .020 .015£ .010 csr. .005# ROLLING MOMENT COEFFICIENT, 0--.005 -.010 -.015 -.020 -.025 -.030 -.035<u>L</u> արա 12

-8 -4 0 4 SIDESLIP ANGLE, BETA, DEGREES FIG.156 747+0RB(8 DEG)+STD SP, VERT TAIL OFF, FLAPS UP, ALPI=2, DE=0/0 (S=-1) TOTAL PAGE 571

16

8

-i6

-12

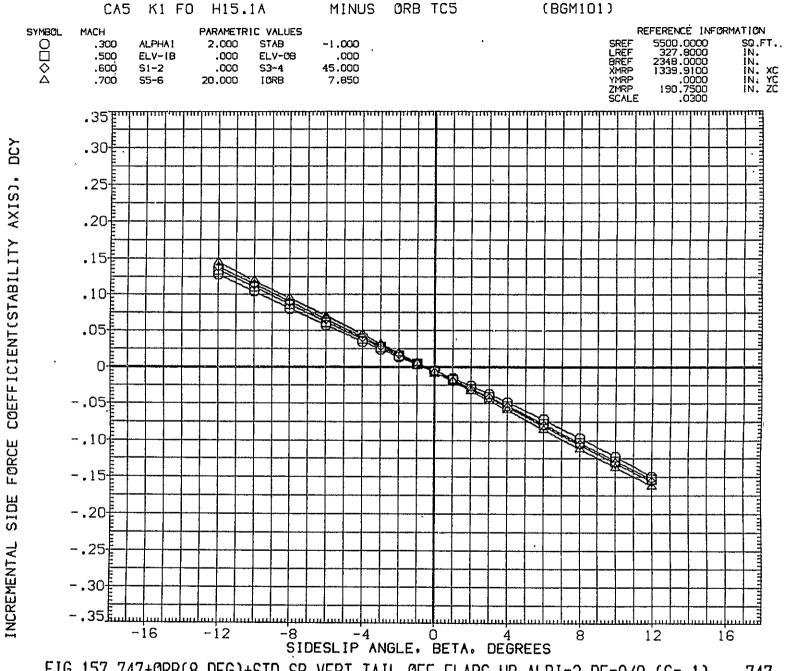


FIG.157 747+0RB(8 DEG)+STD SP, VERT TAIL OFF, FLAPS UP, ALPI=2, DE=0/0 (S=-1)

PAGE 572

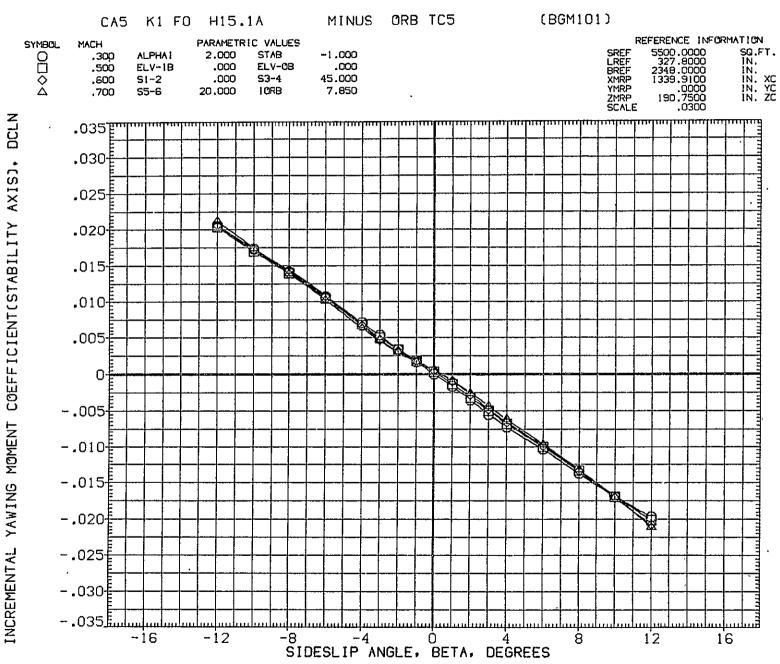


FIG.157 747+0RB(8 DEG)+STD SP, VERT TAIL OFF, FLAPS UP, ALPI=2, DE=0/0 (S=-1) 747

PAGE 573

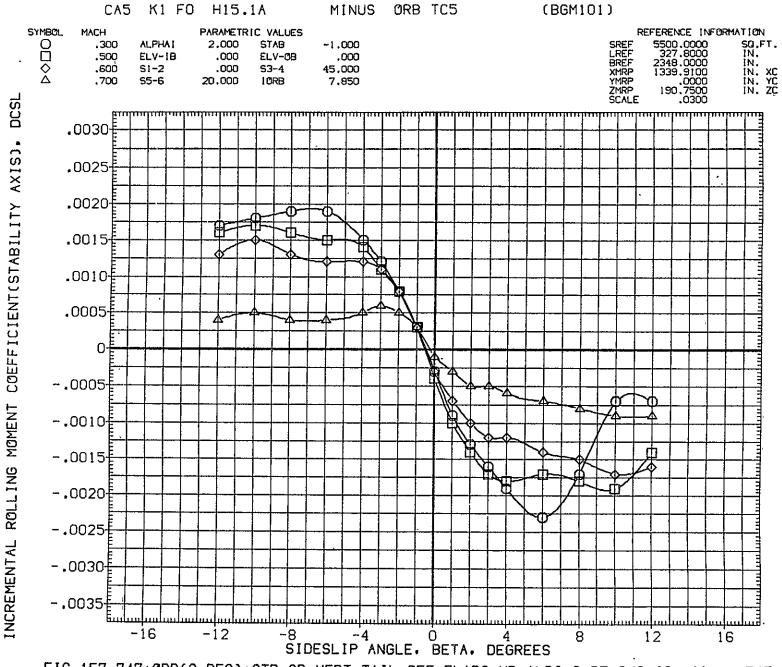
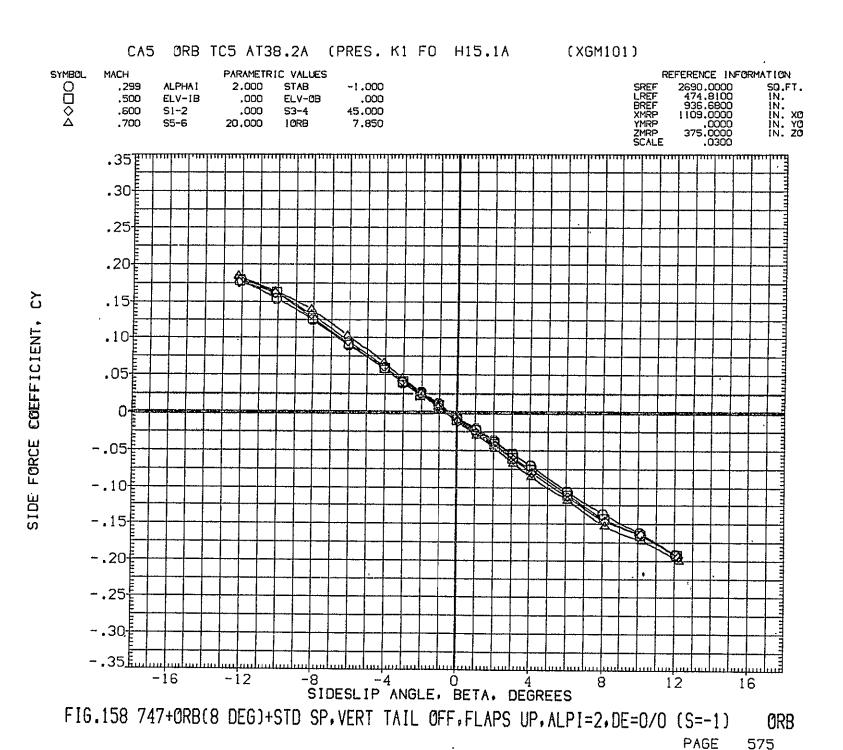


FIG.157 747+0RB(8 DEG)+STD SP, VERT TAIL OFF, FLAPS UP, ALPI=2, DE=0/0 (S=-1)

PAGE 574



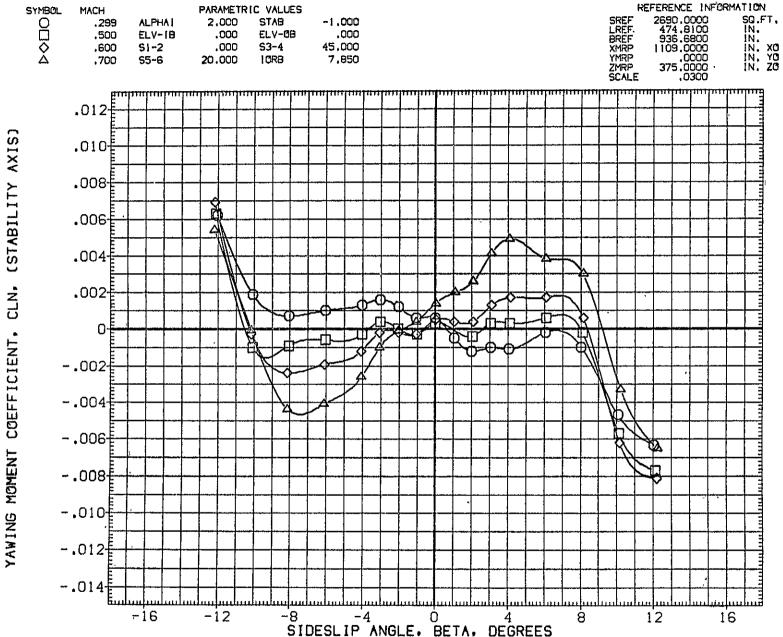


FIG.158 747+0RB(8 DEG)+STD SP, VERT TAIL OFF, FLAPS UP, ALPI=2, DE=0/0 (S=-1) ORB
PAGE 576



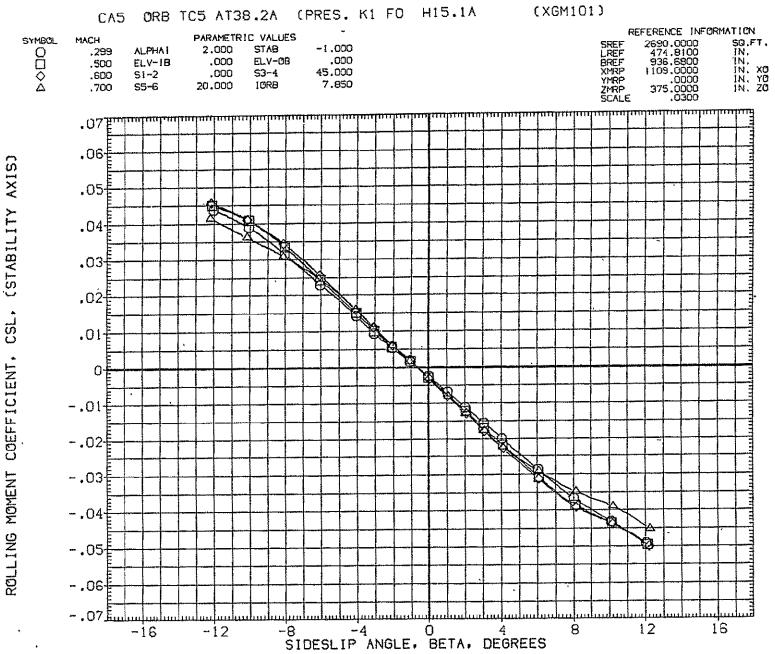


FIG.158 747+0RB(8 DEG)+STD SP, VERT TAIL OFF, FLAPS UP, ALPI=2.DE=0/0 (S=-1) ORB
PAGE 577

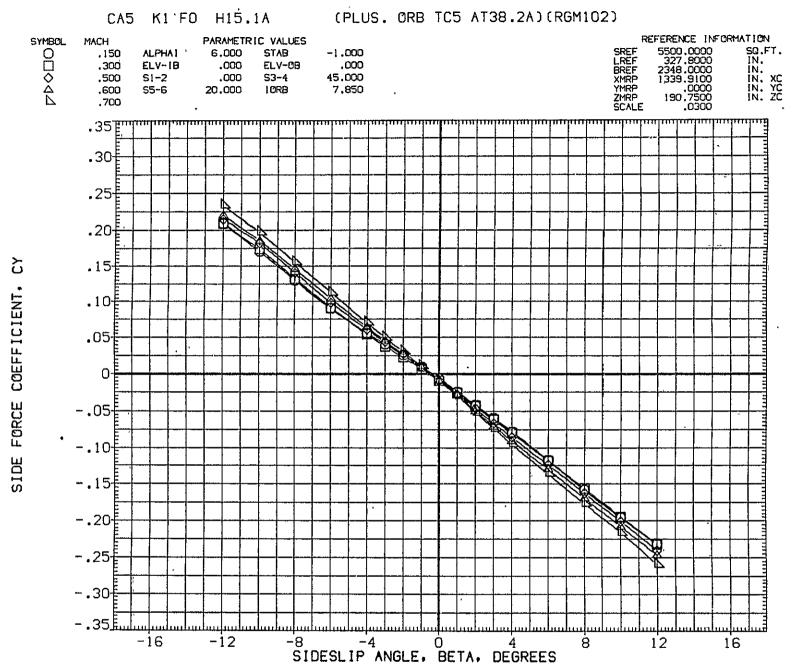


FIG.159 747+0RB(8 DEG)+STD SP.VERT TAIL OFF, FLAPS UP, ALPI=6, DE=0/0 (S=-1) TOTAL
PAGE 578



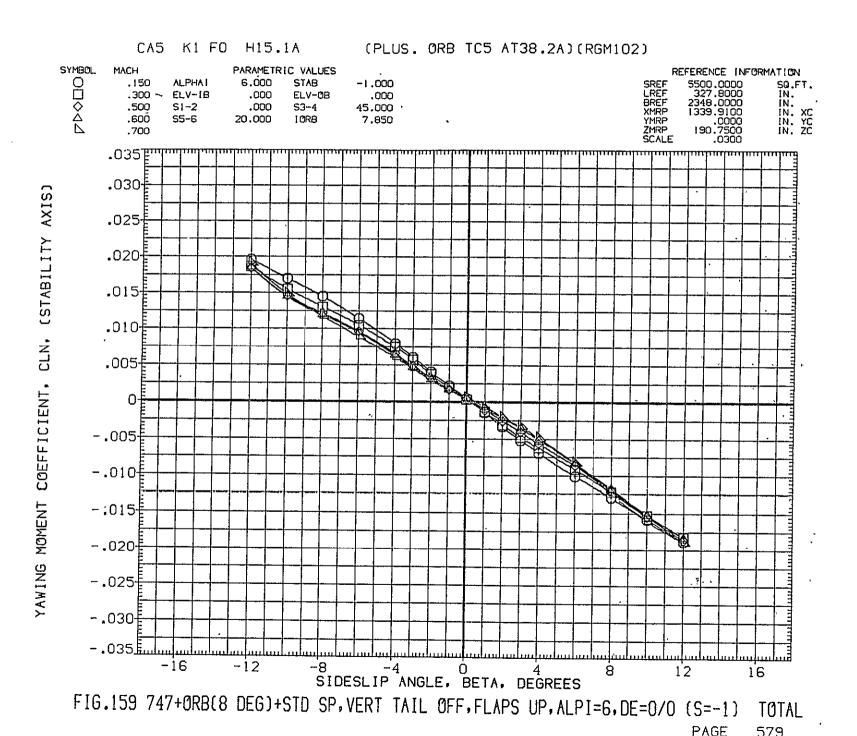
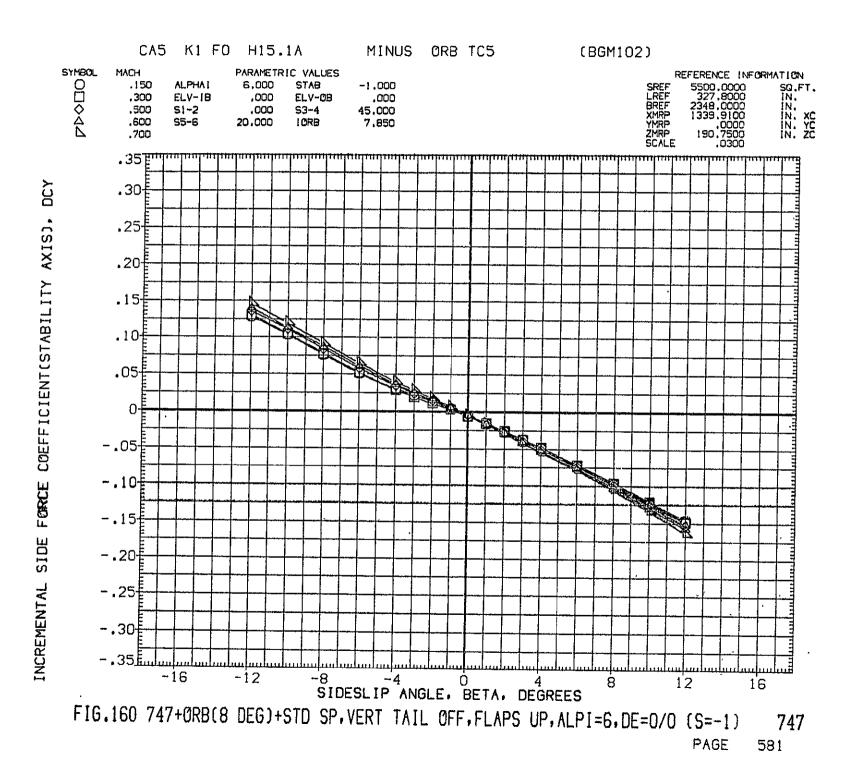


FIG.159 747+0RB(8 DEG)+STD SP, VERT TAIL OFF, FLAPS UP, ALPI=6, DE=0/0 (S=-1) PAGE 580



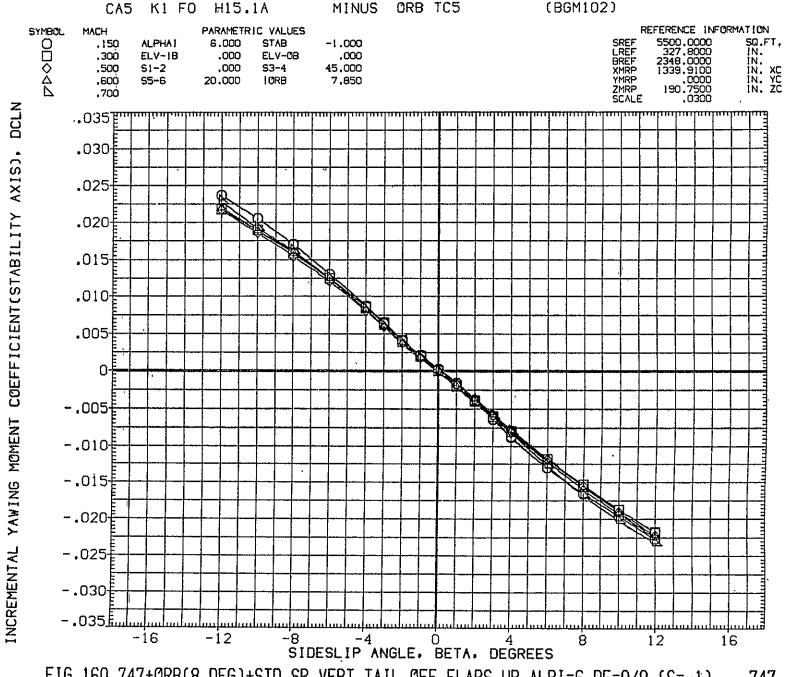


FIG.160 747+0RB(8 DEG)+STD SP, VERT TAIL OFF, FLAPS UP, ALPI=6, DE=0/0 (S=-1)

PAGE 582

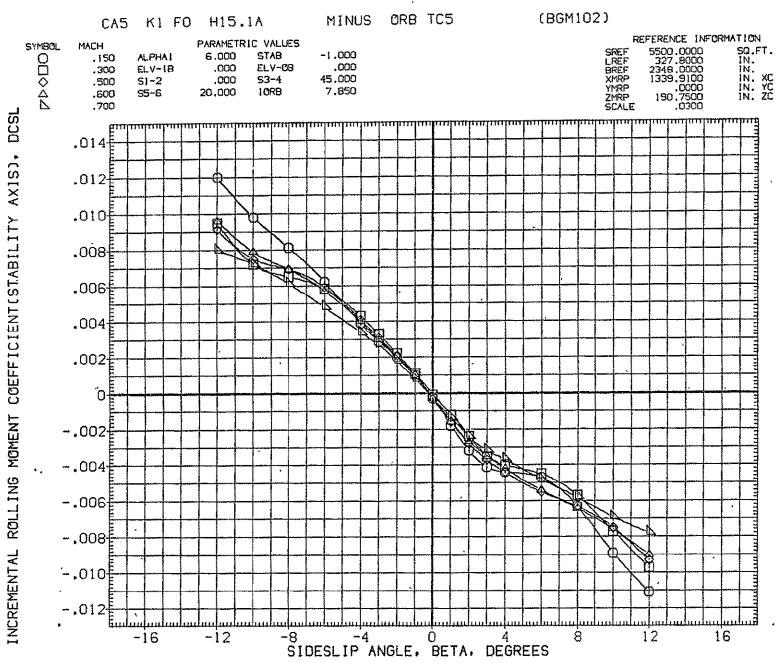


FIG.160 747+0RB(8 DEG)+STD SP, VERT TAIL OFF, FLAPS UP, ALPI=6, DE=0/0 (S=-1)
PAGE 583

FIG.161 747+0RB(8 DEG)+STD SP.VERT TAIL OFF.FLAPS UP.ALPI=6.DE=0/0 (S=-1) ORB
PAGE 584

FIG.161 747+0RB(8 0EG)+SID SP.VERT TAIL OFF.FLAPS UP.ALPI=5.0E=0/0 (S=-1) ORB

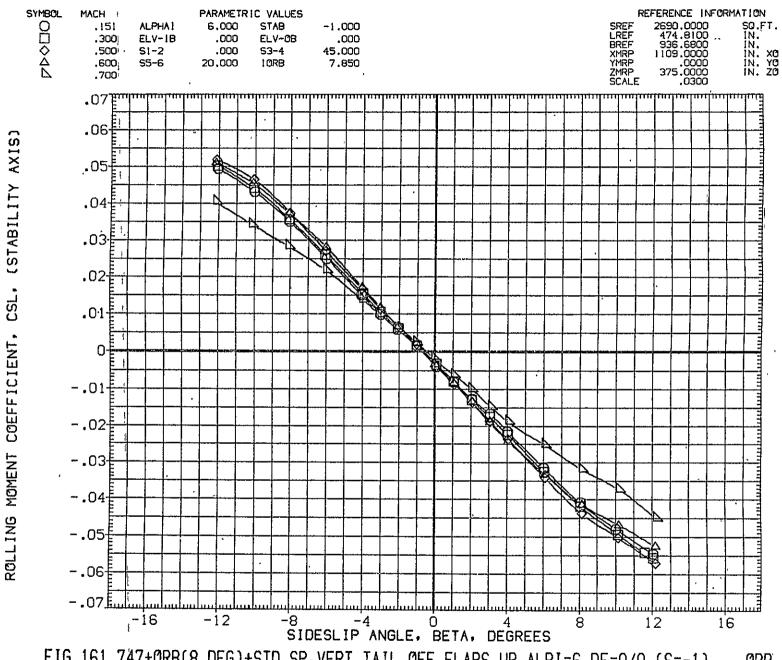


FIG.161 747+0RB(8 DEG)+STD SP, VERT TAIL OFF, FLAPS UP, ALPI=6, DE=0/0 (S=-1) ORB
PAGE 586

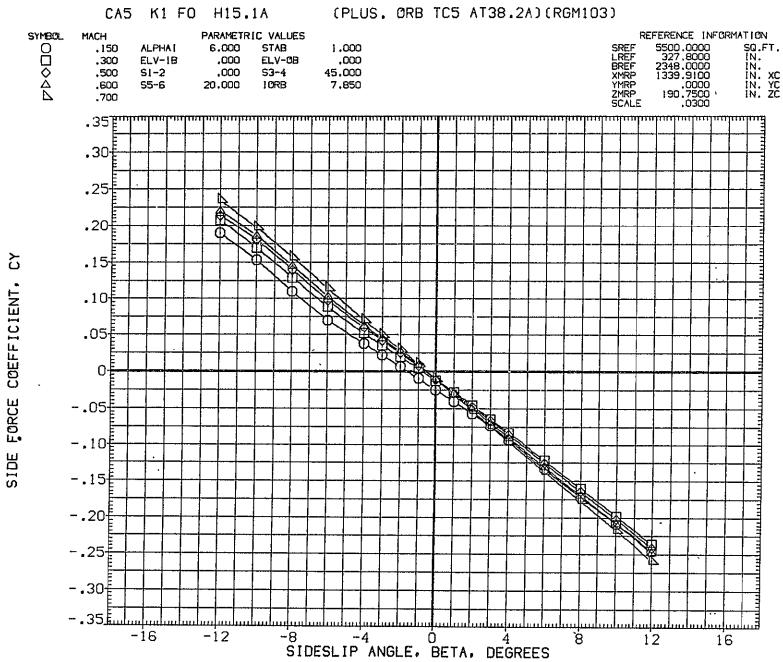


FIG.162 747+0RB(8 DEG)+STD SP, VERT TAIL OFF, FLAPS UP, ALPI=6, DE=0/0 (S=+1) TOTAL PAGE 587

PAGE

588

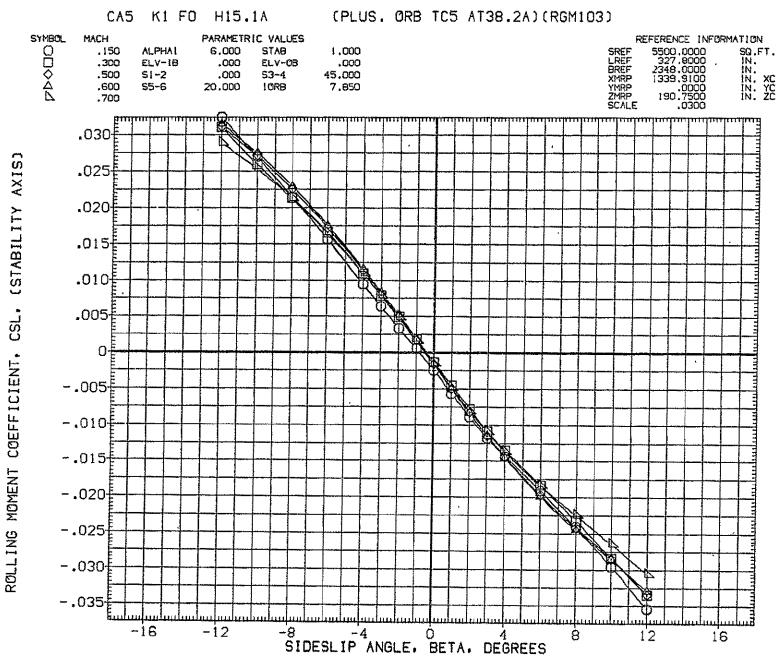


FIG.162 747+0RB(8 DEG)+STD SP, VERT TAIL OFF, FLAPS UP, ALPI=6, DE=0/0 (S=+1) TOTAL PAGE 589

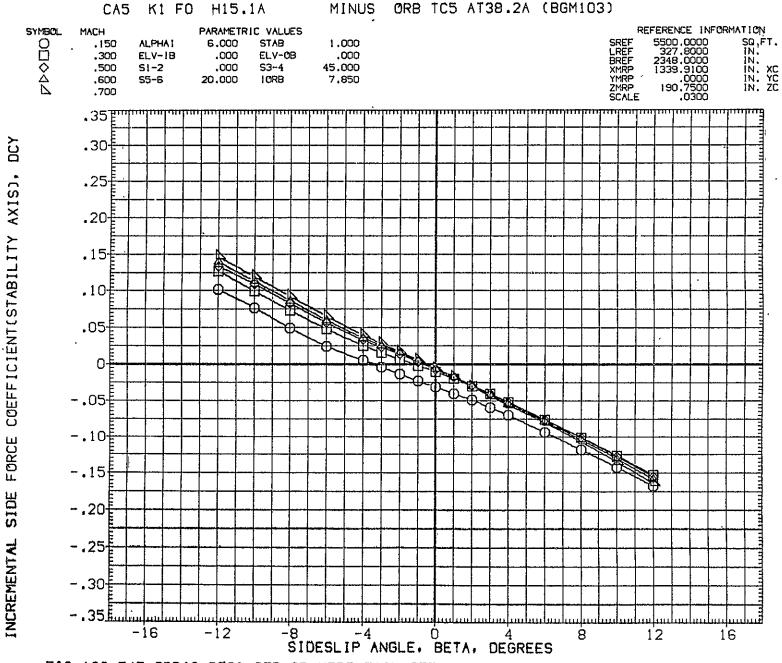


FIG.163 747+0RB(8 DEG)+STD SP, VERT TAIL OFF, FLAPS UP, ALPI=6, DE=0/0 (S=+1) 747

PAGE 590



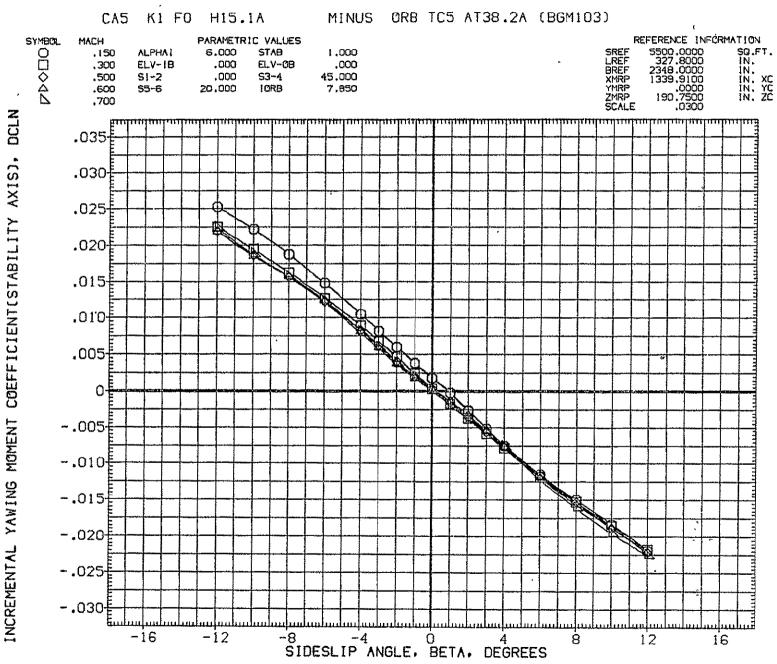


FIG.163 747+0RB(8 DEG)+STD SP.VERT TAIL OFF, FLAPS UP, ALPI=6, DE=0/0 (S=+1) 747

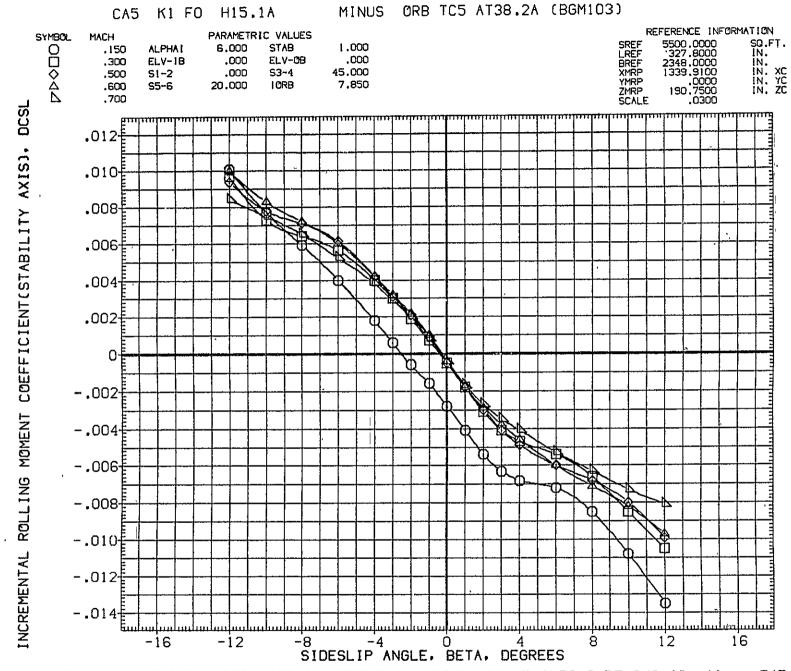


FIG.163 747+0RB(8 DEG)+STD SP, VERT TAIL OFF, FLAPS UP, ALPI=6, DE=0/0 (S=+1) 747

PAGE 592

ORB

593

PAGE

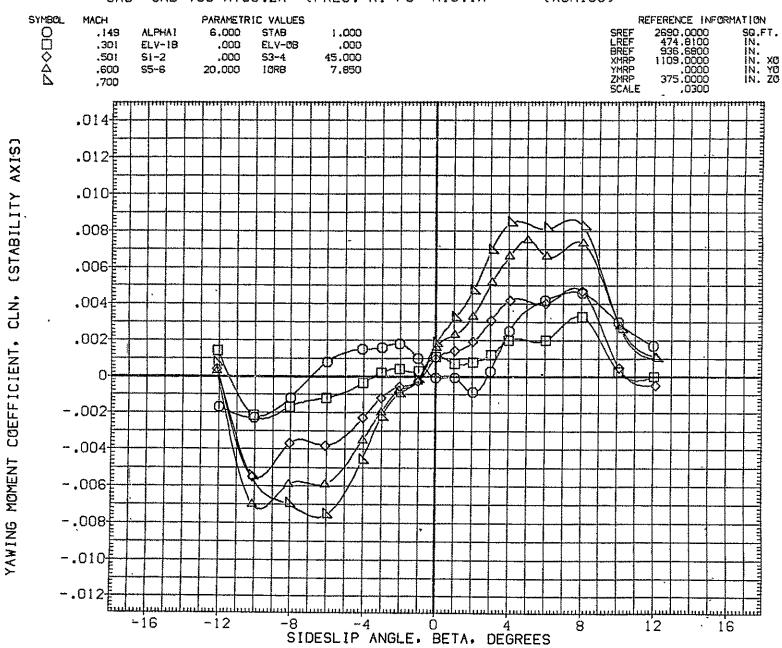
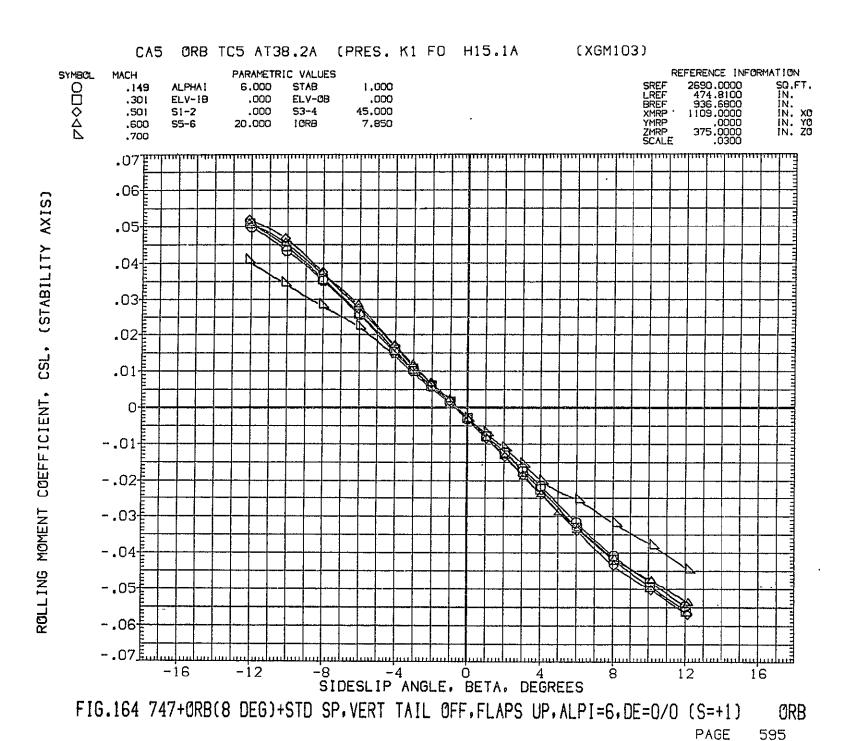
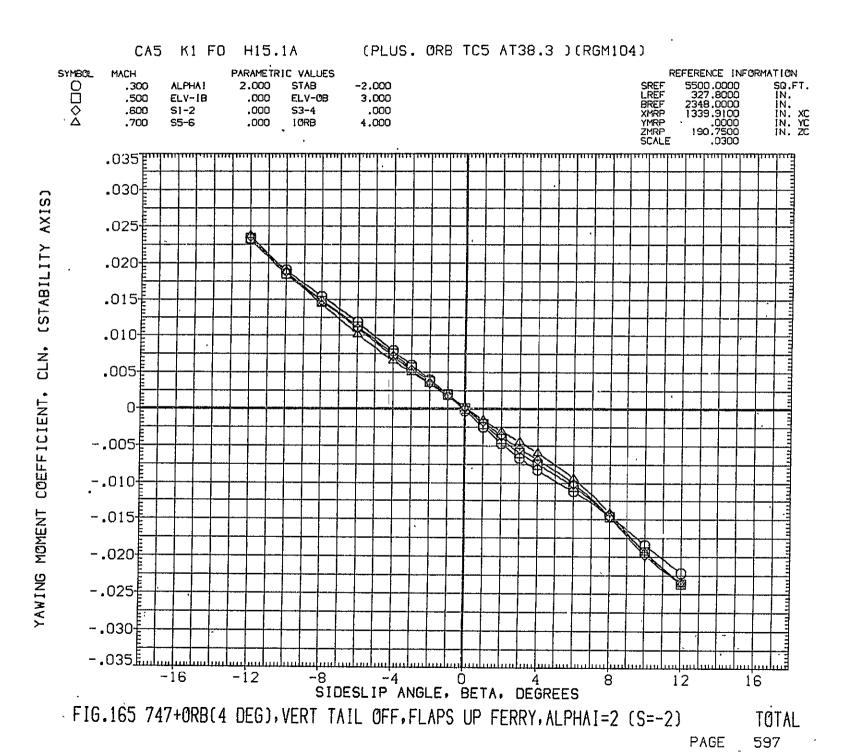
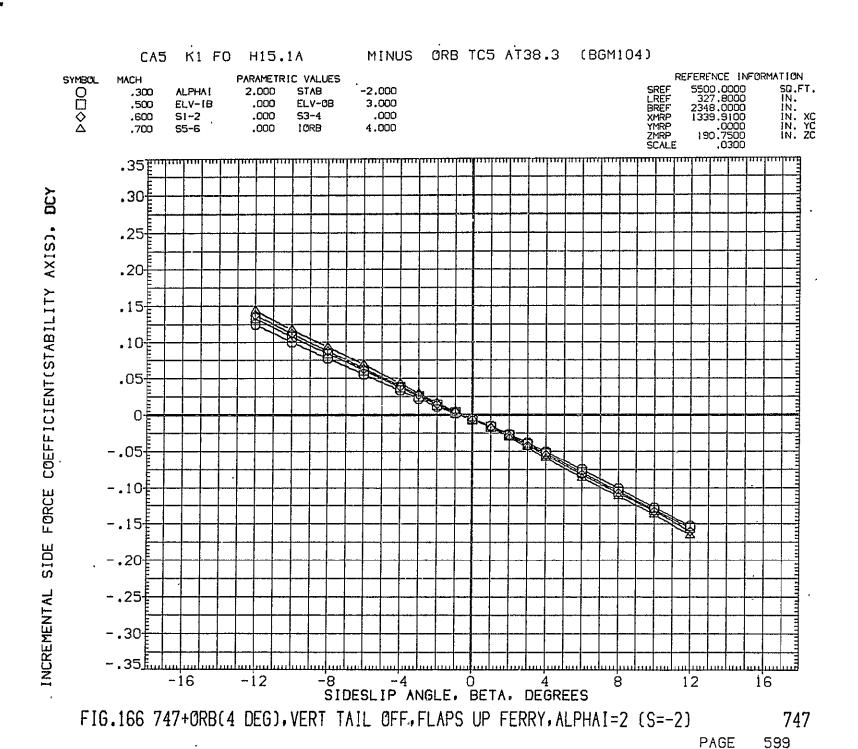
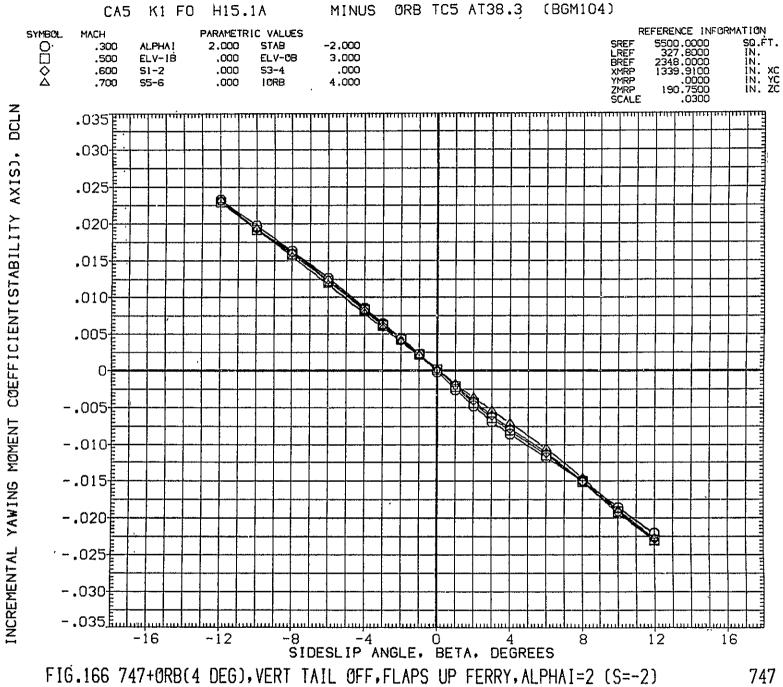


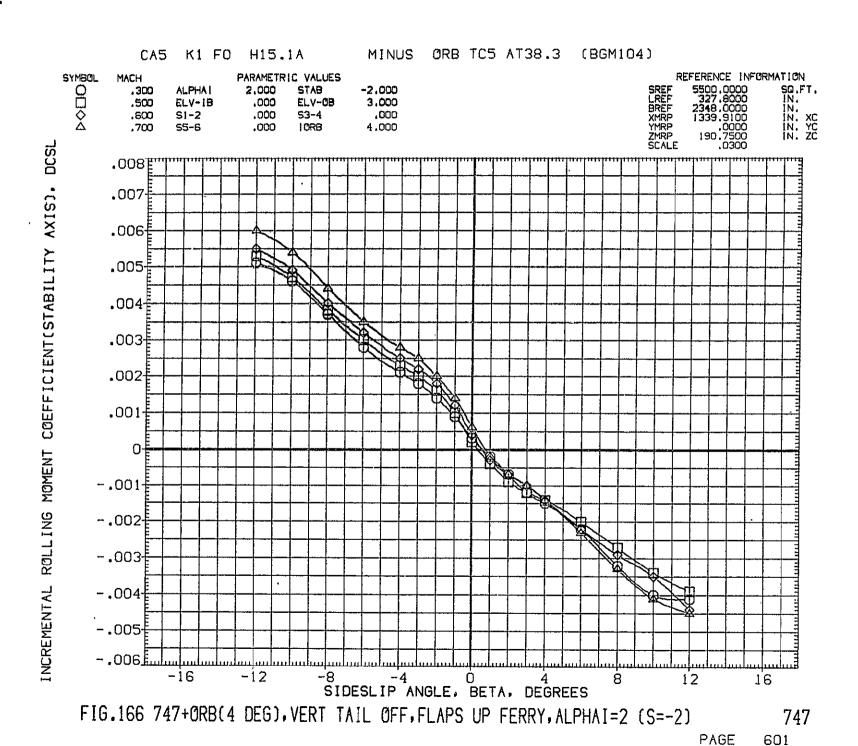
FIG.164 747+0RB(8 DEG)+STD SP.VERT TAIL OFF.FLAPS UP.ALPI=6.DE=0/0 (S=+1) ORB
PAGE 594











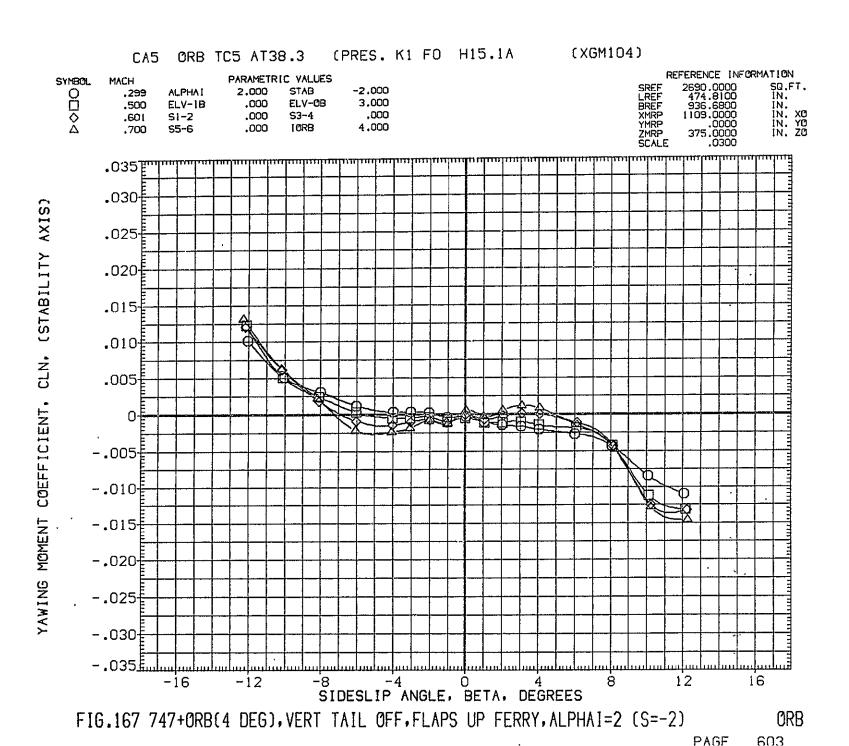
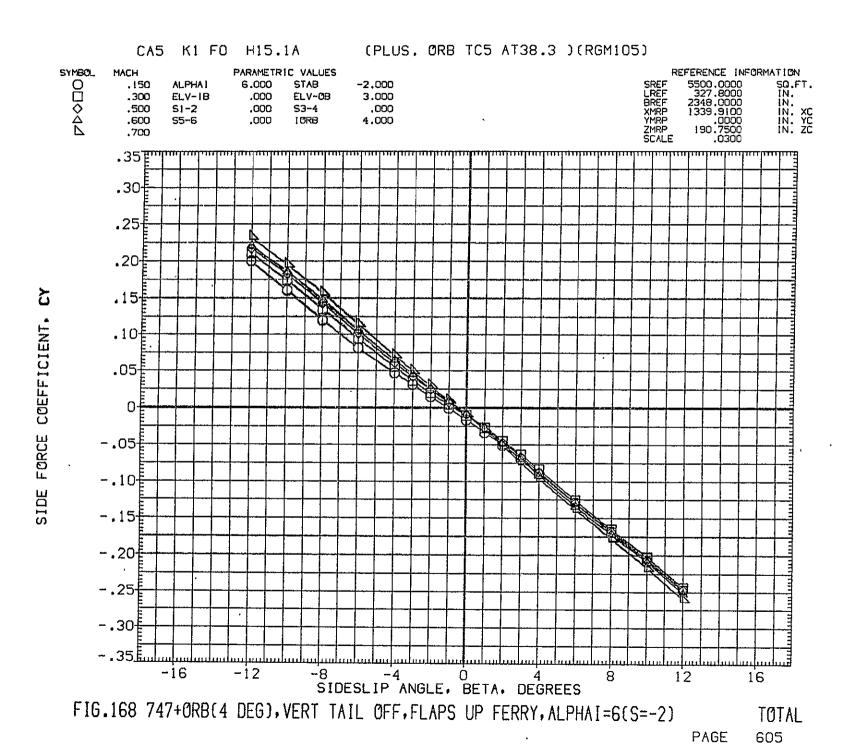
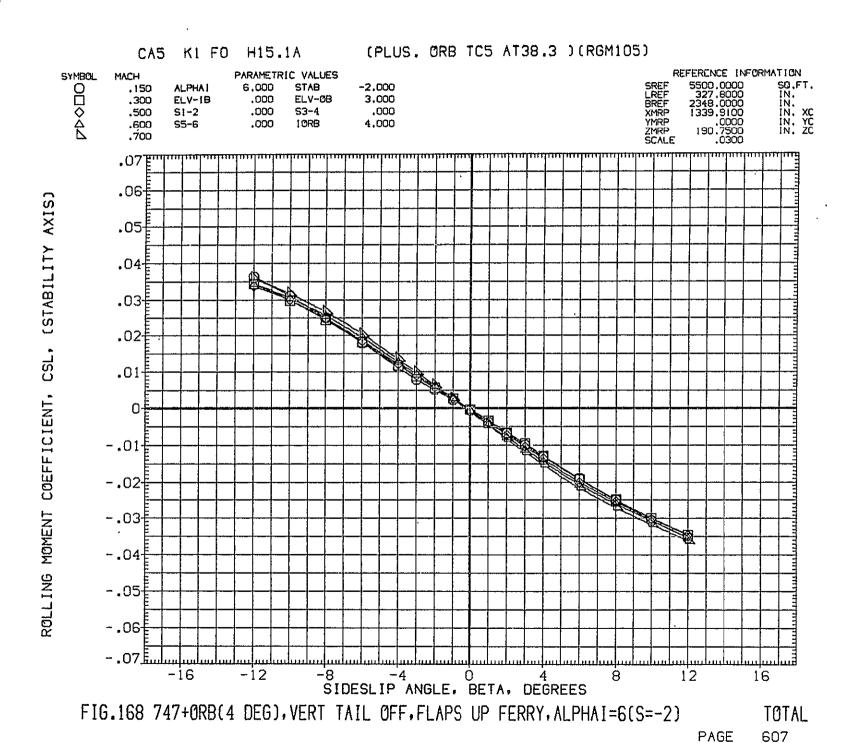


FIG.167 747+0RB(4 DEG), VERT TAIL OFF, FLAPS UP FERRY, ALPHAI=2 (S=-2)
PAGE 604





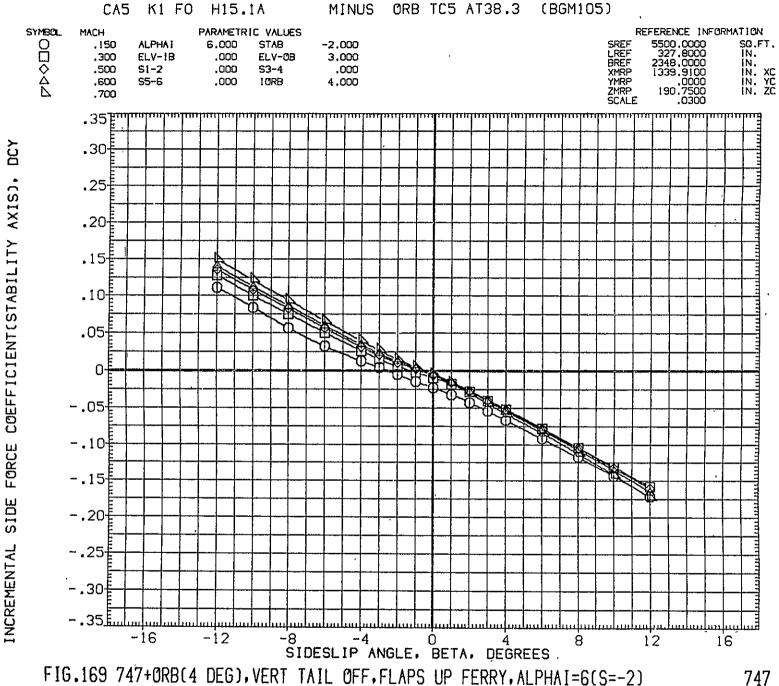
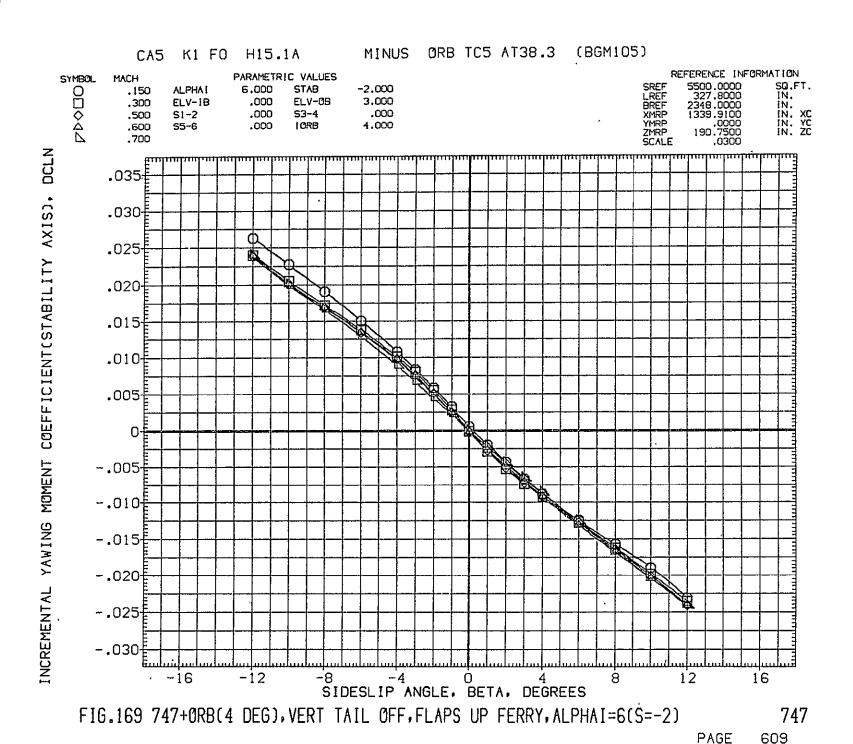


FIG.169 747+ORB(4 DEG), VERT TAIL OFF, FLAPS UP FERRY, ALPHAI=6(S=-2)



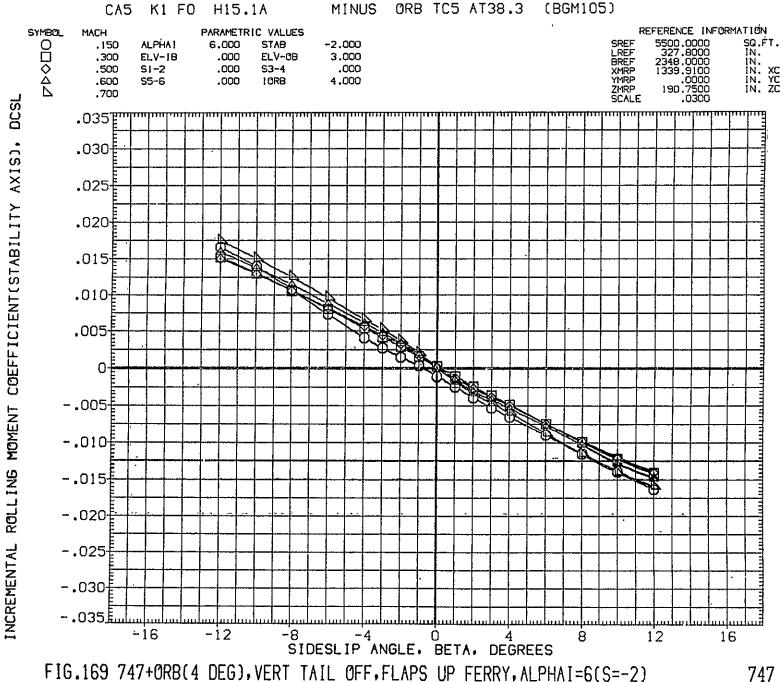
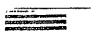
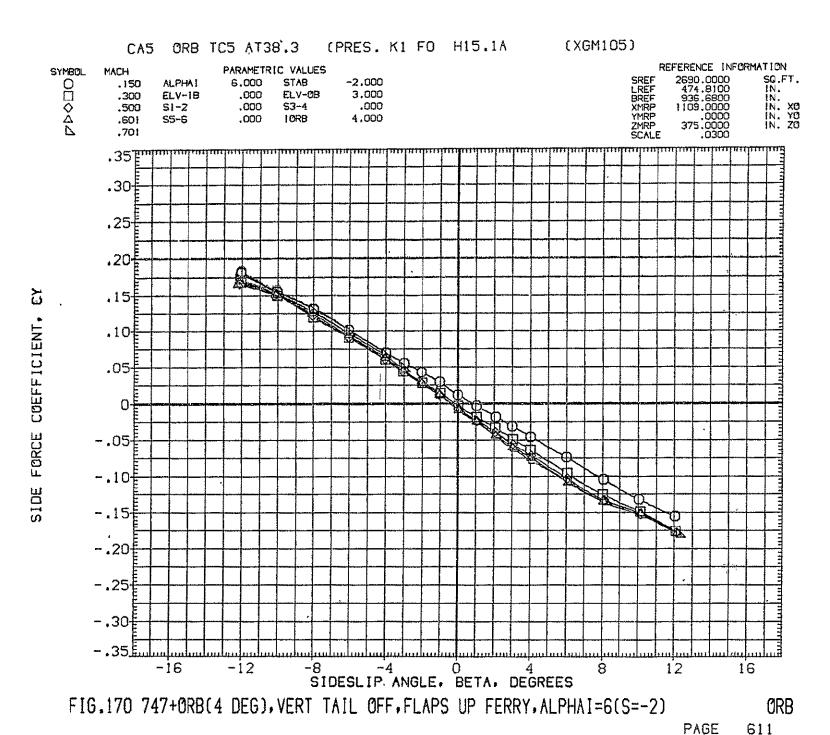


FIG.169 747+0RB(4 DEG), VERT TAIL OFF, FLAPS UP FERRY, ALPHAI=6(S=-2)





CA5 ORB TC5 AT38.3 (PRES. K1 FO H15.1A (XGM105) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH 2690.0000 474.8100 936.6800 1109.0000 .0000 375.0000 .0300 SQ.FT. IN. IN. IN. XO IN. YO IN. ZO 000044 6,000 STAB -2.000 .150 ALPHA1 .300 ELV-IB .000 ELV-0B 3,000 BREF XMRP YMRP ZMRP SCALE .500 S1-2 .000 S3-4 .000 IORB 4.000 .601 S5-6 .000 .701 .0147" .012[AXIS) .010 (STABIL 1TY .008 .006- .004 CLN. .002 COEFFICIENT. 0--.002 -.004 YAWING MOMENT -.006 -.008 -.010 -.012[-.014<u>.</u>E... -i2 12 16 SIDESLIP ANGLE, BETA, DEGREES

FIG.170 747+ORB(4 DEG), VERT TAIL OFF, FLAPS UP FERRY, ALPHAI=6(S=-2)

ORB

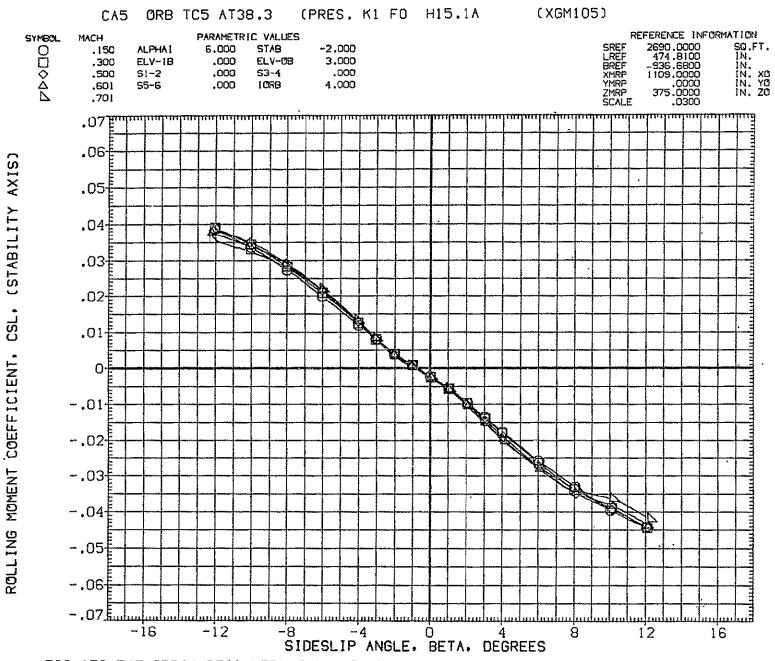


FIG.170 747+0RB(4 DEG), VERT TAIL OFF, FLAPS UP FERRY, ALPHAI=6(S=-2)

ORB PAGE 613

SIDESLIP ANGLE. BETA. DEGREES

FIG.171 747+ORB(4 DEG).BASIC TAIL, FLAPS UP FERRY, ALPHAI=8(S=-2)

-8

-16

-12

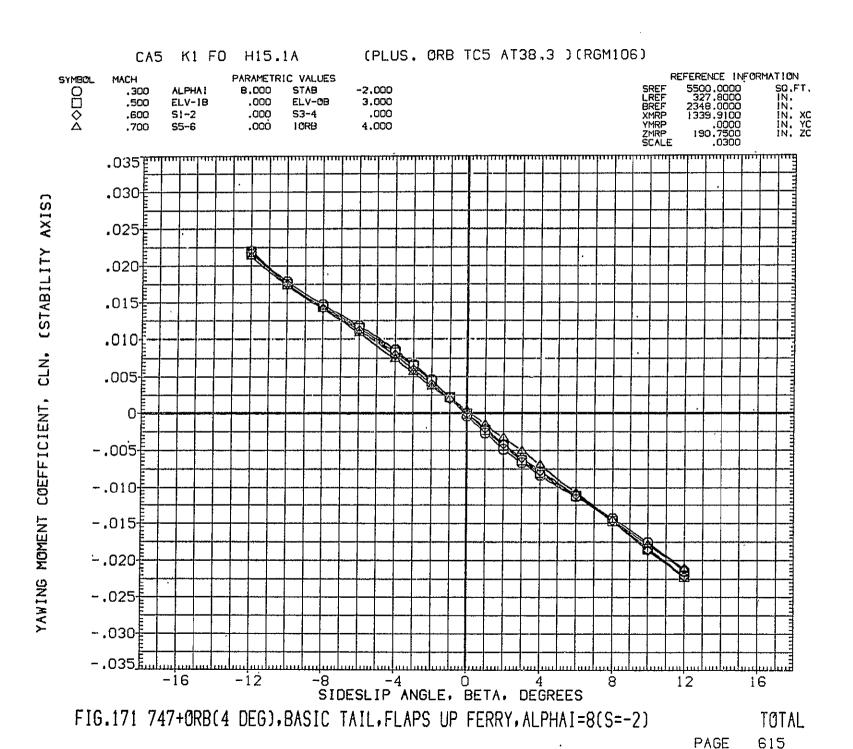
TOTAL

16

PAGE 614

12

8



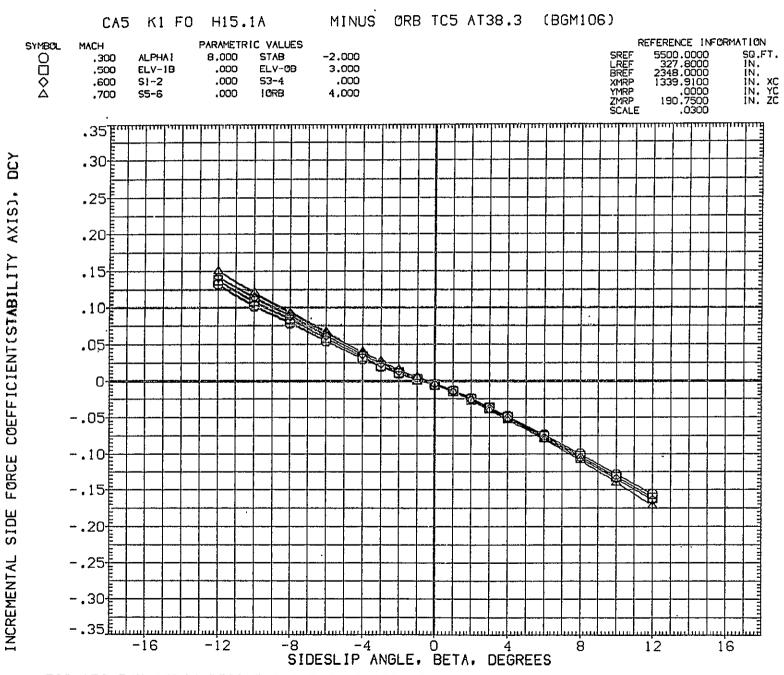
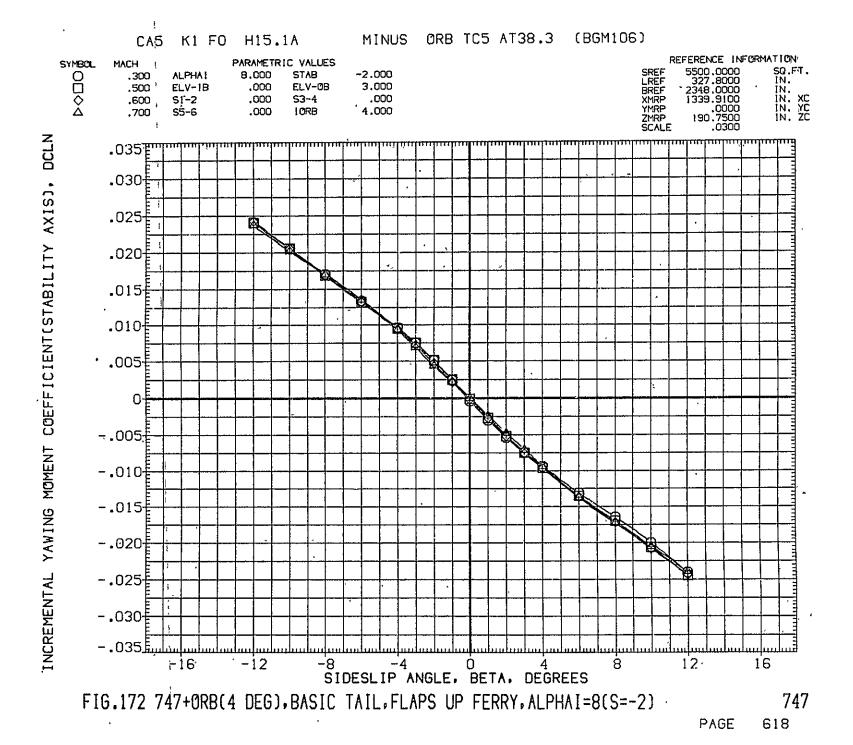


FIG.172 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=8(S=-2)

747



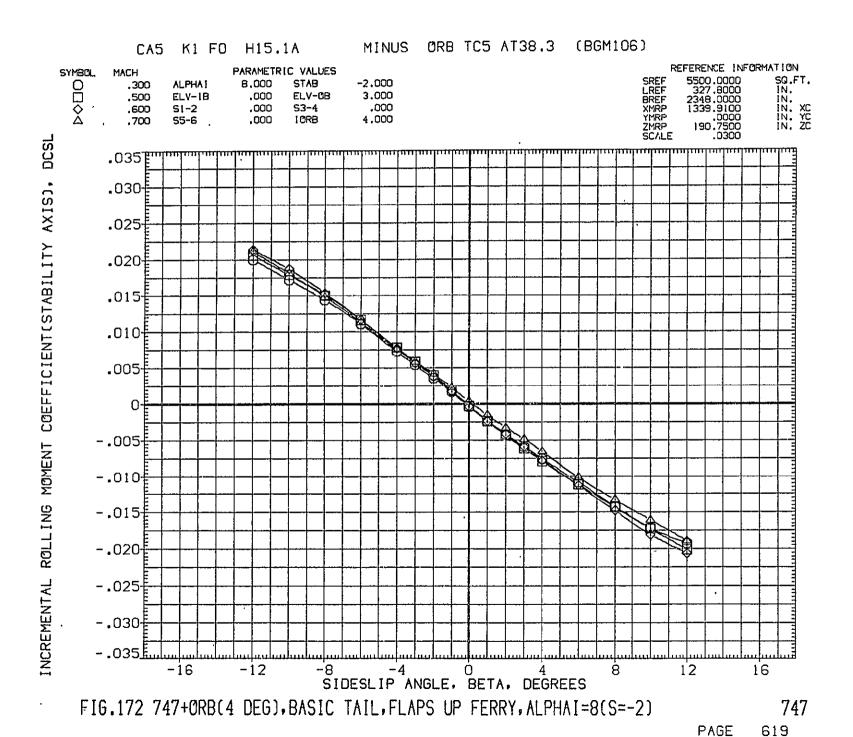
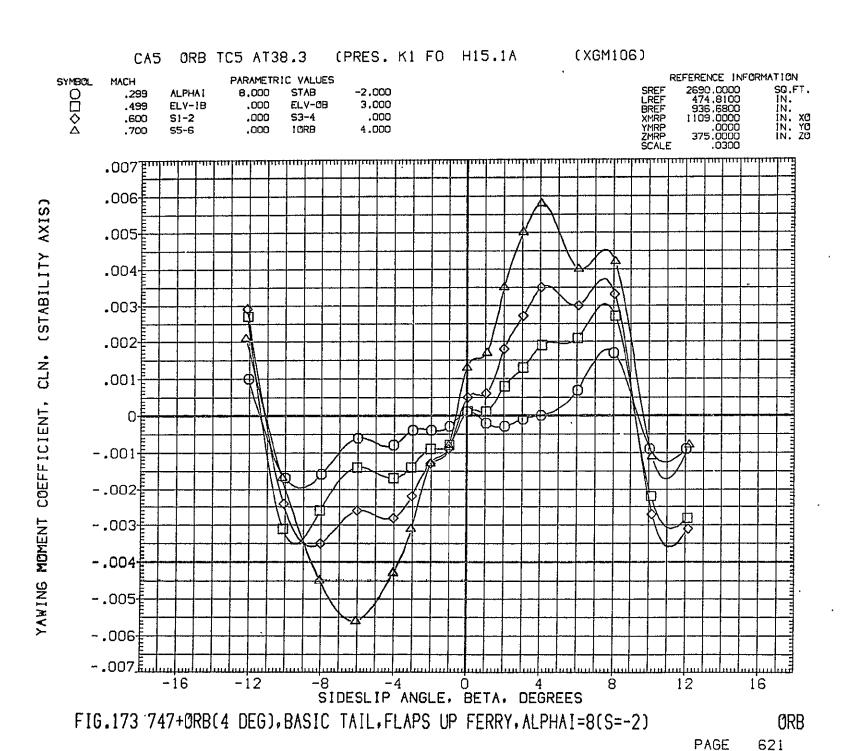


FIG.173 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=8(S=-2)



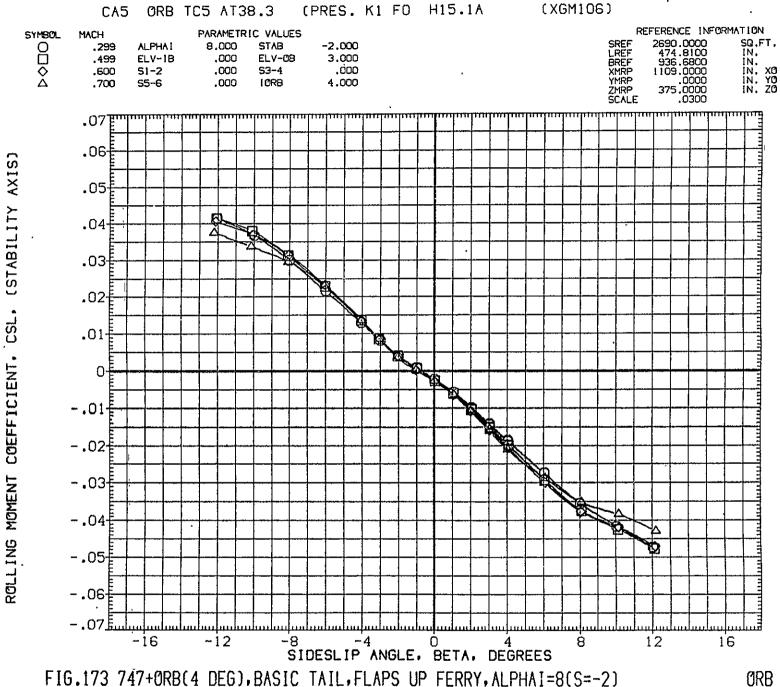


FIG.173 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=8(S=-2)

CA5 K1 FO H15.1A V9.1 (PLUS. ORB TC5 AT38.3)(RGM109) PARAMETRIC VALUES

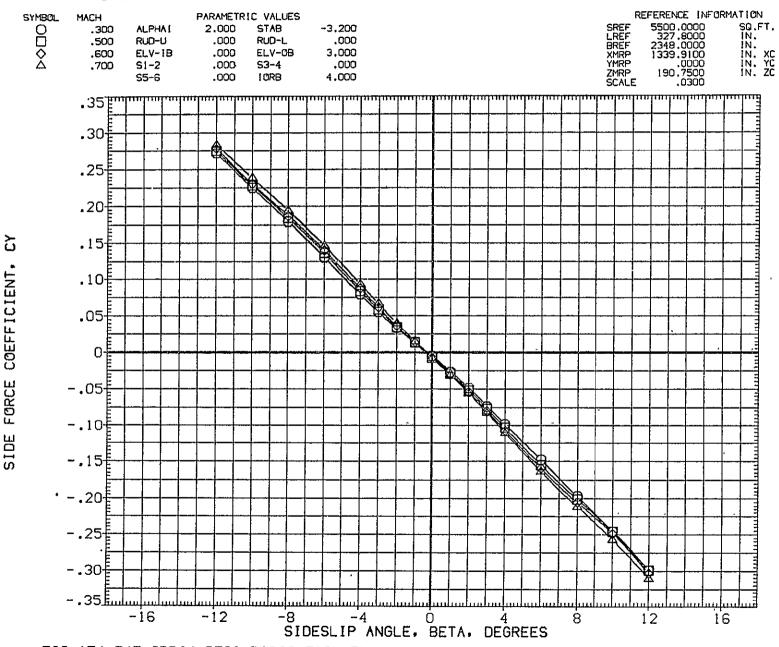


FIG.174 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=2(S=-3.2)

TOTAL

CA5 K1 FO H15.1A V9.1 (PLUS. ORB TC5 AT38.3)(RGM109)

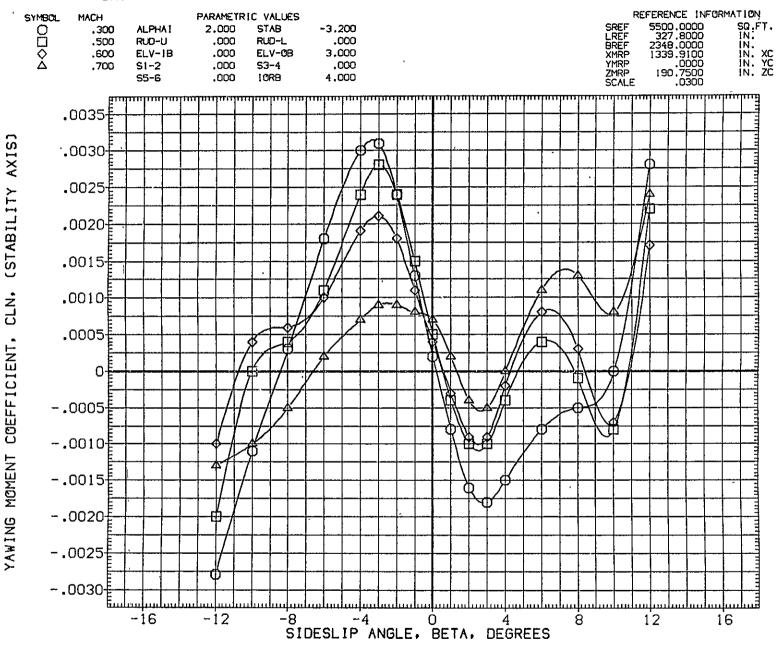


FIG.174 747+0RB(4 DEG).BASIC TAIL.FLAPS UP FERRY.ALPHAI=2(S=-3.2)

TOTAL



CA5 K1 FO H15.1A V9.1 (PLUS. ORB TC5 AT38.3)(RGM109) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH 5500.0000 327.8000 2348.0000 1339.9100 SO.FT. IN. IN. IN. XC IN. YC IN. ZC SREF LREF BREF XMRP YMRP ZMRP SCALE 0000 STAB .300 **ALPHAI** 2,000 -3.200 .000 .000 .500 RUD-U RUD-L .600 ELV-IB .000 ELV-0B 3,000 .000 .700 S1-2 .000 S3~4 4.000 S5-6 .000 IORB .035F .030[.025 .020 .015 .010[.005 0--.005 -.010[-.015 -.020 ·-.025 -.030[-.035<u>L</u>...l -12 -16 12 8 16

AXIS)

(STABILITY

CSL,

ROLLING MOMENT COEFFICIENT,

-16 -12 -8 -4 SIDESLIP ANGLE, BETA, DEGREES
FIG.174 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=2(S=-3.2)

TOTAL

CA5 K1 FO H15.1A V9.1 MINUS ORB TC5 AT38.3 (BGM109)

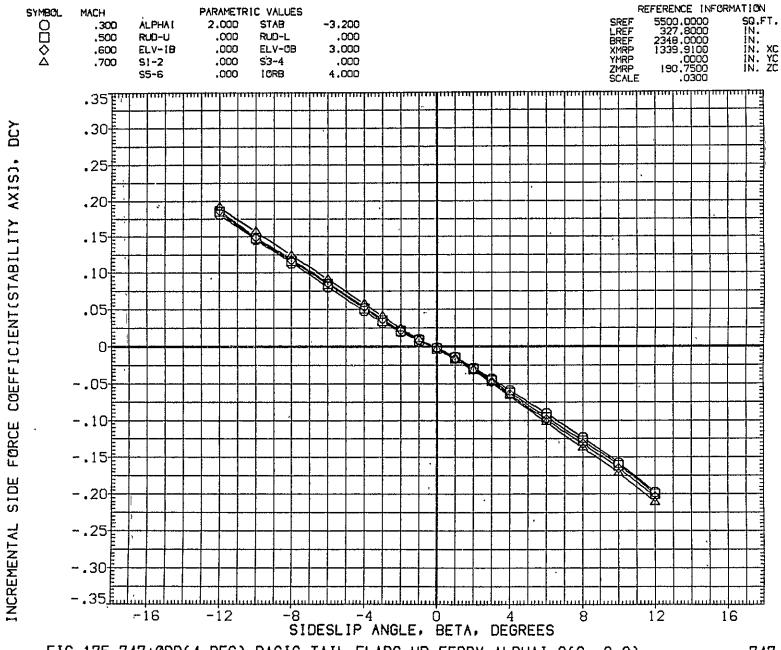


FIG.175 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=2(S=-3.2)

747

ORB TC5 AT38.3 (BGM109) MINUS K1 F0 H15.1A V9.1 REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH 5500.0000 327.8000 2348.0000 1339.9100 .0000 190.7500 .0300 SQ.FT. IN. IN. XC IN. YC IN. ZC SREF LREF BREF XMRP YMRP ZMRP SCALE .300 ALPHA1 STAB -3.200 .000 .000 RUD-L .500 RUD-U 3.000 000. ELV-1B .000 ELV-0B .600 .000 S3-4 .700 S1-2 10RB 4.000 .000 \$5**-**6 DCLN .0040<u>Fudunlunlundun</u> .0035£ AXIS), .0030# COEFFICIENT(STABILITY .0025 .0020[.0015[.0010[.0005[O-E YAWING MOMENT -.0005 -.0010E -.0015[INCREMENTAL -.0020[-.0025[~ · 0030 gm -8 -4 0 4
SIDESLIP ANGLE, BETA, DEGREES -16 -i2 12 16 8

FIG.175 747+ORB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=2(S=-3.2)

747

627

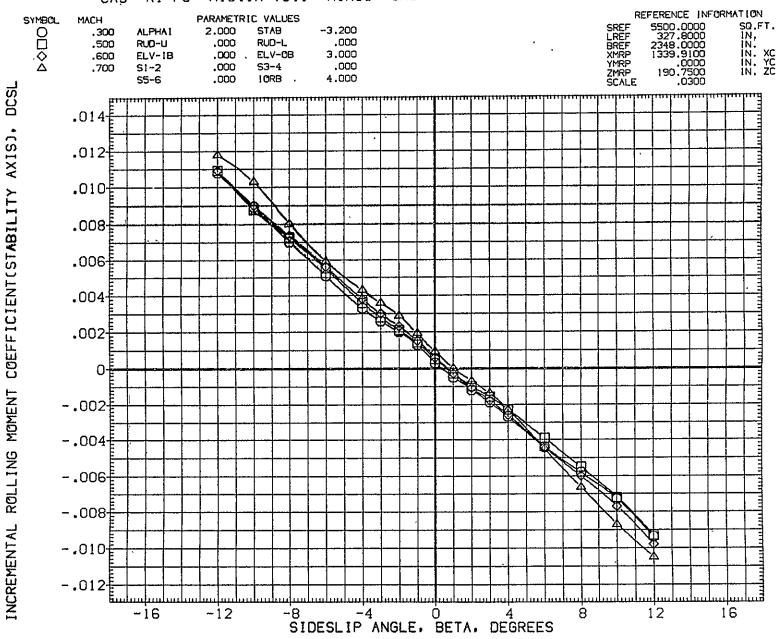


FIG.175 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=2(S=-3.2)

747

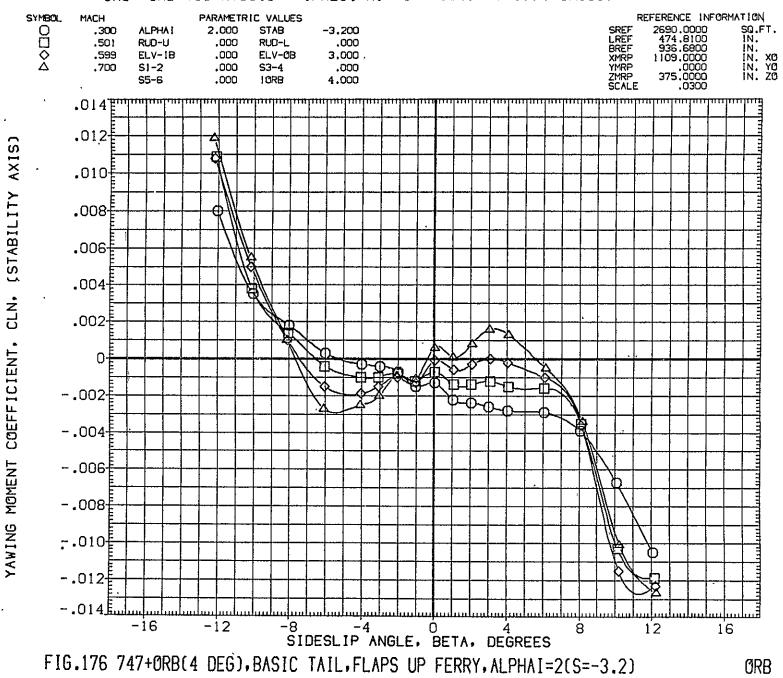


FIG.176 747+ORB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=2(S=-3.2)

ORB

629

CA5 ORB TC5 AT38.3 (PRES. K1 FO H15.1A V9.1)(XGM109)



(PRES. K1 FO H15.1A V9.1)(XGM109) ORB TC5 AT38.3 CA5 REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH 2690.0000 474.8100 936.6800 1109.0000 .0000 375.0000 .0300 SQ.FT: IN. IN. IN. XO IN. YO IN. ZO SREF LREF BREF XMRP YMRP ZMRP SCALE 0000 -3.200 2,000 STAB .300 ALPHA1 .000 RUD-U .000 RUD-L .501 3.000 ELV-08 ELV-18 .000 .599 .000 .000 \$3-4 S1-2 4,000 10RB .000 S5~6 .071 .06‡ (STABILITY AXIS) .05 .04 .03 .02 CSL, .01 COEFFICIENT, 0--.01 -.02 ROLLING MOMENT -.03 -.04 -.05 -.06 -.07<u>E</u>... ահակա 12 -8 -4 0 4
SIDESLIP ANGLE, BETA, DEGREES 16 -16 -12 -8 8 **ORB** FIG.176 747+ORB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=2(S=-3.2)

PAGE

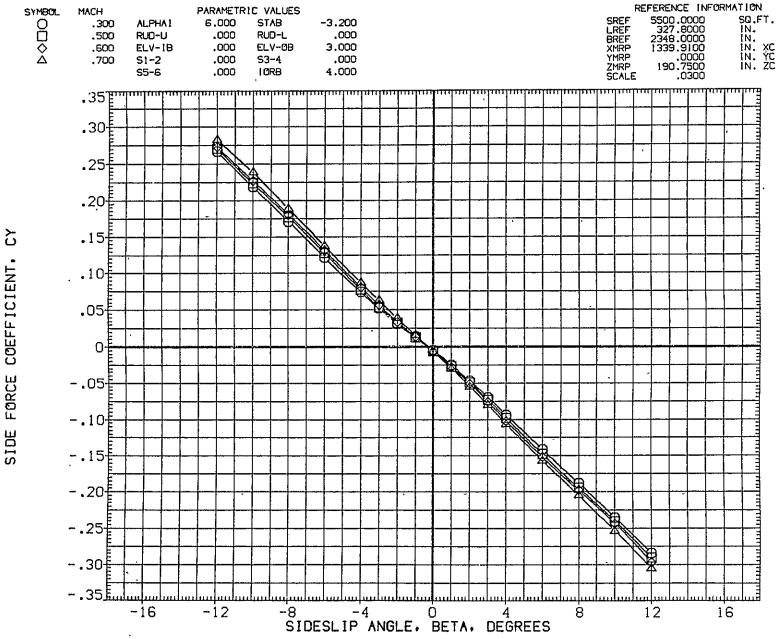
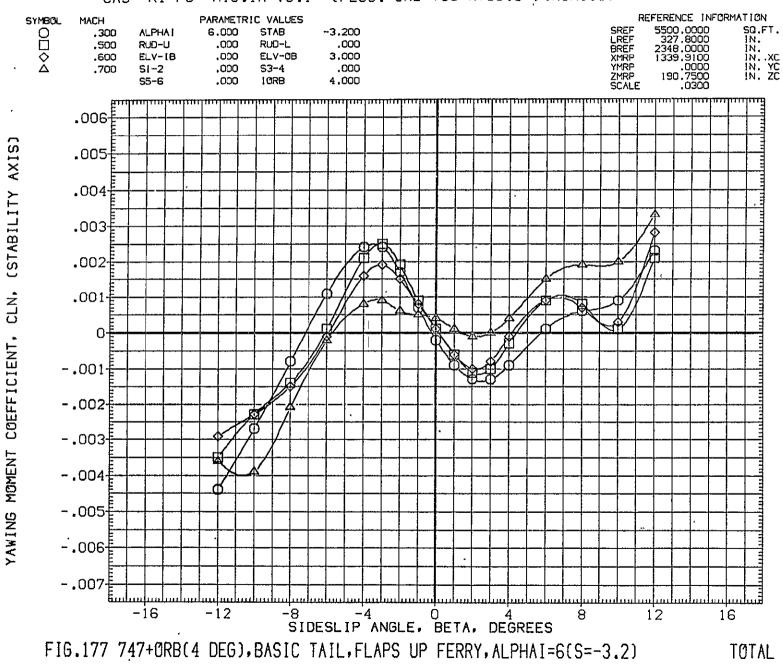


FIG.177 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=6(S=-3.2)

CA5 K1 FO H15.1A V9.1 (PLUS. ORB TC5 AT38.3)(RGM110)



633

CA5 K1 FO H15.1A V9.1 (PLUS. ORB TC5 AT38.3)(RGM110)

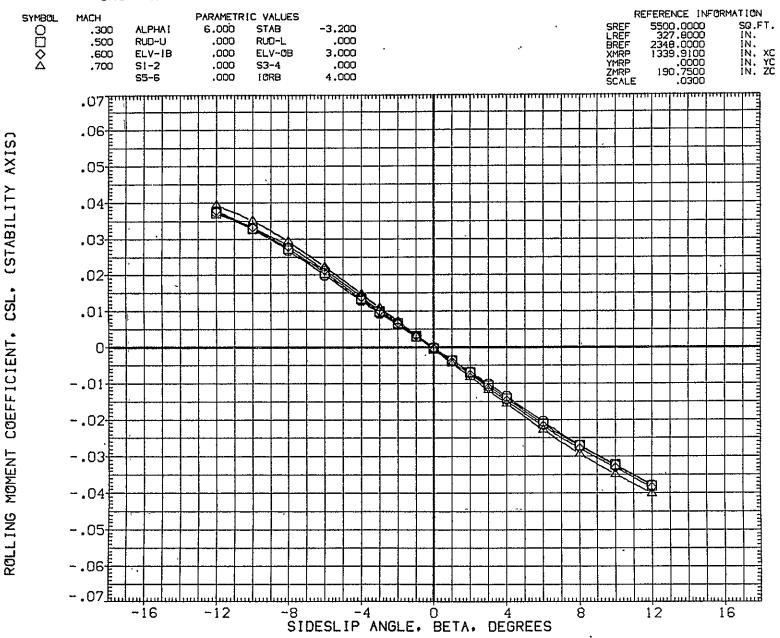
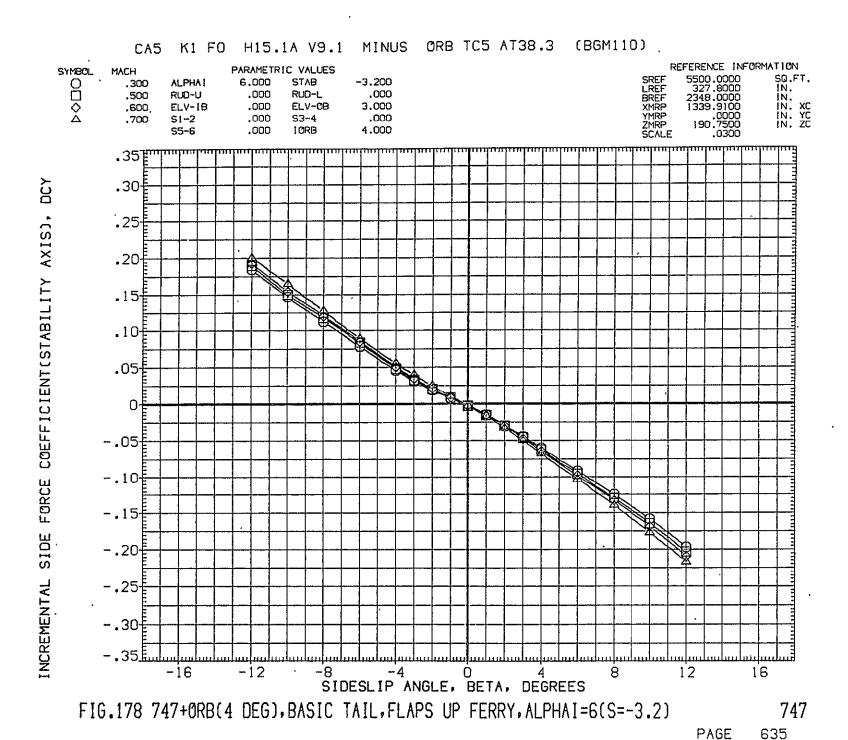
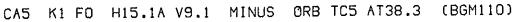


FIG.177 747+ORB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=6(S=-3.2)





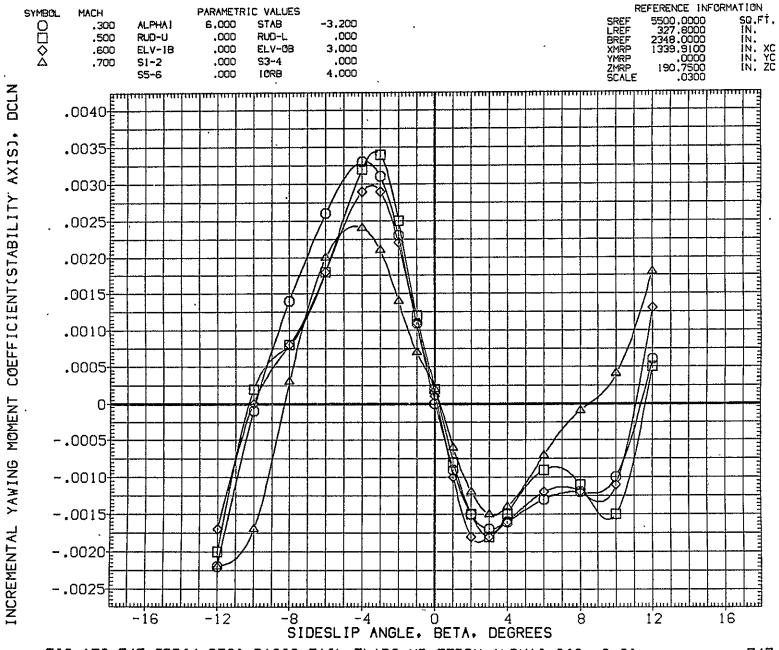


FIG.178 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=6(S=-3.2)

747

FIG.178 747+ORB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=6(S=-3.2)

CA5 ORB TC5 AT38.3 (PRES. K1 FO H15.1A V9.1)(XGM110)

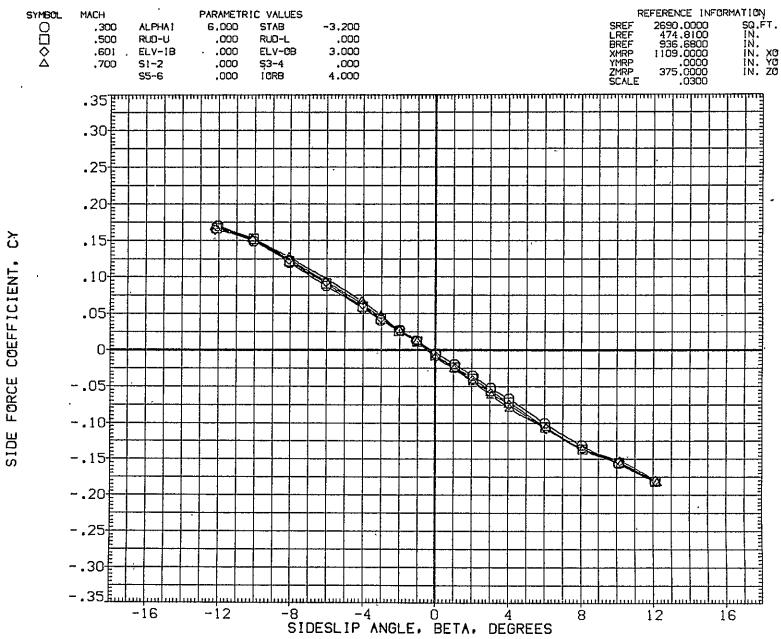
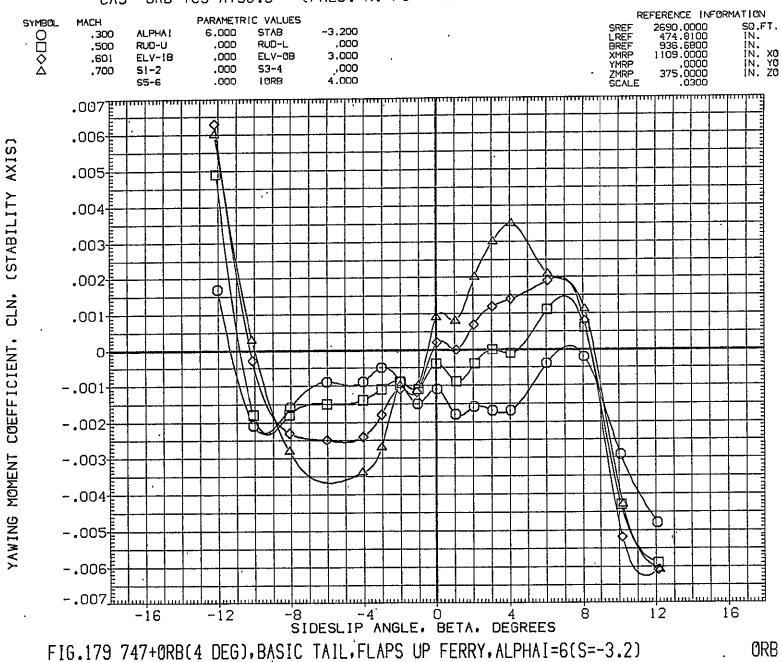


FIG.179 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=6(S=-3.2)

ORB



CA5 ORB TC5 AT38.3 (PRES. K1 FO H15.1A V9.1)(XGM110)



ORB TC5 AT38.3 (PRES. K1 FO H15.1A V9.1)(XGM110) SYMBOL MACH PARAMETRIC VALUES REFERENCE INFORMATION 0000 2690,0000 474,8100 936,6800 1109,0000 .0000 375,0000 .0300 SREF LREF BREF XMRP YMRP ZMRP SCALE .300 1 **ALPHAI** 6.000 STAB -3,200 SQ.FT. .500 RUD-U .000 RUD-L ,000 IN, .601 ELV-IB .000 ELV-08 3,000 N. XO IN. YO IN. ZO .700 | \$1-2 .000 53-4 .000 S5-6 .000 IORB 4.000 .06[(STABILITY AXIS) .05‡ .04‡ .03[.02 CSL. .01 ROLLING MOMENT COEFFICIENT, -.01 -.03 -.04E -.05£ -.06

SIDESLIP ANGLE, BETA, DEGREES
FIG.179 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=6(S=-3.2)

-4

-8

-16

-12

PAGE 640

16

ORB

12



CA5 K1 FO H15.1A V9.1 (PLUS. ORB TC5 AT38.3)(RGM111)

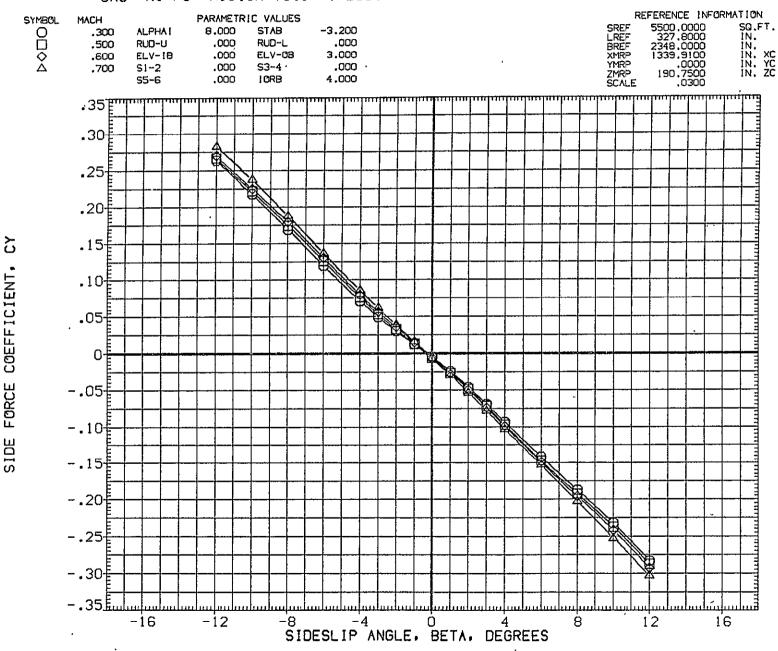


FIG.180 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=8(S=-3.2)

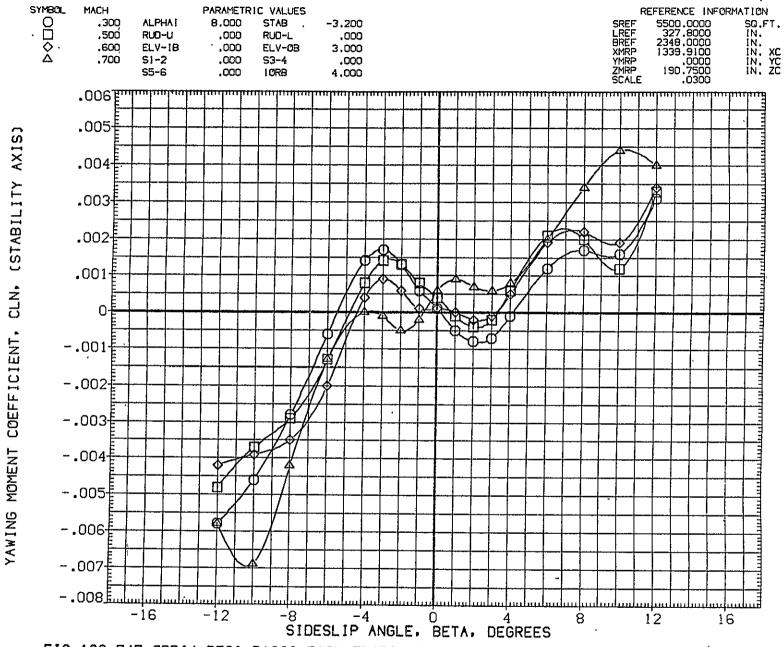


FIG.180 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=8(S=-3.2)

TOTAL PAGE 642

CA5 K1 FO H15.1A V9.1 (PLUS. ORB TC5 AT38.3)(RGM111)

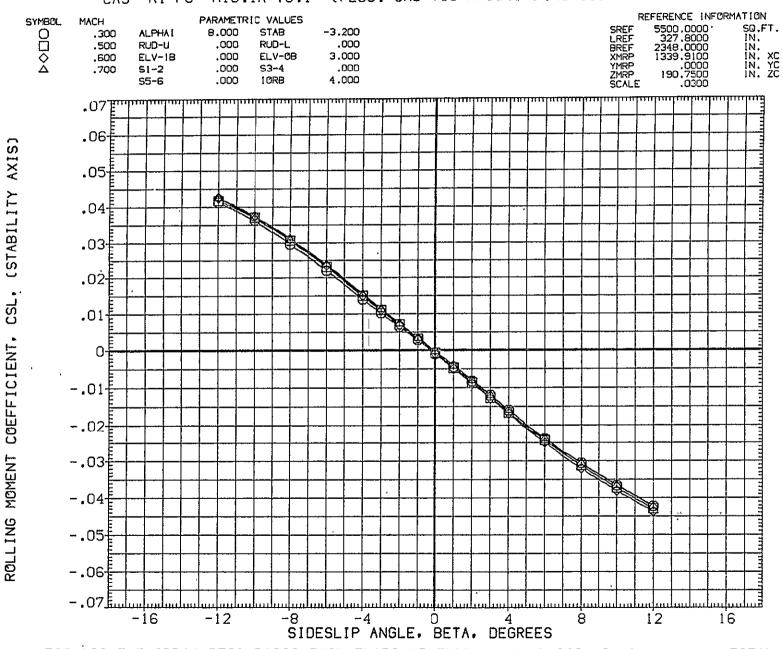


FIG.180 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=8(S=-3.2)

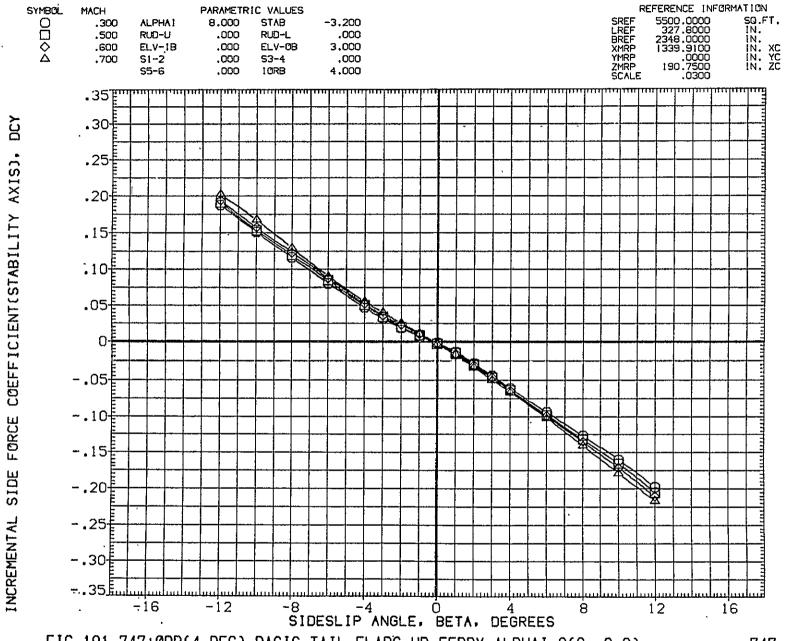


FIG.181 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=8(S=-3.2)

747



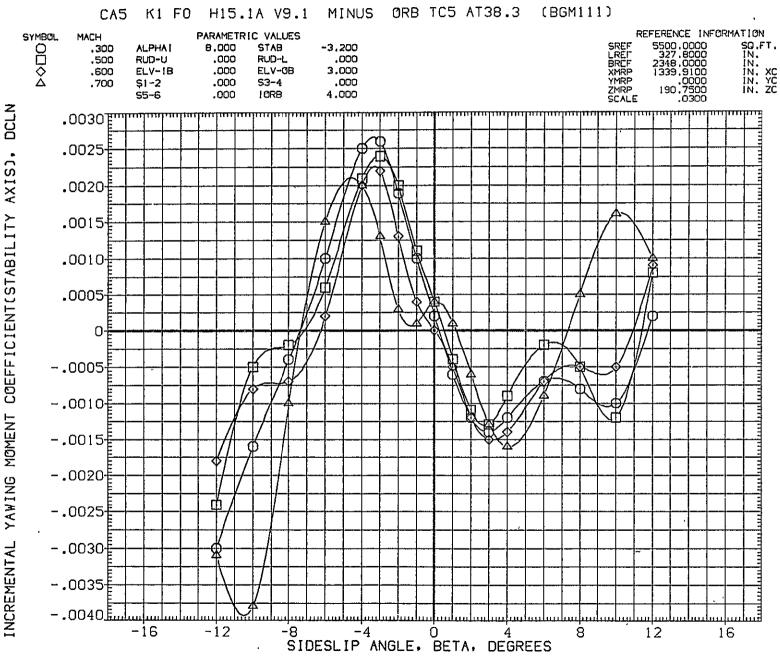


FIG.181 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=8(S=-3.2)

FIG.181 747+0RB (4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=8(S=-3.2)

PAGE' 646

(PRES. K1 FO H15.1A V9.1)(XGM111) ORB TC5 AT38.3 CA5 REFERENCE INFORMATION PARAMETRIC VALUES SREF LREF BREF XMRP YMRP ZMRP SCALE SQ.FT. IN. IN. IN. XO IN. YO IN. ZO 2690.0000 474.8100 936.6800 1109.0000 **ALPHAI** 8,000 STAB -3.200 .300 .500 RUD-U .000 RUD~L .000 .000 ELV-0B 3.000 .600 ELV-IB .000 .700 S1-2 .000 S3-4 .0000 10RB 4.000 .35 [[[]] .30[.25-.20[.15 .10 .05 0 -.05[-.10 -.15[-.20 -.25 -.30 -.35<u>£</u>...

FIG.182 747+0RB(4 DEG), BASIC TAIL, FLAPS UP FERRY, ALPHAI=8(S=-3.2)

-4

SIDESLIP ANGLE, BETA, DEGREES

-8

SYMBOL

 \mathcal{C}^{d}

COEFFICIENT,

SIDE FORCE

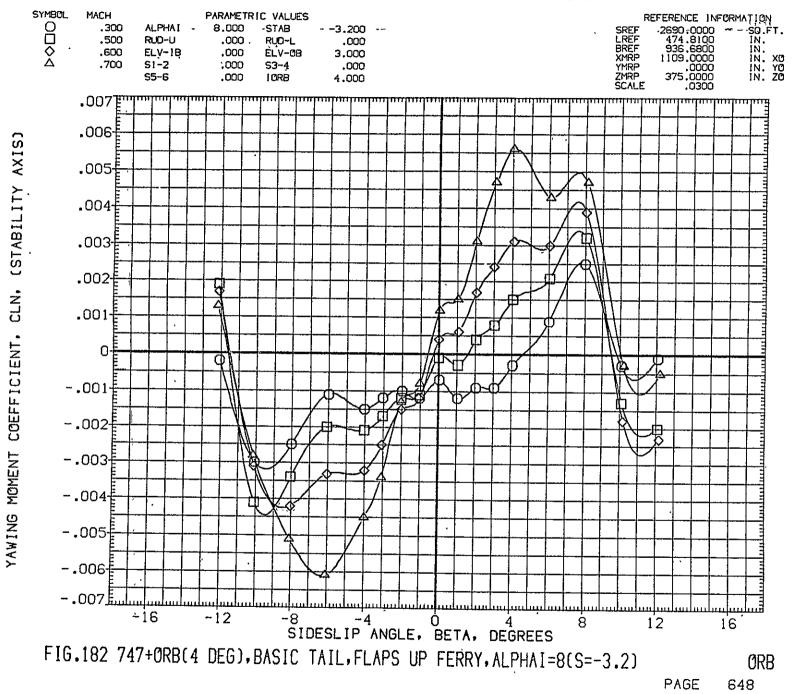
-i6

-i2

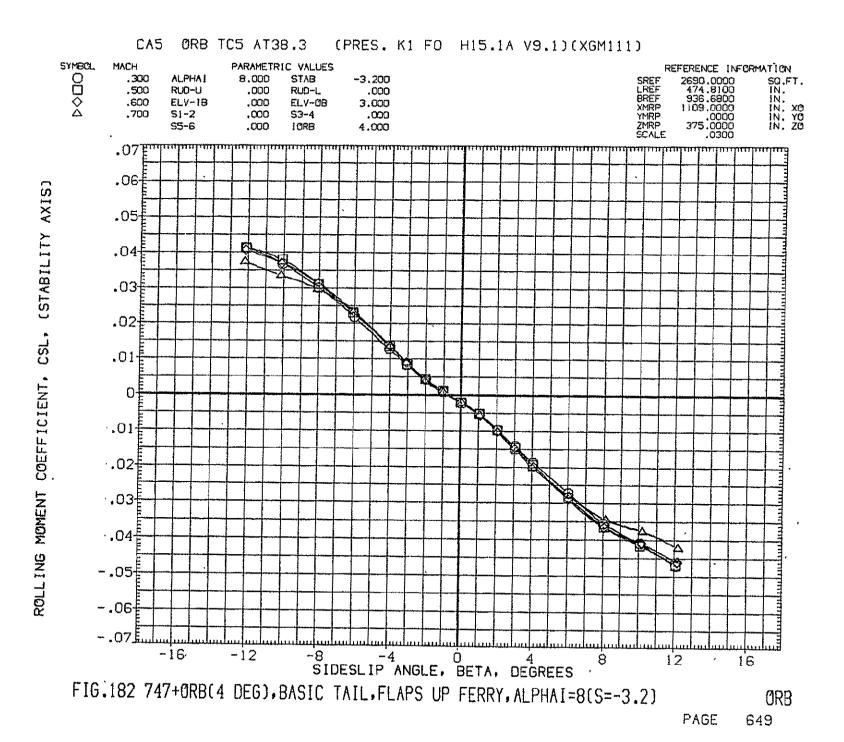
ORB

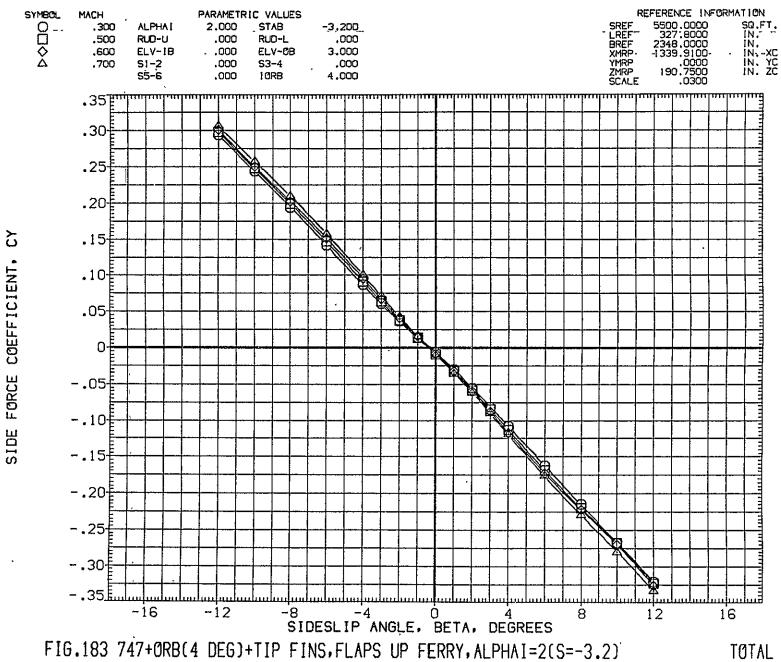
16

12











K1 F0 H15.6 V9.1 (PLUS. ORB TC5 AT38.3)(RGM113) CA5 PARAMETRIC VALUES REFERENCE INFORMATION MACH SO.FT. IN. IN. XC IN. YC IN. ZC SREF LREF BREF XMRP YMRP ZMRP SCALE 5500,0000 327,8000 2348,0000 1339,9100 ALPHA I 2.000 STAB -3,200 .000 .000 ,500 RUD-U RUD-L 3.000 .000 ELV-IB ELV-08 .0000 190.7500 .0300 \$1-2 .000 S3-4 .000 IORE 4,000 S5-6 .000 .035 .030 .025 .020 .015 .010 .005 -.005 -.010 -.020 -.025 -.030[

-8 -4 0 4
SIDESLIP ANGLE, BETA, DEGREES

FIG.183 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ALPHAI=2(S=-3.2)

SYMBOL

(STABILITY AXIS)

CLN,

YAWING MOMENT COEFFICIENT,

-16

-i2

TOTAL

16

12

CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.3)(RGM113)

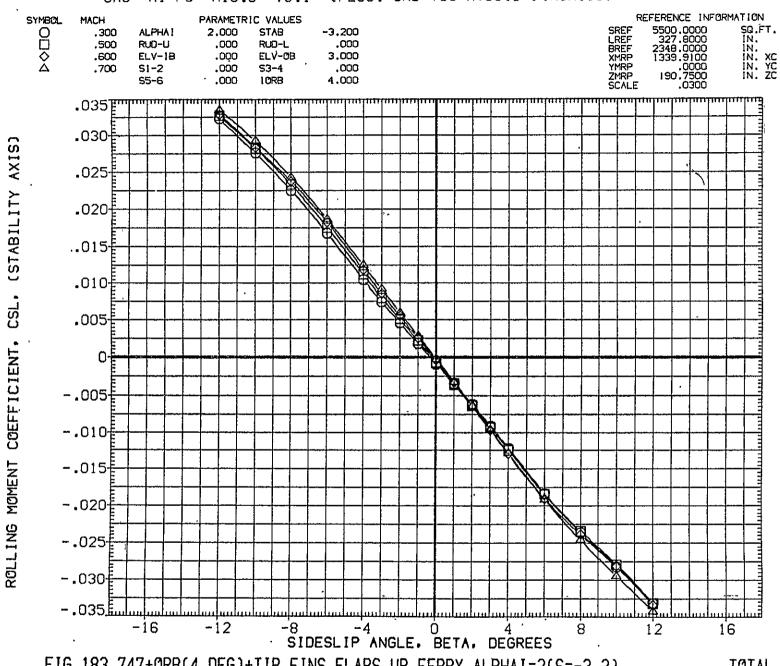
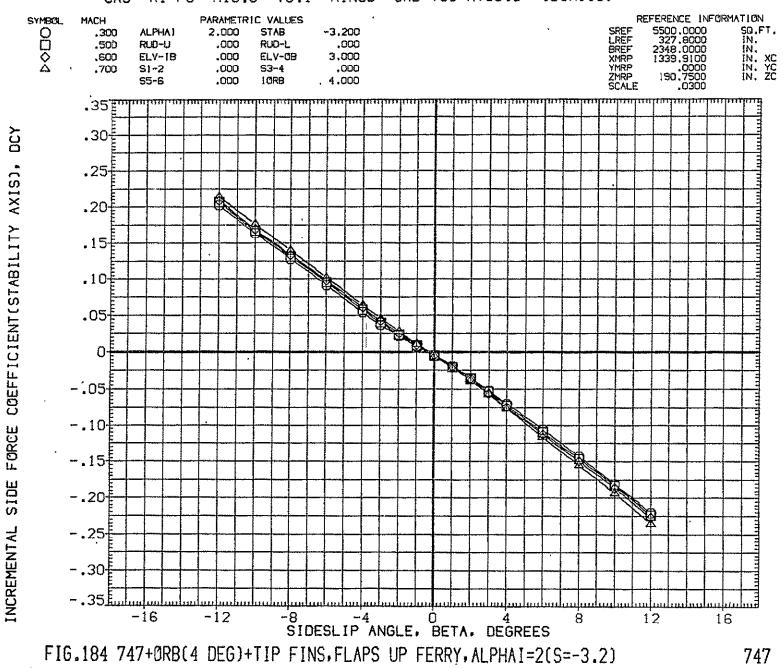


FIG.183 747+0RB(4 DEG)+TIP FINS.FLAPS UP FERRY.ALPHAI=2(S=-3.2)

TOTAL



CA5 K1 FO H15.6 V9.1 MINUS ORB TC5 AT38.3 (BGM113)



A5 K1 F0 H15.6 V9.1 MINUS ORB TC5 AT38.3 (BGM113) REFERENCE INFORMATION PARAMETRIC VALUES SREF LREF BREF XMRP YMRP ZMRP SCALE 5500.0000 327.8000 2348.0000 -3,200 3 ALPHAI 2.000 STAB IN. XC IN. YC IN. ZC .000 .000 RUD÷L RUO-U .000 1339.9100 .0000 190.7500 3.000 ELV-08 ELV-1B S1-2 .,000 S3-4 .000 4,000 .000 IORB S5-6 ∸16 -12 -8 12 16 -4

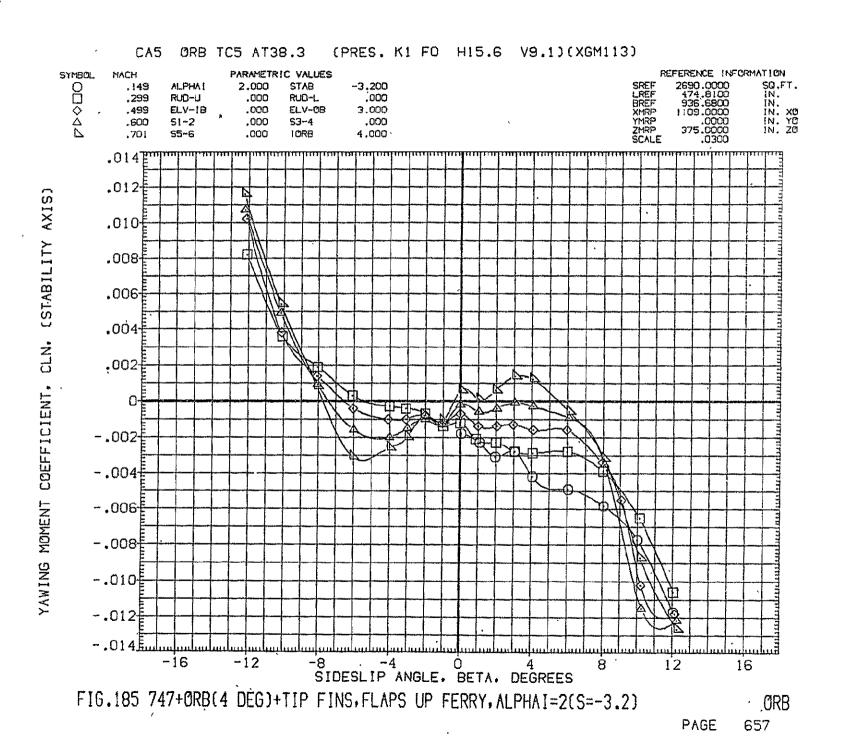
747+ORB(4 DEG)+TIP FINS, FLAPS UP FERRY, ALPHAI=2(S=-3.2)

SIDESLIP ANGLE, BETA, DEGREES

747

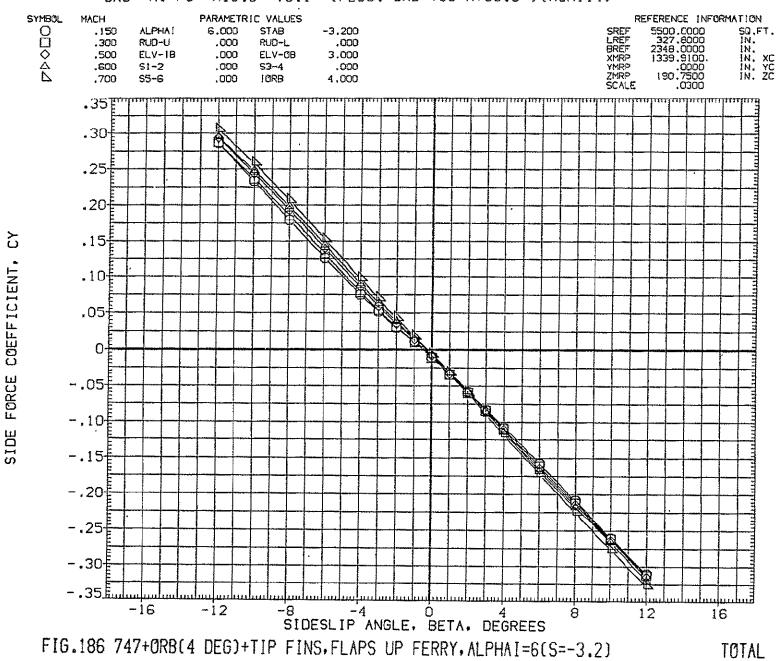
ORB TC5 AT38.3 CA5 K1 FO H15.6 V9.1 MINUS (BGM113) SYMBOL O O O O MACH PARAMETRIC VALUES REFERENCE INFORMATION SREF LREF BREF XMRP YMRP ZMRP ZMRP SCALE 5500,0000 327,8000 2348,0000 1339,9100 ,0000 190,7500 SO.FT. IN. IN. XC IN. YC IN. ZC .300 ALPHAI" 2,000 STAB -3.200 .500 RUD-U .000 RUD-L ,000 .600 ELV-18 .000 ELV-OB 3.000 .700 \$1-2 .000 53-4 .000 SS-6 .000 ICRB 4.000 DCSL .035[""] INCREMENTAL ROLLING MOMENT COEFFICIENT(STABILITY AXIS). .030+ .025-.020 .015# .010# .005 -.005 -.010 ·-.015 -.020 -.025 -.030 -.035駈 -8 -4 0 4 SIDESLIP ANGLE, BETA, DEGREES -16 -12 12 16 FIG.184 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ALPHAI=2(S=-3.2) 747





CA5 ORB TC5 AT38.3 (PRES. K1 FO H15.6 V9.1)(XGM113) PARAMETRIC VALUES REFERENCE INFORMATION SYMBOL MACH SREF LREF BREF XMRP YMRP ZMRP SCALE .149 , ALPHAI, 2,000 STAB -3.200 .299 RUD-U .000 RUD-L .000 .499 | ELV-IB ELV-0B 3,000 N. XO IN. YO IN. ZO .000 1600 - S1-2 .000 S3-4 .000 .70! | 55-6 .000 LORB 4.000 .06[(STABILITY AXIS) .05[.03[.02 CSL, .01[ROLLING MOMENT COEFFICIENT, -4 12 16 SIDESLIP ANGLE, BETA, DEGREES FIG.185 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ALPHAI=2(S=-3.2) 658 . PAGE

CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.3)(RGM114)



CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.3)(RGM114) SYMBOL PARAMETRIC VALUES REFERENCE INFORMATION 5500.0000 327.8000 2348.0000 1339.9100 000044 .150 ALPHA I 6.000 STAB SREF LREF BREF XMRP -3.200 .300 RUD~U .000 IN. RUD-L .000 .500 - - ELV-18 .000 ELV-0B 3,000 IN. XC IN. YC IN. ZC .600 i S1-2 .000 S3-4 .000 190.7500 YMRP ZMRP SCALE .700 S5-6 :000 IORB 4.000 ·035 Pri .030# AX1S) .025 (STABILITY .020 .015 .010[CLN, .005 YAWING MOMENT COEFFICIENT, -.005 -.010[-.015E -.020 -.025 ~.030

-8 -4 0 4 SIDESLIP ANGLE, BETA, DEGREES FIG.186 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ALPHAI=6(S=-3.2)

-8

-i2

TOTAL

16

PAGE 660

12

8



-.035<u>L</u>

CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.3) (RGM114) SYMBOL O O O O PARAMETRIC VALUES MACH REFERENCE INFORMATION 5500,0000 327,8000 2348,0000 1339,9100 ,0000 190,7500 ,0300 SREF LREF BREF XMRP YMRP ZMRP SCALE .150 ALPHA I 6.000 SQ.FT. STAB -3.200 RUD-U .000 .000 .300 RUD-L IN. XC IN. YC IN. ZC 3.000 .500 ELV-1B .000 ELV-0B .600 \$1-2 .000 S3-4 .000 \$5**-6** .000 IORB 4,000 .072" .06 (STABILITY AXIS) .05 .04 .03 .02 CSL. .01 COEFFICIENT. -.01[-.02 ROLLING MOMENT -.03# -.04 -.05 -.06 -.07.15... -8 -4 0 4 SIDESLIP ANGLE, BETA, DEGREES -i6 -i2 12 16 FIG.186 747+0RB(4 DEG)+TIP FINS.FLAPS UP FERRY.ALPHAI=6(S=-3.2) TOTAL

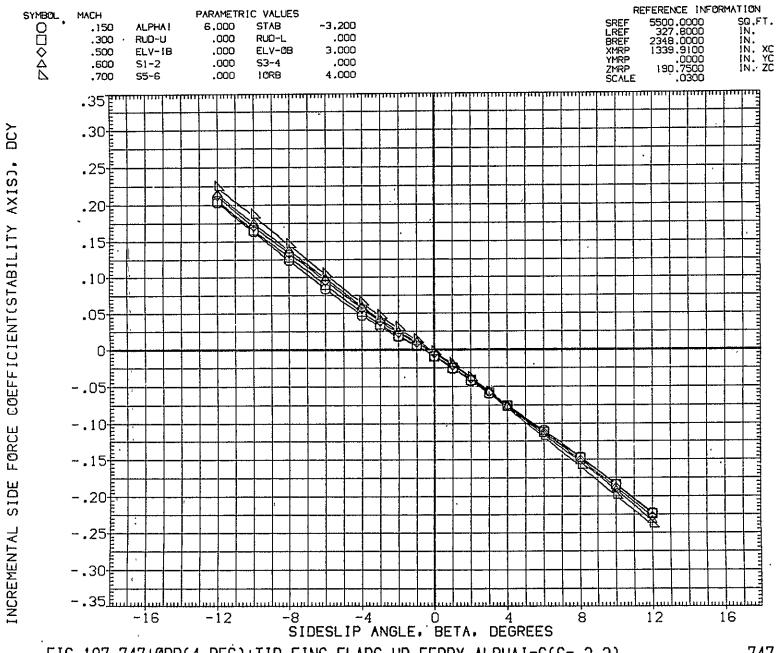


FIG.187 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ALPHAI=6(S=-3.2)

747

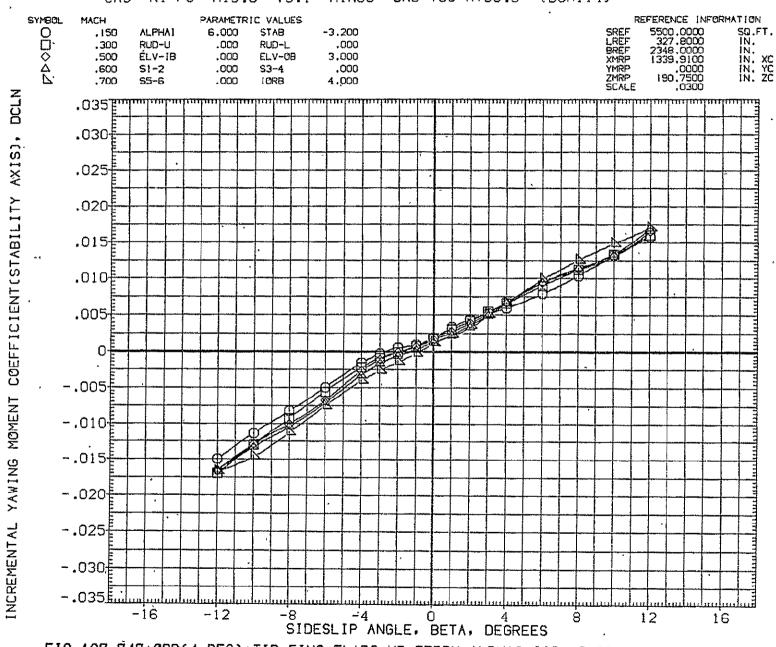


FIG.187 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ALPHAI=6(S=-3.2)

CA5 K1 FO H15.6 V9.1 MINUS ORB TC5 AT38.3 (BGM114) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL 5500.0000 327.8000 2348.0000 1339.9100 .0000 190.7500 .0300 SO.FT. IN. IN. XC IN. YC IN. ZC 000 6.000 .000 STAB -3.200 .150 r ALPHA1 RUD-L .000 .300 RUD-U BREF 3.000 .500 ELV-18 .000 ELV-0B XMRP YMRP ZMRP SCALE S1-2 .600 ' S3-4 · .000, .000 .700 🤈 55-6 IORB. 4.000 .000 DCSL .035世前 AXIS), .030€ .025 INCREMENTAL ROLLING MOMENT COEFF.ICIENT(STABILITY .020 .015 .010 .005 -.005 -.010[-.015 -.020 -.025 -.030

-8 -4 0 4
SIDESLIP ANGLE, BETA, DEGREES FIG.187 747+0RB(4 DEG)+TIP FINS.FLAPS UP FERRY.ALPHAI=6(S=-3.2)

-12

-.035<u>E</u>...l

PAGE 664_

16

747

· 12

CA5 ORB TC5 AT38.3 (PRES. K1 FO H15.6 V9.1)(XGM114) SYMBOL MACH PARAMETRIC VALUES REFERENCE INFORMATION 2690.0000 474.8100 936.6800 1109.0000 .0000 375.0000 .0300 SREF LREF BREF XMRP YMRP ZMRP SCALE SO.FT. IN. IN. IN. YO IN. YO ALPHA! 6.000 000044 .149 STAB -3.200 .299 RUD-U .000 RUD-L .000 .500 ELV-18 ELV-08 .000 3,000 .600 'S1-2 .000 53~4 ,000 .699 S5-6 .000 I ORB 4.000 .35Em .30E .25[.20-ដ .15- COEFFICIENT, .10-.05 0= SIDE FORCE -.05 -.10-₽ -.15# -.20 -.25 -.30 - .35島 -i2 -8 -4 12 8 16 SIDESLIP ANGLE, BETA, DEGREES FIG.188 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ALPHAI=6(S=-3.2) ÖRB

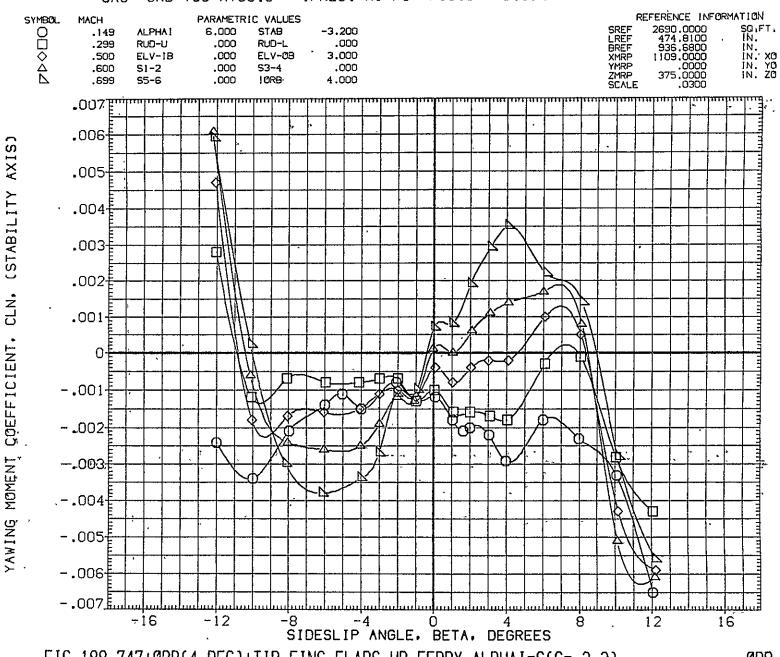
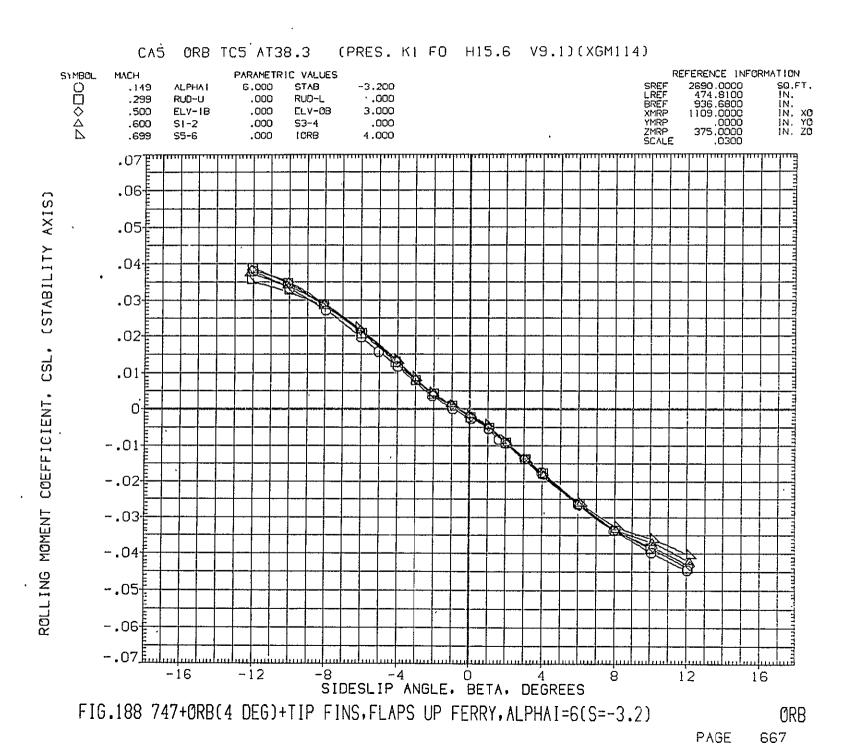
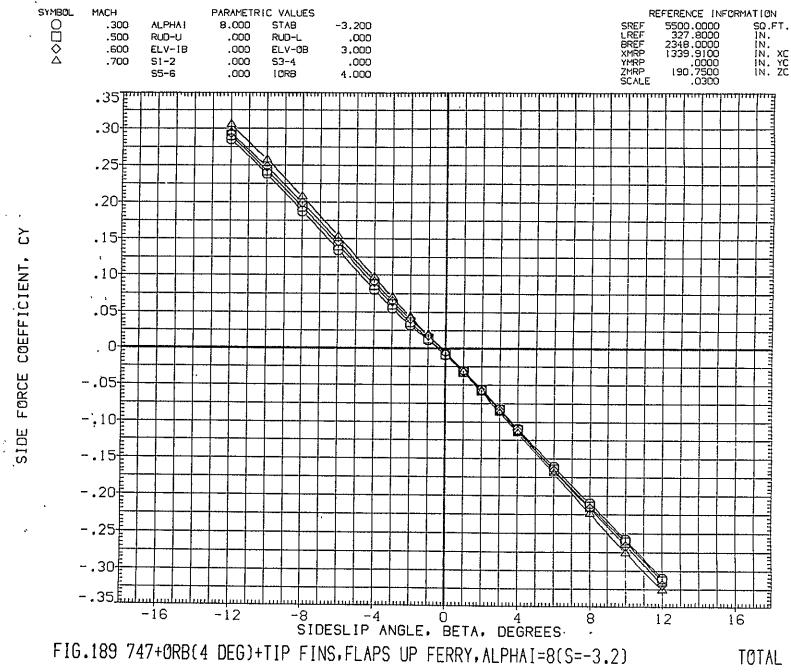


FIG.188 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ALPHAI=6(S=-3.2)

ORB





CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.3)(RGM115) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH 5500.0000 327.8000 2348.0000 1339.9100 .0000 190.7500 SQ.FT. IN. IN. XC IN. YC IN. ZC SREF LREF BREF XMRP YMRP ZMRP SCALE 0000 .300 ALPHA1 8.000 STAB -3.200 .000 RUD-L ,000 .500 RUD-U .600 ELV-1B .000 ELV-0B 3.000 S3-4 .000 .700 S1-2 .000 .000 ICRB 4.000 S5-6 .035ET •030£ (STABILITY AXIS) .025 .020 .015 .010 CLN, .005 YAWING MOMENT COEFFICIENT, 0+ -.005± -.010 -.015€ -.020--.025 -.030# سطَ503. – -12 12 -16 16 8 SIDESLIP ANGLE, BETA, DEGREES

FIG.189 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ALPHAI=8(S=-3.2)

CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.3) (RGM115)

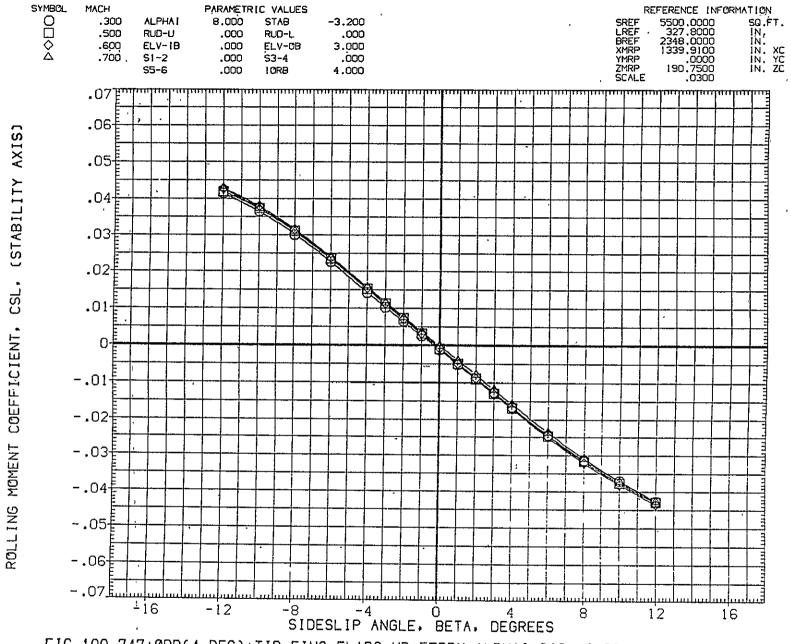
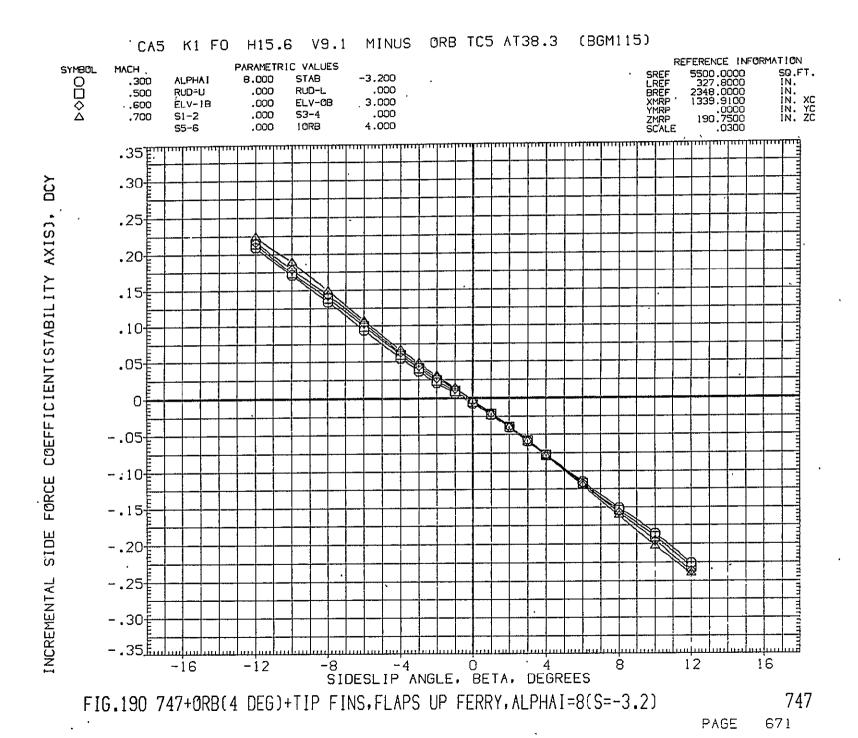


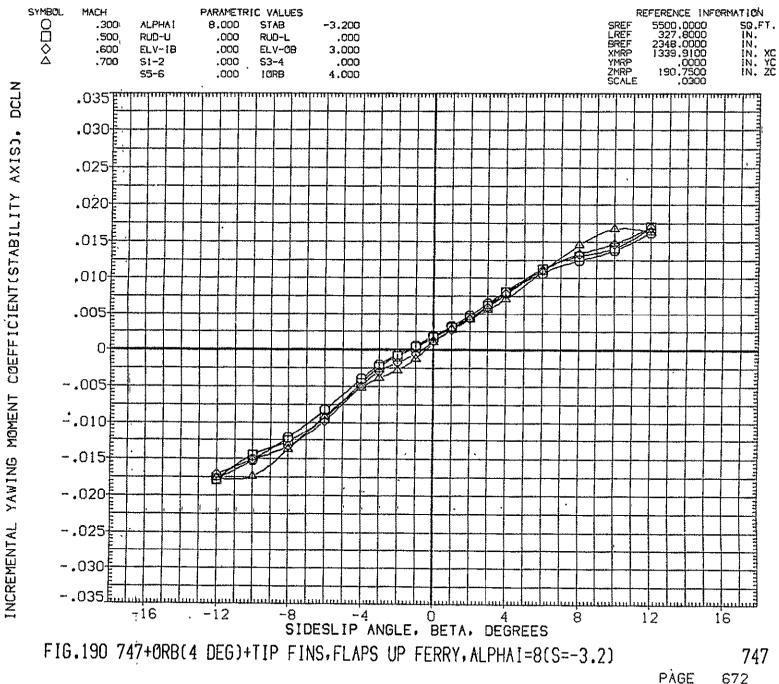
FIG.189 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ALPHAI=8(S=-3.2)

TOTAL 670





CA5 KI FO HI5.6 V9.1 MINUS ORB TC5 AT38.3 (BGM115)



CA5 K1 FO H15.6 V9.1 MINUS ORB TC5 AT38.3 (BGM115) PARAMETRIC VALUES REFERENCE INFORMATION SYMBOL MACH 5500.0000 327.8000 2348.0000 SREF LREF BREF SQ.FT. 0000 -3.200 .300 **ALPHAI** 8.000 STAB .000 RUD-L .000 .500 RUD-U IN. XC IN. YC IN. ZC 1339.9100 .0000 190.7500 .0300 ELV-0B 3.000 ELV-IB .000 XMRP .600 .000 S3-4 .000 YMRP .700 S1-2 ZMRP SCALE 10RB 4,000 55-6 .000 DCSL .035ETT AXIS), .030£ .025 MOMENT COEFFICIENTISTABILITY .020 .015 .010 .005 . 0--.005[-.010 ROLL ING -.015[-.020 INCREMENTAL -.025 -.030 -.035 <u>Lui</u> -16 -12 12 16 -8 0 SIDESLIP ANGLE, BETA, DEGREES

FIG.190 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ALPHAI=8(S=-3.2)

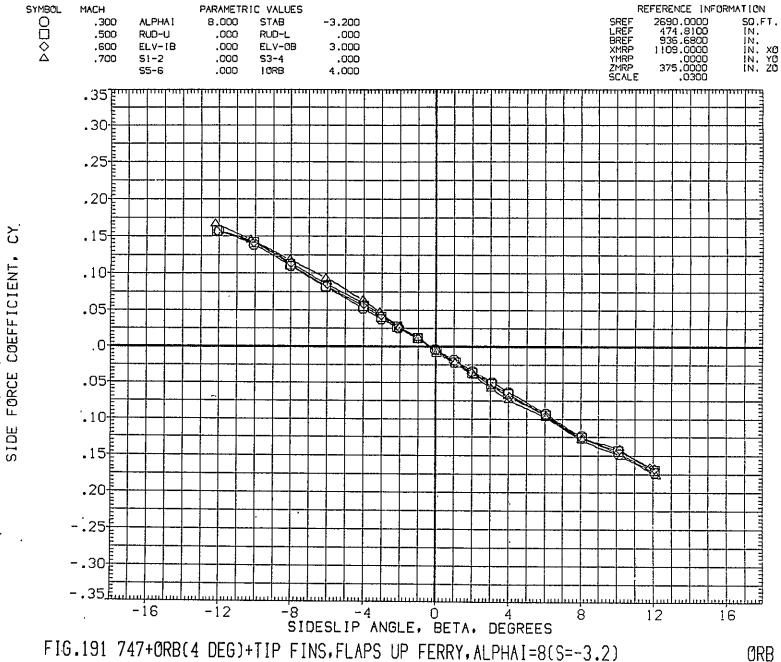
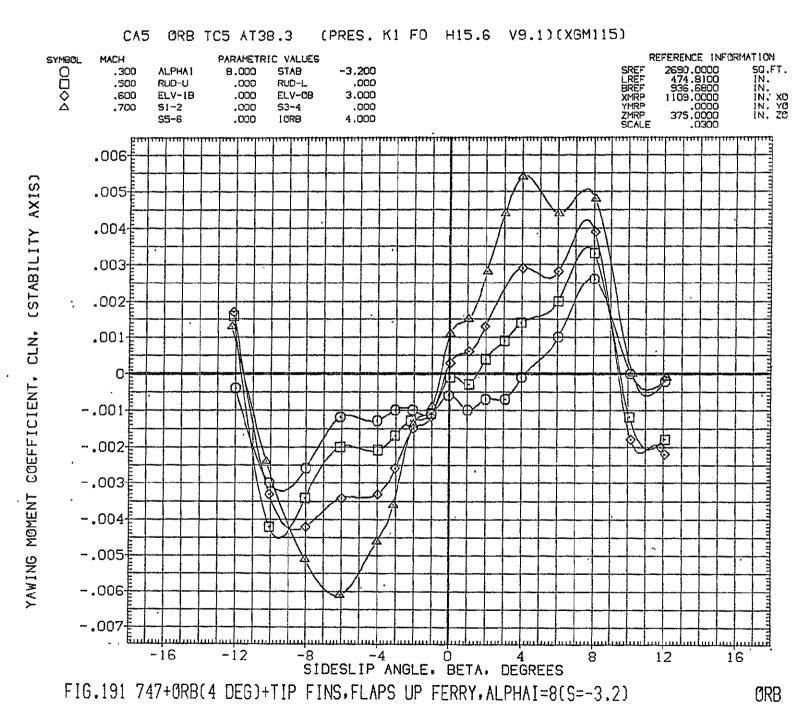


FIG.191 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ALPHAI=8(S=-3.2)



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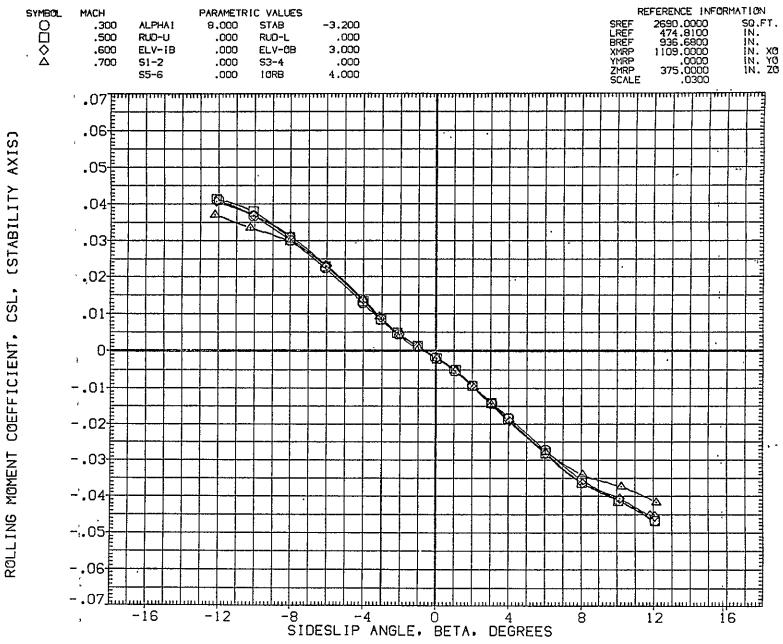


FIG.191 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ALPHAI=8(S=-3.2)

ORB

CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.3) (RGM118)

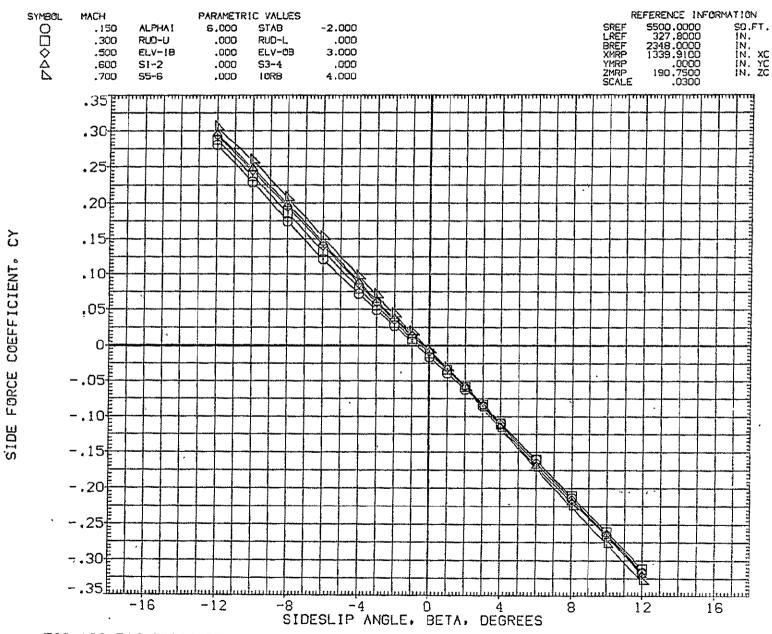


FIG.192 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ALPHAI=6(S=-2)

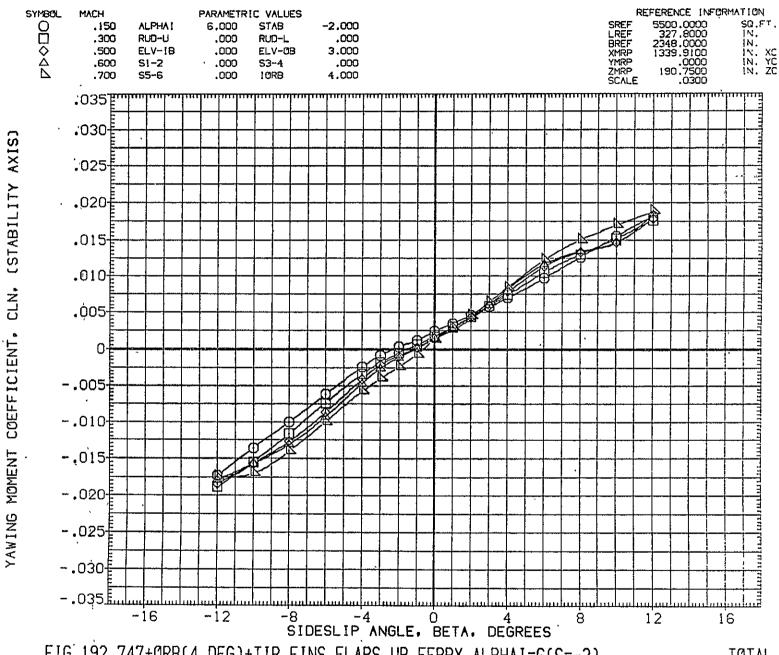


FIG.192 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ALPHAI=6(S=-2)

CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.3) (RGM118) REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH SO.FT. IN. IN. XC IN. YC IN. ZC 6,000 STAB -2.000 000044 ALPHA! .150 RUD-U .000 RUD-L .000 3.000 ` EL,V-0B ELV-1B .000 .000 .600 S1-2 .000 53-4 10R8 4.000 .07E .06[(STABILITY AXIS) .05ŧ .04€ •03‡ .02 CSL, .01 COEFFICIENT, 0--.01[-.02[MOMENT -.03 -.04 ROLL ING -.05 -.06 - .07 1.... -8 -4 0 4
SIDESLIP ANGLE, BETA, DEGREES -i2-16 12 16

FIG.192 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ALPHAI=6(S=-2)

SIDESLIP ANGLE, BETA, DEGREES FIG.193 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ALPHAI=6(S=-2)

-4

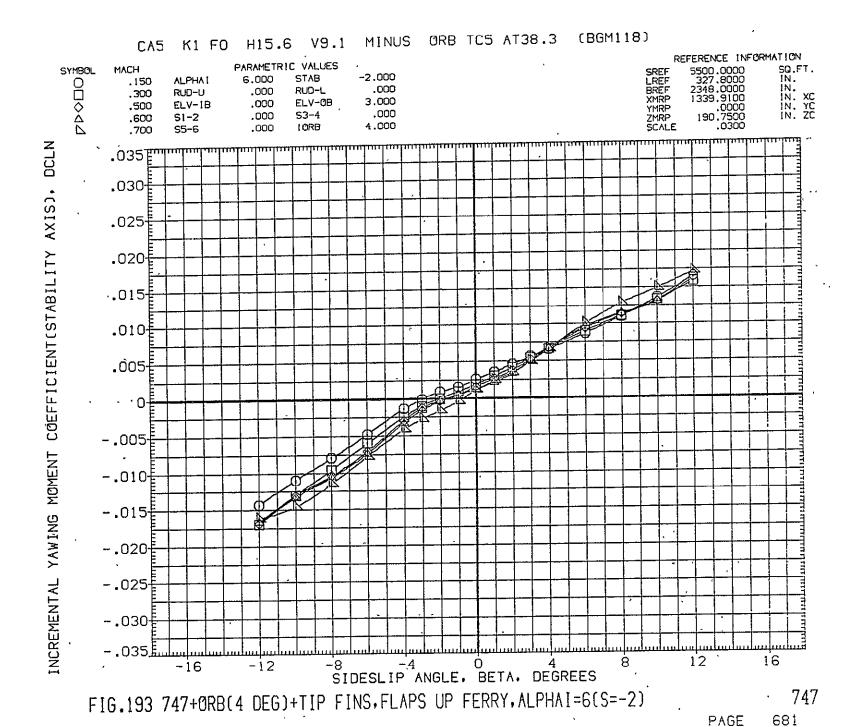
-8

-12

747

16





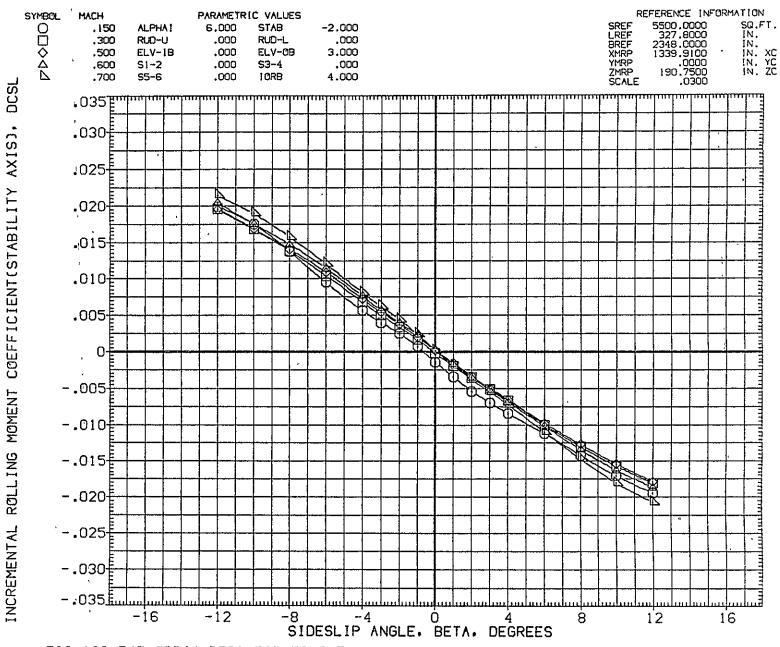
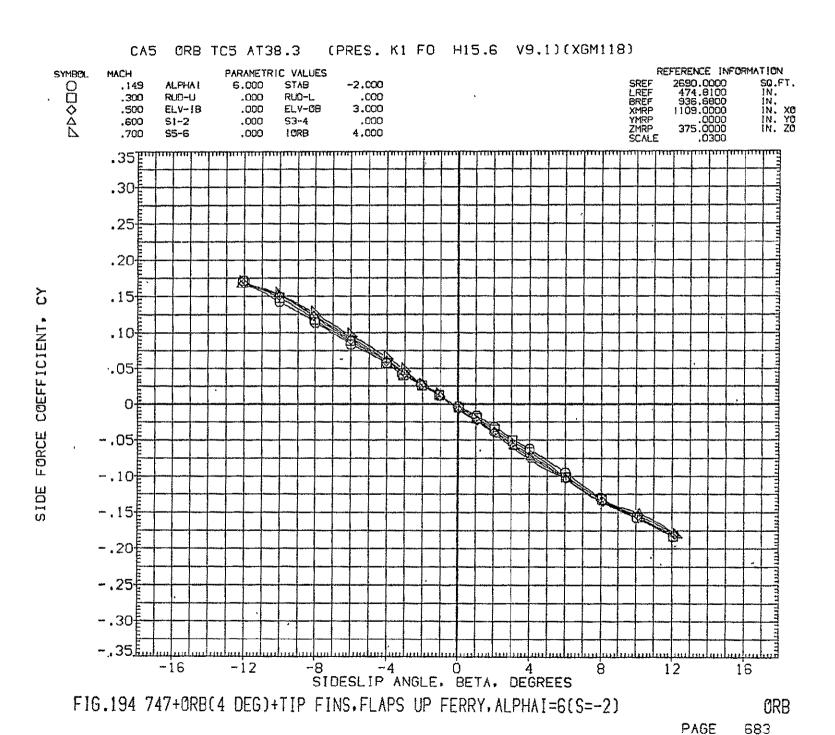


FIG.193 747+0RB(4 DEG)+TIP FINS, FLAPS UP FERRY, ALPHAI=6(S=-2)





CA5 ORB TC5 AT38.3 (PRES. K1 FO H15.6 V9.1)(XGM118)

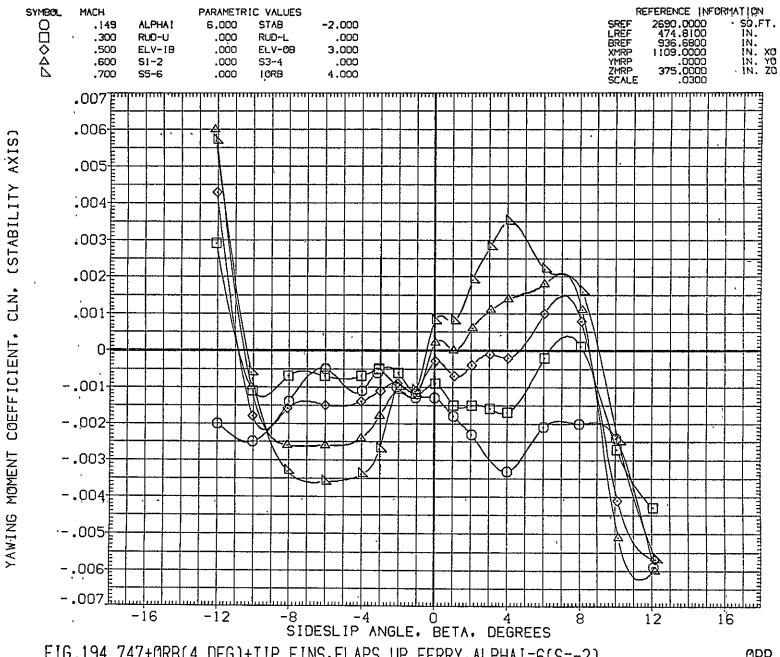


FIG.194 747+ORB(4 DEG)+TIP FINS, FLAPS UP FERRY, ALPHAI=6(S=-2)

ORB

ORB TC5 AT38.3 (PRES. K1 FO H15.6 V9.1)(XGM118) CA5 REFERENCE INFORMATION PARAMETRIC VALUES SYMBOL MACH SO.FT. IN. IN. XO IN. ZC SREF LREF BREF XMRP YMRP ZMRP SCALE 2690,0000 474,8100 936,6800 000044 ALPHA! 6,000 STAB -2,000 .149 RUD-L .000 .000 .300 RUD-U 3.000 .500 ELV-18 .000 ELV-08 1109,0000 375,0000 375,0000 .000 SI-2 .000 \$3-4. .600 LORB 4.000 S5-6 .000 .700 .07T .06E AXIS) .05- (STABIL 1TY .04€ .03£ .02₽ CSL. .01 COEFFICIENT, -.01 -.02 ROLLING MOMENT -.03E -.04 -.05 -.06[-.07<u>E</u> -8
SIDESLIP ANGLE, BETA, DEGREES -12 -16 12 16

FIG.194 747+ORB(4 DEG)+TIP FINS, FLAPS UP FERRY, ALPHAI=6(S=-2)

ORB

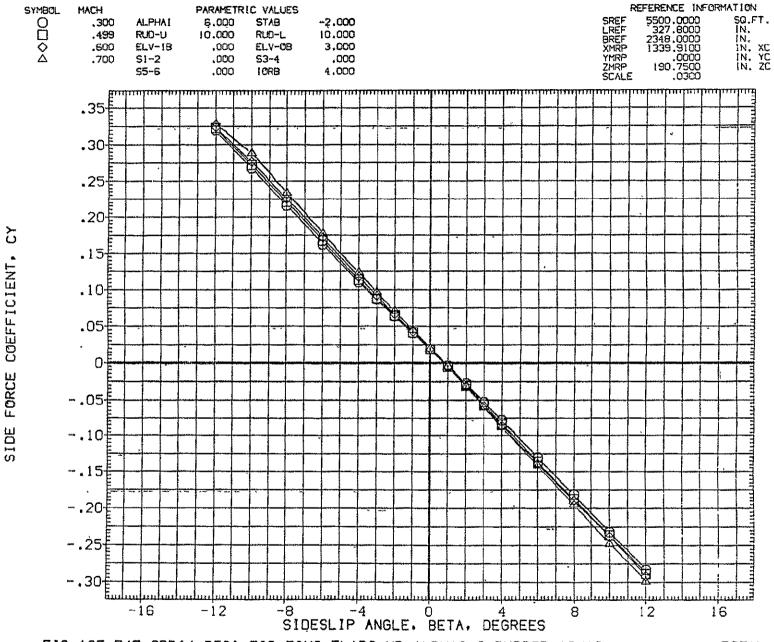


FIG.195 747+ORB(4 DEG)+TIP FINS, FLAPS UP, ALPHAI=6, RUDDER=10/10

CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.3)(RGM121) REFERENCE INFORMATION PARAMETRIC VALUES 5500.0000 327.8000 2348.0000 1339.9100 .0000 190.7500 .0300 SYMBOL MACH SREF LREF BREF XMRP YMRP ZMRP SCALE SQ.FT. -2.000 STAB 0000 ALPHA I 6.000 IN. XC IN. YC IN. ZC RUD-L 10.000 10.000 RUD-U .499 3.000 .000 ELV-OB ELV-18 S3-4 .000 S1-2 .000 4.000 10RB S5-6 .000 .020Emlmlmlmlmlmlmlmlml .015 AXIS) .010[**CSTABILITY** .005 -.005 CLN; -.010 COEFFICIENT, -.015[-.020 -.025 YAWING MOMENT -.030 -.035[-.040[-.045[سلسطِ 050. -.-8 -4 0 4 SIDESLIP ANGLE. BETA. DEGREES 12 16 -12 TOTAL FIG.195 747+ORB(4 DEG)+TIP FINS, FLAPS UP, ALPHAI=6, RUDDER=10/10

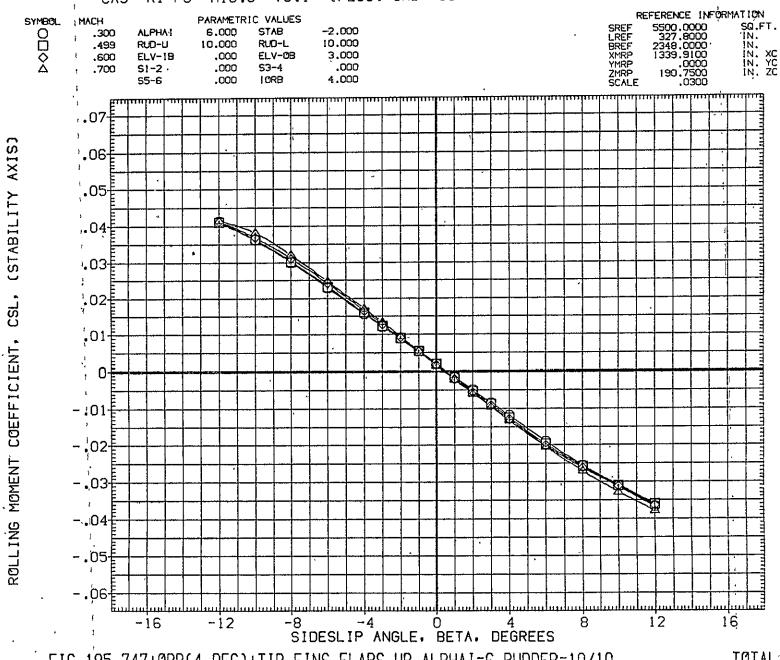
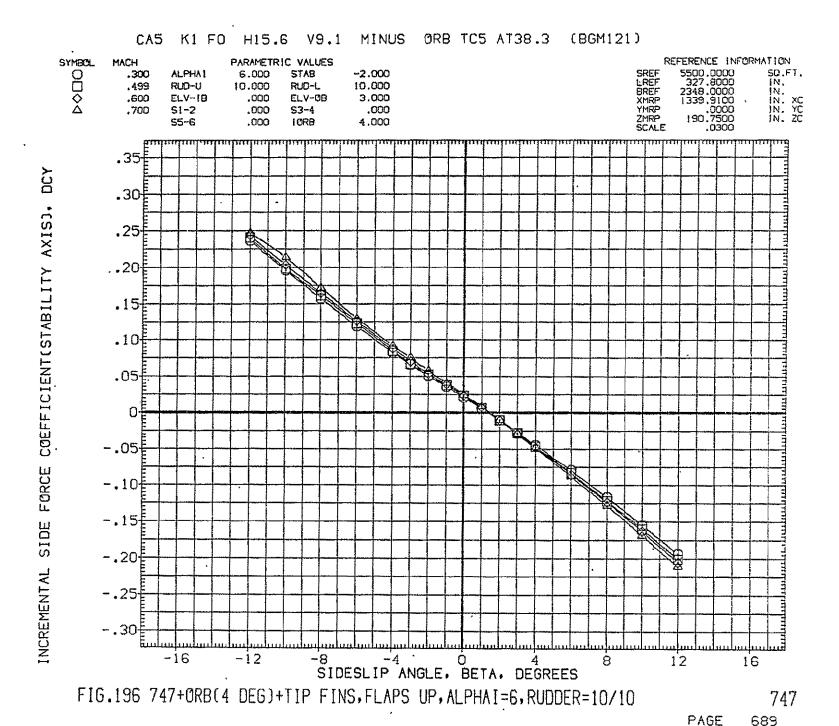


FIG.195 747+ORB(4 DEG)+TIP FINS, FLAPS UP, ALPHAI=6, RUDDER=10/10

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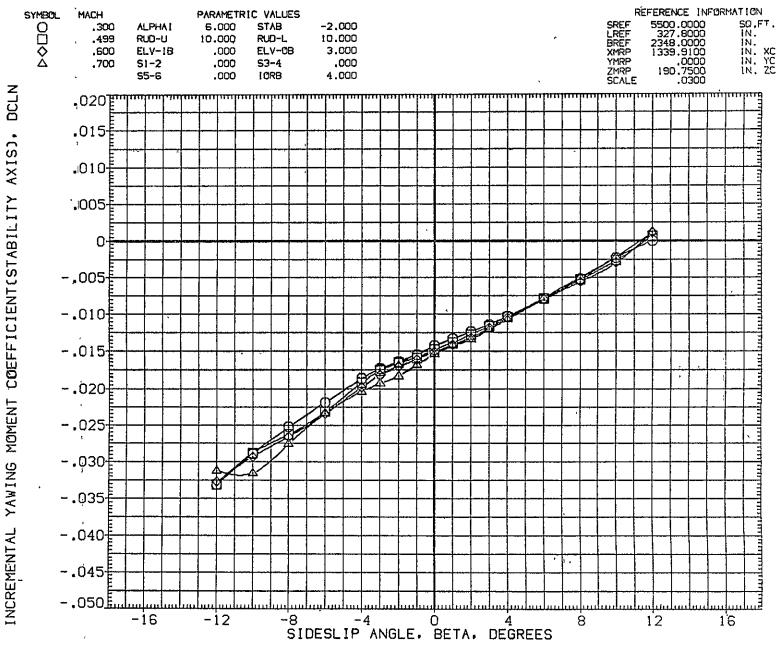
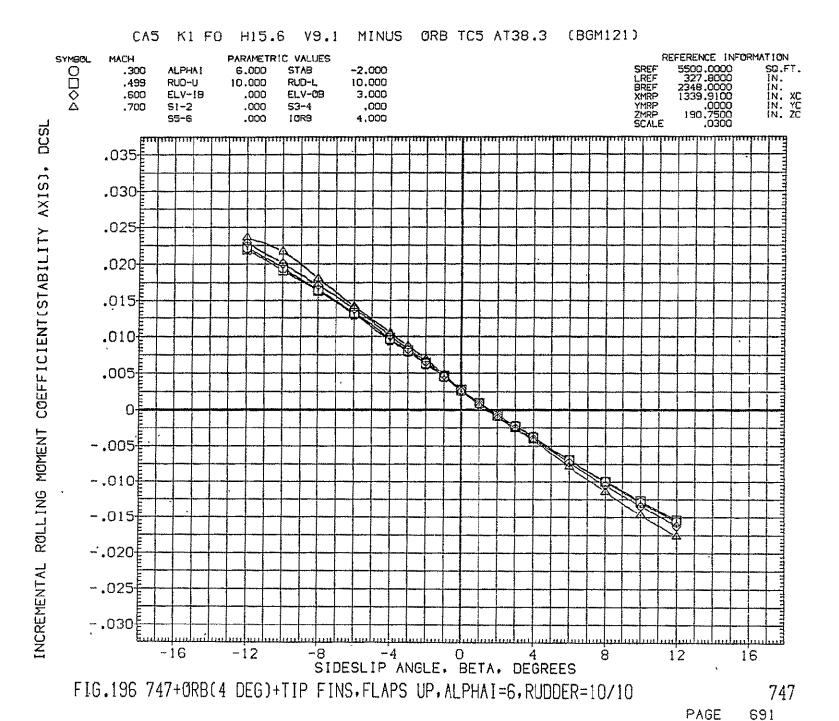


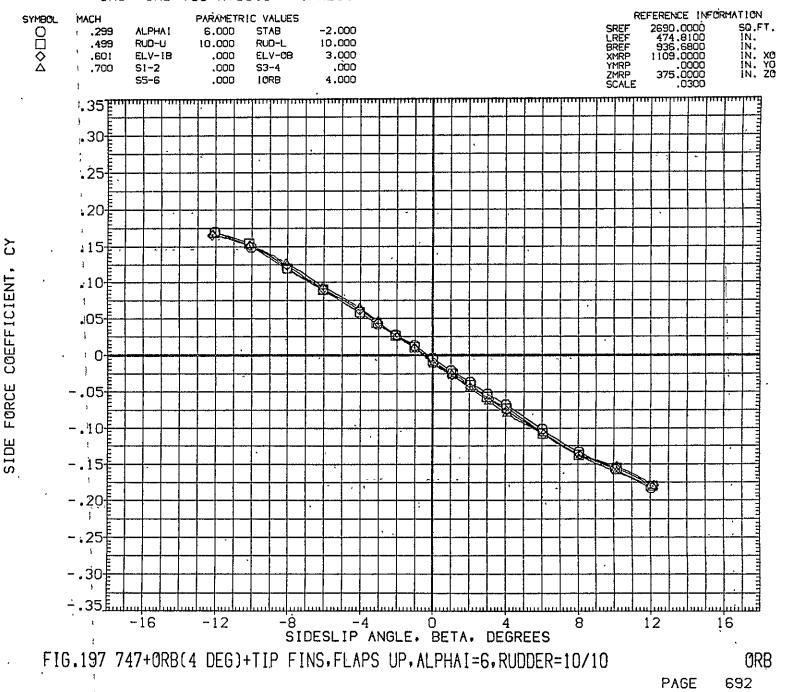
FIG.196 747+0RB(4 DEG)+TIP FINS,FLAPS UP, ALPHAI=6, RUDDER=10/10

747





CA5 ORB TC5 AT38.3 (PRES. K1 FO H15.6 V9.1)(XGM121)





(PRES. K1 FO H15.6 V9.1)(XGM121) CA5 · ORB TC5 AT38.3 REFERENCE INFORMATION PARAMETRIC VALUES MACH SYMBOL, 2690.0000 474.8100 SQ.FT. -2.000 .299 ALPHA1 6,000 STAB ĪÑ. 10,000 RUD-U 10,000 RUD-L 936.6600 1109.0000 .0000 375.0000 .499 IN. XO IN. YO IN. ZO .601 ELV-18 .000 ELV-08 3.000 XMRP YMRP ZMRP SCALE S3-4 .000 ,700 \$1-2 .000 55-6 .000 TORB 4,000 .007E .006f .005 (STAB1LITY .004 剛 .003[.002 CLN, .001[COEFF ICIENT 0 -.001E Ø -.002[MOMENT -.003[-.004 YAWING -.005 -.006 -.007<u>E</u>...l -16 -12 -8 12 --4 16 SIDESLIP ANGLE, BETA, DEGREES

FIG.197 747+0RB(4 DEG)+TIP FINS, FLAPS UP, ALPHAI=6, RUDDER=10/10

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ORB

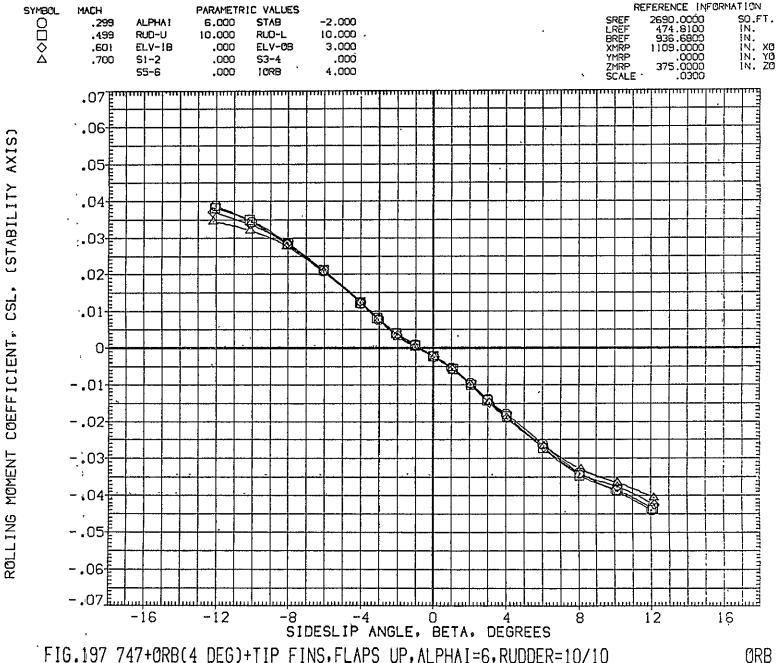


FIG.197 747+0RB(4 DEG)+TIP FINS, FLAPS UP, ALPHAI=6, RUDDER=10/10

CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.3) (RGM122) REFERENCE INFORMATION SYMBOL MACH PARAMETRIC VALUES SQ.FT. IN. IN. XC IN. YC IN. ZC SREF LREF BREF XMRP 5500.0000 327.8000 -2.000 ALPHA1 2.000 STAB .300 .500 RUD-U 10.000 RUD-L 10,000 2348.0000 1339.9100 ELV-0B 3.000 .600 ELV-IB .000 .0000 190.7500 .0300 YMRP .000 .700 51-2 .000 S3-4 ZMRP SCALE IOR8 4.000 55-6 .000 .35 .30 .25[.20‡ \sim .15[COEFFICIENT, .10[.05 o. FORCE -.05[SIDE -.10[-.15[-.30 -8 -4 0 4 SIDESLIP ANGLE, BETA, DEGREES -16 -12 12 16

FIG.198 747+0RB(4 DEG)+TIP FINS,FLAPS UP, ALPHAI=2, RUDDER=10/10

CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.3)(RGM122)

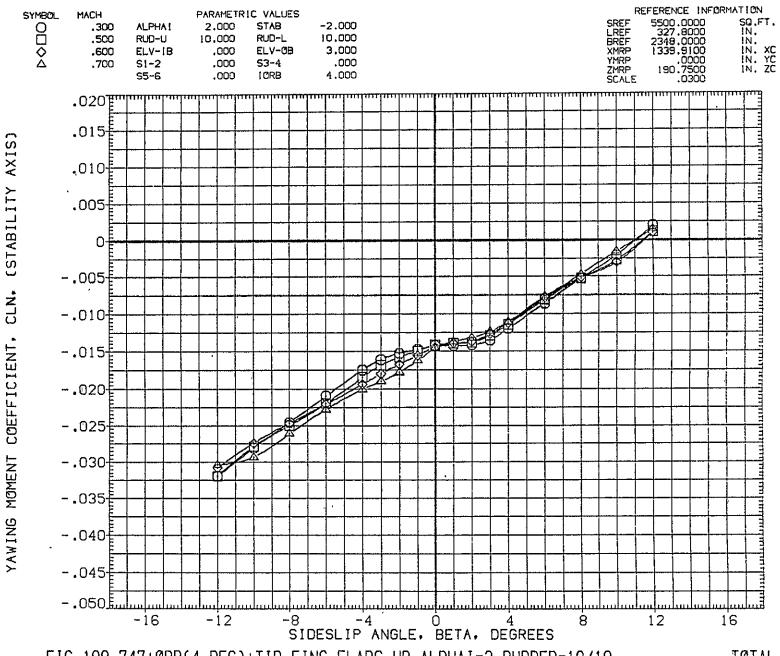


FIG.198 747+0RB(4 DEG)+TIP FINS,FLAPS UP,ALPHAI=2,RUDDER=10/10

SYMBOL

(STABILITY AXIS)

CSL,

COEFFICIENT.

ROLLING MOMENT

-.025

-.030[

CA5 K1 FO H15.6 V9.1 (PLUS. ORB TC5 AT38.3)(RGM122) MACH PARAMETRIC VALUES REFERENCE INFORMATION 5500.0000 327.8000 SQ.FJ. 2.000 ALPHAI STAB -2.000 SREF RUD-U 10.000 RUD-L 10.000 .500 2348.0000 1339.9100 .0000 190.7500 .0300 IN. XC IN. YC IN. ZC BREF XMRP YMRP .600 ELV-18 .000 ELV-0B 3.000 .700 .000 S3-4 .000 S1-2 ZMRP SCALE \$5-6 .000 !ORB 4.000 .035-.030 .025 .020 .015 .010+ .005 -.005 -.010 -.015‡ -.020

SIDESLIP ANGLE, BETA, DEGREES FIG.198 747+0RB(4 DEG)+TIP FINS,FLAPS UP, ALPHAI=2, RUDDER=10/10

-4

-12

TOTAL

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16

12

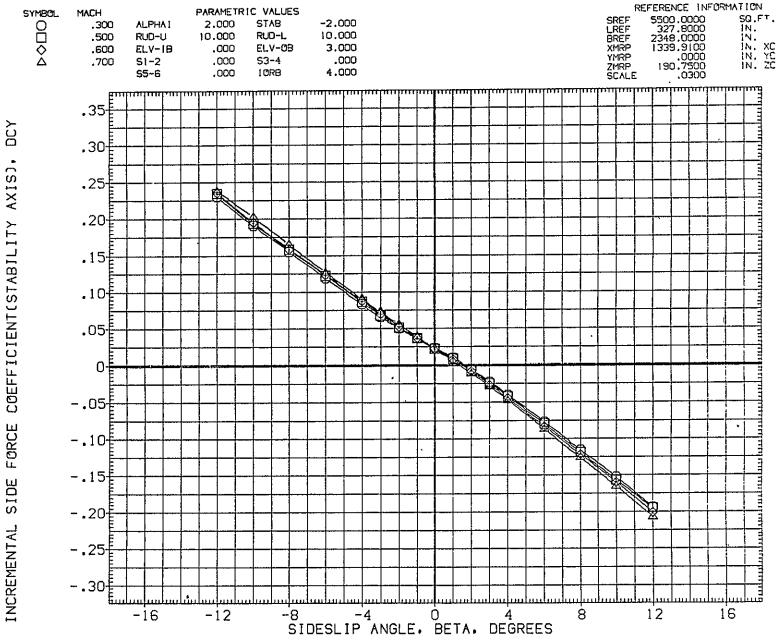


FIG.199 747+ORB(4 DEG)+TIP FINS, FLAPS UP, ALPHAI=2, RUDDER=10/10

747

CA5 K1 FO H15.6 V9.1 MINUS ORB TC5 AT38.3 (BGM122) SYMBOL MACH PARAMETRIC VALUES REFERENCE INFORMATION 0000 5500.0000 327.8000 2348.0000 1339.9100 .0000 190.7500 .0300 .300 ALPHA I 2.000 STAB -2.000 SO.FT. LREF .500 RUD-U 10,000 RUD-L 10,000 N. XC IN. XC IN. YC BREF ELV-IB .600 .000 ELV-OB 3,000 ,700 YMRP ZMRP SCALE Si-2 .000 S3-4 .000 95-B .000 IORB 4,000 DCLN .020ETT .015 AXIS3, .010년 COEFFICIENTCSTABILITY .005£ 0--.005# -.010 -.015E -.020-YAWING MOMENT -.025 -.030 -.035 INCREMENTAL -.040 -.045 -.050£.... -16 -i2 -8 -4 12 16 SIDESLIP ANGLE, BETA, DEGREES FIG.199 747+ORB(4 DEG)+TIP FINS, FLAPS UP, ALPHAI=2, RUDDER=10/10 747

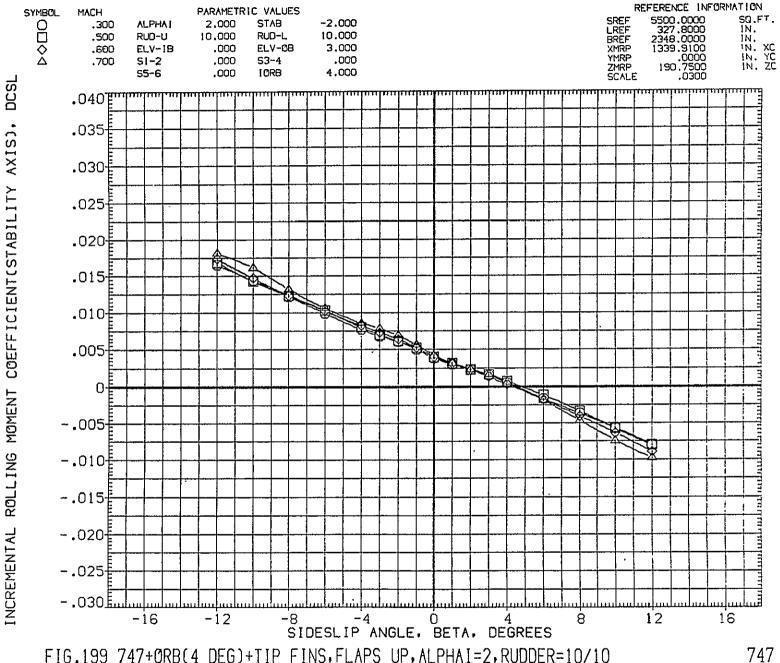


FIG.199 747+ORB(4 DEG)+TIP FINS, FLAPS UP, ALPHAI=2, RUDDER=10/10



ORB TC5 AT38.3 (PRES. K1 FO H15.6 V9.1)(XGM122) CA5 SYMBOL O O A MACH PARAMETRIC VALUES REFERENCE INFORMATION 2690.0000 474.8100 936.6800 1109.0000 .0000 375.0000 .0300 SQ.FT. IN. IN. XQ IN. YQ IN. ZQ .301 ALPHA! 2.000 -2.000 STAB SREF .499 RUD-U 10.000 RUD-L 10.000 BREF XMRP YMRP ZMRP SCALE .600 ELV-IB .000 ELV-08 3.000 .700 51-2 .000 S3-4 .000 S5-6 .000 TORB 4.000 .35Em .30₽ .25[.20 ۲ .15[FORCE COEFFICIENT, .10E .05₺ 0 -.05[-.10[SIDE -.15 -.20<u>₽</u> -.25 -.30 -.35<u>L</u> -8 -4 0 4 SIDESLIP ANGLE, BETA, DEGREES -16 -12 12 18 8 FIG.200 747+0RB(4 DEG)+TIP FINS,FLAPS UP,ALPHAI=2,RUDDER=10/10 **ORB**

CA5 ORB TC5 AT38.3 (PRES. K1 FO H15.6 V9.1)(XGM122)

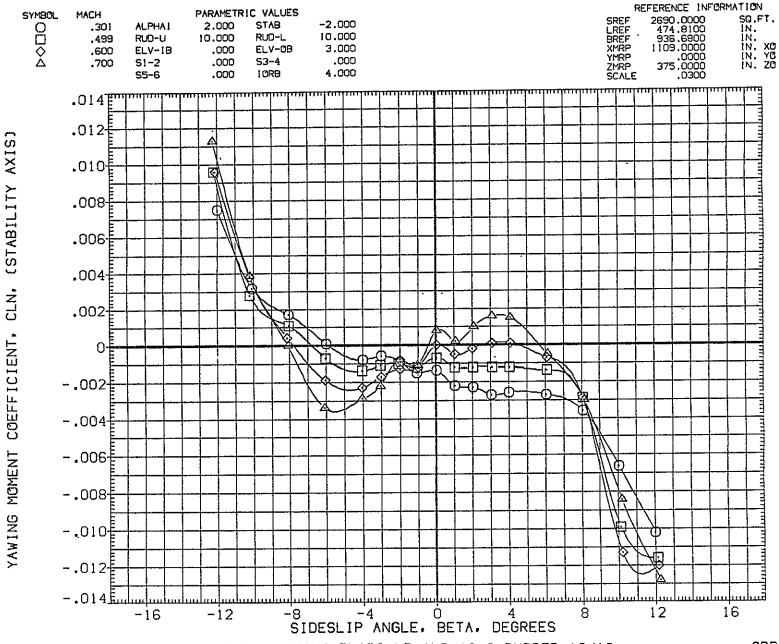


FIG.200 747+ORB(4 DEG)+TIP FINS, FLAPS UP, ALPHAI=2, RUDDER=10/10

ØRB

